TWO - DIMENSIONAL FINITE ELEMENT ANALYSIS OF TRANSIENT FLOW AND TRACER MOVEMENT IN CONFINED AND PHREATIC AQUIFERS

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This is to certify that the

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OF TRANSIENT FLOW AND TRACER MOVEMENT
IN CONFINED AND PHREATIC AQUIFERS

presented by

Sirous Haji-Djafari

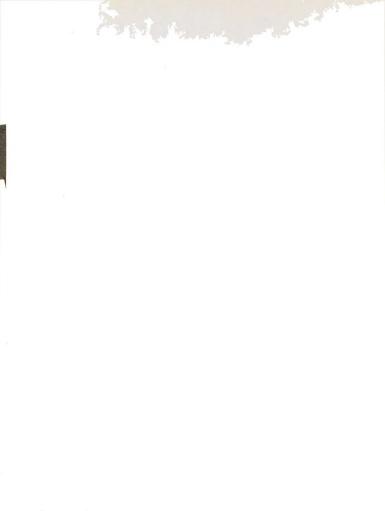
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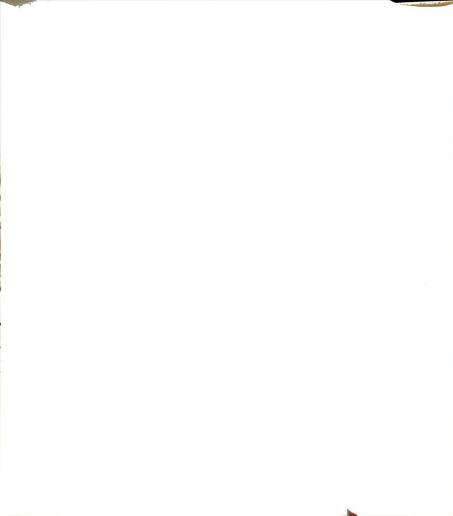
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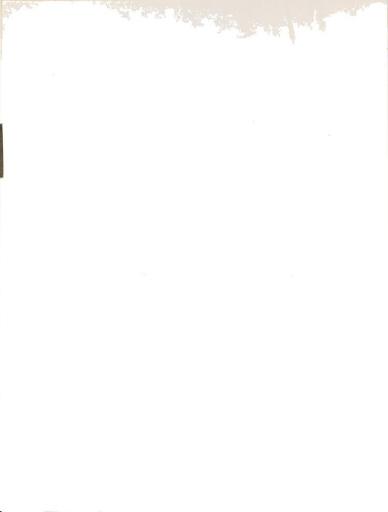
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ABSTRACT

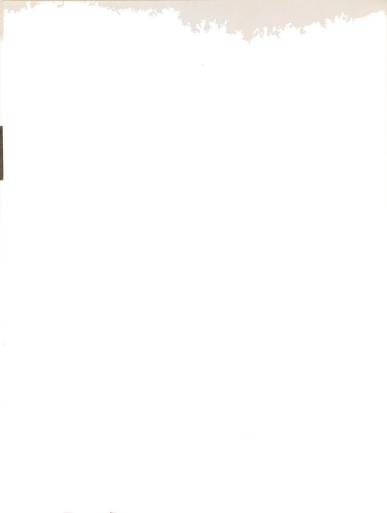
TWO-DIMENSIONAL FINITE ELEMENT ANALYSIS
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Ву

Sirous Haji-Djafari

In this study the movement of a tracer in an aquifer with transient flow conditions is investigated, both on a regional as well as local scale. For the regional scale the two-dimensional horizontal plane is considered, while for the local scale a vertical cross section of a site is chosen. Special emphasis is placed upon solving the flow and mass transfer phenomena in a phreatic aquifer with a time variable boundary.

Finite element formulation of the flow and convective-dispersion equations leads to a set of first order partial differential relations. In addition, with use of the finite element concept, higher order time approximations for the system of equations are derived. In order to obtain continuous flow across elements and at the nodes, the Galerkin formulation of the Darcy law is constructed and velocity vectors are calculated simultaneously at the nodes. These transient velocities are



subsequently used in shifting the phreatic surface and in computing dispersion coefficients and convective terms of the mass-transport equation. A procedure is adapted which locates the phreatic boundary of an aquifer without repositioning the nodal coordinates of the elements.

The validity of the proposed techniques is established by first comparing the numerical flow results with existing analytical, experimental, and field data. Upon verification of the solution of the flow equation, the prediction of the movement of a tracer in an unconfined aquifer with a time-variable phreatic boundary, and in a confined aquifer with a transient flow condition, is conducted. Numerical examples are presented to demonstrate the capability of the proposed techniques.

It is shown that the Galerkin finite element method can be used to solve the flow and convectivedispersion equations, both in confined and phreatic aguifers under time-dependent flow situations.



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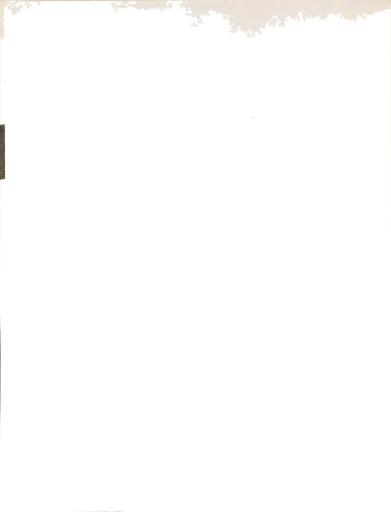
Sirous Haji-Djafari

A DISSERTATION

Submitted to
Michigan State University
in partial fulfillment of the requirements
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DOCTOR OF PHILOSOPHY

Department of Civil and Sanitary Engineering



To my mother and in memory of my father



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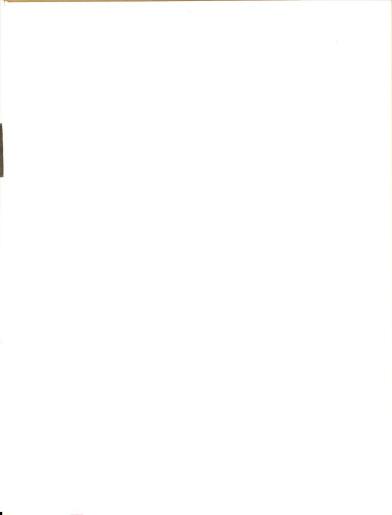
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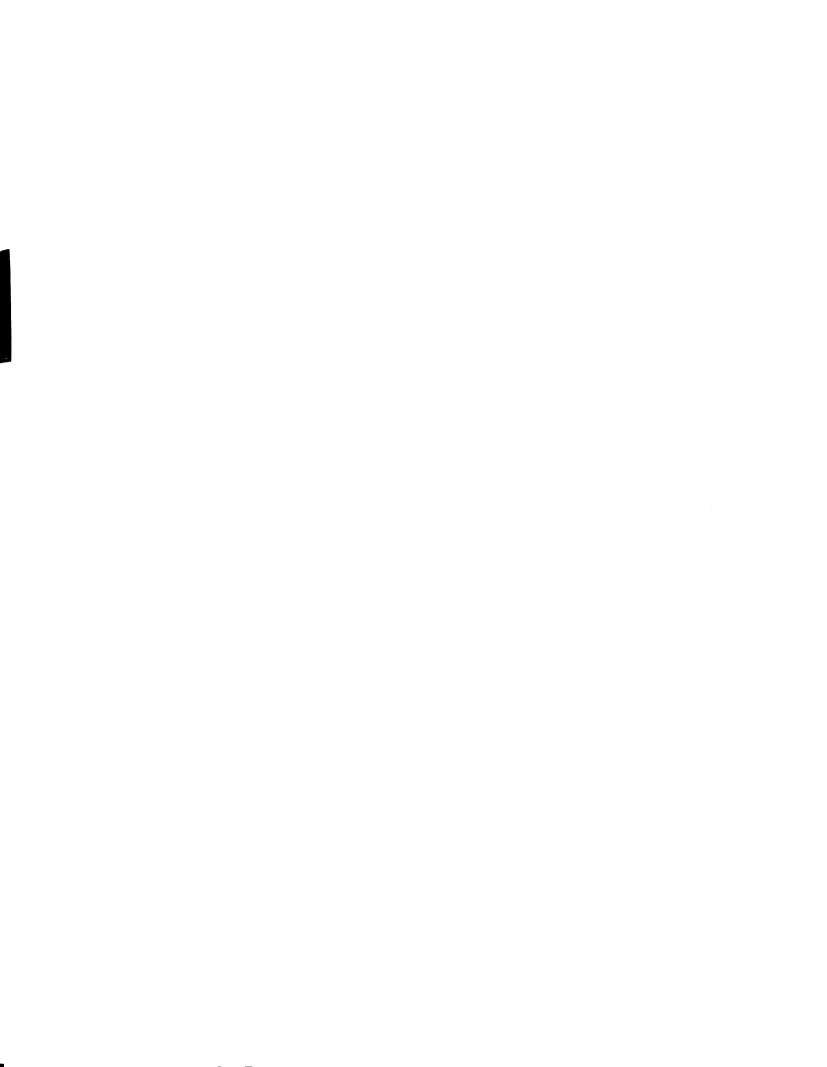


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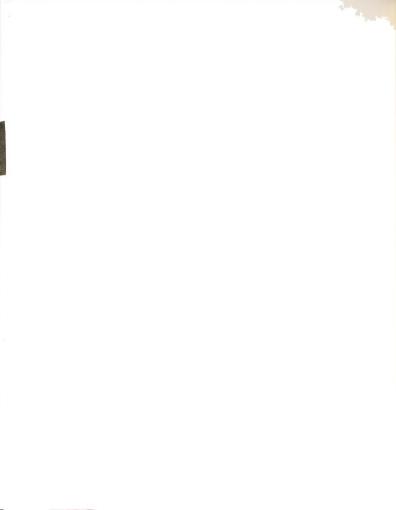
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NOMENCLATURE

a _I	longitudinal dispersivity	L
a _{II}	transversal dispersivity	L
a _{rk}	coefficients in Equation (5.1.5)	
b	thickness of aquifer	L
В	dimensionless length	
С	mass concentration	M/L^3
С	unknown variable (in Chapter V)	
c c*	initial value of C	M/L^3
c *	analytical solution of C	
^d 1' ^d 3	shifting distances of phreatic surface in the \mathbf{x}_1 and \mathbf{x}_3 directions	L
ď	shifting distance along the normal line	
d _n	shifting distance along the normal to the free surface	L
ā	average shifted distance	
D	domain of interest	
DTMAX	maximum specified time step	T
$^{\mathrm{D}}\mathrm{d}$	molecular diffusivity	L^2/T
D _{ij}	mechanical dispersion coefficient	L^2/T
D'ij	hydrodynamic dispersion coefficient	L^2/T
$D_{\mathbf{L}}$	longitudinal dispersion coefficient	L^2/T
$D_{\mathbf{T}}$	transversal dispersion coefficient	L^2/T
(D*d) _{ij}	coefficient of molecular diffusion	L^2/T



е	superscript representing an element	
FNT	Fixed Node Technique	
g	acceleration due to gravity	L/T^2
i	subscript	
i	superscript representing iteration	
î	unit normal along x ₁	
I	accretion	L/T
I _n	magnitude of accretion along the normal to the phreatic surface	L/T
j	subscript	
k	subscript	
ƙ	unit normal along x_3	
k _{ij}	intrinsic permeability	L ²
K _{ij}	hydraulic conductivity	L/T
l _i	directional cosine	
L	length	L
LA	length of a segment	L
m	an integer	
М	number of nodes in one element	
MA	order of approximation	
MAXBW	upper bandwidth of matrix plus one	
MNT	Movable Node Technique	
n	subscript	
n _e	effective porosity	
N	shape function	
NELS	number of elements in whole grid system	

NNDS	number of nodes in whole grid system	
NNFS	number of nodes on the phreatic surface	
NT	number of available data	
NZ	number of available data	
N _n	shape function for a specific point	
р	pressure	M/LT ²
P	discharge per unit area	L^3/L^2T
$P_{\mathbf{w}}$	discharge per unit area at a specific point	L ³ /L ² T
$\mathtt{q}_\mathtt{i}$	specific discharge	L/T
$q_{\mathbf{n}}$	seepage flux along the unit normal	L/T
ф́с	mass flux of source or sink	M/L^3T
Q_2	known flux along boundary	L/T
r	subscript	
r	distance from pumped well to observation point	L
s	drawdown	L
S	storage coefficient	LO
S	surface boundary	
Ss	elastic specific storage	1/L
s _y	specific yield	Γ_{O}
t	time	T
т _{іј}	transmissivity	L ² /T
T [*] ij	medium's tortuosity	
υ _j	velocity of propagation of the free surface	L/T
Un	propagation of the phreatic surface surface along the unit normal	L/T

v	magnitude of velocity	L/T
$\mathtt{v_i}$	seepage velocity	L/T
v _i ^c	simultaneous solution of the seepage velocity vectors	L/T
W	weighting function	
×j	Cartesian coordinates	L
x'1, x'3	location of the phreatic surface at time t + Δt	L
x ₁ ,x ₃	temporary location of the phreatic surface at time t + Δt	L
x ₃	elevation head	L
ZT	averaged sum of the square of residu- als using different time steps	
ZX	averaged sum of the square of residuals using different $\Delta \mathbf{x}$	
lpha i	known functions for boundary conditions of dispersion	
lpha i	coefficients which are used to approximate unknown variable (Chapter V)	
β	an angle equal to $(\pi/2)$ - ω + θ	
β _i	known functions for boundary condition of flow	
β _i	coefficients which are used to approximate first derivative of unknown variable (Chapter V)	
Υį	coefficients which are used to approximate second derivative of unknown variable (Chapter V)	
δ	Dirac delta function	
$^{\delta}$ ij	Kronecker delta	
Δt	time step	T
Δ× _i	grid size	L

η	dimensionless curvilinear and local coordinates in x_2^- or x_3^- direction	
θ	dimensionless time parameter	
θ	angle that the tangent to the phreatic surface makes with the positive $\mathbf{x}_1\text{-}\text{direction}$	
λ	independent variable	
ν	kinematic viscosity of fluid	L ² /T
ξ	dimensionless curvilinear and local coordinates along \mathbf{x}_1 -direction	
ρ	density of the fluid	M/L^3
ф	piezometric head	L
ϕ_{av}	average piezometric head	L
Φ	known piezometric head	L
ω	angle that the tangent to the phreatic surface makes with positive $\mathbf{x}_1\text{-}\text{direction}$	
^	a superscript representing the numerical approximation of unknown variable such as φ or C	

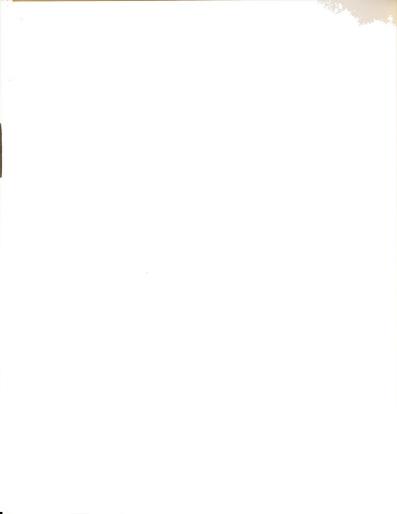
CHAPTER I

INTRODUCTION

1.1 Motivation

In recent years the demand for fresh water has increased drastically due to population increase, industrial growth, and agricultural expansion. However, fresh water sources are limited and the bulk of this vital resource lies underground in the form of groundwater. It is now more or less generally accepted that man's future will depend on his ability to conserve this valuable resource for human consumption.

Almost every pollutant that leaves a sewage plant or is part of runoff from rural lands or urban areas will reach some water basin such as a lake or an aquifer. Besides the customary disposal of treated effluent in nearby streams, rivers and lakes, irrigation of farmlands with secondary and tertiary wastewater is being sought as one of the possible ways to replenish groundwater sources while maintaining their high quality. In one type of system, treated sewage effluent is passed through a series of ponds before transferal to farmlands for use as irrigation water [see, e.g., Bahr 1974]. The tertiary



wastewater slowly infiltrates into the soil while providing intake water and nutrients for the plants. During this passage the constituents are mixed, dispersed,
and diffused through the flowing mass, and some of them
such as phosphate, sulfate, and certain heavy metals are
adsorbed by the soil [Ellis 1973]. After the water
leaves the biologically active zone of the soil it enters
the groundwater reservoir and eventually appears in wells
and springs in the region.

Although studies of nutrient intake by plants and studies of pollutants deposited along the course of the water's flow deserve great attention, it is important to know the concentration of the dissolved chemicals which travel easily with the water through the porous medium at different stages of the movement. The analysis of the effects of possibly contaminated recharged effluent on the quality and quantity of the transient groundwater is extremely worthwhile and beneficial.

1.2 Objective and Plan

The objective of this thesis is to investigate the effects of treated wastewater recharge on the quality and quantity of the groundwater resources. Consideration is given to regional problems, but primary emphasis is focused upon the local region, that is, the region beneath and in the vicinity of the recharge site. This

will be done by considering a two-dimensional horizontal plane for the regional scale (on the order of one square kilometer) and a vertical cross section for the local scale (one hundred square meters). For the purpose of this study the pollutants of concern are dissolved chemical substances such as chloride and chromium, which remain unaltered during the transport process. The dispersion and convection of a tracer through a confined or unconfined aquifer with transient conditions are simulated by a Galerkin formulated finite element method.

Calculated transient velocity vectors are used to obtain the dispersion coefficient, and they are introduced into the convective-dispersion equation. The specific objectives include:

- Calculating the location of a transient phreatic surface in unconfined aguifers.
- Investigating the tracer movement in a confined and an unconfined aquifer with a time-dependent flow domain.

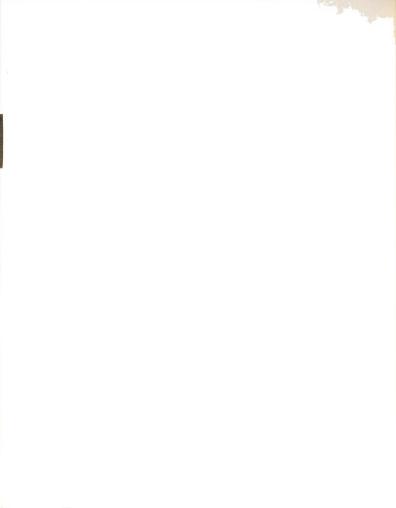
1.3 Scope of Study

The scope of the study presented in this thesis includes a detailed description of the mathematical equations for both flow and convective-dispersion in confined and unconfined aquifus, solutions of the above equations by the finite element method, comparison of the numerical results of this investiation with existing



data, and finally the application of the employed techniques for more complex problems.

The mathematical equation describing the magnitude of the piezometric head in an unconfined aguifer is nonlinear, because of the existing phreatic boundary. The difficulties of solving related equations will increase when the boundary is transient due to accretion or other events. Furthermore, the mass transport equation is also a nonlinear equation because of the dependence of the convective terms and dispersion coefficients on the velocity components. Both the dispersion and flow equations have to be solved simultaneously or consecutively for each time step in order to predict mass distribution in a porous medium. It is customary to modify the numerical grid systems such that the movement of the phreatic surface can be handled. However, because of the changing location of the nodes, it is difficult to obtain the values of velocity components within the grid system so that they might be used in computing the convective terms and dispersion coefficients. In this study this difficulty has been overcome by adapting a technique proposed by France [1971] which can locate the phreatic surface within the grid system without repositioning the nodal coordinates. The method provides a feasible way to solve the convective-dispersion equation for unconfined phreatic aquifers.



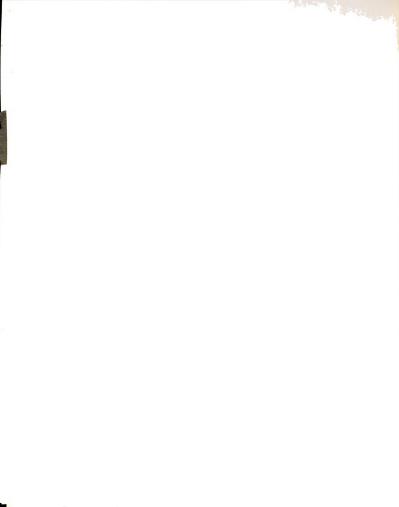
The velocity vectors play a dominant role in the accuracy of the predictions governed by the dispersion model. A simplified procedure for solving velocity components is introduced; this procedure provides continuous velocity values at the nodes. The technique also enables one to predict the tracer movement with a transient flow condition.

The finite element formulation of the field problem leads to a set of ordinary differential equations of the form

[A]{C(t)} + [H]
$$\{\frac{\partial C}{\partial t}\}$$
 = {F(t)}

where C(t) is an unknown variable such as concentration or piezometric head. Discretization of the time derivative of this equation is one of the major concerns for many investigators. With the finite element concept, recurrence formulae for the above equation for three different orders of time approximation are derived. In this process a simple procedure to obtain the finite difference relation for a variable and its first and second derivatives is also shown.

In this study the validity of the proposed techniques is established by first comparing the numerical flow results with existing analytical, experimental, and field data. Upon verification of the solution of the



flow equation, the prediction of the movement of a tracer in an unconfined aquifer with a transient phreatic boundary is conducted. Numerical examples are presented to demonstrate the capability of the proposed techniques. Sensitivity analyses are made to explore the effects of time steps and element size on the accuracy of the numerical results.



CHAPTER II

LITERATURE REVIEW

2.1 Introduction

The major objective of this study is to investigate the movement of a tracer in a confined or an unconfined aguifer which is experiencing a transient flow regime. The piezometric head and the velocity of groundwater flow must be known in order to predict the rate and direction of movement of dispersive substances. The simulation of mass transport through porous media in twodimensional horizontal flow in a confined aguifer has received considerable attention, but little attention has been given to the solution of the dispersion equation for an unconfined aquifer with a phreatic boundary. The difficulties associated with calculation of the convective term and dispersion coefficient with a timedependent phreatic boundary might be considered the major obstacles in the way of progress in this area. There is a vast number of publications available concerning the solutions of the flow equation and the convectivedispersion equation; only those publications relevant to this study are referenced herein.

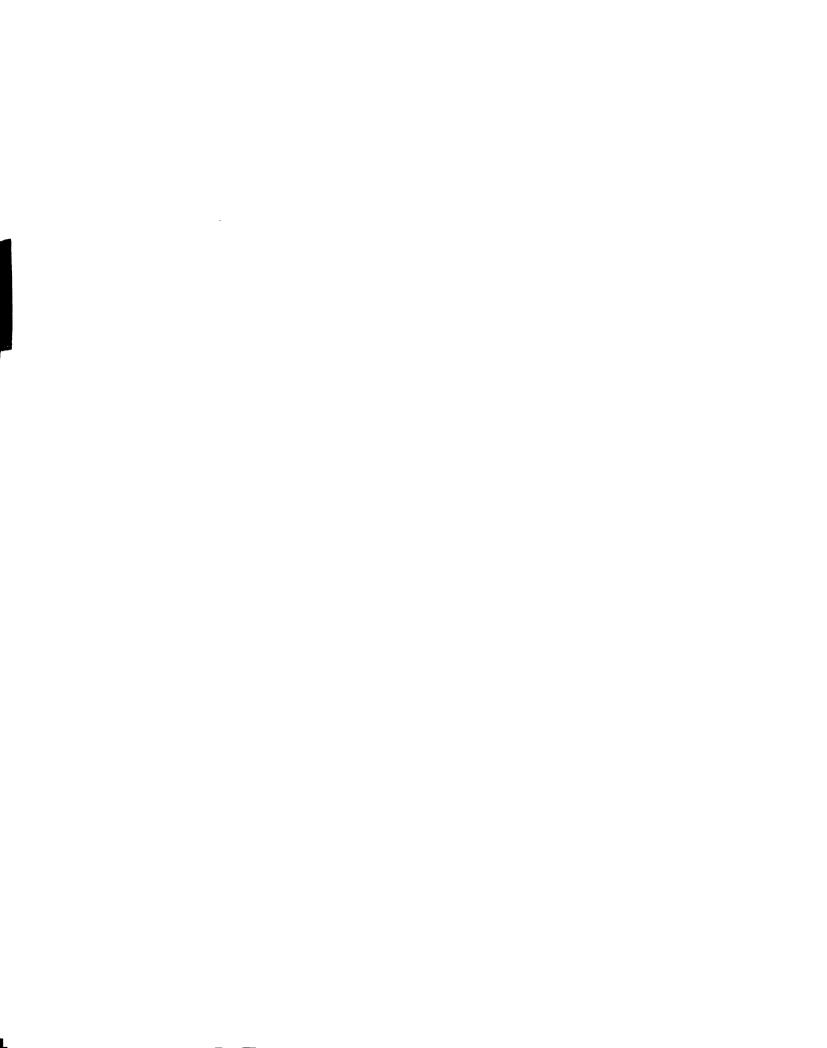
2.2 Mechanics of Flow

2.2.1 Confined Aquifer

The finite element solution of the differential equation describing flow in a two-dimensional horizontal plane for a nonhomogeneous and anisotropic confined aquifer is well-established and is available in the literature. Zienkiewicz et al. [1966] employed the method to obtain a steady-state solution for the heterogeneous and anisotropic seepage problem. Javandel and Witherspoon [1968] used the Rayleigh-Ritz procedure to solve groundwater problems with linear triangular elements. Pinder and Frind [1972] utilized Galerkin's technique to analyze groundwater problems with isoparametric elements. The numerical model developed in this study to calculate the piezometric head in a confined aquifer is similar to the procedure presented by Pinder and Frind.

2.2.2 Unconfined Aquifer With Phreatic Surface

The equations governing boundary and initial value problems of the phreatic surface for liquid flow through porous media are known [Polubarinova-Kochina 1962, Bear 1972]. The exact analytical solutions of these equations are extremely complex. To simplify the treatment of such problems, Dupuit in 1863 assumed that the gradient of the phreatic surface in the vertical plane away from the wells and mounds is very small, thus



the groundwater flow is essentially horizontal and can be considered as a uniform flow. This assumption led to the well-known Boussiness equation. Because of the nonlinearity of the Boussinesq equation, only a small number of analytical solutions are known to date [e.g., Polubarinova-Kochina 1962, and Bear 1972]. An approach to overcome the problem is to linearize either the partial differential equation describing the phreatic surface boundary or the Boussinesq equation. Bear [1972] has outlined several linearization techniques with related references. Marino [1967], following Hantush [1963, 1967], employed a linearization method to solve the problem of the rise and decay of a groundwater mound below a spreading site, and justified the solution with experimental study. All analytical solutions are limited to flow systems in which the boundary conditions are simple, the porous medium is relatively uniform, and the Dupuit approximation is valid, i.e., the vertical gradients throughout are not too large and hence are negligible.

Models and analogues are tools for achieving the solutions of problems where the direct analytical solution is not possible because of the complexity of the system. The Hele-Shaw or viscous flow analog and resistance network analogues are most commonly used. Bear [1960] discussed the scales of viscous analog models for groundwater studies. Bear [1972] presents extensive



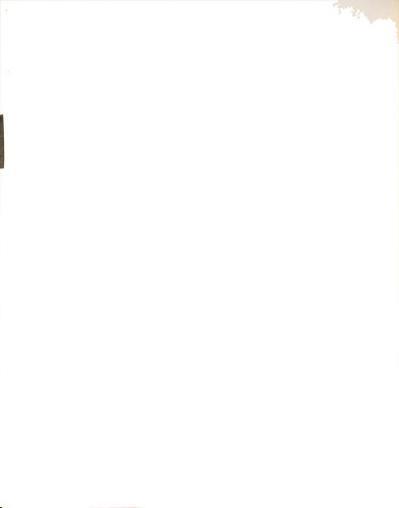
bibliographies of past applications of the Hele-Shaw model to studies of groundwater flow. Marino [1967] used the viscous analog model to study the growth and decay of groundwater ridges. Tinsley and Ragan [1968] employed the Hele-Shaw model to investigate the response of an unconfined aquifer to localized recharge. Herbert [1968] used a resistance network analog to study the time-variable movement of the water table in unconfined saturated strata. His work is based on the assumption that for any time-variant system of water, the flow can be approximated as a series of steady-state solutions, each of slightly varying shape and satisfying the Laplace equation. In his technique the location of the water table is known at a given time, and the new position of the free surface is predicted for a chosen finite time interval. This process is repeated for several time intervals until the maximum required time has been reached or the system has reached the steady state.

In the solution of groundwater flow problems, the digital computer offers enormous advantages and has become a dominant computational tool. There are several numerical techniques available for solving the governing equation for groundwater flow. Of these, the finite difference and the finite element techniques are most commonly used. Todsen [1971] used the finite difference method for solving the free surface flow problems. Amar

[1975] investigated the two-dimensional hydrodynamic behavior of recharge of an unconfined aquifer with the finite difference technique.

The finite difference technique is simple to program, but manipulation of curved boundary conditions which most likely appear in nature is difficult. The finite element technique eliminates this problem and the computer model can be developed in such a way that it can be used for any type of boundary without modification of the program.

Taylor and Brown [1967] presented finite element solutions of steady seepage through dams using a network of triangular elements. In their technique, the location of the phreatic surface was guessed and subsequently adjusted until the free surface boundary conditions were satisfied. Neuman and Witherspoon [1970, 1971] improved and extended their technique to problems of steady-state and transient seepage with a free surface using linear triangular elements. Desai [1972] used the finite element procedure with isoparametric elements to analyze transient unconfined seepage under drawdown conditions in porous media. Sandhu et al. [1974] introduced the variable time step analysis of unconfined seepage. France et al. [1971], following Herbert [1968], used the finite element method with isoparametric elements to analyze



the free surface seepage problem. France [1974] has extended this work to three-dimensional problems.

In all numerical solutions it is necessary to modify the elements to accommodate the movement of the phreatic surface. Based on the characteristics of curved isoparametric elements, France [1971] introduced a new method which permits one to locate the phreatic surface without altering the position of the nodes for each element. This procedure will be called "location of the phreatic surface by use of fixed nodes." The technique is adopted and modified in this study to determine the location of the free surface with transient recharge. This method will provide a tool to solve the convective-dispersion equation in unconfined aquifers with a time-dependent phreatic boundary.

2.3 Mechanics of Convective-Dispersion Phenomena

The analytical solution of the convective-dispersion equation, except for a small number of simple one- and two-dimensional cases, is not easy to determine. Ogata and Banks [1961] used Laplace transforms to obtain the solution of the one-dimensional longitudinal dispersion equation. Harleman and Rumer [1963] gave a steady-state solution for two-dimensional dispersion. Bruch and Street [1967] formulated the analytical solution for unsteady dispersion in an idealized study of

one-dimensional seepage flow through an isotropic porous medium. Hoopes and Harleman [1967] introduced an expression for the distribution of dissolved concentration substances which were added to the steady-state flow between a recharging and pumping well in a homogeneous, isotropic aquifer of infinite, horizontal extent. Marino [1974d] gave a mathematical solution to predict the distribution of concentration in saturated porous media resulting from a variable source concentration.

Much experimental work has been attempted to investigate the behavior of dispersion coefficients and their relation to the seepage velocity, porous structure, and concentration gradient; literature concerning this subject is given by Bear [1972]. Numerous investigators such as Bear [1961], and de Josselin de Jong and Bossen [1961], showed that the dispersion coefficient is a function of true velocity and medium properties. Rumer [1962] experimentally determined the longitudinal dispersion coefficient for one-dimensional transient flow within a certain range of values of the Reynolds number. Harleman and Rumer [1963] investigated the dependency of the dispersion coefficient upon the Reynolds number and porous structure. They conducted laboratory experiments to study the convection and dispersion of salt water in a two-dimensional confined aguifer. Shamir and Harleman [1967] developed an analytical solution for two problems

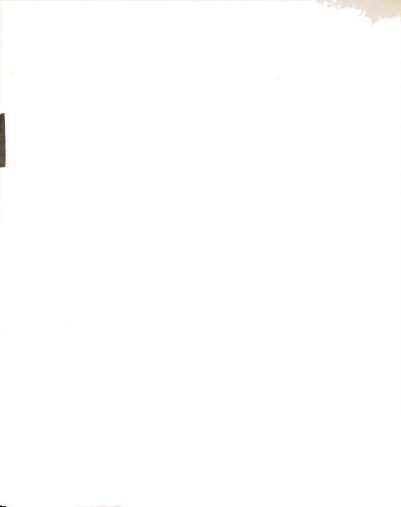


of dispersion in layered porous media and verified their results experimentally. Fattah [1974] investigated and verified a model of the dispersion coefficient tensor in flow through anisotropic and homogeneous porous media.

The conclusion that can be made from the results of these investigations is that the dispersion coefficient is a second-rank tensor and is a function of the true velocity vectors, porous media properties, and the Peclet number. However, there is still no universal agreement regarding the degree of dependency of the dispersion coefficient on these parameters.

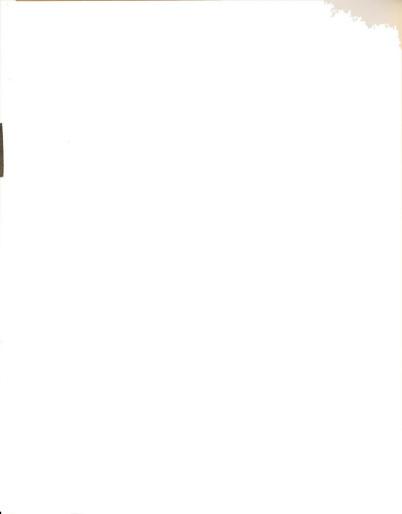
In the simulation of the movement of a tracer in a porous medium, the flow and convective-dispersion equations are solved simultaneously or consecutively. The following paragraphs concern only those techniques which are used to solve the partial differential equation describing the dispersion of a dissolved-chemical constituent in a medium.

The finite difference method is the most commonly used scheme in attempting the numerical solutions of the mass transport equation. Douglas et al. [1959] employed an alternating direction-implicit procedure to solve a two-dimensional, two-phase, incompressible flow model. A similar technique was used by Peaceman and Rachford [1962] in calculating the multidimensional miscible displacement. The Crank-Nicolson approximation



is frequently used in the area of mass displacement. Fried and Combarnous [1971] summarize some of the analytical and numerical methods of resolution of the convective-dispersion equation.

Pinder and Cooper [1970], and Reddell and Sunada [1970], applied the characteristics approach to solve salt water intrusion including the effect of dispersion. Bredehoeft and Pinder [1973] used similar concepts to investigate the groundwater contamination at Brunswick, Georgia. Konikow and Bredehoeft [1974] used the method of characteristics to investigate the chemical quality changes in an irrigated stream-aquifer system. Robertson [1974] used the same approach to model the transport of radioactive and chemical waste in the Snake River Plain Aguifer. The method of characteristics involves placing several moving particles in each cell of the finite difference grid. The location and concentration associated with each particle varies with time. Although this method gives good results compared to the analytical solution and is simple in concept, it is tedious to program and is suitable only for specific situations commonly encountered in the field [Pinder 1973]. In an effort to circumvent difficulties associated with the method of characteristics, Price et al. [1968] introduced a Galerkin-based variational method to



approximate the solution of the dispersion equation, in which various different base functions were used.

The finite element technique has been used recently as a numerical tool to solve the convectivedispersion equation. Guymon et al. [1970] and Nalluswami et al. [1972] employed a finite element integration scheme with triangular elements. Cheng [1973] solved the convective-dispersion equation based on the Galerkin procedure, using a family of triangular elements or quadrilateral isoparametric elements. Pinder [1973] also used the Galerkin-finite element formulation to simulate the groundwater contamination in Long Island, New York. Wang and Cheng [1975], using Dupuit's assumption, solved the convective-dispersion equation by quadratic isoparametric elements for homogeneous and isotropic media with uniform horizontal flow and constant dispersion coefficients. Segol et al. [1975] realized that the velocity vectors used in the mass transport equation should be continuous across elements. Thus, they solved three equations (two components of Darcy and one of mass conservation) simultaneously, and investigated the distribution of salt concentration with steady flow.

In the present work the technique for solving the convective-dispersion equation in unconfined aquifers with a transient phreatic surface is presented, and higher order approximations of time-dependent variables are introduced.

CHAPTER III

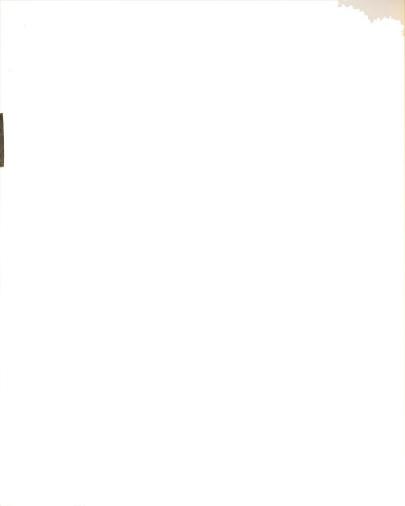
MATHEMATICAL REPRESENTATION OF PHYSICAL SYSTEM

3.1 Background

In this study the dispersion of a tracer on a regional as well as a local scale is investigated. For the regional scale the two-dimensional horizontal plane is considered, while for the local scale a vertical cross section of a site is chosen. To predict the movement of a tracer in a porous medium, the flow regime and its behavior in the medium should be well understood. In this chapter the system is defined, the mathematical descriptions of flow in the aquifers with and without a phreatic boundary are given, the convective-dispersion equation is presented, and the initial and boundary conditions are discussed.

3.2 Darcy Equation

Darcy established a linear relationship between the seepage velocity and the gradient of the piezometric head. This law, which is a consequence of the equation of motion neglecting inertia effects, can be generalized



for either a two- or a three-dimensional situation. It is given by:

where $\mathbf{V_i}$ is the seepage velocity, $\mathbf{q_i}$ is the specific discharge, $\mathbf{K_{ij}}$ is a second order tensor whose elements are called the hydraulic conductivities, $\mathbf{n_e}$ is the effective porosity of the aquifer, ϕ is the piezometric head, and $\mathbf{x_j}$ are the Cartesian coordinates. The hydraulic conductivity is a scalar coefficient which depends on both solid matrix and fluid properties. It is defined as

where k_{ij} is the intrinsic permeability of the porous matrix and depends solely on the properties of the solid matrix, g is the acceleration due to gravity, and ν is the kinematic viscosity of the fluid.

From purely physical considerations, it would seem that the hydraulic conductivity tensor must be symmetric [Eagleson 1970], in which case ${\rm K}_{12} = {\rm K}_{21}$, ${\rm K}_{13} = {\rm K}_{31}$, and ${\rm K}_{23} = {\rm K}_{32}$, and its components reduce to six. Since the principal axes of the symmetric permeability tensor will be orthogonal, it is possible to orient the



coordinate axes $(\mathbf{x}_1,\ \mathbf{x}_2,\ \mathbf{x}_3)$ parallel to the principal axes so that only the three orthogonal terms remain. Thus

$$\begin{aligned} v_1 &= -\frac{\kappa_{11}}{n_e} \frac{\partial \phi}{\partial x_1} \\ v_2 &= -\frac{\kappa_{22}}{n_e} \frac{\partial \phi}{\partial x_2} \\ v_3 &= -\frac{\kappa_{33}}{n_o} \frac{\partial \phi}{\partial x_2} \end{aligned} \tag{3.2.3}$$

For incompressible fluid, the piezometric head is defined (see Figures 3-1 and 3-2)

$$\phi = \frac{p}{\rho g} + X_3 \tag{3.2.4}$$

where p is the pressure deviation from atmospheric pressure, ρ is the fluid density, and X_3 is the elevation above datum. In Equation (3.2.4) the term p/ ρ g is called the pressure head, and X_3 is known as the elevation head.

For confined aquifers the piezometric surface is an imaginary surface to which water rises in a tapped well (Figure 3-1). In unconfined aquifers the piezometric surface coincides with the upper surface of the zone of saturation, called the water table or phreatic surface, where the pressure is atmospheric (Figure 3-2).



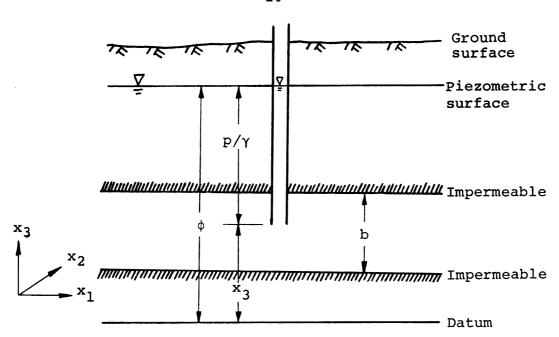


Figure 3-1.--Definition of Piezometric Head in a Confined Aquifer.

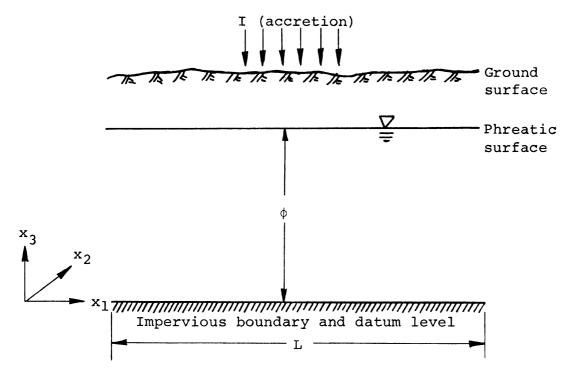
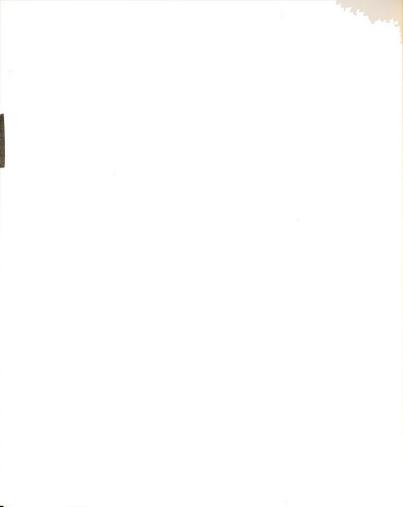


Figure 3-2.--Vertical Cross Section of a Phreatic Aquifer.



3.3 Regional Groundwater Flow

For regional problems, two-dimensional horizontal flow is considered. The governing equations are well established [e.g., see Bear 1972, Pinder and Frind 1972].

3.3.1 Basic Assumptions

 $\label{eq:the_problem} \mbox{The following assumptions are valid for regional} \\ \mbox{groundwater flow:}$

- (a) The flow is essentially horizontal in a twodimensional plane. This assumption is valid when the variation of thickness of the aquifer is much smaller than the thickness itself. This approximation fails in regions where the flow has a vertical component.
- $\begin{tabular}{ll} \begin{tabular}{ll} \beg$
- (c) The aquifer is elastic and generally nonhomogeneous and anisotropic. The consolidating medium deforms during flow due to changes in effective stress with only vertical compressibility being considered.
- (d) For the two-dimensional horizontal flow assumption, an average piezometric head is used where the average is taken along a vertical line extending from the bottom to the top of the aquifer, i.e.,

$$\phi_{av}(x_1, x_2, t) = \frac{1}{b} \begin{cases} b & \phi(x_1, x_2, x_3, t) dx_3 \\ x_3 = 0 & 0 \end{cases}$$

where b is the thickness of the aquifer.



3.3.2 Two-Dimensional Horizontal Flow

The combined equation of motion and continuity for flow in a two-dimensional horizontal plane may be written

$$\frac{\partial}{\partial \mathbf{x_i}} \left[\mathbf{T_{ij}} \frac{\partial \phi}{\partial \mathbf{x_j}} \right] - \mathbf{P} + \mathbf{I} = \mathbf{S} \frac{\partial \phi}{\partial \mathbf{t}} \qquad i, j=1, 2$$
 (3.3.1)

where T_{ij} is the transmissivity tensor equal to the aquifer thickness multiplied by the hydraulic conductivity K_{ij} , S is the storage coefficient, t is time, I is the vertical recharge or infiltration into the aquifer, and P is strength of a sink (or source) function $[Pinder \ and \ Frind \ 1972]$ defined by

$$\mathbf{P} = \sum_{m=1}^{M} \; \mathbf{P_w} [\; (\mathbf{x_1})_m, (\mathbf{x_2})_m] \, \delta \, [\mathbf{x_1} - (\mathbf{x_1})_m] \, [\mathbf{x_2} - (\mathbf{x_2})_m]$$

where P_W is the discharge (or recharge) from the aquifer, M is the number of nodes in one element (details are given in Chapter IV), and δ is the Dirac delta function.

3.3.3 Initial and Boundary Conditions

3.3.3.1 Boundary conditions. -- In order to solve a partial differential equation describing a physical phenomenon, it is necessary to choose certain additional conditions imposed by the physical situation at the



boundaries (S) for the domain (D) under consideration. In general the equation for the boundary condition can be written

where ℓ_1 are the directional cosines, and β_1 , β_2 , and β_3 are given functions of position and possibly time. For flow through an aquifer, three different boundary conditions are applicable:

 (a) Dirichlet or prescribed potential: In this case the potential is specified for all points along the boundary

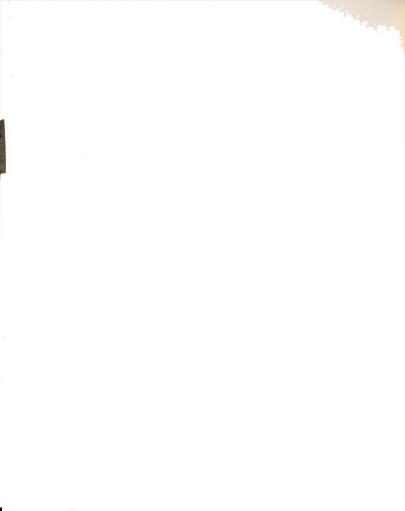
$$\phi = -\frac{\beta_3}{\beta_2} \; ; \qquad \qquad \beta_2 \neq 0$$

(b) Neumann or prescribed flux: Along a boundary of this type, the flux normal to the boundary surface is prescribed for all points of the boundary as a function of position and time

$$T_{ij} \frac{\partial \phi}{\partial x_i} \ell_i = -\frac{\beta_3}{\beta_1}$$
 on S; $\beta_1 \neq 0$

A special case of the Neumann condition is the impervious boundary where the flux vanishes everywhere on the boundary, i.e.,

$$\beta_3 = 0$$



(c) Cauchy boundary: This problem occurs when the potential and its normal derivative are prescribed on the boundary in the combined form, and the entire Equation (3.3.2) is used. Different forms of Equation (3.3.2) for three types of boundary conditions are summarized in Table 3-1.

In general, for a flow problem one will have mixed boundary conditions in which the Dirichlet condition will apply over a part of the boundary and the Neumann condition will be specified for the remaining portion [Bear 1972].

3.3.3.2 Initial conditions.—At the initial time, either the piezometric heads are known in the entire domain (D) or the hydrologic stresses (such as pumping and recharge) are specified and boundary conditions are known. For the second case the system has reached the steady state, so the solution of the equation

$$\frac{\partial}{\partial \mathbf{x_i}} \left[\mathbf{T_{ij}} \frac{\partial \phi}{\partial \mathbf{x_j}} \right] - \mathbf{P} + \mathbf{I} = 0 \qquad \text{i,j=1,2}$$
 (3.3.3)

will yield piezometric heads for the initial time. The procedure of solving Equation (3.3.3) is discussed in Chapter VI.



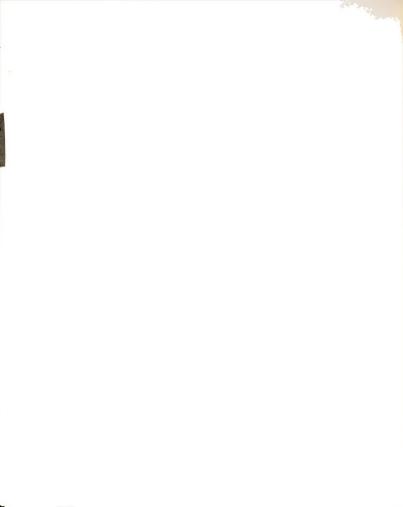
TABLE 3-1. -- Boundary Conditions.

The general equation for the boundary condition can be written

$$\beta_1 T_{ij} \frac{\partial \phi}{\partial x_j} \delta_i + \beta_2 \phi + \beta_3 = 0$$

where ℓ_1 are directional cosines; β_1 , β_2 and β_3 are known functions of space and time.

Remark		If $\beta_3=0$, the flow on the boundary is zero.	
Equation	$\neq 0$ $\phi = -\frac{\beta_3}{\beta_2} = f(\mathbf{x_i}, t)$	$T_{ij} \frac{\partial \phi}{\partial x_j} k_i = -\frac{\beta_3}{\beta_1}$	$T_{ij} \frac{\partial \phi}{\partial x_j} \ell_i + \frac{\beta_2}{\beta_1} \phi = -\frac{\beta_3}{\beta_1}$
β2	0 ≠	0	0 ≠
81	0	0 ≠	0 *
Name	Dirichlet	Neumann	Cauchy
Туре	٦	7	m



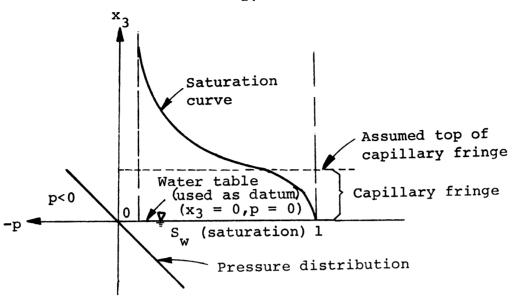
3.4 Phreatic Aquifer With Accretion

A typical cross section of a phreatic aquifer with accretion is illustrated in Figure 3-2. The governing equations are discussed in literature [e.g., Bear 1972, France 1974].

3.4.1 Basic Assumptions

- 1. Usually immediately above the water table (p = 0) there is a zone that is saturated or nearly so. This nearly saturated zone above the phreatic surface is called the capillary fringe or capillary rise, where the pressure is negative. In Figure 3-3 a typical saturation curve and pressure distribution in the capillary fringe at equilibrium are shown. The capillary rise might range from 2-5 cm for coarse sand up to greater than 200 cm for clay [Bear 1972]. In this study it is assumed the aquifer is fully saturated and the capillary fringe can be ignored. The resulting idealized diagram is given in Figure 3-3b.
- 2. When the saturated soil is being drained, the free surface gradually descends and some water is removed from the soil profile. In practice, the amount of water removed per unit volume of soil depends upon the water level, rate of drawdown, temperature, and atmospheric pressure; but for theoretical analysis it is usually taken as a constant and equal to the specific





(a) [After Bear 1972]

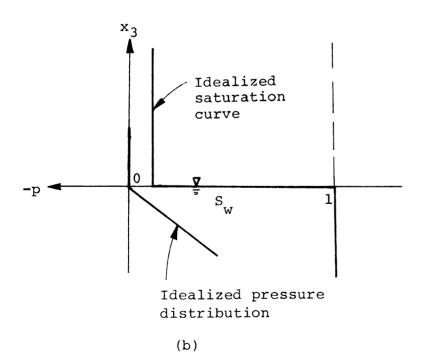
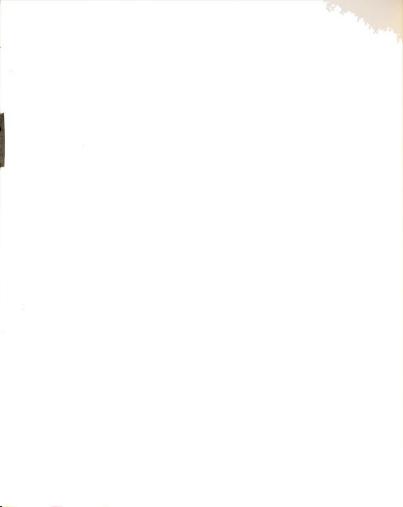


Figure 3-3.--Saturation Curve and Pressure Distribution in the Capillary Zone. (a) Actual form, (b) idealized form.



yield. The specific yield S_y is thus defined to be the volume of water drained over the gross volume of the porous medium. Quantitative information on specific yield is given by Todd [1959, pp. 23-26]. Because of the assumptions made in the above paragraph, in this study the magnitude of the specific yield is assumed constant and equal to the effective porosity which is defined to be the volume of water drained by gravity from a unit volume of saturated soil.

3. In unconfined aquifers the amount of water released from storage is usually small compared to the water available from normal movement through the aquifer and accretion, thus elastic specific storage (S_8) can be ignored [Herbert 1968; Neuman and Witherspoon 1971].

3.4.2 Governing Differential Equations

Consider an unconfined aquifer with its phreatic surface depicted schematically in Figure 3-2. The governing differential equation can be written

$$\frac{\partial}{\partial x_{i}} \left(K_{ij} \frac{\partial \phi}{\partial x_{j}} \right) = S_{s} \frac{\partial \phi}{\partial t} \qquad \text{i,j=1,3}$$
 (3.4.1)

where $\mathbf{S}_{\mathbf{S}}$ is the elastic specific storage, and the other terms are defined previously. As discussed above, if $\mathbf{S}_{\mathbf{S}}$ can be neglected then Equation (3.4.1) will have the form:



$$\frac{\partial}{\partial \mathbf{x}_{i}} \left[\mathbf{K}_{ij} \frac{\partial \phi}{\partial \mathbf{x}_{j}} \right] = 0 \qquad i,j=1,3 \qquad (3.4.2)$$

3.4.3 Boundary and Initial Conditions

Let D represent the flow region, which in general may possess up to four kinds of boundary conditions: in addition to the Dirichlet and Neumann conditions, a phreatic (free surface) boundary and a seepage face.

Referring to Figure 3-2, the following boundary conditions can be written:

1. At \mathbf{x}_1 = 0 and \mathbf{x}_1 = L, the piezometric heads are known functions of time (Dirichlet)

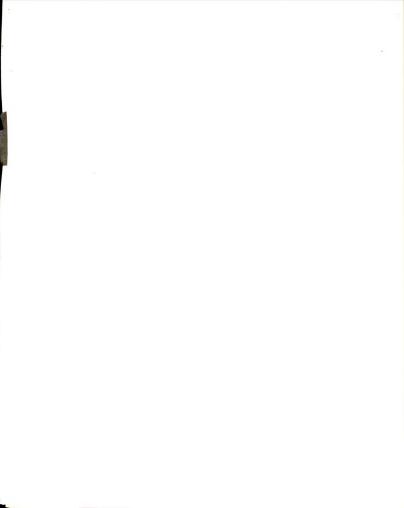
$$\phi(0, x_3, t) = 0$$

$$\phi(L, x_3, t) = 0 \qquad (3.4.3)$$

2. On the impervious boundary B, the normal flow is zero

$$K_{ij} \xrightarrow{\partial \phi (\mathbf{x}_1, 0, t)} \lambda_i = 0 \quad i, j=3 \quad 0 \le x_1 \le L$$
 (3.4.4)

3. In the concept of successive changes of steady-state values, it is assumed the flow at each instant is steady but its boundary condition is time variable [Polubarinova-Kochina 1962]. Therefore, the flow rate for a small time increment is equal to the



change of volume filled with fluid divided by the time interval. Consider Figure 3-4 where the position of the phreatic surface at times t and t + Δt is shown. \boldsymbol{I}_n represents the rate of accretion normal to the phreatic surface. By taking a control volume in the direction of the unit normal between two successive positions of the boundary at times t and t + Δt , and writing the continuity relation, one arrives at

$$U_n n_e = q_n - I_n$$
 (3.4.5)

 $\mathbf{U}_{\mathbf{n}}$ is the propagation of the phreatic surface, $\mathbf{q}_{\mathbf{n}}$ is the seepage flux, both along the unit normal, and $\mathbf{n}_{\mathbf{p}}$ is the

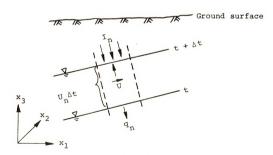


Figure 3-4.--Phreatic Boundary With Accretion [Adapted from Todsen 1971].

effective porosity. Equation (3.4.5) is a nonlinear boundary condition for Equation (3.4.2) because it contains an unknown dependent variable, i.e., ϕ , in the flux term. As will be discussed in detail in Chapter VI, by assuming that at the beginning of each time increment the piezometric heads are known at the phreatic surface, Equation (3.4.5) is linearized. Equation (3.4.5) after multiplying by Δt can be written as

$$U_{j} \ell_{i} \Delta t = U_{n} \Delta t = \frac{\Delta t}{n_{e}} \left(-K_{ij} \frac{\partial \phi}{\partial x_{j}} \ell_{i} - I \cdot \ell_{3}\right) (3.4.6)$$

$$i, j=1,3$$

where I is the accretion (positive downward), and U_j is the velocity of propagation of the free surface at the point of consideration on which the pressure is maintained atmospheric. Todsen [1971] has also derived Equation (3.4.5).

4. On the seepage face,

$$\phi = X_3$$

Initially, the surface configuration and the boundary conditions are known.

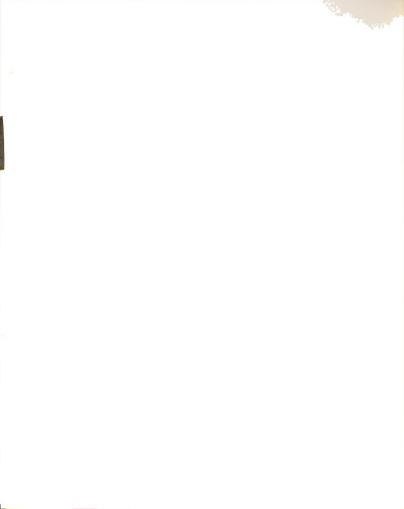
3.5 Convective-Dispersion Phenomena

In this study the movement of a solute in a saturated flow through a porous medium is considered.

This solute will be referred to as a "tracer." The symbol C will be used to denote the concentration of a tracer, i.e., mass of tracer per unit volume of solution. The term tracer will be used to represent any species of interest in a solution.

3.5.1 Basic Assumptions

- 1. It is assumed that no chemical reactions occur between the water and the aquifer or soil material that affect the tracer concentration.
- 2. The porous medium is homogeneous and isotropic with respect to dispersivity.
 - 3. The flow regime is laminar.
- 4. In general, variations in tracer concentration cause changes in the density and viscosity of the liquid. These in turn affect the flow regime (i.e., velocity distribution). At relatively low concentrations it is assumed that the concentration does not affect the liquid properties [Bear 1972]. This assumption leads to the following conclusions:
 - a. the viscosity is constant,
 - b. the concentration does not affect the velocity distribution.



3.5.2 Hydrodynamic Dispersion Coefficient in a Porous Medium

Hydrodynamic dispersion is the macroscopic outcome of the actual movements of individual tracer particles through the pores and includes two processes [Bear 1972]. One mechanism is mechanical dispersion, which depends on both the flow of the fluid and the characteristics of the porous medium through which the flow takes place. The second process is molecular diffusion which basically results from variations in tracer concentration within the liquid phase, and is more significant at low velocities (e.g., less than 1 cm/hr). Thus the coefficient of hydrodynamic dispersion D_{ij}^* includes the effect of both the mechanical (or convective) dispersion D_{ij}^* and molecular diffusion $(\mathrm{D}_{ij}^*)_{ij}^*$. Hence

$$D'_{ij} = D_{ij} + (D_d^*)_{ij}$$
 (3.5.1)

In Equation (3.5.1), $(D_d^*)_{ij} = T_{ij}^*D_d$, where D_d is the molecular diffusivity and T_{ij}^* is the medium's tortuosity. For homogeneous and isotropic media the value of T_{ij}^* is approximately equal to 2/3 [Bear 1972, pp. 109-112]. For most situations the contribution of molecular diffusion to hydrodynamic dispersion is negligible when compared to the mechanical dispersion coefficient. For example, for a gravel with seepage velocity ranging from



0.1 to 0.45 cm/sec, the magnitude of the dispersion coefficient varies from 0.01 to 0.08 cm²/sec [Rumer 1962]. The molecular diffusivity for solutes in water is very small and in the range of 0.5 to 4.0×10^{-5} cm²/sec [Welty 1969, p. 461].

Many investigators have attempted to model the dependence of the hydrodynamic dispersion coefficient on media, fluid properties, and flow characteristics, in order to understand the dispersion process in flow through porous media. A comprehensive discussion of the factors affecting the dispersion coefficient can be found in Bear [1972, pp. 605-616].

The mechanical dispersion coefficient for an isotropic medium in Cartesian coordinates can be written [Bear 1972] as:

In Equation (3.5.2) a_I and a_{II} are the longitudinal and transversal dispersivities of the medium, respectively, V_i and V_j are components of the seepage velocity in the i and j directions, V is the magnitude of the velocity, and δ_{ij} is the Kronecker delta. Its value is one when i=j and is zero, otherwise. Equation (3.5.2) is commonly used by investigators to calculate the mechanical



dispersion coefficient and hence is utilized in this study. It includes the major parameters causing the mechanical dispersion, and for any practical study it is assumed adequate.

3.5.3 Convective-Dispersion Equation in Cartesian Coordinates

The equation describing the mass transport and dispersion of dissolved chemical constituents in a saturated porous medium may be written as

$$\begin{bmatrix} \frac{\partial \mathbf{C}}{\partial \mathbf{t}} + \frac{\partial}{\partial \mathbf{x_i}} & (\mathbf{V_i \cdot c}) \end{bmatrix} - \begin{bmatrix} \frac{\partial}{\partial \mathbf{x_i}} & \left[\mathbf{D_{ij}^t} \frac{\partial \mathbf{C}}{\partial \mathbf{x_j}} \right] + \dot{\mathbf{q}_c} \end{bmatrix} = 0 \qquad (3.5.3)$$

$$(1) \qquad (2) \qquad (3) \qquad (4)$$

$$\mathbf{i}, \mathbf{j} = 1, 2, 3$$

where C is the mass concentration of the tracer; D_{ij}^{l} is the coefficient of hydrodynamic dispersion, discussed in Section 3.5.2; V_{i} is the component of seepage velocity; \dot{q}_{c} is the mass flux of source or sink; and x_{i} is the Cartesian coordinate. The theoretical basis and the derivation of the diffusion-convection equation are discussed in detail by Reddell and Sunada [1970], Bear [1972], and Bredehoeft and Pinder [1973]. In Equation (3.5.3) the first term represents the time rate of change of the tracer concentration. The second term describes the convective transport of C in the x_{i} -direction, which is proportional to the seepage velocity.



The third term is the transport (redistribution) of C due to dispersion and molecular diffusion. Finally, the last term represents the time rate of production or decay of C.

The convective-dispersion equation is a nonlinear partial differential equation of parabolic type. The relation is nonlinear because of the convective term, and because of the transport coefficient which is a function of the dependent variable V. The convective term $((3/3x_{\underline{i}})(V_{\underline{i}}C)) \text{ is nonsymmetric and has been a principal source of difficulty in the numerical solution of the convective-dispersion equation [Guymon et al. 1970].}$

3.5.4 Initial and Boundary Conditions

3.5.4.1 Boundary conditions. -- The general equation of the boundary conditions for the mass transfer equation is similar to the flow equation. As discussed in Section 3.3.3, it can be written:

$$\alpha_{1}D_{ij}^{\dagger} \frac{\partial C}{\partial x_{i}} \lambda_{i} + \alpha_{2}C + \alpha_{3} = 0$$
 (3.5.4)

where α_1 , α_2 , and α_3 are known functions. Three different boundary conditions are:

(a) Dirichlet or prescribed concentration boundary condition:



$$C = -\frac{\alpha_3}{\alpha_2}$$
, $\alpha_1 = 0$; $\alpha_2 \neq 0$

(b) Neumann or prescribed flux:

$$D_{ij}^{\prime} \frac{\partial C}{\partial x_{j}} \ell_{i} = -\frac{\alpha_{3}}{\alpha_{1}}, \quad \alpha_{1} \neq 0 ; \quad \alpha_{2} = 0$$

for α_3 = 0, one has the no-flow boundary.

(c) Cauchy boundary:

$$D_{ij}^{!} \frac{\partial C}{\partial x_{j}} \ell_{i} + \frac{\alpha_{2}}{\alpha_{1}} C = -\frac{\alpha_{3}}{\alpha_{1}}, \alpha_{1} \text{ and } \alpha_{2} \neq 0$$

Again, as in the flow situation, usually along the boundary one has mixed boundary conditions, i.e., the Dirichlet condition applies over a part of the boundary and the Neumann condition applies over the remaining part.

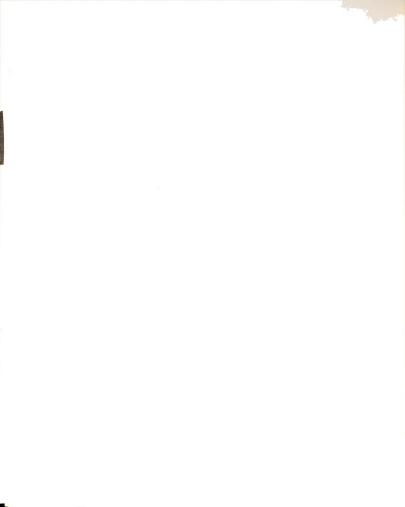
3.5.4.2 Initial conditions. -- As an initial condition, the concentration distribution at some initial time t=0 at all points of the flow domain must be specified:

$$C(x_i,0) = f_1(x_i)$$

where f_1 is a known function of x_i .

3.6 Closure

In this chapter the mathematical description of flow and dispersion phenomena in porous media is presented, and basic assumptions are introduced. Using the concept of successive changes of steady-state values, the nonlinear boundary condition for an unconfined aquifer with a phreatic surface is linearized. For practical purposes the hydrodynamic dispersion coefficient can be replaced by the mechanical dispersion coefficient in predicting the tracer movement. Finally, the flow and convective-dispersion equations can be solved consecutively. In the following chapter, the finite element formulation of the flow and mass transfer equations is given.



CHAPTER IV

FINITE ELEMENT FORMULATION

The finite element method is a numerical technique which is used to approximate a continuous partial differential equation in a given domain D with specified boundary conditions along boundaries S. The key features of the finite element concept are [Norrie and de Vries 1973]:

- The domain is divided into subdomains or finite elements, usually of the same form.
- The trial solution is prescribed (functionally) over the domain in a piecewise fashion, element by element.

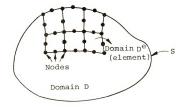
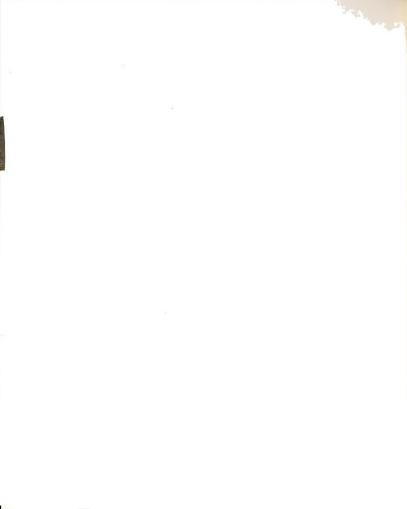


Figure 4-1.--Domain Divided Into Finite Elements.



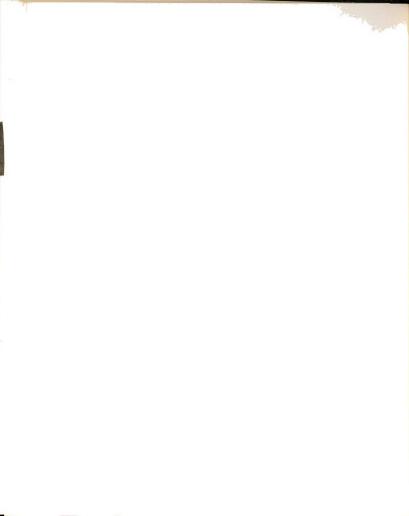
A detailed formulation of the finite element method is given by Zienkiewicz [1971], Norrie and de Vries [1973]. This technique has been utilized by several investigators [Javandel and Witherspoon 1968; Pinder and Frind 1972; Neuman and Witherspoon 1971; Desai 1972; Cheng and Li 1973; and France 1971, 1974] to solve transient flow problems in a confined or unconfined aquifer. Recently the finite element procedure was also used to solve the convective-dispersion equation [Cheng 1973, Pinder 1973, and Segol et al. 1975].

In this chapter a brief discussion of the Galerkin based finite element technique is given and the method is used to discretize the space derivatives of the flow and dispersion equations. The simultaneous solution of velocity vectors is also described, that is, the Galerkin formulation of the Darcy law is constructed and velocity components are calculated at the nodes.

4.1 The Galerkin Finite Element Method

While the approximate minimization of a functional is the most widely accepted means of arriving at a finite element representation, it is by no means the only possible approach. The Galerkin method offers an alternative way to formulate a problem for the finite element solution without using variational principles.

In the finite element technique the domain D is divided into subdomains $\mathbf{D}^{\mathbf{e}}$ which are called elements.



Each element is designated by nodes. In this chapter NELS represents the number of elements, M is the number of nodes in each element, and NNDS stands for the total number of nodes in domain D (Figure 4-1).

Consider a problem of solving approximately a set of differential equations in which the unknown function {C} has to be satisfied in the domain D with the boundary conditions specified along S. The governing equation can be written

$$f(\{C\}) = 0$$

Let the trial solution for this equation be C

$$\hat{C} = [N] \{C\} = \sum_{n=1}^{M} N_n C_n$$
 (4.1.1)

where [N] = [N(x_i)] are shape functions (prescribed functions of coordinates) and {C} = {C(t)} is a set of M unknown parameters. In general, the equation of residual (or error) is formed in the following way:

$$R = f_{D^{e}}(\{C\}) - f_{D^{e}}(\{\hat{C}\}) = -f_{D^{e}}(\{\hat{C}\}) \neq 0 \quad (4.1.2)$$

The best solution will be one in which the residual R has the least value at all points in the domain D^e . An obvious way to achieve this [Zienkiewicz 1971] is to make use of the fact that if R is identically zero elsewhere, then

$$\int_{D^{e}} W R dD = 0$$
 (4.1.3)

where W is any function of the coordinates. If the number of unknown parameters $\{C\}$ is NNDS and NELS linearly independent functions W_k are chosen, one can write a suitable number of simultaneous equations as

$$\int_{D^{e}} W_{k} R dD = \int_{D^{e}} W_{k} f([N]\{C\}) dD = \{0\}$$
 (4.1.4)

$$k = 1, ..., M$$

where W_k is called the weighting function. If the shape function N_k is to be chosen as the weighting function, the process is termed the Galerkin procedure. The element equations can be assembled by

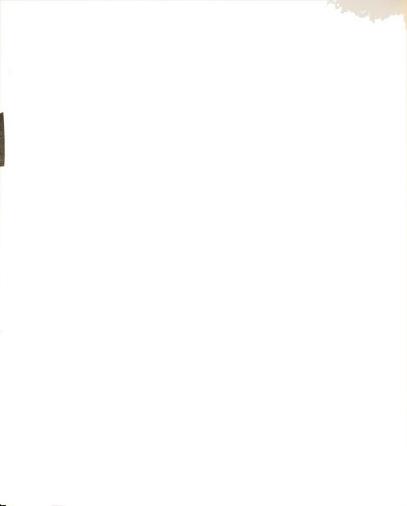
$$\sum_{e=1}^{NELS} \left(\int_{D^e} W R dD \right) = 0$$
 (4.1.5)

to yield the global relations for domain D.

4.2 Finite Element Formulation of Flow Equation

The residual equation for flow in a confined horizontal aquifer (Equation 3.3.1) can be written as

$$R = S \frac{\partial \hat{\phi}}{\partial t} + P - \frac{\partial}{\partial x_{i}} \left(T_{ij} \frac{\partial \hat{\phi}}{\partial x_{j}} \right) \quad i, j=1, 2 \quad (4.2.1)$$



The symbol $\hat{}$ represents the numerical approximation of ϕ . Substituting Equation (4.2.1) into Equation (4.1.4), one obtains

$$\int_{D_{\mathbf{c}}} \left[\mathbf{S} \frac{\partial \hat{\phi}}{\partial \mathbf{t}} + \mathbf{P} - \frac{\partial}{\partial \mathbf{x}_{i}} \left[\mathbf{T}_{ij} \frac{\partial \hat{\phi}}{\partial \mathbf{x}_{j}} \right] \right] \mathbf{N}_{k} dD = 0 \qquad k=1, \dots M \qquad (4.2.2)$$

By use of the Green theorem, the third term can be modified

$$\int_{D} e^{\frac{\partial}{\partial x_{i}}} \left(T_{ij} \frac{\partial \hat{\phi}}{\partial x_{j}} \right) N_{k} dD = - \int_{D} e^{-T_{ij}} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial \hat{\phi}}{\partial x_{j}} dD$$

$$+ \int_{S} e^{-N_{k}} T_{ij} \frac{\partial \hat{\phi}}{\partial x_{j}} \ell_{i} dS (4.2.3)$$

The last term in Equation (4.2.3) is nonzero only for elements which contain the Neumann flux boundary condition

$$-\int_{S^{e}} N_{k} T_{ij} \frac{\partial \hat{\phi}}{\partial x_{j}} \ell_{i} dS = \int_{S^{e}} N_{k} Q_{2} dS \qquad (4.2.4)$$

where Q_2 is known flux along the boundary. Substituting Equation (4.2.3) and Equation (4.1.1) into Equation (4.2.2) and rearranging the terms, one obtains

$$\int_{D} e^{-\phi_{n}} T_{ij} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial N_{n}}{\partial x_{j}} dD + \int_{D} e^{-S} N_{k} N_{n} \frac{\partial \phi}{\partial t} dD + \int_{D} e^{-N_{k}} P dD$$

$$+ \int_{S} e^{-N_{k}} Q_{2} dS = 0 \qquad (4.2.5)$$

•			

Since ϕ_n and its time derivatives are independent of the coordinates, they can be taken out of the integrals. Equation (4.2.5) can be written in matrix form

$$[B]^{e} \{\phi\}^{e} + [H]^{e} \frac{\partial}{\partial f} \{\phi\}^{e} = \{F\}^{e}$$
 (4.2.6)

where

$$[B]^{e} = B_{kn}^{e} = \int_{D^{e}} \left[T_{ij} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial N_{n}}{\partial x_{j}} \right] dD$$

$$[H]^{e} = H_{kn}^{e} = \int_{D^{e}} S N_{k} N_{n} dD$$

$$\{F\}^{e} = F_{k}^{e} = -\int_{S^{e}} N_{k} Q_{2} dS - \int_{D^{e}} P N_{k} dD$$

$$(4.2.7a)$$

$$i, j=1, 2$$

$$(4.2.7c)$$

It is assumed that the storage coefficient is constant throughout the element and that the element coordinate axes coincide with the principal direction of the transmissivity tensor: the transmissivity can be defined either at the nodes or at each element. Evaluation of Equation (4.2.7) for different types of elements is discussed in Appendix II. Upon evaluation of Equation (4.2.7) for all elements and transformation to a global coordinate system, they are assembled by virtue of Equation (4.1.5) into a global relationship

[B]
$$\{\phi\}$$
 + [H] $\{\frac{\partial \phi}{\partial t}\}$ = $\{F\}$ (4.2.8)



The parameter $\{\phi\}$, matrices [B] and [H], and force vector $\{F\}$ are the summation of the corresponding terms in Equation (4.2.7) over all the elements in the domain D. [B] and [H] are banded symmetric matrices. Equation (4.2.8) is a set of first order linear differential equations with unknowns $\{\phi\}$ and can be solved simultaneously at the given nodes in the space domain. The solution of Equation (4.2.8) and a similar equation which is generated from the finite element formulation of the convective-dispersion equation is presented in Chapter V.

The governing equation for the unconfined aquifer with phreatic surface is

$$\frac{\partial}{\partial \mathbf{x}_{i}} \left(\mathbf{K}_{ij} \frac{\partial \phi}{\partial \mathbf{x}_{j}} \right) = 0 \qquad i, j=1, 3$$
 (3.4.2)

The finite element formulation has the form

$$[B']^{e} \{\phi\}^{e} = \{F\}^{e}$$
 (4.2.9)

where

$$[B']^{e} = \int_{D^{e}} K_{ij} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial N_{n}}{\partial x_{j}} dD \quad i,j=1,3 \\ k,n=1,...M \quad (4.2.10a)$$

$$\{F\}^{e} = -\int_{S^{e}} N_{k} Q_{2} dS$$
 (4.2.10b)

Assembling the element matrices leads to a system of equations in the form

$$[B] \{ \phi \} = \{ F \}$$
 (4.2.11)



4.3 Finite Element Formulation of Convective-Dispersion Equation

The residual of Equation (3.5.3) for each element can be written

$$R = \frac{\partial \hat{C}}{\partial t} + \frac{\partial}{\partial x_{i}} (\hat{C} V_{i}) - \frac{\partial}{\partial x_{i}} \left(D_{ij}^{!} \frac{\partial \hat{C}}{\partial x_{j}} \right) - \hat{q}_{C}^{!} \quad i,j=1,2 \quad (4.3.1)$$

Again, \hat{C} represents the numerical approximation of C.

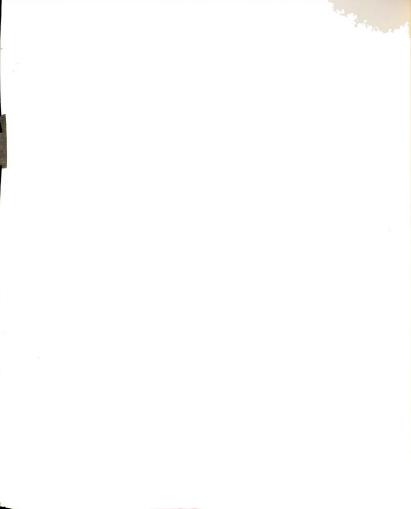
In this development it is assumed that at every small time step the velocity vectors $\mathbf{V_i}$ and the dispersion coefficient $\mathbf{D_{ij}}$ are known functions which are either determined independently or are replaced by values of $\mathbf{V_i}$ and $\mathbf{D_{ij}}$ from the previous time step. Using Equation (4.1.1), $\hat{\mathbf{C}}$ and $\hat{\mathbf{q}}_{\mathbf{C}}$ can be written

$$\hat{C} = \sum_{n=1}^{M} N_n C_n = [N_n] \{C_n\}$$
 (4.3.2)

$$\hat{q}_{c} = \sum_{n=1}^{M} N_{n} (\dot{q}_{c})_{n} = [N_{n}] \{ (\dot{q}_{c})_{n} \}$$
(4.3.3)

In order that \hat{C} be an exact solution of Equation (3.5.3), Equation (3.5.3) must be identically zero when \hat{C} is substituted into it. To minimize the errors of residuals as discussed in Section 4.1, the orthogonality condition requires that

$$\int_{D^{e}} R(\hat{C}) N_{k} dD = \int_{D^{e}} R([N_{n}]\{C_{n}\}) N_{k} dD = 0 \quad (4.3.4)$$



where $D^{\mathbf{e}}$ represents the integral over the domain of an element. Substituting Equation (4.3.1) into Equation (4.3.4) yields

$$\int_{D^{\mathbf{c}}} \left\{ \frac{\partial \hat{\mathbf{c}}}{\partial t} + \frac{\partial}{\partial \mathbf{x_{i}}} \left(\hat{\mathbf{c}} \ \mathbf{v_{i}} \right) \right. \\ \left. - \frac{\partial}{\partial \mathbf{x_{i}}} \left[\mathbf{c}_{i,j}^{t} \ \frac{\partial \hat{\mathbf{c}}}{\partial \mathbf{x_{j}}} \right] - \hat{\mathbf{q}}_{\mathbf{c}} \right\} N_{\mathbf{k}} \ \mathbf{dD} = 0 \tag{4.3.5}$$

By use of the Green theorem,

$$\int_{D^{\mathbf{e}}} \left\{ \frac{\partial}{\partial \mathbf{x}_{i}} \left[D_{ij}^{!} \frac{\partial \hat{\mathbf{c}}}{\partial \mathbf{x}_{j}} \right] N_{k} dD \right\} = - \int_{D^{\mathbf{e}}} D_{ij}^{!} \frac{\partial N_{k}}{\partial \mathbf{x}_{i}} \frac{\partial \hat{\mathbf{c}}}{\partial \mathbf{x}_{j}} dD$$

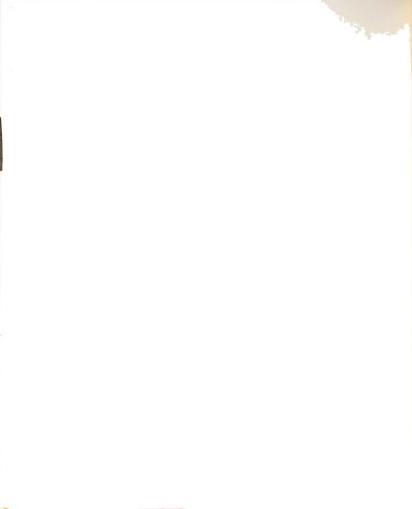
$$+ \int_{\mathbf{c}^{\mathbf{e}}} N_{k} D_{ij}^{!} \frac{\partial \hat{\mathbf{c}}}{\partial \mathbf{x}_{i}} \hat{\mathbf{x}}_{i} dS \quad (4.3.6)$$

where ℓ_1 is the directional cosine of the boundary at the node under consideration. Substituting Equation (4.3.6) into Equation (4.3.5), one arrives at

$$\begin{split} \int_{D^{\mathbf{e}}} \left[\frac{\partial \hat{\mathbf{c}}}{\partial t} + \frac{\partial}{\partial \mathbf{x}_{i}} (\hat{\mathbf{c}} \ \mathbf{v}_{i}) - \hat{\mathbf{q}}_{\mathbf{c}} \right] \mathbf{v}_{k} \ \mathrm{dD} + \int_{D^{\mathbf{e}}} \mathbf{D}_{ij}^{!} \frac{\partial \mathbf{N}_{k}}{\partial \mathbf{x}_{i}} \frac{\partial \hat{\mathbf{c}}}{\partial \mathbf{x}_{j}} \, \mathrm{dD} \\ - \int_{S^{\mathbf{e}}} \mathbf{N}_{k} \mathbf{D}_{ij}^{!} \frac{\partial \hat{\mathbf{c}}}{\partial \mathbf{x}_{j}} \, \hat{\mathbf{x}}_{i} \, \mathrm{dS} \end{split} \tag{4.3.7}$$

Expanding Equation (4.3.7) and substituting Equations (4.3.2) and (4.3.3) yields

$$\int_{D^e} {^{N}_{n}N_{k}} \ \frac{^{\partial C}_{n}}{^{\partial t}} \ ^{dD} \ + \ \int_{D^e} \left[{^{C}_{n}N_{n}N_{k}} \left(\frac{\partial v_{i}}{^{\partial x_{i}}} \right) \ + \ v_{i}c_{n}N_{k}} \left(\frac{\partial N_{n}}{^{\partial x_{i}}} \right) \right] dD$$



$$+ \int_{D^{e}} D_{ij}^{!} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial N_{n}}{\partial x_{j}} C_{n} dD = \int_{S^{e}} N_{k} D_{ij}^{!} \frac{\partial \hat{C}}{\partial x_{j}} \ell_{i} dS$$

$$+ \int_{D^{e}} (\dot{q}_{c})_{n} N_{n} N_{k} dD \quad (4.3.8)$$

As discussed in Section 3.5.4, the general equation for the boundary condition can be written as

$$\alpha_1 D_{ij}^{\dagger} \frac{\partial C}{\partial x_j} \ell_i + \alpha_2 C + \alpha_3 = 0$$
(3.5.4)

If $\alpha_1 = 0$, $C = \frac{\alpha_3}{\alpha_2} = f(x_i)$, provided $\alpha_2 \neq 0$

This boundary is the "Dirichlet" boundary. For nonzero α_1 , the Galerkin formulation of Equation (3.5.4) will be

$$\int_{S^{e}} N_{k} D_{ij}^{!} \frac{\partial \hat{C}}{\partial x_{j}} \ell_{i} dS = \int_{S^{e}} N_{k} \left[-\frac{\alpha_{2}}{\alpha_{1}} \hat{C} - \frac{\alpha_{3}}{\alpha_{1}} \right] dS =$$

$$- \int_{S^{e}} N_{k} \hat{C} \frac{\alpha_{2}}{\alpha_{1}} dS - \int_{S^{e}} N_{k} \frac{\alpha_{3}}{\alpha_{1}} dS \qquad (4.3.9)$$

Substituting Equation (4.3.9) into Equation (4.3.8) yields

$$\int_{D} e^{i\hat{N}_{n}} N_{k} \frac{\partial C_{n}}{\partial t} dD + \int_{D} e^{i\hat{N}_{n}} C_{n} N_{n} N_{k} \left(\frac{\partial V_{i}}{\partial x_{i}}\right) + V_{i} C_{n} N_{k} \left(\frac{\partial N_{n}}{\partial x_{i}}\right) dD$$

$$+ \int_{D} e^{i\hat{N}_{i}} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial N_{n}}{\partial x_{j}} C_{n} dD = - \int_{S} e^{i\hat{N}_{k}} C \frac{\alpha_{2}}{\alpha_{1}} dS$$



$$-\int_{S^{e}} N_{k} \frac{\alpha_{3}}{\alpha_{1}} dS + \int_{D^{e}} (\dot{q}_{C})_{n} N_{n} N_{k} dD \qquad (4.3.10)$$

Equation (4.3.10) may be written in matrix form [Cheng 1973]

$$[H]^{e} \{ \frac{\partial C}{\partial t} \} + ([K]^{e} + [S]^{e}) \{C\} + [E]^{e^{T}} [I] \{C\} =$$

$$[H]^{e} \{\dot{q}_{c}\} + \{F\}^{e}$$
 (4.3.11)

where

$$[H]^{e} = H_{k,n}^{e} = \int_{D^{e}} N_{k} N_{n} dD$$
 (4.3.12a)

$$[K]^{e} = K_{k,n}^{e} = \int_{D^{e}} D_{ij}^{i} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial N_{n}}{\partial x_{j}} dD \qquad (4.3.12b)$$

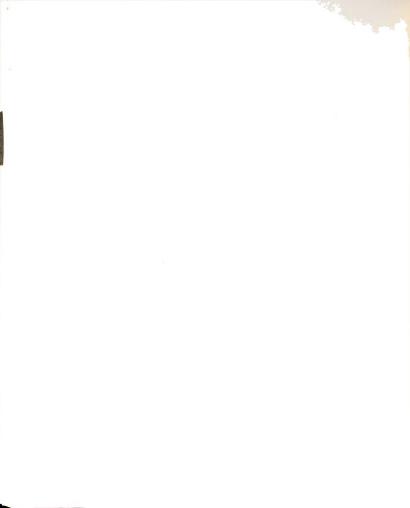
$$[S]^{e} = S_{k,n}^{e} = \int_{D^{e}} \left[N_{n} N_{k} \left(\frac{\partial V_{i}}{\partial x_{i}} \right) + V_{i} N_{k} \left(\frac{\partial N_{n}}{\partial x_{i}} \right) \right] dD \qquad (4.3.12c)$$

$$\{E\}^{e} = E_{k}^{e} = \int_{S^{e}} \frac{\alpha_{2}}{\alpha_{1}} N_{k} dS \qquad \alpha_{1} \neq 0$$
 (4.3.12d)

$$\{F\}^{e} = F_{k}^{e} = -\int_{S^{e}} \frac{\alpha_{3}}{\alpha_{1}} N_{k} dS \qquad \alpha_{1} \neq 0$$
 (4.3.12e)

If $\alpha_1 = 0$, then $E_k = F_k = 0$.

[I] is the identity matrix, and $[E]^{e^{T}}$ is the transpose of the vector $\{E\}^{e}$.



It is assumed that the velocity vectors are known at the nodes and that the dispersion coefficients are calculated for each integration point using Equation (3.5.1).

Calculation of Equation (4.3.12) for different types of elements is given in Appendix II. Upon evaluation of Equation (4.3.11) for all elements, they are assembled to obtain the global relations

$$[H] \{\frac{\partial C}{\partial t}\} + ([K]+[S])\{C\} + [E]^{T}[I]\{C\} = [H]\{\dot{q}_{C}\} + \{F\}$$
(4.3.13)

[H] and [K] are banded symmetric matrices, while [S] is a nonsymmetric matrix.

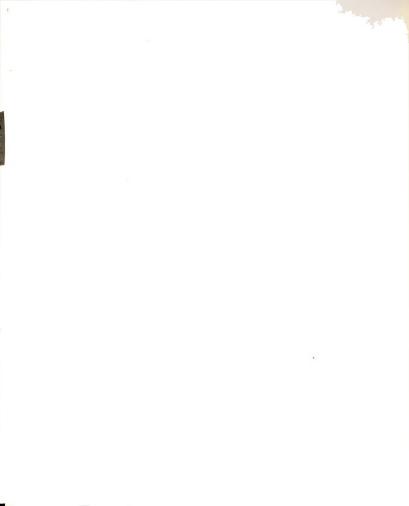
4.4 Finite Element Computation of Velocity Vectors

Mathematical equations of velocity vectors are discussed in Section 3.2. When the coordinate axes coincide with the principal direction of the hydraulic conductivity matrix, the Darcy equation can be written:

$$V_i = -\frac{K_{ij}}{n_e} \frac{\partial \phi}{\partial x_j}$$
 i,j=1,2,3; $K_{ij}=0$ when $i \neq j$ (4.4.1)

In two-dimensional Cartesian coordinates,

$$v_1 = -\frac{\kappa_{11}}{n_e} \frac{\partial \phi}{\partial x_1}$$
 (4.4.2a)



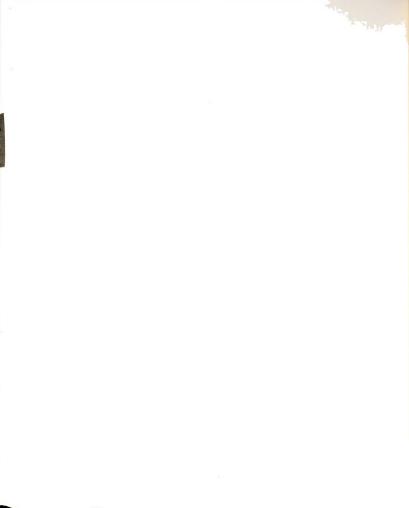
$$V_2 = -\frac{K_{22}}{n_p} \frac{\partial \phi}{\partial x_2}$$
 (4.4.2b)

Velocity components play an important role in predicting the tracer movement in a porous medium, because they appear in the convective term and are used in the calculation of dispersion coefficients. Special attention thus has to be given for the evaluation of the velocity vector. Two techniques are discussed below:

- 1. Direct calculation in this study is defined as the technique in which the velocity vectors are calculated using only the gradient of the shape functions multiplied by the corresponding piezometric heads, element by element, and is outlined in Section 4.4.1.
- 2. Simultaneous calculation is defined to be a procedure where the continuity of the velocity vectors is maintained and the gradient of the piezometric heads is multiplied by a weighted coefficient, as it will be shown in Section 4.4.2. The conjugate function concept is also used when referring to a similar method.

4.4.1 Direct Calculation

One of the most common methods to calculate the velocity vectors is to substitute Equation (4.1.1) into Equation (4.4.2). The resulting equation will be



$$V_1 = -\frac{K_{11}}{n_e} \frac{\partial N_n}{\partial x_1} \phi_n \qquad (4.4.3a)$$

$$v_2 = -\frac{\kappa_{22}}{n_0} \frac{\partial N_n}{\partial \kappa_2} \phi_n \qquad (4.4.3b)$$

where ϕ_n 's are piezometric heads at the nodes and are known. In Equation (4.4.3), $\partial N_n/\partial x_1$ and $\partial N_n/\partial x_2$ are first derivatives of the shape functions and are evaluated at the point of interest. Subscripts for x represent the coordinates, and subscripts for shape functions represent the node numbers. For example, for the element in Figure 4-2 with four nodes, V_1 and V_2 at point A are

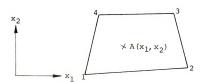


Figure 4-2.-- A Typical Finite Element.

$$\left. V_1 \right|_A \; = \; - \; \frac{\kappa_{11}}{n_e} \; \left[\frac{\partial N_1}{\partial x_1} \; \phi_1 + \frac{\partial N_2}{\partial x_1} \; \phi_2 + \frac{\partial N_3}{\partial x_1} \; \phi_3 + \frac{\partial N_4}{\partial x_1} \; \phi_4 \right]$$

$$\left. \boldsymbol{v_2} \right|_{\boldsymbol{A}} = \left. - \frac{\kappa_{22}}{n_3} \left[\frac{\partial \boldsymbol{N_1}}{\partial \boldsymbol{x_2}} \; \boldsymbol{\phi_1} + \frac{\partial \boldsymbol{N_2}}{\partial \boldsymbol{x_2}} \; \boldsymbol{\phi_2} + \frac{\partial \boldsymbol{N_3}}{\partial \boldsymbol{x_2}} \; \boldsymbol{\phi_3} + \frac{\partial \boldsymbol{N_4}}{\partial \boldsymbol{x_2}} \; \boldsymbol{\phi_4} \right]$$



with the $\partial N_n/\partial x_1$ and $\partial N_n/\partial x_2$ computed at point A. This procedure is easy to apply and provides accurate results at the centroid of the element, but the velocity components become discontinuous along the element edge as will now be demonstrated.

Consider linear rectangular elements as shown in Figure 4-3a. The piezometric head φ and its first derivative with respect to x_1 along lines AB_1 and B_2C are plotted in Figure 4-3. Figure 4-3b shows that the piezometric heads are continuous between two elements, and Figure 4-3C demonstrates how $\partial \varphi/\partial x_1$ is discontinuous between the elements. The gradient of φ takes different values depending on which element is used to compute its value. The discontinuity of the first or higher derivatives is also discussed by Zienkiewicz [1971]. To overcome the difficulties associated with discontinuity of the first derivative at the nodes, different techniques such as the conjugate function concept [Gallagher 1975] and Hermitian shape function [Zienkiewicz 1971] are used.

In the next section the technique of simultaneous calculation of the velocity vectors at the nodes is presented.

4.4.2 Simultaneous Calculation of Velocity Vectors

In order to provide continuous velocity functions between elements at the nodes, a smoothing technique,

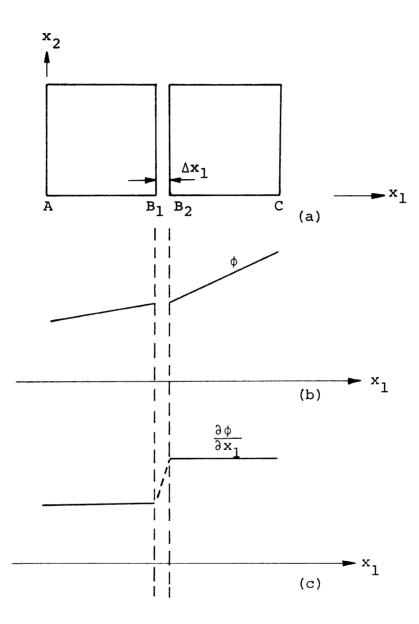


Figure 4-3.--Interelement Zone Depicting How Continuous Function φ May Have Discontinuous Gradients as $\Delta x_1 \to 0$.



namely, the Galerkin-based finite element formulation of Equations (4.4.2), is developed. The concept is similar to that presented by Zienkiewicz [1971, pp. 44-46] and also used by Segol et al. [1975]. In the proposed technique which is outlined below, the computer storage and the complication of the solution process are reduced considerably.

Let V_1^c , V_2^c represent the simultaneous calculation of V_1 and V_2 . The piezometric head and velocity vectors can be written in terms of nodal parameters.

$$\hat{\phi} = [N] \{\phi\} = \sum_{n=1}^{M} N_n \phi_n \qquad (4.4.4a)$$

$$\hat{V}_{1}^{C} = [N] \{V_{1}^{C}\} = \sum_{n=1}^{M} N_{n} (V_{1}^{C})_{n}$$
 (4.4.4b)

$$\hat{V}_{2}^{c} = [N] \{V_{2}^{c}\} = \sum_{n=1}^{M} N_{n} (V_{2}^{c})_{n}$$
 (4.4.4c)

Equation (4.4.4) will assure that continuity of the variables is maintained along the element boundaries, regardless of the type of elements used (by definition of the shape function). The technique of calculating the velocity vector along the x_1 -coordinate is presented below. The residual of Equation (4.4.2a) is

$$R(\hat{\mathbf{v}}_{1}^{\mathbf{c}}) = \hat{\mathbf{v}}_{1}^{\mathbf{c}} + \frac{\mathbf{K}_{11}}{\mathbf{n}_{\mathbf{c}}} \frac{\partial \hat{\phi}}{\partial \mathbf{x}_{1}}$$
 (4.4.5)



The definition of V_1 is given by Equation (4.4.2a) where K_{11} , n_e , and ϕ are known parameters. Substituting Equation (4.4.5) into Equation (4.1.4) yields

$$\int_{D^{\mathbf{e}}} N_{\mathbf{k}} \left[\hat{\mathbf{v}}_{1}^{\mathbf{c}} + \frac{K_{11}}{n_{\mathbf{e}}} \frac{\partial \hat{\boldsymbol{\phi}}}{\partial \mathbf{x}_{1}} \right] dD = 0$$
 (4.4.6)

Expanding Equation (4.4.6),

$$\int_{D^{e}} N_{k} \hat{V}_{1}^{c} dD + \int_{D^{e}} N_{k} \frac{K_{11}}{n_{e}} \frac{\partial \hat{\phi}}{\partial x_{1}} dD = 0 \qquad (4.4.7)$$

Introducing Equation (4.4.4) into Equation (4.4.7) gives

$$\int_{D} e^{-N_k N_n (V_1^C)} n^{dD} = -\int_{D} e^{-N_k \frac{K_{11}}{n_e}} \frac{\partial N_n}{\partial x_1} \phi_n dD \qquad (4.4.8)$$

Since $(V_1^C)_n$ is independent of the space coordinate, it can be taken out of the integral

$$\{(v_1^c)_n\} \int_{D^c} N_k N_n dD = -\frac{K_{11}}{n_e} \int_{D^c} N_k \frac{\partial N_n}{\partial x} \phi_n dD \qquad (4.4.9)$$

The right-hand side of Equation (4.4.9) is known and becomes a column matrix. Thus

$$[H]^{e} \{ (V_{1}^{c}) \}^{e} = \{F_{1}\}^{e}$$
 (4.4.10)

where

$$[H]^{e} = H_{k,n}^{e} = \begin{cases} \int_{D^{e}} N_{k} N_{n} dD & (4.4.11) \\ k, n=1, 2, ... M \end{cases}$$



$$\{F_1\}^e = (F_1)_k^e = -\frac{K_{11}}{n_e} \int_{D^e}^{N_k} \frac{\partial N_n}{\partial x_1} \phi_n dD$$
 (4.4.12)

Constructing a global matrix yields

$$[H] \{v_1^c\} = \{F_1\}$$
 (4.4.13)

[H] is a banded square symmetric matrix, and $\{F_1\}$ is a known column force. The solution of Equation (4.4.13) will give the x_1 -component of velocity at each node simultaneously. Similarly, for the x_2 -component of velocity the resulting equation will be

$$[H] \{V_2^C\} = \{F_2\}$$
 (4.4.14)

$$\{F_2\}^e = (F_2)_k^e = -\frac{K_{22}}{n_e} \int_{D^e} N_k \frac{\partial N_n}{\partial x_2} \phi_n dD$$
 (4.4.15)

The similarity of this method to the conjugate function concept is shown below. Let $\{V_1^C\}$ be the simultaneous solution of the x_1 -component of velocity and $\{V_1\}$ be the values of the velocity defined by Equation (4.4.3a) at specified points. Substituting Equation (4.4.3a) into Equation (4.4.8) yields

$$\{v_1^c\} \cdot \int_{D^c} v_k v_n dD = \int_{D^c} \{v_1\} v_k dD$$

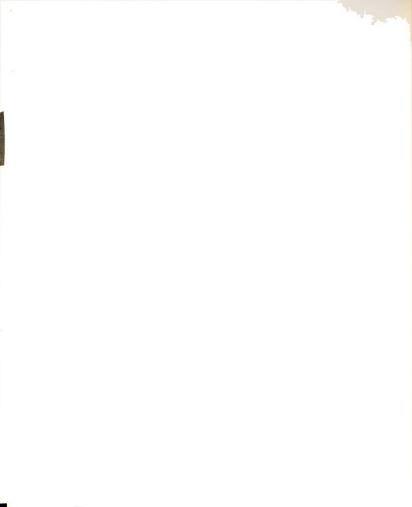
Constructing the global relations and solving for $\{\boldsymbol{v}_{1}^{c}\}$ yields



$$\{v_1^C\} = \left(\int_D \{v_1\} N_k dD\right) \left(\int_D N_k N_n dD\right)^{-1}$$
 (4.4.16)

This expression is identical to Oden's conjugate function concept [see Gallagher 1975, pp. 261-268]. Oden employed this concept to calculate the stress field. He used superscript c to represent the conforming of the stress field.

Oden states that the stress calculated by the above mentioned method minimizes the mean square error between the nonconforming stress field (in our case, V_1 , i.e., direct calculation of the velocity vector) and the conjugate stress field (in our case, V_1^{C} , i.e., simultaneous calculation of the velocity vector).



CHAPTER V

FINITE ELEMENT FORMULATION OF TIME DERIVATIVE

5.1 Introduction

The variation of transient field variables

described by the finite element method results in a system of first order linear differential equations:

[A]
$$\{C(t)\} + [H] \{\frac{\partial C(t)}{\partial t}\} = \{F(t)\}\$$
 (5.1.1a)

$$\{C(0)\} = C_0$$
 a known function (5.1.1b)

where [A] and [H] are usually banded symmetric or nonsymmetric matrices and $\{C(t)\}$ the unknown variable such as temperature, piezometric head, or mass concentration. In Equation (5.1.1), $\{\frac{\partial C(t)}{\partial t}\}$ is the time derivative of $\{C(t)\}$, and $\{F(t)\}$ is the known force vector. The symbol $\{\}$ represents a column matrix, and [] represents a square matrix. In Equation (5.1.1), t also can be replaced by other independent variables such as x_1 , x_2 , and x_3 .

The differential Equation (5.1.1) is normally integrated numerically using a finite difference method with the aid of a digital computer [Donea 1974]. In the



derivation of finite difference relations it is customary to expand the first or second order derivatives by the Taylor series expansion, and then by truncating higher order terms, find an approximation for the desired derivative [Carnahan et al. 1969]. In this chapter the finite element concept is used to obtain the finite difference relations for unknown C and its first and second derivatives. The results will have the following form:

$$C = \alpha_1 C(t+\Delta t) + \alpha_2 C(t) + \alpha_3 C(t-\Delta t) + \cdot \cdot \cdot$$
$$+ \alpha_r C(t+[1-m]\Delta t) \qquad (5.1.2)$$

$$\frac{\partial C}{\partial t} = \frac{1}{\Delta t} \left[\beta_1 C(t + \Delta t) + \beta_2 C(t) + \beta_3 C(t - \Delta t) + \cdots \right]$$

$$+ \beta_r C(t + [1 - m] \Delta t)$$
(5.1.3)

$$\frac{\partial^2 C}{\partial t^2} = \frac{1}{(\Delta t)^2} \left[\gamma_1 C(t + \Delta t) + \gamma_2 C(t) + \gamma_3 C(t - \Delta t) + \cdots + \gamma_r C(t + [1 - m] \Delta t) \right]$$

$$(5.1.4)$$

In Equations (5.1.2) through (5.1.4), α , β , and γ are coefficients which will be evaluated; r and m are defined below. Using Equations (5.1.2) and (5.1.3) the recurrence formula for Equation (5.1.1) will take the form



$$(a_{11}[A] + \frac{a_{12}}{\Delta t}[H]) \{C(t+\Delta t)\} = (a_{21}[A] + \frac{a_{22}}{\Delta t}[H]) \{C(t)\}$$

$$+ (a_{31}[A] + \frac{a_{32}}{\Delta t}[H]) \{C(t-\Delta t)\} + \cdot \cdot \cdot$$

$$+ (a_{r1}[A] + \frac{a_{r2}}{\Delta t}) \{C(t-(m-1)\Delta t)\} + a_{13} \{F(t+\Delta t)\}$$

$$+ a_{23} \{F(t)\} + a_{33} \{F(t-\Delta t)\} + \cdot \cdot \cdot + a_{r3} \{F(t-(m-1)\Delta t)\}$$

$$(5.1.5)$$

The second order time derivative usually appears in dynamic problems and thus its approximation is given in this chapter. But no recurrence formula which involves the second derivative is derived in this study. In Equation (5.1.5), $\{C(t+\Delta t)\}$ is a set of unknown variables and the right-hand side of the equation are all known values. The integer r = m + 1, m = 1, 2, ..., MA, where MA is the order of approximation. For example, for first order approximation MA is one, i.e., only one previous value of the unknown variable is used. If one uses two previous values of the unknown variable it is termed second order approximation and MA will be two. In this chapter the related equations up to MA = 3 have been developed and in Chapters VIII and IX the effects of using first and second order approximations on the accuracy of the numerical results are discussed.



5.2 Finite Element Formulation

In the finite element technique the unknown variable within an element can be evaluated by

$$\hat{\mathbf{C}}(\lambda) = \sum_{n=1}^{M} \mathbf{C}_{n} \mathbf{N}_{n}(\lambda)$$
 (5.2.1)

where C is the unknown variable, C_n is the value of C at node n, M is the number of nodes at each element, N_n is called the shape function, and λ is the independent variable such as time or x_1 , x_2 , x_3 . First and second derivatives of C with respect to λ are

$$\frac{\partial \hat{C}(\lambda)}{\partial \lambda} = \sum_{n=1}^{M} c_n \frac{\partial N_n}{\partial \lambda}$$
 (5.2.2)

$$\frac{\partial^2 \hat{C}(\lambda)}{\partial \lambda^2} = \sum_{n=1}^{M} C_n \frac{\partial^2 N_n}{\partial \lambda^2}$$
 (5.2.3)

In the following sections the basic relations will be derived in terms of λ and then replaced by t.

5.3 First Order Approximation

The simplest one-dimensional finite element is a linear element (Figure 5-1) with two nodes. Let the distance between these nodes be $\Delta\lambda$.



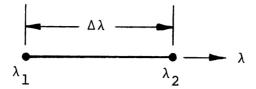


Figure 5-1.--One-Dimensional Finite Element.

The shape functions for this element are

$$N_{1} = \frac{\lambda - \lambda_{2}}{\lambda_{1} - \lambda_{2}} = \frac{\lambda_{2} - \lambda}{\Delta \lambda}$$

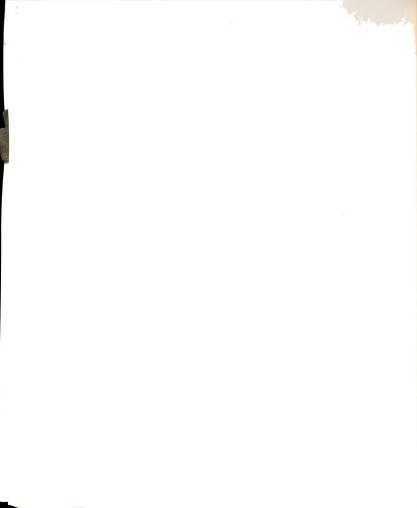
$$N_{2} = \frac{\lambda - \lambda_{1}}{\lambda_{2} - \lambda_{1}} = \frac{\lambda - \lambda_{1}}{\Delta \lambda}$$
(5.3.1)

where λ_1 and λ_2 are the values of the independent variables at nodes 1 and 2. Let C_1 and C_2 be the values of the dependent variable at nodes 1 and 2, respectively. Substituting Equation (5.3.1) into Equations (5.2.1) and (5.2.2) yields

$$C = N_1 C_1 + N_2 C_2 = \left(\frac{\lambda_2 - \lambda}{\Delta \lambda}\right) C_1 + \left(\frac{\lambda - \lambda_1}{\Delta \lambda}\right) C_2 \qquad (5.3.2)$$

$$\frac{\partial C}{\partial \lambda} = \frac{\partial N_1}{\partial \lambda} C_1 + \frac{\partial N_2}{\partial \lambda} C_2 = \frac{1}{\Delta \lambda} (C_2 - C_1)$$
 (5.3.3)

In order that this development be consistent and might be compared with existing methods in the literature, define the variable θ :



$$\theta = \frac{\lambda - \lambda_1}{\lambda_2 - \lambda_1} \tag{5.3.4}$$

such that

$$\theta = 0$$
 for $\lambda = \lambda_1$

$$\theta = 1$$
 for $\lambda = \lambda_2$

and $0 \le \theta \le 1$

Introducing θ in Equations (5.3.2) and (5.3.3) yields

$$C = \theta C_2 + (1-\theta)C_1$$
 (5.3.5)

$$\frac{\partial C}{\partial \lambda} = \frac{1}{\Delta \lambda} \left[C_2 - C_1 \right] \tag{5.3.6}$$

Equations (5.3.5) and (5.3.6) can be written in the form

$$C = \alpha_1 C_2 + \alpha_2 C_1$$
 (5.3.5a)

$$\frac{\partial C}{\partial \lambda} = \frac{1}{\Delta \lambda} \left[\beta_1 C_2 + \beta_2 C_1 \right] \tag{5.3.6a}$$

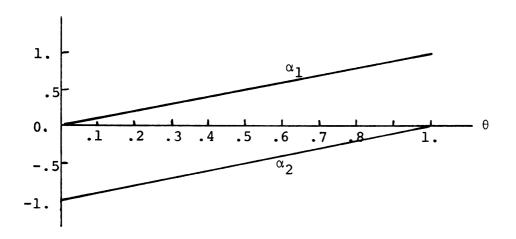
where α_1 = θ , α_2 = 1- θ , β_1 = 1, β_2 = -1, and the values of γ are zero. The variations of α and β with respect to θ are shown in Figure 5-2.

Equations (5.3.5a) and (5.3.6a) in terms of time have the form

$$C = \alpha_1 C(t + \Delta t) + \alpha_2 C(t)$$
 (5.3.7)

$$\frac{\partial C}{\partial t} = \frac{1}{\Delta t} \left[\beta_1 C(t + \Delta t) + \beta_2 C(t) \right]$$
 (5.3.8)





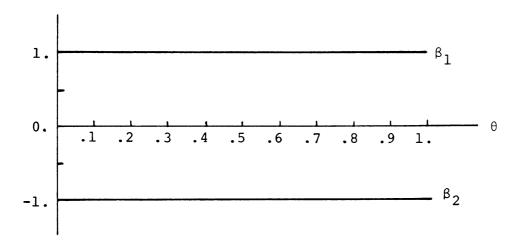


Figure 5-2.--Variation of α and β With θ for First Order Approximation.



Substituting Equations (5.3.7) and (5.3.8) into Equation (5.1.1) yields the first order recurrence formula for Equation (5.1.1). Some of the possible discretizations of Equation (5.1.1) for different θ are given below.

(a)
$$\lambda = t + \Delta t$$
; $\theta = 1$

$$([A] + \frac{1}{\Delta t} [H]) \{C(t+\Delta t)\} = \frac{1}{\Delta t} [H] \{C(t)\} + \{F(t+\Delta t)\} (5.3.9)$$

In Equation (5.3.9), $\{C(t+\Delta t)\}$ is unknown and $\{C(t)\}$ and $\{F(t+\Delta t)\}$, the forces at time $t+\Delta t$, are known. Equation (5.3.9) is a simple form of the pure implicit approximation of Equation (5.1.1).

(b)
$$\lambda = t$$
; $\theta = 0$

$$\frac{1}{\Delta t} [H] \{C(t+\Delta t)\} = (\frac{1}{\Delta t} [H] - [A]) \{C(t)\} + \{F(t)\}$$
 (5.3.10)

Equation (5.3.10) is an explicit approximation of Equation (5.1.1).

(c) $\lambda = t + (\Delta t)/2$; $\theta = 1/2$ (i.e., at the center of the element)

$$\left(\frac{1}{2} [A] + \frac{1}{\Delta t} [H]\right) \{C(t+\Delta t)\} = \left(\frac{1}{\Delta t} [H] - \frac{1}{2} [A]\right) \{C(t)\}
+ \frac{1}{2} \left(\{F(t+\Delta t)\} + \{F(t)\}\right)$$
(5.3.11)



This relation is identical to the Crank-Nicolson recurrence formula [Donea 1974].

(d)
$$\lambda = t + \frac{2}{3} \Delta t$$
; $\theta = \frac{2}{3} = 0.6667$

$$\left[\frac{2}{3} [A] + \frac{1}{\Delta t} [H]\right] \{C(t+\Delta t)\} = \left[\frac{1}{\Delta t} [H] - \frac{1}{3} [A]\right] \{C(t)\}$$

$$+ \frac{1}{3} \{F(t)\} + \frac{2}{3} \{F(t+\Delta t)\}$$
 (5.3.12)

This equation is identical to the Galerkin recurrence linear interpolation given by Donea [1974], except that the coefficients of the related formula given by Donea [Equation 4] need to be multiplied by two.

The recurrence formulas introduced by Equations (5.3.9) through Equation (5.3.12) are based on the backward approximation. Moreover, with changing θ one can derive a different recurrence relation for Equation (5.1.1).

5.4 Second Order Approximation

In a quadratic element (Figure 5-3) there are three nodes with positions λ_1 , λ_2 , and λ_3 such that λ_2 = λ_1 + $\Delta\lambda$ and λ_3 = λ_2 + $\Delta\lambda$.

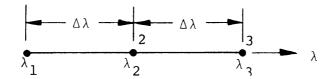
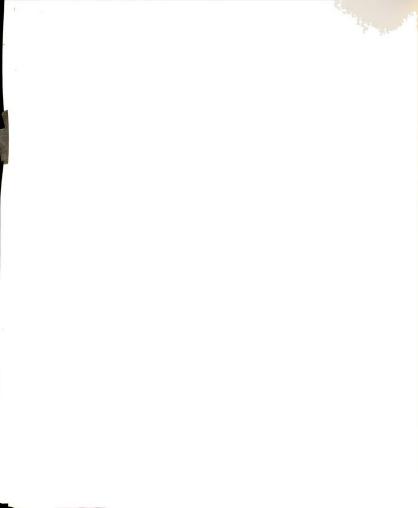


Figure 5-3.--One-Dimensional Quadratic Element.



The shape functions for this element can be written

$$N_{1} = \frac{(\lambda - \lambda_{2})(\lambda - \lambda_{3})}{(\lambda_{1} - \lambda_{2})(\lambda_{1} - \lambda_{3})} = \frac{(\lambda - \lambda_{2})(\lambda - \lambda_{3})}{2(\Delta \lambda)^{2}}$$

$$N_2 = \frac{(\lambda - \lambda_1)(\lambda - \lambda_3)}{(\lambda_2 - \lambda_1)(\lambda_2 - \lambda_3)} = -\frac{(\lambda - \lambda_1)(\lambda - \lambda_3)}{(\Delta \lambda)^2}$$

$$N_{3} = \frac{(\lambda - \lambda_{1})(\lambda - \lambda_{2})}{(\lambda_{3} - \lambda_{1})(\lambda_{3} - \lambda_{2})} = \frac{(\lambda - \lambda_{1})(\lambda - \lambda_{2})}{2(\Delta \lambda)^{2}}$$
(5.4.1)

Again define the variable

$$\theta = \frac{\lambda - \lambda_2}{\lambda_3 - \lambda_2} \tag{5.4.2}$$

such that

$$\theta = 0$$
 for $\lambda = \lambda_2$

$$\theta = 1$$
 for $\lambda = \lambda_3$

and

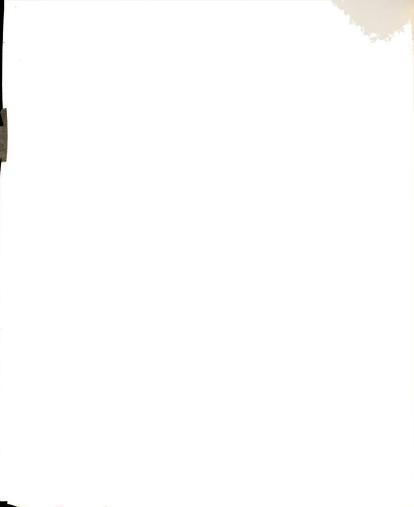
$$0 < \theta < 1$$

The shape functions in terms of θ can be written

$$N_1 = \frac{1}{2} (\theta^2 - \theta)$$

$$N_2 = 1 - \theta^2$$

$$N_3 = \frac{1}{2} (\theta^2 + 1) \tag{5.4.3}$$



Substituting the shape functions and their derivatives in Equations (5.2.1) through (5.2.3) yields

$$C = \alpha_1 C_3 + \alpha_2 C_2 + \alpha_3 C_1 \tag{5.4.4}$$

$$\frac{\partial C}{\partial \lambda} = \frac{1}{\Delta \lambda} \left[\beta_1 C_3 + \beta_2 C_2 + \beta_3 C_1 \right] \tag{5.4.5}$$

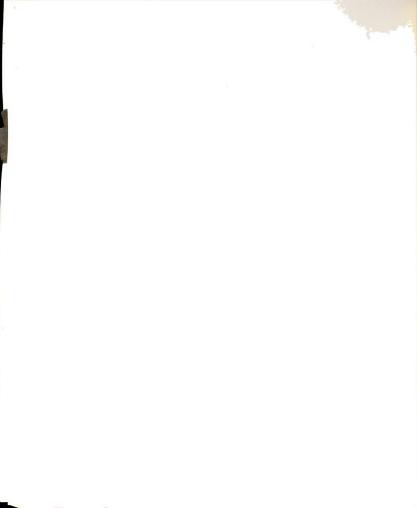
$$\frac{\partial^2 C}{\partial \lambda^2} = \frac{1}{(\Delta \lambda)^2} [\gamma_1 C_3 + \gamma_2 C_2 + \gamma_3 C_1]$$
 (5.4.6)

The values of α , β , and γ in terms of θ are given in Table 5-1.

TABLE 5-1.--Values of α , β , and γ for Second Order Approximation in Terms of θ .

n	α _n	β _n	Ϋ́n
1	.5 $(\theta^2 + \theta)$	θ + .5	1
2	1 - 0 ²	-2 0	-2
3	.5 $(\theta^2 - \theta)$	θ5	1

Equations (5.4.4) through (5.4.6) represent the finite difference approximation for C, $\partial C/\partial \lambda$, and $\partial^2 C/\partial \lambda^2$, respectively, which have been derived using the finite element technique. The variations of α and β with respect to θ are shown in Figure 5-4. For any given θ , the values of the coefficients can be read directly from Figure 5-4 and substituted in the desired equation.



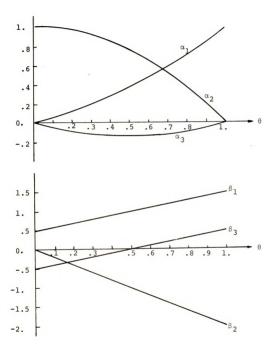


Figure 5-4.--Variation of α and β With θ for Second Order Approximation.



For the case where the independent variable is time, C_3 , C_2 , C_1 , and λ can be replaced in Equations (5.4.4) through (5.4.6) by $C(t+\Delta t)$, C(t), $C(t-\Delta t)$, and t, respectively. Thus

$$C = \alpha_1 C(t+\Delta t) + \alpha_2 C(t) + \alpha_3 C(t-\Delta t) \qquad (5.4.7)$$

$$\frac{\partial C}{\partial t} = \frac{1}{\Delta t} \left[\beta_1 C(t + \Delta t) + \beta_2 C(t) + \beta_3 C(t - \Delta t) \right]$$
 (5.4.8)

$$\frac{\partial^2 C}{\partial t^2} = \frac{1}{(\Delta t)^2} \left[\gamma_1 C(t + \Delta t) + \gamma_2 C(t) + \gamma_3 C(t - \Delta t) \right]$$
 (5.4.9)

With the aid of Equations (5.4.7) and (5.4.8), it is possible to obtain a second order time approximation for Equation (5.1.1). The recurrence formula has the form of Equation (5.1.5), and values of the coefficients for the first and second order time approximation for different θ are given in Table 5-2. It is possible to obtain a set of recurrence formulas for Equation (5.1.1) by changing the θ value.

An interpretation can be made from Figure 5-4 by noting that the coefficients of $C(t+\Delta t)$, C(t), and $C(t-\Delta t)$ vary with respect to θ . At $\theta=0$, C(t) plays a dominant role, and as θ increases the effect of C(t) decreases while that of $C(t+\Delta t)$ increases. At $\theta=2/3$, the value of α_1 is equal to α_2 . This means that at this θ , $C(t+\Delta t)$ and C(t) are almost equally weighted.

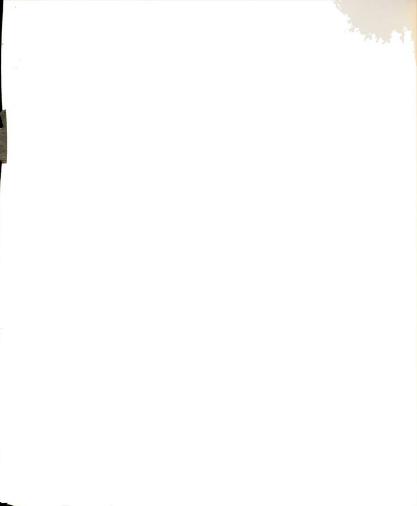


TABLE 5-2.--Values of the Coefficients of Equation (5.1.5) for the First and Second Order Time Approximation.

Method (θ)	a 11	^a 12	a 13	a ₂₁	a ₂₂	a ₂₃	a 31	a ₃₂	a 33
First Order									
$\theta = \frac{1}{2}$ (Crank-Nicolson)	•5	1.	.5	 5	1.	.5	0.	0.	0.
$\theta = \frac{2}{3}$ (Galerkin)	$\frac{2}{3}$	1.	$\frac{2}{3}$	$-\frac{1}{3}$	1.	$\frac{1}{3}$	0.	0.	0.
θ = 1 (Implicit)	1.	1.	1.	0.	1.	0.	0.	0.	0.
Second Order									
$\theta = \frac{1}{2}$	3 8	1.	<u>3</u>	$-\frac{3}{4}$	1.	$\frac{3}{4}$	$\frac{1}{8}$	0.	$-\frac{1}{8}$
$\theta = \frac{2}{3}$	<u>5</u> 9	7 6	<u>5</u> 9	$-\frac{5}{9}$	<u>8</u>	<u>5</u> 9	<u>1</u> 9	$-\frac{1}{6}$	$-\frac{1}{9}$
θ = 1	1.	$\frac{3}{2}$	1.	0.	2.	0.	0.	$-\frac{1}{2}$	0.
Steady State	1.	0.	0.	0.	0.	1.	0.	0.	0.

C(t- Δ t) also appears in the calculation. For θ = 1, α_2 and α_3 both are zero, and all β values are nonzero. In Chapters VIII and IX, the numerical results obtained by using different θ 's are compared.

5.5 Third Order Approximation

For the cubic element with four nodes as shown in Figure 5-5, the shape functions are:



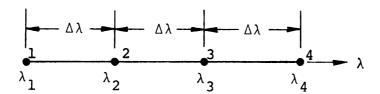


Figure 5-5.--One-Dimensional Cubic Element.

$$N_{1} = \frac{(\lambda - \lambda_{2})(\lambda - \lambda_{3})(\lambda - \lambda_{4})}{(\lambda_{1} - \lambda_{2})(\lambda_{1} - \lambda_{3})(\lambda_{1} - \lambda_{4})} = \frac{(\lambda - \lambda_{2})(\lambda - \lambda_{3})(\lambda - \lambda_{4})}{-6(\Delta \lambda)^{3}}$$

$$N_{2} = \frac{(\lambda - \lambda_{1})(\lambda - \lambda_{3})(\lambda - \lambda_{4})}{(\lambda_{2} - \lambda_{1})(\lambda_{2} - \lambda_{3})(\lambda_{2} - \lambda_{4})} = \frac{(\lambda - \lambda_{1})(\lambda - \lambda_{3})(\lambda - \lambda_{4})}{2(\Delta \lambda)^{3}}$$

$$N_{3} = \frac{(\lambda - \lambda_{1})(\lambda - \lambda_{2})(\lambda - \lambda_{4})}{(\lambda_{3} - \lambda_{1})(\lambda_{3} - \lambda_{2})(\lambda_{3} - \lambda_{4})} = \frac{(\lambda - \lambda_{1})(\lambda - \lambda_{2})(\lambda - \lambda_{4})}{-2(\Delta \lambda)^{3}}$$

$$N_{4} = \frac{(\lambda - \lambda_{1})(\lambda - \lambda_{2})(\lambda - \lambda_{3})}{(\lambda_{4} - \lambda_{1})(\lambda_{4} - \lambda_{2})(\lambda_{4} - \lambda_{3})} = \frac{(\lambda - \lambda_{1})(\lambda - \lambda_{2})(\lambda - \lambda_{4})}{6(\Delta \lambda)^{3}}$$

$$(5.5.1)$$

Define the variable θ :

$$\theta = \frac{\lambda - \lambda_3}{\lambda_4 - \lambda_3} \tag{5.5.2}$$



such that

$$\theta = 0$$
 for $\lambda = \lambda_3$

$$\theta = 1$$
 for $\lambda = \lambda_4$

and

$$0 \le \Theta \le 1$$

The shape functions in terms of $\boldsymbol{\theta}$ can be written

$$N_1 = \frac{1}{6} [\theta(\theta+1)(1-\theta)]$$

$$N_2 = \frac{1}{2} [\theta(\theta+2)(\theta-1)]$$

$$N_3 = \frac{1}{2} [(\theta+1)(\theta+2)(1-\theta)]$$

$$N_4 = \frac{1}{6} [\theta (\theta+1) (\theta+2)]$$
 (5.5.3)

The value of C and its first and second derivatives with respect to λ can be written in the following form:

$$C = \alpha_1 C_4 + \alpha_2 C_3 + \alpha_3 C_2 + \alpha_4 C_1 \tag{5.5.4}$$

$$\frac{\partial C}{\partial \lambda} = \frac{1}{\Delta \lambda} \left[\beta_1 C_4 + \beta_2 C_3 + \beta_3 C_2 + \beta_4 C_1 \right] \tag{5.5.5}$$

$$\frac{\partial^2 C}{\partial \lambda^2} = \frac{1}{(\Delta \lambda)^2} \left[\gamma_1 C_4 + \gamma_2 C_3 + \gamma_3 C_2 + \gamma_4 C_1 \right]$$
 (5.5.6)

The values of α , β , and γ in terms of θ are given in Table 5-3.



TABLE 5-3.--Values of α , β and γ for Third Order Approximation in Terms of θ .

n	^α n	β _n	Ϋ́n
1	$\frac{1}{6}$ θ (0+1) (1- θ)	$\frac{1}{6}$ (1-3 θ^2)	- θ
2	$\frac{1}{2} \theta (\theta+2) (\theta-1)$	$\frac{1}{2} (3\theta^2 + 2\theta - 2)$	(30+1)
3	$\frac{1}{2}$ (0+1) (0+2) (1-0)	$-\frac{1}{2} (3\theta^2 + 4\theta - 1)$	- (3θ+2)
4	$\frac{1}{6}$ θ (θ +1) (θ +2)	$\frac{1}{6} (3\theta^2 + 6\theta + 2)$	(θ + 1)

The variations of α and β with respect to θ are given in Figure 5-6.

For a time-dependent variable, Equations (5.5.4) through (5.5.6) can be written

$$C = \alpha_1 C(t+\Delta t) + \alpha_2 C(t) + \alpha_3 C(t-\Delta t)$$

$$+ \alpha_4 C(t-2\Delta t) \qquad (5.5.7)$$

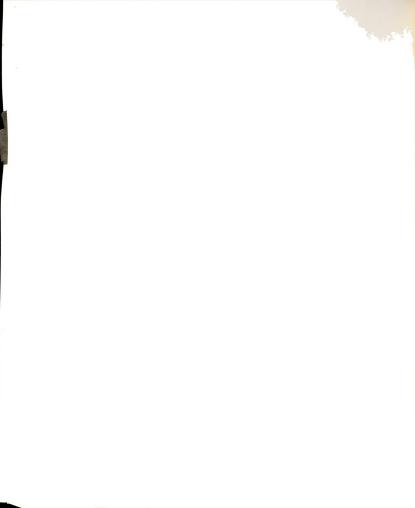
$$\frac{\partial C}{\partial t} = \frac{1}{\Delta t} \left[\beta_1 C(t + \Delta t) + \beta_2 C(t) + \beta_3 C(t - \Delta t) + \beta_4 C(t - 2\Delta t) \right]$$

$$(5.5.8)$$

and

$$\frac{\partial^2 c}{\partial t^2} = \frac{1}{(\Delta t)^2} \left[\gamma_1 C(t + \Delta t) + \gamma_2 C(t) + \gamma_3 C(t - \Delta t) + \gamma_4 C(t - 2\Delta t) \right]$$

$$(5.5.9)$$



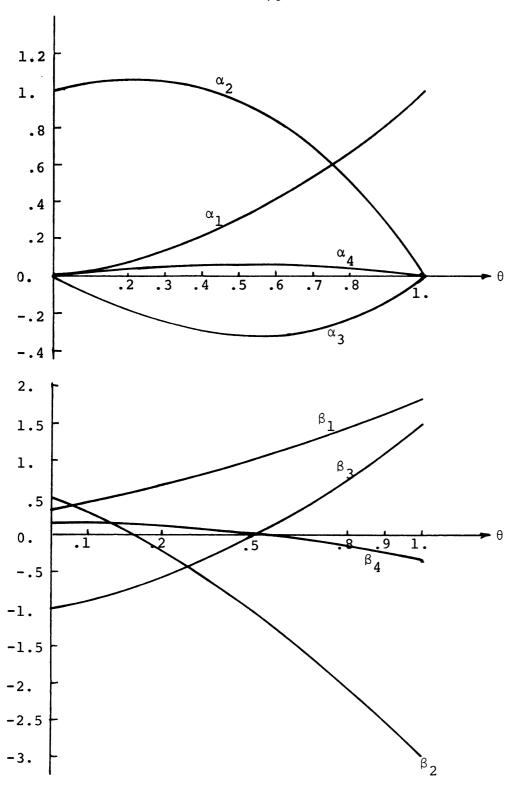
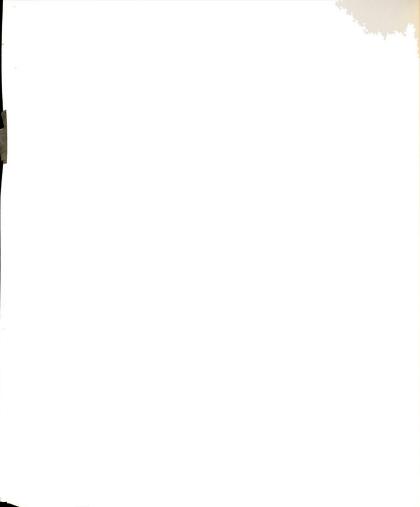


Figure 5-6.--Variation of α and β With θ for Third Order Approximation.



To obtain a third order recurrence formula for a set of first order partial differential equations, Equations (5.5.7) and (5.5.8) need to be substituted in Equation (5.1.1). The coefficients of Equation (5.1.5) for different θ for third order time approximation are given in Table 5-4.

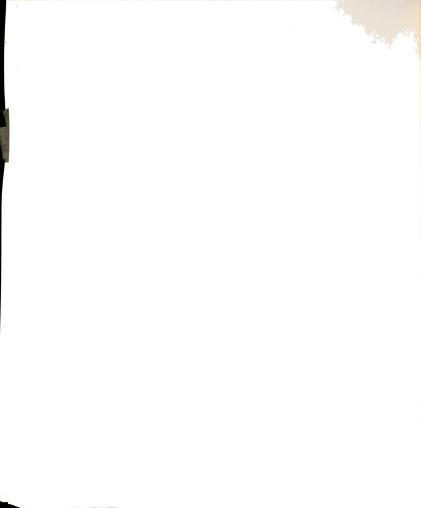
5.6 Summary

In this chapter the recurrence formula for the first, second, and third order approximations for a system of first order differential equations is derived, and the procedure of obtaining the finite difference relations using the finite element concept is presented. In this study the first order approximation was used in the solution of the flow equation, while the first and second orders were employed for solving the convective-dispersion equation. The third order was not examined in this work but for sake of completeness it was introduced.

It will be shown in Chapters VIII and IX that θ = 2/3 provides less oscillatory and more reasonable results for the flow and mass transport equations. However, choosing the value of θ depends on the nature of the equation, the numerical technique, and the required use of previous known values. Giving any specific value for θ at this stage would be premature. More work needs to be done in this area.



_	a43	48	81	0
tor Third Order Time Approximation	a42	1 9	18	чю
Approx	a41	ا 4 <u>8</u>	- <u>5</u>	0
Tıme	a ₃₃	- <u>15</u>	- <u>24</u> - <u>81</u>	0
Order	a ₃₂	ကျဖ	- 18	n a
Third	a ₃₁	<u>15</u>	24 81	0
o) for	a ₂₃	<u>45</u>	60 81	0
(5.1.	a ₂₂	$\frac{21}{6}$	$\frac{27}{18}$	ĸ
TABLE 5-4Coefficients of Equation (5.1.5) for Different θ .	a ₂₁	- <u>45</u>	- <u>60</u>	0
	a ₁₃	<u>15</u>	40 81	1
	a ₁₂	<u>23</u> 6	22 18	11 6
	a ₁₁	<u>15</u>	40 81	1
3LE 5-4	Method (θ)	2 1	2 6	. 1
TAE	Met	П	ΙΙ θ	Π



CHAPTER VI

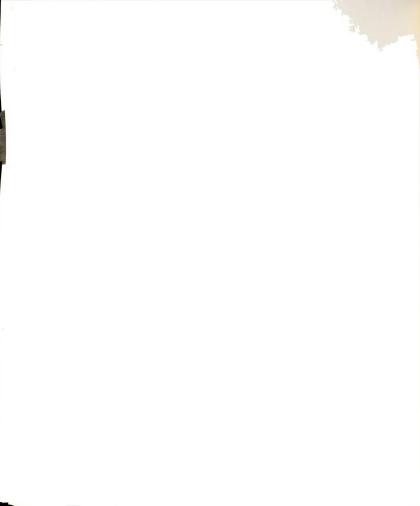
SOLUTION OF SYSTEM EQUATIONS CONCERNING THE FLOW IN CONFINED AND UNCONFINED AQUIFERS

In this chapter the procedure of solving the system of equations concerning the flow in confined and unconfined aquifers and the alternate methods, i.e., direct and simultaneous, of calculating the velocity vectors will be discussed. For locating the free surface in the unconfined aquifers two procedures, namely, modification of the elements and use of fixed nodes, are presented. The latter technique will provide a tool to solve the convective-dispersion equation in the phreatic aquifers.

6.1 Regional Groundwater Flow

Finite element formulation of the combined equation of motion and continuity for flow in a two-dimensional horizontal plane leads to a set of first-order partial differential equations of the form

[B]
$$\{\phi\} + [H] \{\frac{\partial \phi}{\partial t}\} = \{F\}$$
 (4.2.8)



with the unknowns $\{\phi\}$, which is to be solved simultaneously at a given time for a series of nodes in the (x_1, x_2) domain. The [B] and [H] matrices and $\{F\}$ vectors are known; their definitions are given in Section 4.2, and methods of evaluating these functions are outlined in Appendix I.

6.1.1 Computation of Piezo-metric Heads

The discretization of the time derivative of Equation (4.2.8) and similar equations is discussed in detail in Chapter V. Equation (4.2.8) can be written as

$$[\bar{B}] \{ \phi (t + \Delta t) \} = \{ G \}$$
 (6.1.1)

where

$$[\bar{B}] = a_{11}[B] + \frac{a_{12}}{\Lambda +}[H]$$
 (6.1.2)

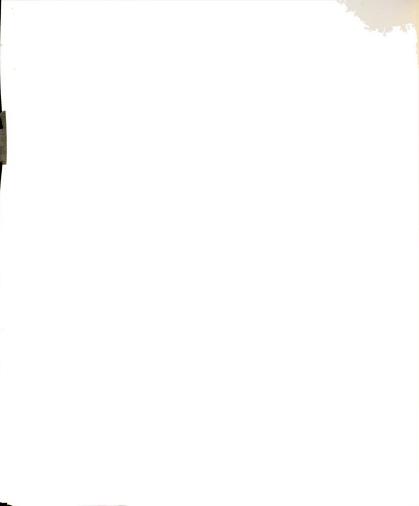
and

$$\{G\} = \left(a_{21}[B] + \frac{a_{22}}{\Delta t}[H]\right) + a_{13}\{F(t+\Delta t)\}$$

$$+ a_{23}\{F(t)\}$$
(6.1.3)

The values of the coefficients are given in Table 5-2.

Typical boundary conditions are discussed in Section 3.3.3. From these conditions the lateral recharge \mathbf{Q}_2 as well as the Dirichlet boundary are assumed to be time invariant. For many applications, time variable pumping must be considered at various nodes throughout



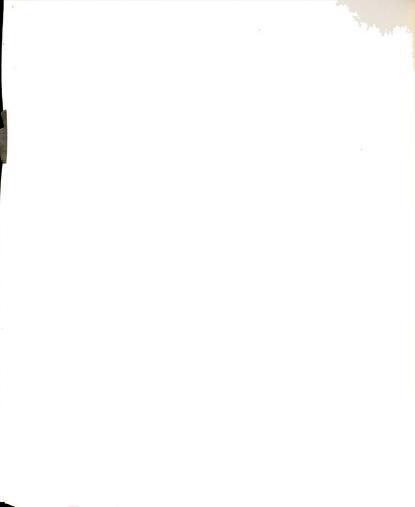
the flow domain. In the computation it is convenient to handle the pumping as a series of step functions discretized with respect to time as shown in Figure 6-1.

Thus at any time interval recharge or discharge will be constant.

There are different numerical techniques which can be employed to solve the system of Equations (6.1.1). Since $[\bar{B}]$ is a banded symmetric matrix with nonzero terms in the diagonal, it is possible to use Cholesky's square root procedure to decompose the matrix $[\bar{B}]$ (upper band) and solve with the companion subroutine for the unknown $\{\phi\}$ as outlined by Weaver [1967], and employed by Pinder and Frind [1972] for groundwater flow. The required storage space for $[\bar{B}]$ in the computer will reduce to NNDS × MAXBW, where NNDS is the number of unknowns $\{\phi\}$ and MAXBW is equal to the upper bandwidth plus one.

The parameters such as transmissibility, storage coefficient, and time step participate in the construction of the $[\bar{B}]$ matrix. Thus if one desires to change any of the above mentioned parameters, it will be necessary to regenerate the $[\bar{B}]$ matrix again.

Equation (6.1.1) can also be considered to be a "steady-state" problem if the coefficients for the steady-state condition from Table 5-2 are used. The resulting equation for this state is



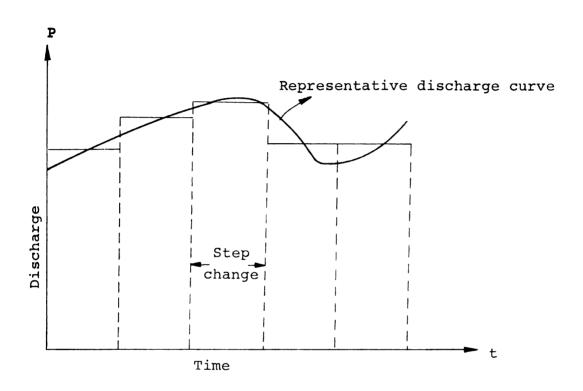


Figure 6-1.--Dividing the Actual Discharge Curve Into a Series of Step Functions.



$$[B] \{ \phi \} = \{ F \} \tag{6.1.4}$$

6.1.2 Introducing the Dirichlet Boundary Condition

The placement of the Dirichlet boundary condition into the global equations can be accomplished by the deletion of rows and columns [Norrie and de Vries 1973].

Assume in Equation (6.1.1) that the piezometric head at node k is known. Equation (6.1.1) in expanded form can be written

$$\begin{bmatrix} \bar{B}_{11} & \cdots & \bar{B}_{1k} & \cdots & \bar{B}_{1n} \\ \vdots & & \ddots & & \vdots \\ \vdots & & \ddots & & \vdots \\ \bar{B}_{k1} & \cdots & \bar{B}_{kk} & \cdots & \bar{B}_{kn} \\ \vdots & & & \ddots & & \vdots \\ \bar{B}_{n1} & \cdots & \bar{B}_{nk} & \cdots & \bar{B}_{nn} \end{bmatrix} \begin{pmatrix} \phi_1 \\ \vdots \\ \vdots \\ \phi_k \\ \vdots \\ \vdots \\ \phi_n \end{pmatrix} = \begin{pmatrix} G_1 \\ \vdots \\ \vdots \\ G_k \\ \vdots \\ \vdots \\ G_n \end{pmatrix}$$
(6.1.1a)

Since ϕ_k is known, all coefficients in the $[\bar{B}]$ matrix at the k^{th} row can be set equal to zero except the diagonal terms (i.e., \bar{B}_{kk}) which will remain unchanged. The G_k is replaced by $\bar{B}_{kk}\phi_k$. Equation (6.1.1a) will have the following terms:



$$\begin{bmatrix} \bar{B}_{11} & \cdots & \bar{B}_{1k} & \cdots & \bar{B}_{1n} \\ \vdots & & \ddots & & \vdots \\ 0 & \cdots & \bar{B}_{kk} & \cdots & 0 \\ \vdots & & \ddots & & \ddots \\ 0 & \cdots & \bar{B}_{kk} & \cdots & 0 \\ \vdots & & \ddots & & \ddots \\ \vdots & & \ddots & & \ddots \\ \bar{B}_{n1} & \cdots & \bar{B}_{nk} & \cdots & \bar{B}_{nn} \end{bmatrix} \begin{bmatrix} \phi_1 \\ \phi_k \\ \phi_k \\ \phi_n \end{bmatrix} = \begin{bmatrix} G_1 \\ \bar{B}_{kk} \phi_k \\ \bar{B}_{kk} \phi_k \\ G_n \end{bmatrix} (6.1.5)$$

Usually \bar{B}_{kk} has a positive value. In the computer program a check has to be made to be certain that the value of \bar{B}_{kk} does not fall below a certain small value, e.g., one. Otherwise \bar{B}_{kk} can be changed to any large value. In the computer program, \bar{B}_{kk} is replaced by the average value of the diagonal terms. Equation (6.1.5) can handle the Dirichlet boundary condition, but the matrix is then banded and nonsymmetric. Since Cholesky's method is used to solve the system of equations, the matrix of Equation (6.1.5) must be symmetric. This can be accomplished by subtracting $\bar{B}_{ik}\phi_k$; k, i = 1 . . . n, i \neq k from both sides of Equation (6.1.). The final form becomes



$$\begin{bmatrix}
\bar{B}_{11} & \cdots & 0 & \cdots & \bar{B}_{1n} \\
\vdots & \vdots & \ddots & \vdots \\
0 & \cdots & \bar{B}_{kk} & \cdots & 0 \\
\vdots & \vdots & \ddots & \vdots \\
\bar{B}_{n1} & \cdots & 0 & \cdots & \bar{B}_{nn}
\end{bmatrix}
\begin{pmatrix}
\phi_{1} \\
\phi_{k} \\
\phi_{k}
\end{pmatrix} = \begin{pmatrix}
G_{1}^{-\bar{B}}_{1k}\phi_{k} \\
\bar{B}_{kk}\phi_{k} \\
\phi_{n}
\end{pmatrix} (6.1.6)$$

This procedure is carried out for all specified piezometric heads prior to decomposition of the [B] matrix.

6.2 Solution of Flow Vectors

The mathematical equation of the velocity vectors is discussed in Section 3.2 and the methods of solution are described in Section 4.4. Once the piezometric heads have been determined, flow (or velocity) vectors can be computed.

6.2.1 Direct Calculation

In the direct method, Equation (4.4.3), i.e.,

$$V_{i} = -\frac{K_{ij}}{n_{e}} \frac{\partial N_{n}}{\partial x_{j}} \phi_{i} \qquad i, j=1, 2$$
 (4.4.3)

is employed to compute the velocity vectors. This technique provides accurate results at the centroid of each element, but the velocity components calculated by this



procedure are discontinuous across element boundaries.

The magnitude and direction of the velocity vectors

(Figure 6-2) can be obtained by

$$v = |v_j| = \sqrt{v_1^2 + v_2^2}$$
 $j=1,2$
 $\alpha = \arctan(v_2/v_1)$ (6.2.1)

6.2.2 Simultaneous Calculation of Velocity Vectors

Simultaneous calculation provides continuous velocity components at the nodes. The finite element formulation of this technique is presented in Section 4.4.2. For example, for V_1^{C} the element equations are

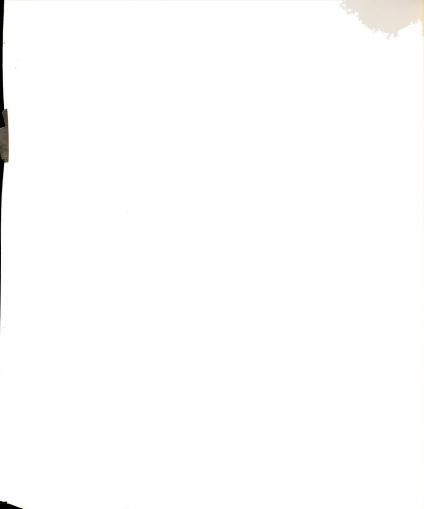
$$H_{kn}^{e} = \int_{D^{e}} N_{k} N_{n} dD$$
 (4.4.11)

$$(F_1)_k^e = \int_{D^e} \frac{K_{11}}{n_e} N_k \frac{\partial N_n}{\partial x_1} \phi_n dD$$
 (4.4.12)

The global matrix is

$$[H] \{v_1^c\} = \{F_1\}$$
 (4.4.13)

Integrated forms of Equation (4.4.11) and $\int_D e^{N_k (\partial N_n/\partial x_1) dD}$ for different types of elements are given in Appendix II. A similar global relation is obtained for V_2^c :



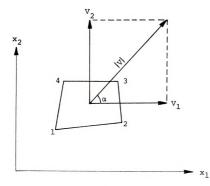
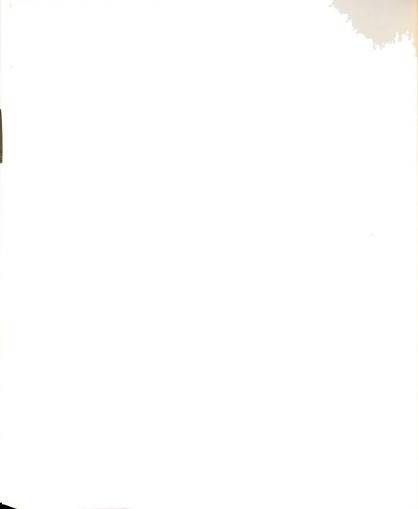


Figure 6-2.--Velocity Vectors in the \mathbf{x}_1 , \mathbf{x}_2 Plane.

$$[H] \{V_2^C\} = \{F_2\}$$
 (4.4.14)

The [H] matrix is a banded symmetric matrix. Cholesky's square root procedure (see Section 6.1.1) is used to decompose the matrix. As long as the nodal coordinates are fixed, the [H] matrix can be decomposed and stored with no need to recalculate it.

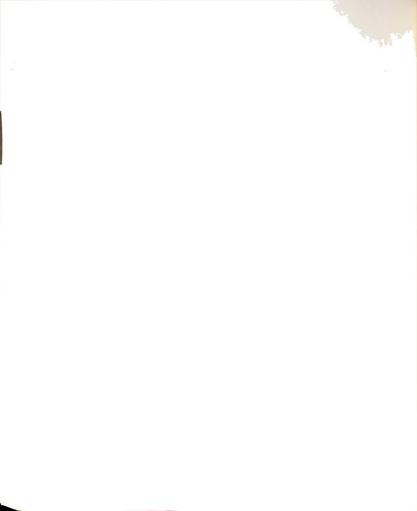
If piezometric heads and velocity vectors were to be calculated at the same time in a computing scheme, the size of both the [H] and [B] matrices would increase up to 3NNDS × 3NNDS with many zero terms, NNDS being the number of nodes in the system. This procedure requires a great amount of computer



memory, and the transient solution of the problem is tedious. In this method, besides the boundary conditions for the piezometric heads the conditions for the velocity vectors need to be specified. But the procedure outlined in the preceding paragraph, based on the simple realistic assumption that the piezometric heads can be calculated independently of the velocity vectors, provides great simplification. With the construction of the [H] matrix which requires only MAXBW × NNDS storage core, the velocity vectors can be calculated at the nodes. Equation (4.4.12) can be calculated for each element and added to $\{F_1\}$, a column matrix with NNDS rows. Similarly, $\{\mathbf{F_2}\}$ will be evaluated. The [H] matrix for $\mathbf{V_1^c}$ and $\mathbf{V_2^c}$ is identical. Thus the solution of Equation (4.4.13) and Equation (4.4.14) will provide the velocity vectors at each node. Equation (6.2.1) can be used to obtain the magnitude of the velocity and its direction at each node. Known velocities at the node can be handled by deletion of rows and columns, as outlined in Section 6.1.2. Section 8.2 the two methods for calculation of velocity vectors are compared.

6.3 Solution of System Equation for Flow in Unconfined Aquifer With Transient Phreatic Surface

In this study the flow in porous media on a regional as well as local scale is studied. For a



regional scale the two-dimensional horizontal flow is considered and the solution of related equations is given in Section 6.1. For a localized scale a vertical cross section of an unconfined aquifer is chosen and the solution of the system of equations is presented in this section. A technique of locating the phreatic surface with fixed nodes is also shown.

6.3.1 Background

Based on the assumption made in Chapter III, if the specific storage coefficient can be neglected, Equation (3.4.1) will be reduced to

$$\frac{\partial}{\partial \mathbf{x_i}} \left[\mathbf{K_{ij}} \frac{\partial \phi}{\partial \mathbf{x_j}} \right] = 0 \qquad i,j=1,3$$
 (6.3.1)

The finite element formulation of Equation (6.3.1) leads to the global matrix of the form

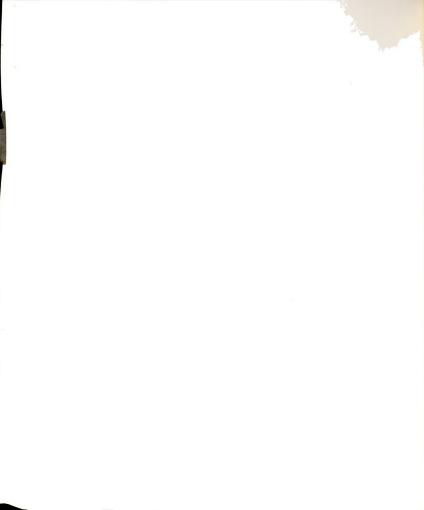
$$[B'] \{ \phi \} = \{ F \} \tag{6.3.2}$$

with its element components defined as

$$[B']^{e} = \int_{D^{e}} K_{ij} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial N_{n}}{\partial x_{j}} dD \qquad (6.3.3a)$$

and

$$\{F\}^{e} = -\int_{S} e N_{k} Q_{2} ds$$
 (6.3.3b)



Equation (6.3.2) is solved with the related boundary conditions discussed in Section 3.4.3. In addition to the known potential and known flux boundary, there is the phreatic boundary condition which is described as

$$\Delta t \ U_{j} \cdot \ell_{i} = U_{n} \Delta t = \frac{\Delta t}{n_{e}} \left(-K_{ij} \frac{\partial \phi}{\partial x_{j}} \ell_{i} - I \ell_{3} \right) \quad i, j=1,3 \quad (3.4.6)$$

Equation (3.4.6) contains a time derivative of the free surface and can therefore be used to determine the height at the later time when the other terms in the equation are known [France 1971]. An iterative technique is used to replace the original transient problem by a discrete number of steady-state problems, based on the assumption that the flow at each instant is steady but the boundary of the flow is time variable [Poluborinova-Kochina 1962, p. 572]. At the beginning of each time interval the position of the free surface and boundary conditions are known. Using Equation (3.4.6), the phreatic surface is then shifted along its normal to a new position. Two possible methods are considered One technique most commonly used (e.g., Desai 1972 and France 1974) requires the modification of the elements such that the free surface always is the upper boundary of the grid system. Another method, presented by France [1971] and modified and improved herein, accomplishes the movement of the phreatic



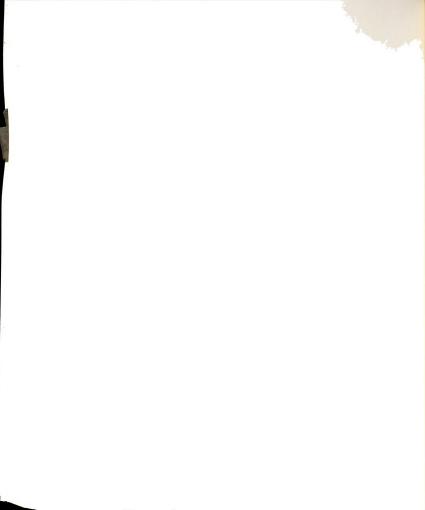
surface within the grid system without repositioning the nodal coordinates of the elements.

6.3.2 Locating the Phreatic Surface by Modifying the Elements

The movement of the phreatic surface is determined in the following manner:

Step 1. At the beginning of each time interval the surface configuration and boundary conditions are known. Equation (6.3.2) is used to find the value of the piezometric heads for all nodal points except those at the phreatic surface. If there is a known flux from boundaries other than the phreatic surface, the values of {F} in Equation (6.3.2) are calculated; otherwise {F} will be zero. At this step the nodal points at the phreatic surface act as the Dirichlet boundary. In Section 6.1.3 it is shown how to introduce the known potential into the global equations. Equation (6.3.2) is similar to Equation (6.1.4) and is solved the same way, as outlined in Section 6.1.1.

Step 2. At the beginning of each time the piezometric heads are known in the system, and from Equation (3.4.6) it is possible to compute the location of the phreatic surface at a time $t + \Delta t$. The distance a point on the phreatic surface will propagate in the direction of the normal to this surface at that point



is equal to $U_n\Delta t$. If $P(x_1,x_3)$ in Figure 6-3 is a point of the phreatic surface at time t and $P'(x_1',x_3')$ is the location of that point at time t + Δt , then the shifting distance along the normal to the free surface d_n , PP' in Figure 6-3, is

$$\mathbf{d_n} = \frac{\Delta t}{n_e} \left(-K_{11} \frac{\partial \phi}{\partial \mathbf{x}_1} \hat{\mathbf{i}} - K_{33} \frac{\partial \phi}{\partial \mathbf{x}_3} \hat{\mathbf{k}} - \mathbf{I} \hat{\mathbf{k}} \right) \cdot \left(k_1 \hat{\mathbf{i}} + k_3 \hat{\mathbf{k}} \right)$$
 (6.3.4)

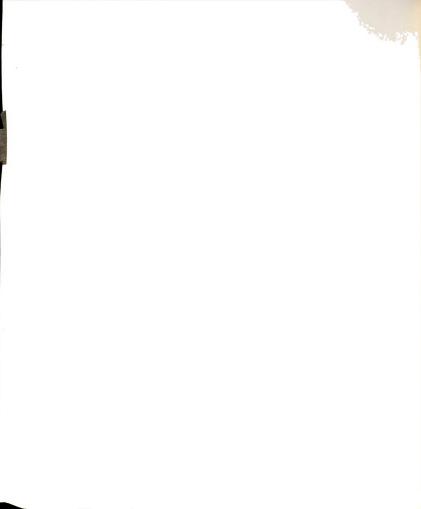
where \hat{i} and \hat{k} are unit vectors along x_1 and x_3 , respectively; and ℓ_1 and ℓ_3 are directional cosines. I is the accretion term, positive downward. If θ is the angle that the tangent to the phreatic surface makes with the positive x_1 -direction, then

$$\ell_1 = \sin \theta$$
 & $\ell_3 = \cos \theta$

$$d_n = \Delta t (V_1 \sin \theta + V_3 \cos \theta - I \cos \theta/n_e)$$
 (6.3.5)

Thus d_n can be evaluated by Equation (6.3.5). It is convenient to calculate the shifting distance along the nodal lines (d_{ℓ}) . Let ω_i be the angle that the nodal line (i) makes with the positive x_1 -direction, and define $\beta = \pi/2 - \omega + \theta$ [Desai 1972]. Then

$$d_{\ell} = \frac{d_{n}}{\cos \beta} \tag{6.3.6}$$



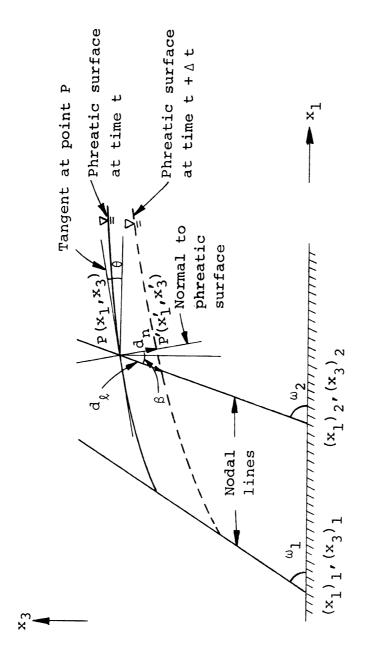


Figure 6-3.--Movement of the Phreatic Surface.



$$d_1 = d_0 \cos \omega \tag{6.3.7a}$$

$$d_3 = d_\ell \sin \omega \tag{6.3.7b}$$

In Equation (6.3.7), d_1 and d_3 are the shifting distances in the \mathbf{x}_1 and \mathbf{x}_3 directions, respectively. Summarizing,

$$d_{1} = \frac{(V_{1} \sin \theta + V_{3} \cos \theta - I \cos \theta/n_{e})\Delta t \cos \omega}{\cos \beta}$$
 (6.3.8a)

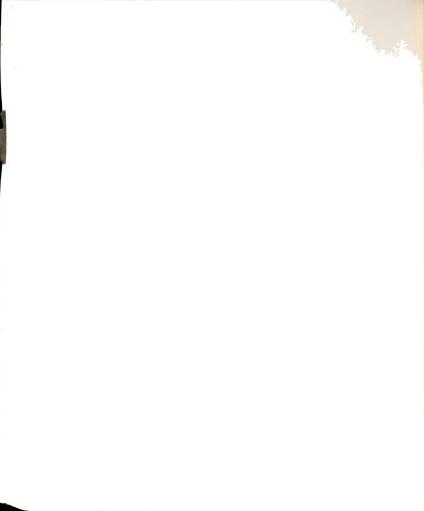
$$d_3 = \frac{(v_1 \sin \theta + v_3 \cos \theta - I \cos \theta/n_e)\Delta t \sin \omega}{\cos \beta}$$
 (6.3.8b)

Thus the location of the phreatic surface at time t + Δt will be

$$x_1' = x_1 + d_1$$
 (6.3.9a)

$$x_3' = x_3 + d_3$$
 (6.3.9b)

In order to complete step 2, one has to know the velocity components and the angle 0 at each phreatic node. As discussed in Section 6.2, direct calculation of the velocity vectors usually does not provide continuous results at the nodes. Most authors, e.g., France et al. [1971] and Desai [1973], have realized this deficiency and have used the average velocity components at the node calculated from two adjacent elements. Although this procedure reduces the errors significantly, because of



discontinuity of the velocity functions it is still not the best way to evaluate the velocity vectors at the nodes (see Section 4.4.1 for details). The proposed simultaneous calculation of the velocity components provides continuity at the nodes and is used in this study.

The slope of the element lines along the phreatic boundary is usually discontinuous at the nodes. For example, at node 8 the slope of line 4-8 in Figure 6-4 differs from line 8-12. In order to obtain a better estimate of θ at the phreatic nodes, a polynomial of degree n is passed through the phreatic nodes of two elements while the node under consideration is almost

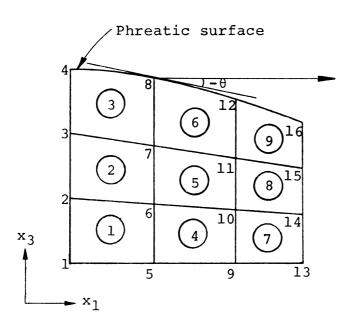


Figure 6-4.--Simple Linear Quadrilateral Isoparametric Elements.

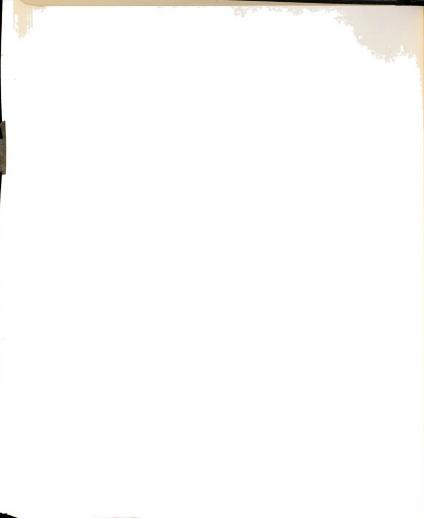
the middle point. When the polynomial is passed through the nodes the slope of the free surface is evaluated at the node. The value of n is dependent on the type of element. For linear and quadratic quadrilateral elements n=2 will be sufficient, while for a cubic element n=3 is recommended. For corner nodes (e.g., nodes 4 and 16 in Figure 6-4) the slope of the phreatic line is used. Because the phreatic nodes might not be equally spaced, Newton's divided-difference method [see Carnahan et al. 1969, pp. 9-26] is used to pass a polynomial for the desired points.

Step 3. Equation (6.3.8) assures that the shifted points always lie along the nodal line. In this case, Equation (6.3.9) represents the location of the new nodes at a new time. In some instances the nodal lines cannot be straight and the value of ω changes along the line, such as for quadratic or cubic elements. Then the new point x_1^*, x_3^* will be calculated by

$$\mathbf{x}_{1}^{\star} = \mathbf{x}_{1} - \mathbf{d}_{n} \sin \theta \tag{6.3.10a}$$

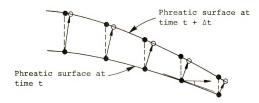
$$x_3^* = x_3 - d_n \cos \theta$$
 (6.3.10b)

where \mathbf{x}_1^\star and \mathbf{x}_3^\star represent the temporary location of the nodes at time t + Δ t, and \mathbf{d}_n is defined in Figure 6-3. To find the location of the actual nodes, it is necessary to fit a polynomial to these temporary nodes and then



find the location of new points with the given x₁-values (Figure 6-5). Polynomial curve-fitting procedures such as Newton's divided-difference method (Carnahan et al. 1969], or other methods such as the Newton-Raphson iteration technique or the least square curve-fitting [Pennington 1970, pp. 408-417], can be used. In this study the first method is employed, and the second technique is used by France et al. [1971].

Step 4. At the start of this step the location of the phreatic surface is known and can act as the Dirichlet boundary condition for solving Equation (6.3.2) to compute the piezometric heads in domain D. But since the free surface has moved, the nodes on the phreatic



O Temporary nodes

• Actual nodes

Figure 6-5.--Location of the Temporary and Actual Nodes
[After France 1971].



surface do not coincide with the nodes of elements. There are two possible ways to handle this problem:

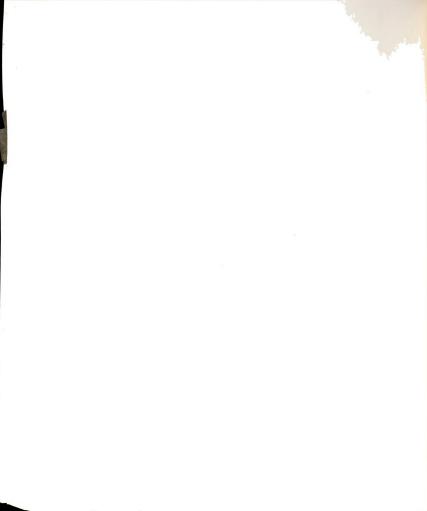
- (a) modify the elements such that the phreatic surface becomes the upper boundary of the grid system, or
- (b) define a new Dirichlet boundary condition for the nodes above the phreatic surface such that it is not necessary to modify the elements.

Procedure (a) is discussed below, and details concerning procedure (b) are postponed until Section 6.3.3.

The location of the phreatic surface is known, so the upper nodes of the phreatic elements are transformed to coincide with the phreatic surface. Since the coordinates of some nodes have changed, it is necessary to re-evaluate the [B'] matrix (Equation 6.3.3a). In order to reduce computation time the whole grid system can be divided into two groups (Figure 6-6):

- (a) fixed elements--their nodal coordinates will not alter during the entire calculation,
- (b) moveable elements--their nodal coordinates will be affected by changing the phreatic surface.

Another important point is that the rise or fall of the phreatic surface is not uniform, and in some nodes the phreatic surface moves several times more than other nodes. In order to keep the elements in reasonable shape it is recommended that all moveable elements be modified rather than only shifting the nodes of phreatic



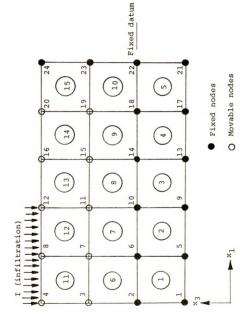


Figure 6-6.--Typical Grid System for Solving Phreatic Aquifer Problem.



elements. When the nodal points are shifted the program is ready to start a new time step.

Steps 1 to 4 are repeated until the system reaches the maximum specified time or the steady state.

6.3.3 Location of Phreatic Surface Using Fixed Elements

As discussed in Section 6.3.2, it is possible to handle the movement of the phreatic surface without altering the nodal coordinates of the elements. In the following method the phreatic surface travels within a fixed finite element grid system.

6.3.3.1 Computation of piezometric heads at the nodes above phreatic surface. -- In order to illustrate the method, a plane linear element, Figure 6-7, is employed. The first part of this development is adopted from France [1971]. Referring to Figure 6-7 which simulates a small portion of the flow domain, the phreatic surface represented by the broken line cuts through elements numbered (1), (3), and (4) at points a, b, c, and d.

For element number (1) the piezometric head ϕ at any point within or along its boundaries is given by:

$$\phi = N_1 \phi_1 + N_2 \phi_2 + N_5 \phi_5 + N_4 \phi_4 \tag{6.3.11}$$



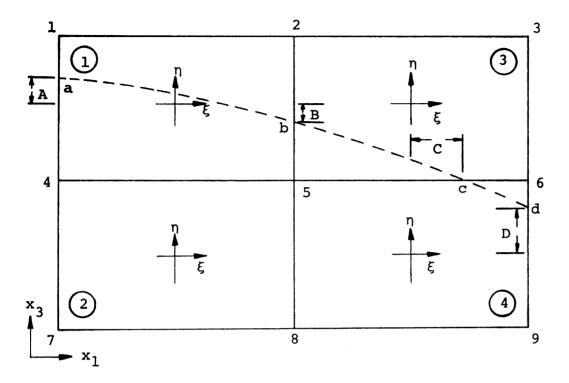


Figure 6-7.--Phreatic Surface Passing Through the Elements [After France 1971].

The shape function for a plane linear element can be written [Zienkiewiez 1971, p. 109]

$$N_{n} = \frac{1}{4} (1 + \xi_{0}) (1 + \eta_{0})$$
 (6.3.12)

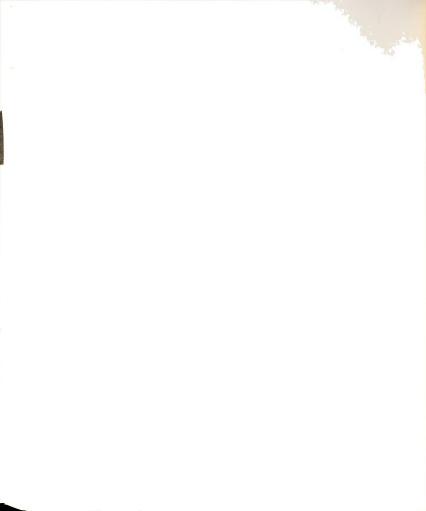
where

$$\xi_0 = \xi \xi_j$$
 and $\eta_0 = \eta \eta_j$

For element number (1) the shape functions are

$$N_1 = \frac{1}{4} (1-\xi) (1+\eta)$$
 $N_2 = \frac{1}{4} (1+\xi) (1+\eta)$

$$N_4 = \frac{1}{4} (1-\xi) (1-\eta)$$
 $N_5 = \frac{1}{4} (1+\xi) (1-\eta)$



At point (a) along line 1-4 (ξ = -1), the piezometric head is

$$\phi_{a} = \frac{1}{4} \left\{ 2(1+A)\phi_{1} + (0)(1+A)\phi_{2} + 2(1-A)\phi_{4} + (0)(1-A)\phi_{5} \right\} \quad (6.3.13)$$

where A is the value of η at point (a), and $-1 \leq A \leq 1$. Finally,

$$\phi_{a} = \frac{1}{2} \{ (1+A) \phi_{1} + (1-A) \phi_{4} \}$$
 (6.3.14)

Similarly for point (b) along line 2-5,

$$\phi_{b} = \frac{1}{2} \{ (1+B) \phi_{2} + (1-B) \phi_{5} \}, -1 \le B \le 1$$
 (6.3.15)

For element number (3) the piezometric head is given as

$$\phi = N_2 \phi_2 + N_3 \phi_3 + N_6 \phi_6 + N_5 \phi_5 \tag{6.3.16}$$

At point (c) on line 5-6 the head is

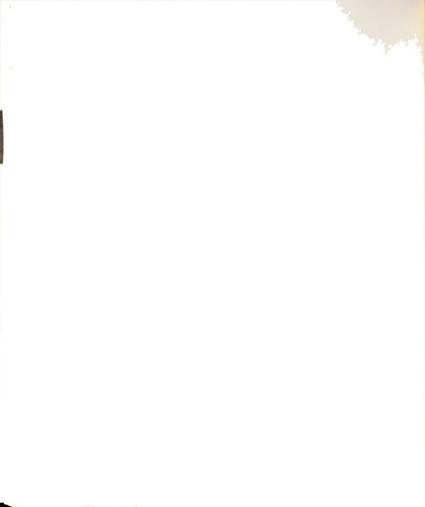
$$\phi_{c} = \frac{1}{2} \{ (1+C)\phi_{6} + (1-C)\phi_{5} \}, -1 \le C \le 1$$
 (6.3.17)

For element number (4) the head is given by

$$\phi = N_5 \phi_5 + N_6 \phi_6 + N_9 \phi_9 + N_8 \phi_8 \tag{6.3.18}$$

At point (d) on line 6-9,

$$\phi_{d} = \frac{1}{2} \{ (1+D)\phi_{6} + (1-D)\phi_{9} \}, -1 \le D \le 1$$
 (6.3.19)



The objective is to determine the piezometric head distribution on and below the phreatic surface. However, the finite element method yields values only at the nodal points, including those points above the phreatic surface. It is therefore necessary to define the values of ϕ_1 , ϕ_2 , ϕ_3 , and ϕ_6 in terms of ϕ_a , ϕ_b , and ϕ_d . Since ϕ_3 does not affect values along 2-5 and 5-6, it can be determined arbitrarily as will be discussed later. The piezometric head at node 6, ϕ_6 , can be specified either in terms of ϕ_c or ϕ_d .

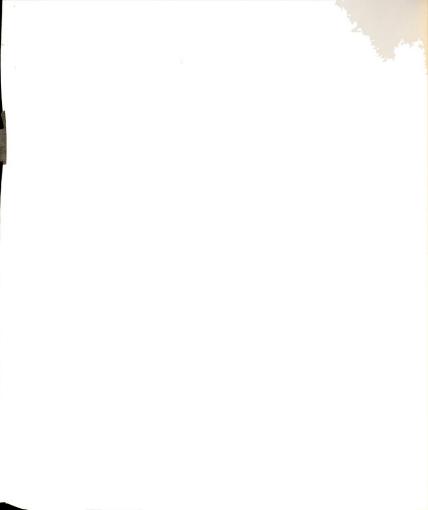
From Equations (6.3.14), (6.3.15), and (6.3.19); ϕ_1 , ϕ_2 , and ϕ_6 can be derived:

$$\phi_{1} = [2\phi_{a} - (1-A)\phi_{4}]/(1+A)$$

$$\phi_{2} = [2\phi_{b} - (1-B)\phi_{5}]/(1+B)$$

$$\phi_{6} = [2\phi_{d} - (1-D)\phi_{9}]/(1+D)$$
(6.3.20)

It is necessary to know the values of ϕ_a , ϕ_b , and ϕ_d . This presents no difficulty since on the phreatic surface the piezometric head equals the elevation head. Thus Equation (6.3.20) can be used to interpolate the values of the piezometric heads at the nodes above the phreatic surface, based on the location of the free surface and known piezometric head at the nodes beneath the surface. Equation (6.3.20) which has been derived by France [1971]



is simply a linear interpolation between three points where the distance between the nodes and the values of two nodes are known. It is desired to find the unknown value at the third point. This can easily be shown using a one-dimensional, two-node element.

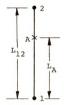


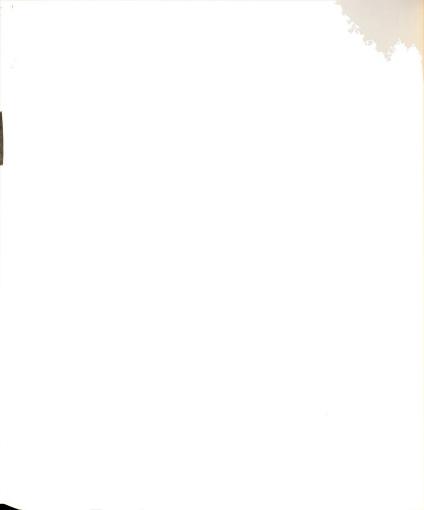
Figure 6-8.--One-Dimensional Linear Element.

In Figure 6-8, let ϕ_1 and ϕ_A be known with the intent to find ϕ_2 . The distances between the nodes are defined in the above figure. Shape functions for nodes 1 and 2 are

$$N_1 = \frac{L_{12} - L_A}{L_{12}}$$
; $N_2 = \frac{L_A}{L_{12}}$

By definition,

$$\phi_{A} = N_{1} \phi_{1} + N_{2} \phi_{2} \tag{6.3.21a}$$



$$\phi_2 = (\phi_A - N_1 \phi_1)/N_2$$
 (6.3.21b)

Substituting the shape functions into Equation (6.3.21) yields

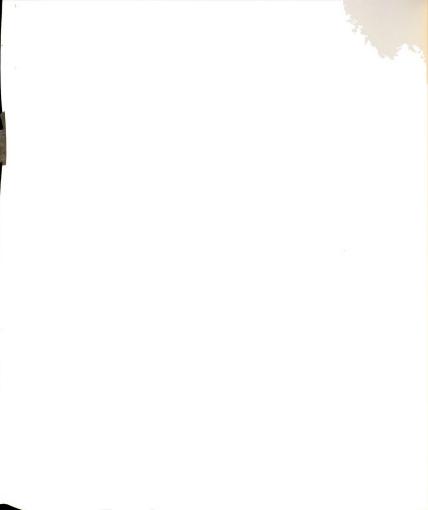
$$\phi_2 = \frac{L_{12}\phi_A - (L_{12} - L_A)\phi_1}{L_A}$$
 (6.3.22)

Define B = $\frac{2L_A}{L_{12}}$ - 1 such that -1 \leq B \leq 1.

Substituting A in Equation (6.3.22) and simplifying,

$$\phi_2 = \frac{2\phi_A - (1 - B)\phi_1}{(1 + B)}$$
 (6.3.23)

Equation (6.3.23) is identical to Equation (6.3.20) which has been derived using the properties of an isoparametric element. Usually, $\phi_{\rm A}$ represents the piezometric head at the phreatic surface, ϕ_1 is a known head within the system, and calculated ϕ_2 is a Dirichlet boundary condition. The problem associated with Equation (6.3.21b) or Equation (6.3.23) is that it is singular when N₂ $^{\rightarrow}$ 0 or B $^{\rightarrow}$ -1, i.e., when the point A is close to point 1. In practice, when the phreatic surface rises due to infiltration or other hydraulic stresses, Equation (6.3.23) does not provide a reasonable estimate for ϕ_2 .



One way to obtain a better estimate for phreatic nodes (Figure 6-10) is to use two interior nodes besides ϕ_A to evaluate unknown ϕ . Let ϕ_1 and ϕ_2 be known and ϕ_A represent the value of the piezometric head at the phreatic surface. It is desired to find ϕ_3 (Figure 6-9). Using the properties of a one-dimensional quadratic element (see Section 5.4),

$$\phi_3 = \frac{\phi_A - N_1 \phi_1 - N_2 \phi_2}{N_3} ; N_3 \neq 0$$
 (6.3.24)

where

$$N_{1} = \frac{(L_{A} - L_{12}) (L_{A} - L_{13})}{(L_{12}) (L_{13})}$$

$$N_2 = \frac{-(L_A)(L_A - L_{13})}{(L_{12})(L_{23})}$$

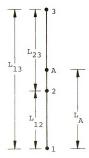


Figure 6-9. -- One-Dimensional Quadratic Element.



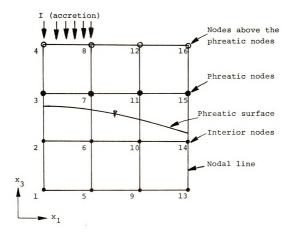
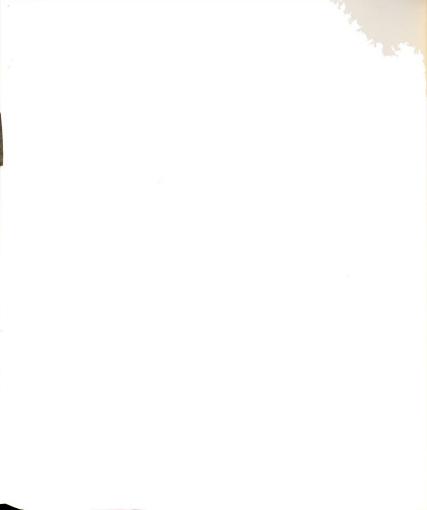


Figure 6-10.--Location of the Phreatic Surface and Definition of the Terms.



$$N_3 = \frac{(L_A)(L_A - L_{12})}{(L_{13})(L_{23})}$$

Although Equation (6.3.24) provides justifiable results it still becomes singular when $N_3 \rightarrow 0$. A possible way to compensate for the singularity of N_2 in Equation (6.3.21) or N_3 in Equation (6.3.24) is to use the node below point 1 of Figure 6-8, when point A is close to point 1. The scheme will be similar to Figure 6-9 but

$$\phi_3 = \frac{\phi_A - N_1' \phi_1}{N_2'} \tag{6.3.25}$$

where

$$N_1' = \frac{L_{13} - L_A}{L_{13}}$$
; $N_2' = \frac{L_A}{L_{13}}$

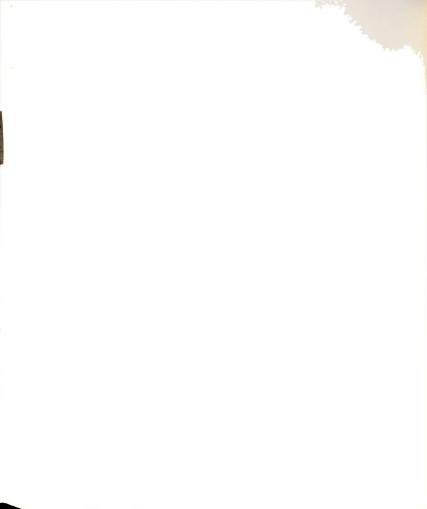
Equation (6.3.25) will never be singular, but since it is using ϕ_1 rather than ϕ_2 it underestimates ϕ_3 because usually $\phi_1 < \phi_2$.

Returning to Figure 6-8, there is a third way to compute ϕ_2 and it is based on Darcy's Law:

$$V_3 = -\frac{K_{33}}{n_e} \frac{\partial \phi}{\partial x_3}$$

or

$$\frac{\partial \phi}{\partial x_3} = -\frac{V_3}{K_{33}} n_e$$



$$\frac{(\phi_2 - \phi_A)}{(x_3)_2 - (x_3)_A} = -\frac{v_3}{\kappa_{33}} n_e$$

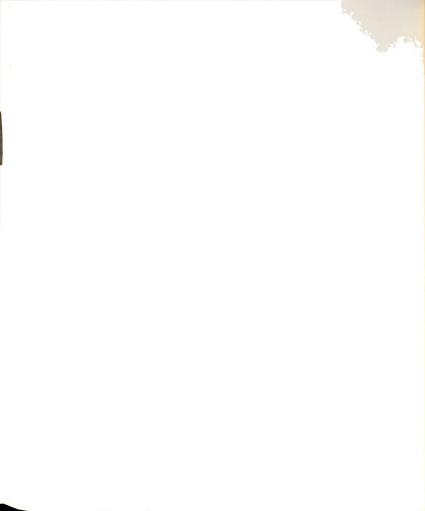
Since $(x_3)_A \equiv \phi_A$, then

$$\phi_2 = \phi_A - \frac{V_3^n e}{K_{33}} \left((x_3)_2 - \phi_A \right)$$
 (6.3.26)

 V_3 and K_{33} are known values at point A.

In Equation (6.3.26), ϕ_2 is calculated based on the location of the phreatic surface. If the values of V_3 and K_{33} are evaluated correctly, then it is believed that Equation (6.3.26) will provide accurate results for ϕ_2 without under or overestimating its values. When the phreatic surface is horizontal, V_3 is zero and thus Equation (6.3.26) is not applicable. In this case, for the first time interval Equation (6.3.25) is employed to compute the value of the piezometric head at the points above the phreatic surface. For the second time interval and later times, one will instead use Equation (6.3.26). It can be concluded that among the Equations (6.3.23) through (6.3.26), Equation (6.3.25) gives a reasonable value at the first time interval and Equation (6.3.26) provides a better estimate at later times.

6.3.3.2 Computation of velocity vectors at phreatic surface.--When calculating the shifting distance for the phreatic surface (Equation 6.3.8), it is necessary



to know the velocity components on that surface. Since the phreatic surface moves within fixed nodes, in most cases it will intersect the element line as shown in Figure 6-7. The velocity components are known at the nodes. Consider again Figure 6-8 to calculate \mathbf{V}_1 and \mathbf{V}_3 at point A between nodes 1 and 2. Using an equation similar to Equation (6.3.21a) to compute velocity vectors, one will have

$$(v_1)_A = \frac{(1-B)(v_1)_1 + (1+B)(v_1)_2}{2}$$
 (6.3.27)

where

$$B = \frac{2L_A}{L_{12}} - 1$$

$$\mathbf{L}_{\mathbf{A}} = \sqrt{\left[\left(\mathbf{x}_{1} \right)_{1} - \left(\mathbf{x}_{1} \right)_{\mathbf{A}} \right]^{2} + \left[\left(\mathbf{x}_{3} \right)_{1} - \left(\mathbf{x}_{3} \right)_{\mathbf{A}} \right]^{2}}$$

$$\mathbf{L}_{12} = \sqrt{\left[\left(\mathbf{x}_{1}\right)_{2} - \left(\mathbf{x}_{1}\right)_{1}\right]^{2} + \left[\left(\mathbf{x}_{3}\right)_{2} - \left(\mathbf{x}_{3}\right)_{1}\right]^{2}}$$

Similar equations can be written for $(v_3)_A$.

6.3.3.3 The procedure for solving piezometric heads with fixed nodes. -- The procedure for solving the piezometric heads in an unconfined aquifer with this method is as follows:

Step 1. The phreatic surface is initially chosen to coincide with the element boundaries. It is not



necessary that this boundary be the upper limit of the grid system, i.e., the phreatic surface might be specified within the system. This is especially important when a rise of the water table is expected. For example, in Figure 6-6 it is expected that the phreatic surface will rise, and line 2-6-10-14-18-22 can represent the free surface at the initial time. A finite element solution is performed, velocity components are calculated, and the phreatic surface is shifted as in the fixed element method.

Step 2. The piezometric head values at points such as a, b, and d of Figure 6-7 are stored. Their values are simply the elevation heads at the respective points. The piezometric head values at nodes 1, 2, and 6 are calculated using Equation (6.3.25), and an arbitrary value is assigned to node 3 as will be discussed later.

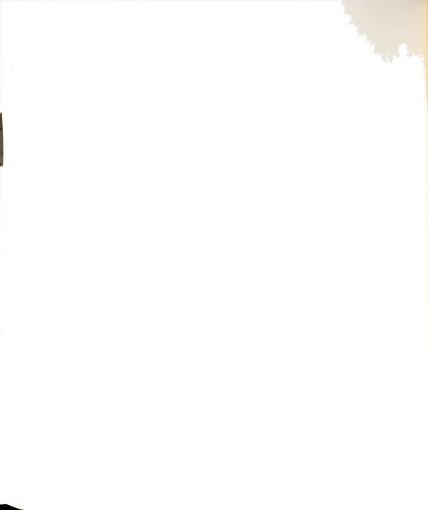
Step 3. Since the element stiffness matrices are unaltered, all that is necessary is to insert these new Dirichlet boundary values in the governing set of simultaneous equations and solve for a new set of piezometric heads.

Step 4. The velocity components are evaluated and the phreatic surface is shifted as mentioned in Section 6.3.3.2, using Equation (6.3.5).

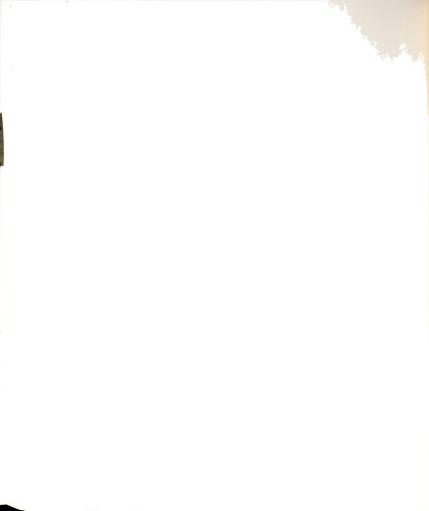


Steps 2, 3, and 4 are repeated until the maximum required time is achieved or until the system has reached the steady-state condition. For the second time interval or later times Equation (6.3.26) is used in step 2.

6.3.3.4 Assigning the value of piezometric heads above phreatic nodes .-- Assigning an arbitrary value of piezometric head to node 3 (Figure 6-7) or to similar nodes is an important task. To clarify the discussion consider Figure 6-10 where the position of the water table is shown, and the phreatic nodes and the nodes above the phreatic nodes are defined. The piezometric heads at the phreatic nodes, i.e., nodes 3, 7, 11, and 15, are calculated using Equation (6.3.26) based on the location of the phreatic surface. The task now is to assign the values of the piezometric heads to nodes above the phreatic nodes, i.e., nodes 4, 8, 12, and 16. In the calculation of the piezometric heads within the system, these values do not play any major role. But they have a great importance particularly when the velocity vectors are calculated simultaneously. It is assumed that there is no capillary fringe above the water table, that an abrupt interface exists between the saturated and unsaturated soil, and the Pressure above the phreatic surface is atmospheric. Due



to the physical characteristics of the porous media there is little or no flow movement above the phreatic surface, except the infiltration due to recharge which is assumed to travel directly to the water table. This concept, together with the above assumptions, directs one to assign the piezometric heads above the nodes such that the physical conditions are satisfied. This is done by equating the piezometric heads in the nodes above the phreatic nodes to the piezometric heads of the phreatic nodes along each nodal line. For example, ϕ_{16} = ϕ_{15} and $\phi_{12} = \phi_{11}$, etc. This technique will provide the desired condition for velocity vectors along x_3 , i.e., V_3 above the phreatic node will be zero, but there will be a gradient along x_1 . This situation will not cause any major problem since the slope of the phreatic surface is small and its effect will be minimal in calculating the shifting distance (Equation 6.3.8). However, to improve the technique it is possible to equate the velocity components at and above the phreatic nodes (Figure 6-10) to zero, which is done in this study. Summarizing, the piezometric head at and above the phreatic nodes will be the same for each nodal line; refer to Figure 6-10. After the velocity components are calculated simultaneously, the velocity vectors along \mathbf{x}_1 at and above the phreatic nodes will be set to zero.



6.3.4 Steady-State Condition

When the system has reached the steady-state condition the phreatic surface will not move considerably, i.e., the normal velocity will approach zero. From Equation (3.4.6),

$$-\kappa_{ij} \frac{\partial \phi}{\partial x_{j}} \ell_{i} - i \ell_{3} = 0$$

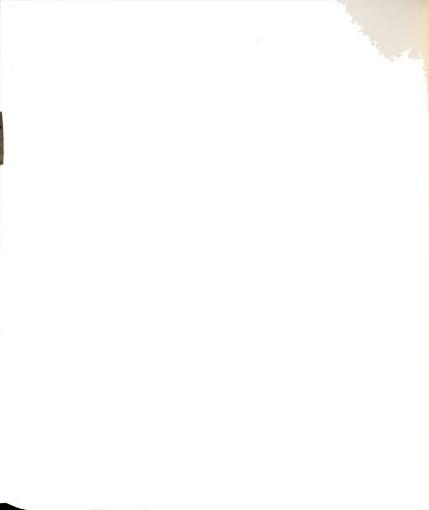
This means that at steady state, normal discharge is equal to the normal component of accretion. In the computer program Equation (6.3.5) is used. The average shifting distance is defined by

$$\bar{d} = \frac{k^{\frac{\Sigma}{2}} (d_n)_k}{NNFS}$$

where NNFS is the number of nodes in the phreatic surface. When \overline{d} is less than or equal to a small value the steady-state condition is reached and the program is halted.

6.3.5 Reasonable Time Step

There is no general agreement for choosing the time step. Rushton and Herbert [see France 1974] have suggested that if the time interval is chosen so that the velocity at which the phreatic surface moves changes by less than 30 percent between successive steps, then it is not necessary to iterate within the time step.



Based on the Lipschitz criteria [Isaacson and Keller 1966, pp. 86-91], Sandhu et al. [1974] have introduced the variable time step. They state that in choosing Δt the following criteria should be satisfied:

$$\frac{\max \left| (x_3)_{k}^{i+1} - (x_3)_{k}^{i} \right|}{\max \left| (x_3)_{k}^{i} - (x_3)_{k}^{i-1} \right|} = \lambda(\Delta t) < 1$$
 (6.3.28)

where k = 1, 2, ..., NNFS, i represents the number of iterations (see Section 6.3.6), and the value of λ is dependent upon Δt . If λ exceeds one, the time step will be reduced by some factor until the time interval satisfies Equation (6.3.28). If, on the other hand, λ is found to be very small compared with one, Δt will be increased by the factor. In the computer program $(x_3)_k^0$ is estimated, the value of the previous time is used, and only two computations of x_3 are required to obtain an estimate for λ and to choose the proper time step.

6.3.6 Iteration Within Time Step

Usually, if one chooses a small time step there will not be any need to iterate within a time step, especially when the rise or fall of the phreatic surface is small compared to the size of the system, such as with



actual field problems. However, Sandhu et al. [1974], by using the mean value theorem and assuming a smooth change of geometry in the time domain, have introduced the following equations for iteration within the time step:

$$\begin{split} \left[\mathbf{x}_{1}(\mathsf{t}+\Delta\mathsf{t})\right]_{k}^{i} &= \left[\mathbf{x}_{1}(\mathsf{t})\right]_{k} + \frac{1}{2}\left\{d_{1}\left[\mathbf{x}_{1}(\mathsf{t})\,,\,\,\mathbf{x}_{3}(\mathsf{t})\right]\right. \\ &+ \left.d_{1}\left[\mathbf{x}_{1}(\mathsf{t}+\Delta\mathsf{t})\,,\,\,\mathbf{x}_{3}(\mathsf{t}+\Delta\mathsf{t})\right]^{i-1}\right\}_{k} \\ \left[\mathbf{x}_{3}(\mathsf{t}+\Delta\mathsf{t})\right]_{k}^{i} &= \left[\mathbf{x}_{3}(\mathsf{t})\right]_{k} + \frac{1}{2}\left\{d_{2}\left[\mathbf{x}_{1}(\mathsf{t})\,,\,\,\mathbf{x}_{3}(\mathsf{t})\right]\right. \\ &+ \left.d_{2}\left[\mathbf{x}_{1}(\mathsf{t}+\Delta\mathsf{t})\,,\,\,\mathbf{x}_{3}(\mathsf{t}+\Delta\mathsf{t})\right]^{i-1}\right\}_{k} \end{split}$$

i and k are defined in Section 6.3.5. The inside bracket means that d_1 and d_2 [see Equation (6.3.8)] are calculated at $[x_1(t), x_3(t)]$ and $[x_1(t+\Delta t), x_3(t+\Delta t)]$. Since the change of x_3 is greater than the x_1 -value after every iteration, a check is made where

$$[\,(\mathbf{x}_{3})_{\,k}^{\,i}\,-\,(\mathbf{x}_{3})_{\,k}^{\,i-1}]/\,|\,(\mathbf{x}_{3})_{\,k}^{}|$$

is less than or equal to ϵ (say, 10^{-5}). If this condition is satisfied the iteration is complete. Usually one or two iterations will be sufficient. The same technique is used in this study.



6.3.7 Comparison Between the Two Methods of Locating the Phreatic Surface

Figures 6-11 and 6-12 depict the schemes for solving the piezometric heads in an unconfined aquifer with the method of modifying elements (method one) and the method of fixed nodes (method two). It is extremely difficult to compare the two methods comprehensively. However, each technique has specific advantages as outlined below.

- (a) In the first method the locations of nodal points are changing and the system is shrinking or expanding with time, in the second procedure the grid system is fixed and the coordinates of nodes are time invariant. The second method enables one to introduce convective-dispersion equations into the system, a major advantage of this technique.
- (b) In the first method it is necessary to reconstruct the global matrices and decompose them for every time step, while it is not required in the second method. Thus the fixed node procedure needs less computer time than the method of modifying elements.
- (c) Another advantage of using the fixed grid technique is that the position of the nodal point is fixed and the shape of each element does not change with time. Thus it is possible to calculate the location of the phreatic surface in any anisotropic and heterogeneous



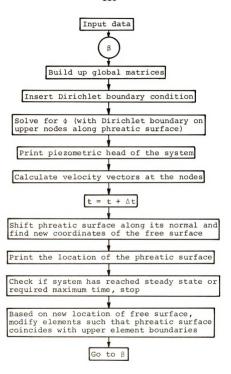
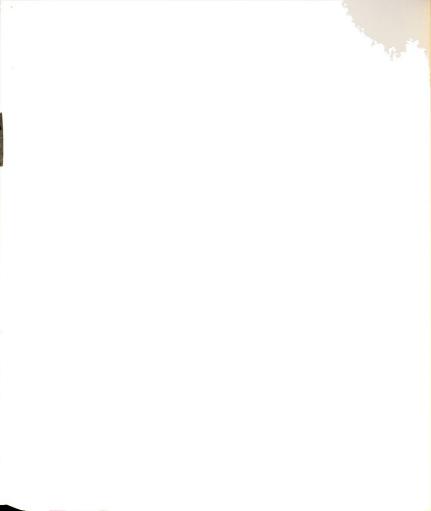


Figure 6-11.--Scheme for Solving the Piezometric Heads in Unconfined Aquifers With the Movable Node Technique.



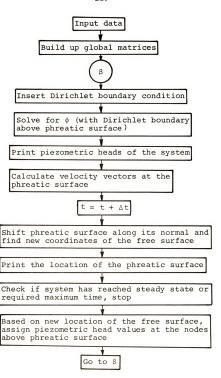
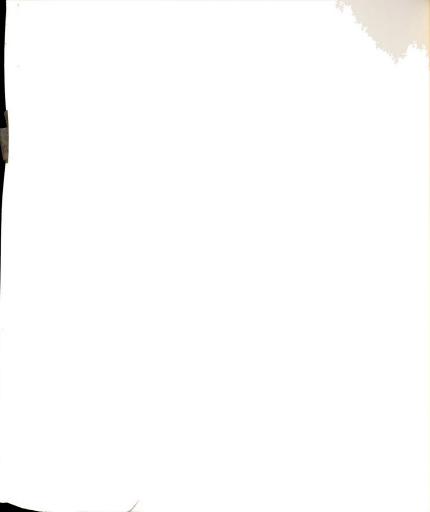


Figure 6-12.--Scheme for Solving the Piezometric Heads in Unconfined Aquifers With the Fixed Node Technique.



porous media. But in the modifiable element method, nonhomogeneity can be handled only for fixed node sections or between the columns of nodal lines.

(d) In the first method the boundary condition is evaluated once to yield the location of the phreatic surface. In the second method the Dirichlet boundary condition at the nodes above the free surface is computed based on the position of the phreatic surface. As discussed in Section 6.3.3.1 there is no concrete method to evaluate the piezometric heads above that surface. Among the equations which are presented, Equation (6.3.26) gives reasonable results. Still, interpolation of the values of V_3 and K_{33} , and assigning the value of piezometric heads for nodes above the phreatic nodes, need to be investigated more.

6.3.8 Summary

The calculation of the piezometric heads in confined and unconfined aquifers is discussed. A modified technique of calculating the phreatic surface with fixed nodes is presented, and it is shown that the velocity vectors can be calculated simultaneously to provide continuous functions at the nodes without using large computer memory. The feasibility of modeling the dispersion phenomena in an unconfined aquifer with a transient phreatic surface, which has become of universal interest, is apparent.

CHAPTER VII

SOLUTION OF SYSTEM EQUATIONS FOR CONVECTIVE-DISPERSION PHENOMENA

7.1 Introduction

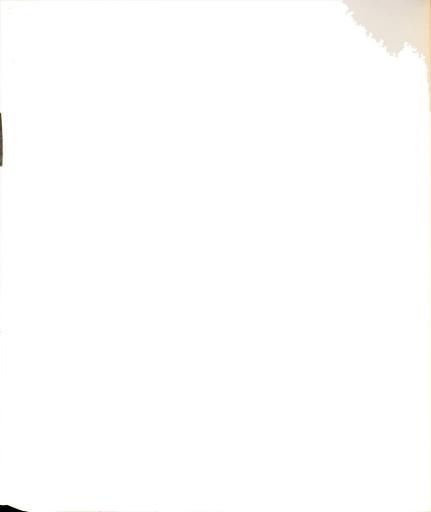
In this chapter the procedure of describing the convective-dispersion phenomena is discussed. Related mathematical equations are described in Section 3.5 and the finite element formulation is given in Section 4.3. For most practical purposes at relatively low concentrations, it can be assumed that the concentration of a tracer does not affect the velocity distribution [Bear 1972]. Hence the solution of a dispersion problem is made up of two independent subproblems. First, the velocity distribution is determined for all points of the flow domain. Second, the resulting velocity distribution is inserted into the dispersion equation, which in turn is solved to vield the concentration distribution in the flow domain. Velocity vectors play a dominant role in the convective-dispersion equation. They appear in the convective term and dispersion coefficients, and hence realistic and accurate evaluation of the velocity components is very important. Proposed simultaneous solution of the velocity vectors from piezometric heads



(Section 4.4.2) provides continuous velocities at each node. The concept is also presented by Segol et al. [1975], where the pressure and velocity components are computed simultaneously. For an almost homogeneous liquid, the proposed method (in Section 4.4) has some advantages over their procedure: (1) the required storage space in the computer program is significantly reduced; (2) the transient solution of the velocity vectors is feasible; and (3) the boundary conditions are simple to apply.

7.2 Calculation of Dispersion Coefficients

The hydrodynamic dispersion coefficient defined in Section 3.5.2 regulates the degree of spreading of the contact zone between two miscible fluids. For most practical purposes the molecular diffusion coefficient is negligible compared with the mechanical dispersion coefficient. The dispersion coefficient is given by Equation (3.5.1). The longitudinal and transversal dispersivities are the two components of the mechanical dispersion coefficient, and are considered as porous media properties. In this study they are assumed to be constant over the entire domain. The reported values of a_I and a_{II} both range between 4 and 135 meters [Pinder 1973, Robertson 1974].



Dispersion coefficients can be either evaluated at a node or assumed to be constant over each element at a given instant. For a transient flow where velocity components are also time variant, it is necessary to compute the dispersion coefficient at each node for every time step and reconstruct the global matrices (see Section 4.3). Consequently, the time dependency of the dispersion coefficient will increase the computation time considerably.

7.3 Computation of Tracer Concentration

The finite element formulation of the convectivedispersion equation leads to a system of ordinary differential equations of the form

$$[H] \{ \frac{\partial C}{\partial t} \} + ([K] + [S]) \{C\} + [E]^T [I] \{C\} = [H] \{\dot{q}_C\} + \{F\}$$

$$(4.3.13)$$

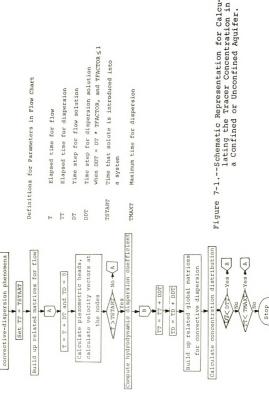
The two most common boundary conditions, namely, the Dirichlet and the Neumann boundary conditions, are used in solving tracer concentration. The Dirichlet boundary condition is handled by specifying the known concentration at the appropriate boundary nodes and factoring out rows and columns in the coefficient matrix associated with those nodes as discussed in Section 6.1.2. The known mass flux along the boundary line is incorporated in the {F} vector. The procedure of the allocation of



a constant line source to nodal points is given in Appendix I. Since it is assumed that at the boundaries either the concentration or the flux of the tracer is defined, the matrix [E] in this study is assumed to be zero. In order to construct Equation (4.3.13), three parameters (i.e., dispersion coefficient, velocity vectors, and velocity gradients) should be known. Velocity vectors are calculated at the nodes, and the other two parameters are evaluated at the integration point. All three are then introduced into Equation (4.3.12).

The dispersion equation for flow in a confined or an unconfined aquifer is similar. With the fixed node technique described in Section 6.3.3, the piezometric heads in a phreatic aguifer can be obtained without repositioning the nodal coordinates of the elements; thus the velocity vectors are known for a given point at any time step. The movement of a tracer is predicted by introducing the calculated velocity vectors in the convective-dispersion equation. The nodal coordinates of the grid system for the tracer will coincide with the nodal points of flow. Thus, either the grid system for flow and dispersion will be identical, or the grid system for predicting the tracer concentration will lie within the grid system of flow prediction. The procedure for calculating the tracer concentration in a confined or unconfined aguifer is given in Figure 7-1.





Input data for flow and

Figure 7-1.--Schematic Representation for Calcu-

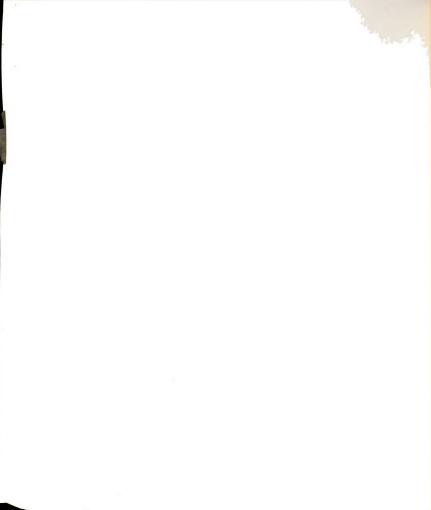


7.4 Discretization of Time Derivatives and Solution of System Equation

Equation (4.3.13) is a set of first order ordinary differential equations. Different methods to discretize the time derivative of the family of equations similar to Equation (4.3.13) were discussed in detail in Chapter V. Using Equation (5.1.5), Equation (4.3.13) takes the following form:

$$[\bar{B}]\{C(t+\Delta t)\} = \{G\}$$
 (7.4.1)

where [B] is a nonsymmetric banded matrix, C is the unknown concentration for time $t + \Delta t$, and $\{G\}$ is a known force vector. Since [B] is nonsymmetric, the Cholesky method (see Section 6.1.1) is not applicable. The Gauss elimination technique [Carnahan 1969, pp. 270-272] is used to solve the system of simultaneous equations. The subroutine GELB [IBM Application Program, 1968] has been adopted for solving the system of equations for the dispersion equation. Initially, a two-point backward finite-difference scheme is used, where the initial condition of C at time t is given and $C(t + \Delta t)$ is sought. As the computation progresses, solutions of C at a few prior time steps become known and they may be stored in the core. Then a higher order time approximation of Equation (4.3.13) is used as derived in Chapter V, to give a more accurate solution or to permit a larger time



step without additional penalty of excessive computer time. As can be seen from Equation (5.1.5) the [A] and [H] matrices are fixed, and only their coefficients will change if the order of approximation or the time step changes. The additional required memory for each increased order of approximation will be 2 × NNDS, where NNDS is the number of nodes. The results of using different methods of discretization of the time step are discussed in Chapter IX.

7.5 Stability and Convergence Criteria

There are two important concepts closely associated with the convergence of a particular numerical procedure, namely, those of consistency and stability.

Carnahan et al. [1969] define the stability and consistency as the following: "In general, a solution is said to be unstable if errors introduced at some stage in the calculation . . . are propagated without bound throughout subsequent calculations." The term consistency means that the numerical procedure may in fact approximate the solution of the partial differential equation under consideration and not the solution of some other equations. There is no definite rule for defining stability criteria by the finite element method. For the finite difference explicit scheme, Fried and Combarnous [1971] give a



relation for the stability of longitudinal dispersion which can be written as follows:

$$\sqrt{2} D_{L} \Delta t \leq \Delta x_{1} \leq 2 D_{L} / V_{1}$$
 (7.5.2)

The stability function for the two-dimensional dispersion equation solved by the explicit finite difference method was obtained by Reddell and Sunada [1970], and is summarized as

$$D_{11}$$
; $D_{22} > 0$ (7.5.3a)

$$4 D_{11} D_{22} > (D_{12} + D_{21})^2$$
 (7.5.3b)

$$\frac{D_{11} \Delta t}{(\Delta x_1)^2} + \frac{D_{22} \Delta t}{(\Delta x_2)^2} \le \frac{1}{2}$$
 (7.5.3c)

Although Equation (7.5.3) was derived for finite differences, it reveals the limitation of choosing Δx_1 and Δt . As discussed in Chapter VIII, for $\theta \geq 1/2$ (Chapter V) the result of exceeding the time step restriction is a stable but oscillatory solution, and as θ approaches 2/3, the oscillation decreases substantially. However, giving any specific θ for stability criteria will be premature.

Using the higher order time derivative approximation of Equation (4.3.13) will improve the convergence of the finite element solution.



CHAPTER VIII

NUMERICAL RESULTS FOR SIMULATION OF GROUNDWATER FLOW

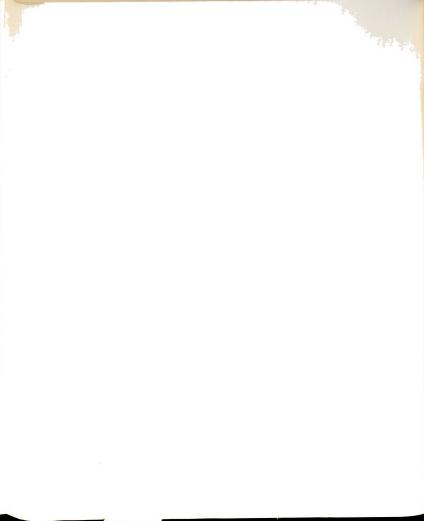
In this chapter the validity of the finite element numerical simulation of the flow in confined and unconfined aquifers with a phreatic surface is discussed. The mechanisms of locating the free surface with fixed nodes and with modifiable elements are verified. Also, the results of the two methods for velocity calculation, i.e., direct and simultaneous procedures, are presented.

8.1 Flow in a Confined Aquifer

The finite element solution of the flow equation for a confined aquifer has been accomplished successfully many times and will not be the subject of detailed review in this study. Since success of the convective-dispersion model is highly dependent on the hydrologic simulation model, one has to be certain that the hydrodynamic simulator provides accurate results.

8.1.1 Pumping in a Single Well Field

Consider a single well pumping from a confined aquifer of nearly infinite area (Figure 8-1). The



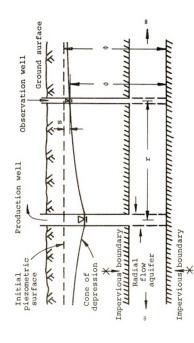


Figure 8-1.--Nonleaky Artesian Aquifer With a Fully Penetrating Well [After Walton 1970].



analytical solution for transient radial flow in an isotropic nonleaky artesian aquifer with fully penetrating well and a constant-discharge condition is:

$$s = \frac{P}{4\pi T} [-E_{i}(-u)]$$

$$u = \frac{r^{2}S}{4Tt}$$
(8.1.1)

where s is drawdown, r is the distance from pumped well to the observation point, P is discharge, t is time, T is transmissibility, S is the storage coefficient, and $-E_{\frac{1}{2}}(-u)$ is the exponential integral [Davis and DeWiest 1966].

Due to symmetry of the flow field, it is sufficient to model only one-fourth of the aquifer, shown in Figure 8-2. No-flow boundaries are taken 4,800 meters from the well located at point W, and initially the drawdown s is zero. The following parameters are used in the modeling: $T = 929 \text{ m}^2/\text{day}$, S = 0.01, and $P = 946 \text{ m}^3/\text{day}$. Figure 8-3 shows dimensionless drawdown versus dimensionless time at locations A and B (Figure 8-2), at distances of 1,697 and 2,163 meters from the well, respectively. Numerical results compare favorably with the analytical solution, and the deviation of numerical results from the theoretical curve is within acceptable range. It has been observed that the size



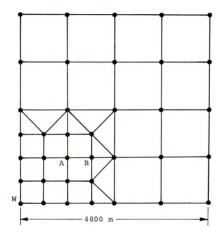


Figure 8-2.--Grid System Which Is Used in Simulating a Single Well Field.

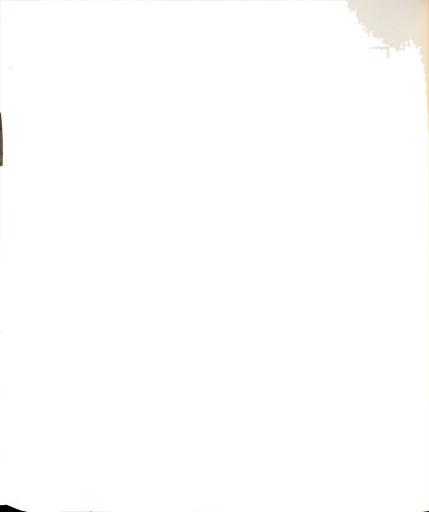


Figure 8-3.--Comparison of the Numerical and Analytical Solutions for Drawdown, 1697 and 2163 Meters Away From the Well.



and type of elements affect the accuracy of the numerical results.

8.1.2 Effects of Various Time Approximations on the Accuracy of the Results

In order to examine the effects of the kinds of time approximation on the results of the numerical simulator, a small confined aquifer, $1828.8 \times 1219.2 \text{ m}$ (6000 × 4000 ft), was modeled (Figure 8-4). Initially, the system is at the steady state with zero piezometric heads. Along boundary lines AC and BD the system is maintained at zero potential, while no-flow boundary conditions are assumed along the AB and CD sides. A well is located at the center of the medium at point W. The following parameters were used in simulating the groundwater movement: $P = 556.4 \text{ m}^3/\text{day}$ (2.0E + 4 ft $^3/\text{day}$), S = 0.01, and $T = 929.6 \text{ m}^2/\text{day}$ (10,000 ft $^2/\text{day}$).

Three different θ 's, i.e., $\theta=1/2$, $\theta=2/3$, and $\theta=7/12$ which is the average of $\theta=1/2$ and $\theta=2/3$, were used for first order time approximation (see Equation 5.1.5). The related coefficients for $\theta=2/3$ and 1/2 are given in Table 5-2, and the coefficients for $\theta=7/12$ are as follows:

$$a_{11} = \frac{7}{12}$$
, $a_{12} = 1$., $a_{13} = 1$., $a_{21} = -\frac{5}{12}$, $a_{22} = 1$.



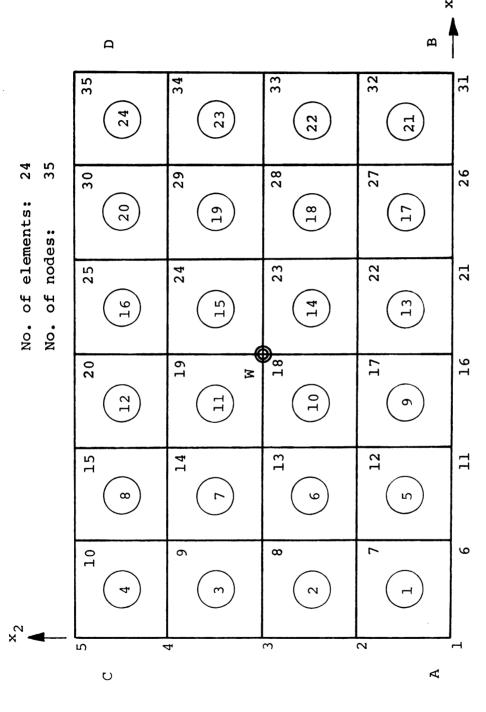
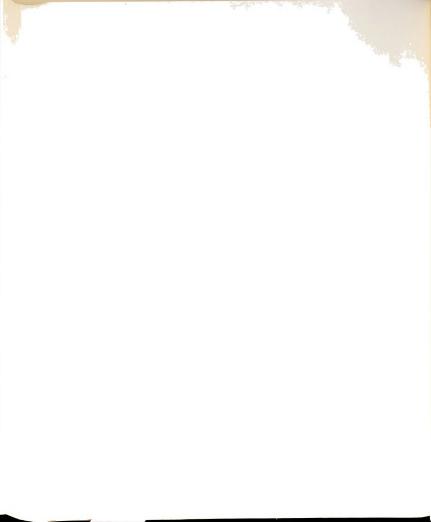


Figure 8-4.--Finite Element Representation of a Confined Aquifer With Linear Isoparametric Quadrilateral Elements.



The other coefficients of Equation (5.1.5) will be zero. The results of the calculated piezometric heads at point W with different θ 's for $\Delta t = 0.05$ day are depicted in Figure 8-5. For $\theta = 1/2$, the piezometric heads oscillate substantially when compared with the results of $\theta = 2/3$. Although $\theta = 7/12$ is the average of the two previously mentioned θ values, the observed oscillation is less than for $\theta = 1/2$. All three θ 's provide the same results for steady state as one might have anticipated. To see the effects of θ more clearly, the time interval is increased 20 times to $\Delta t = 1.0$ day. The piezometric heads for a maximum time of 50 days for point W are shown in Figure 8-6. As can be observed, piezometric heads oscillate at an early time for θ = 2/3 and 7/12; however, the oscillations diminish very rapidly and converge to steady-state condition. In contrast, for $\theta = 1/2$, the piezometric heads oscillate about the mean but the oscillation does not vanish. From this example and similar studies that have been carried out, it is believed that for groundwater problems, θ = 2/3 provides fewer oscillations when compared with θ = 1/2 and 7/12. The θ = 2/3 and 7/12 solutions converge, but $\theta = 1/2$ oscillates about the mean for continued time.



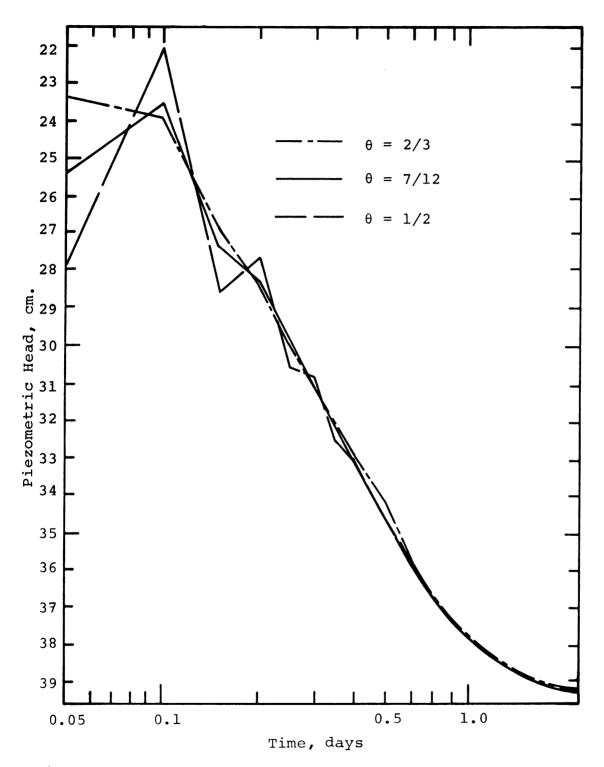


Figure 8-5.--Calculated Piezometric Heads Versus Time at Point W Using Different θ With Δt = 0.05 Day.



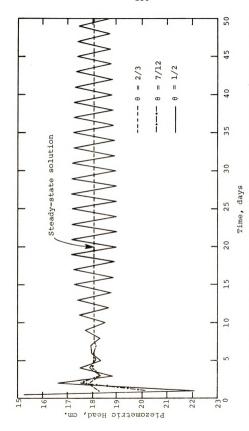


Figure 8-6.--Calculated Piezometric Heads Versus Time at Point W Using Different θ With Δt = 1.0 Day.



8.2 Comparison of Two Different Methods of Solution of the Velocity Vectors

As discussed in Section (6.2), the velocity vectors can be calculated either simultaneously or by the direct method. To show the discontinuity of velocity vectors at element interfaces or nodes calculated by the direct method, a regional aquifer is taken and shown in Figure 8-4. The area and applied boundary conditions are shown in the figure, with a well located at point W. The system has reached a steady-state drawdown condition due to pumping and the piezometric heads are known. Equation (4.4.3) is used to calculate the velocity vectors at the nodes and at the centroid of the elements for the direct method, while Equations (4.4.11) and (4.4.13) provide simultaneous calculation of the velocity vectors at the nodes.

Figures 8-7 and 8-8 show the velocity vectors at the nodes, computed by direct and simultaneous methods for the aquifer depicted in Figure 8-4. The numbers in each box represent the velocity vectors calculated simultaneously, while the ones in the corner of the elements are obtained by the direct method. It is obvious that direct calculation does not provide continuous velocities at the nodes. For example, if one takes element number 11 and evaluates $\rm V_1$ at node 18, a value of 7.6 will be obtained. But $\rm V_1$ for the same node from element number 15



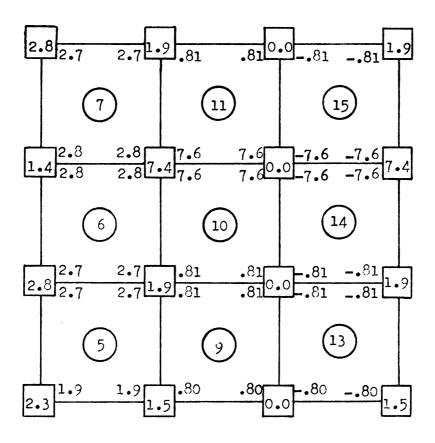


Figure 8-7.--Computed Values of V_1 at the Nodes by Direct and Simultaneous Methods.



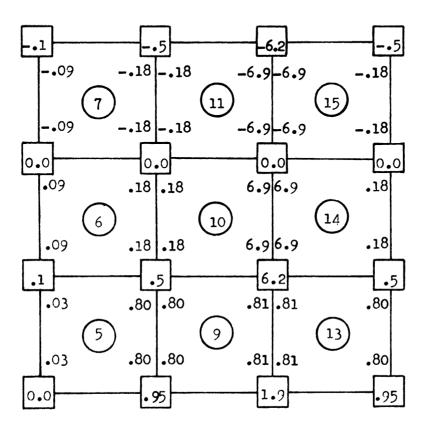


Figure 8-8.--Computed Values of V_2 at the Nodes by Direct and Simultaneous Methods.



will be -7.6. The true velocity at that node is zero, which is obtained by simultaneous calculation. Averaging the velocity vectors computed by the direct method might yield the values obtained by the simultaneous method, but the procedure is not valid for every location. The computation time for both methods is almost the same. The simultaneous method requires more core for storing the capacitance matrix [H] and force vectors $\{F_{\mathbf{x}}\}$ and $\{F_{\mathbf{y}}\}$. It is believed that the Galerkin formulation of the Darcy law will yield a reasonable estimate of velocity vectors (or flux) at the nodes, provided the calculated piezometric heads are good representations of the actual field conditions.

An important point which has been observed in the different numerical results is that the value of the velocity vectors at the boundaries might depart slightly from the true answer. The deviation might be due to the construction of the element matrices, because some nodes along the boundary are weighted less in the formation of the global matrix and force vectors. In practice this would not cause any problem if the hydrodynamic model is used in the prediction of the tracer concentration. The grid system for the dispersion model might be placed within the hydrodynamic model, or taken slightly smaller in order to minimize the effect of this discrepancy.



In this study, simultaneous velocity vectors are used in the convective-dispersion program to calculate convective terms and dispersion coefficients.

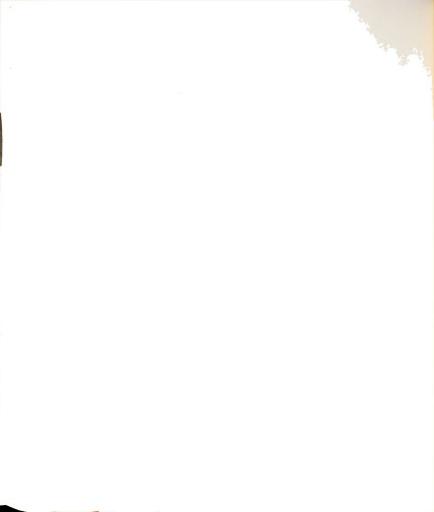
Simultaneous and direct calculated velocity vectors are used for locating the phreatic surface.

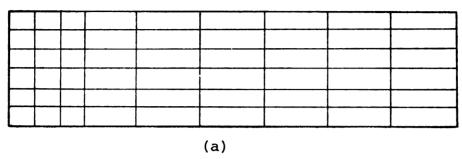
8.3 Flow in a Phreatic Aquifer

The mechanisms of locating the phreatic surface with movable and fixed node techniques are presented in detail in Chapter VI. In this section the existing experimental and field observation data is used to verify the validity of the two methods. In all numerical examples illustrated hereafter, linear isoparametric quadrilateral elements are employed, and DTMAX stands for the maximum specified time step.

8.3.1 Transient Buildup of a Mound Due to Accretion

Using the linearized technique, Marino [1967], following Hantush [1963], gives the analytical solution for growth and decay of a mound due to recharge. He verified the analytical solution with a Hele-Shaw model. A strip of finite height and infinite length was chosen for numerical studies as shown in Figure 8-9. Due to symmetry, only half of the system was modeled with linear isoparametric quadrilateral elements. The following data is employed in the computer program: NELS = 54,





Recharge = 5.6 E-2 cm/sec

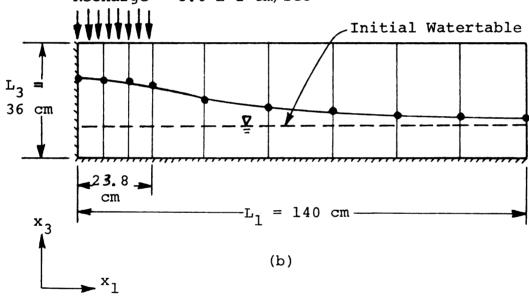
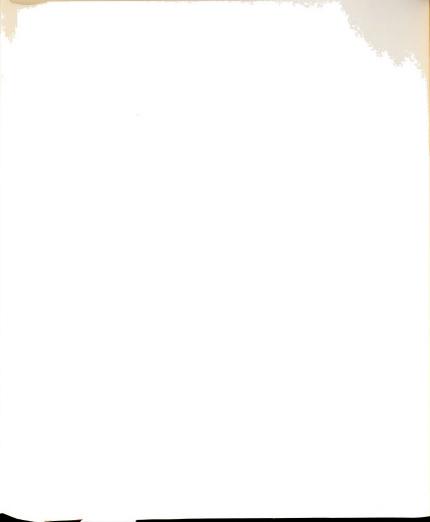


Figure 8-9.--Simulation of Transient Buildup of a Mound Due to Accretion. (a) Grid system representation; (b) scheme of the vertical cross section. The phreatic surface is shifted along the vertical lines.



NNDS = 70, K₁₁ = K₂₂ = 25.2 cm/min, n_e = 1.0, and I = 3.36 cm/min. At the beginning, a small time step equal to 0.01 minute was chosen and increased to DTMAX = 0.2 minute within several time intervals. Both the movable technique (MNT) and the fixed node technique (FNT) were analyzed, and the location of the phreatic surface at different times is shown in Figure 8-10. The results obtained by FNT and MNT compare favorably with existing analytical and experimental results. At early times, both methods provide almost the same results. As time increases, they differ to some extent, but it can be seen that the results are within a reasonable range. In these runs the velocity vectors are computed simultaneously.

The effects due to the use of simultaneous and direct calculated velocity vectors are also examined in this study. Velocity vectors are calculated and averaged at each node using the results obtained from two adjacent elements. For example, consider Figure 8-11 which shows a portion of a grid system which might be used to locate the phreatic surface. For node 6, $(v_1)_6 = 0.5$ $(v_1^2 + v_1^4)$ where the superscripts represent the element numbers. Similarly, $(v_2)_6 = 0.5$ $(v_2^2 + v_2^4)$. The velocity vectors computed in this manner are introduced to FNT. The calculated location of the phreatic surface is also compared with the observed curve as shown in Figure 8-12. The



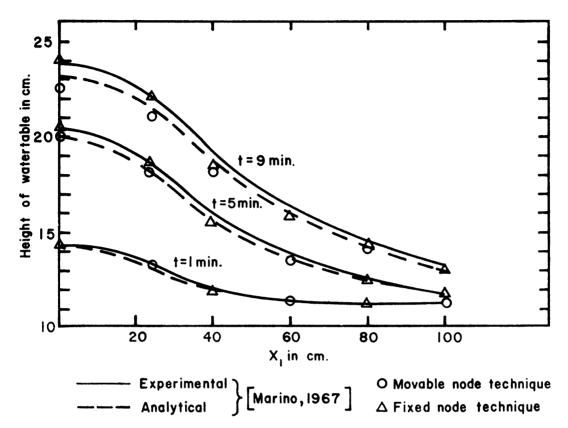


Figure 8-10.--Prediction of Transient Buildup of a Mound Due to Accretion With Different Techniques.

The velocity vectors are calculated by the simultaneous method.

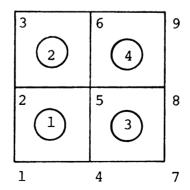


Figure 8-11.--Portion of a Grid System.



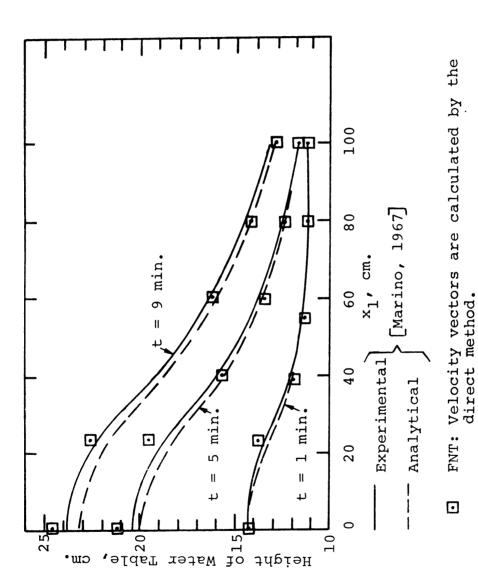


Figure 8-12. -- Prediction of Transient Buildup of a Mound Due to Accretion. The velocity vectors are calculated by the direct method.



results follow the expected curve and are slightly higher than those obtained using the simultaneous velocity solution.

Depending on the technique chosen for calculating the velocity vectors and locating the phreatic surface, one might arrive at slightly different results. However, employed techniques provide reasonable solutions which are within an acceptable range with FNT giving the best results for this case of study. It is also shown in the literature that MNT has the capacity to simulate flow field problems which involve a phreatic surface [France et al. 1971, Desai 1972]. It can be observed that FNT is also capable of solving similar problems, if not with greater accuracy, at least with the same degree of accuracy.

8.3.2 Numerical Modeling of a Field Problem

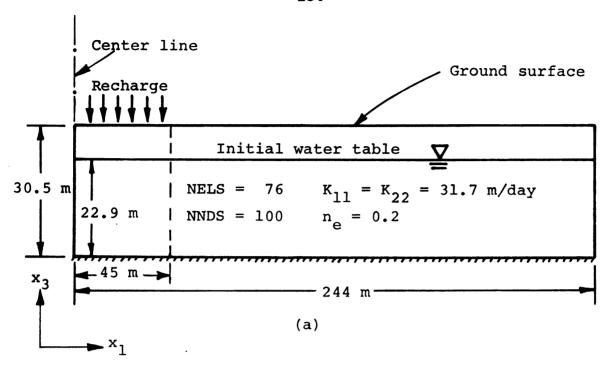
Recently, Bianchi and Haskell [1975] presented a series of field observation data on the shape of groundwater mounds produced by artificial recharge water spreading. Detailed descriptions of the experimental recharge ponds and the location of the observation wells are presented by the authors. In one of the experiments a square pond, 90×90 m (295 \times 295 ft), was chosen and the rise of the mound due to recharge was measured. The initial position of the water table was determined from

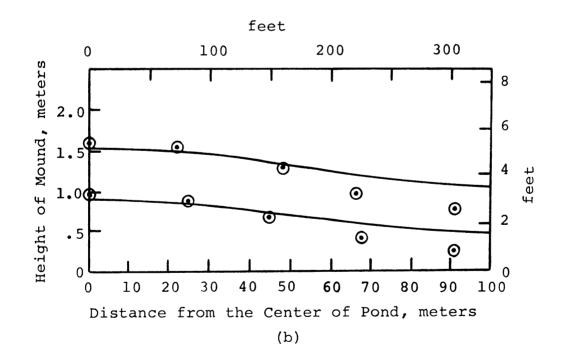


well observations prior to flooding, and corrected for barometric fluctuation. The recharge rate (I) was taken to be the rate that water enters the surface of the pond, and was assumed constant. The aquifer permeability and the "fillable void" were evaluated based on a single pumping test at the pond. In the remainder of this section effective porosity or specific yield is used instead of the term fillable void. The water table was 7.62 m (25 ft), and the impeding layer of lower permeability was observed between 5.18 and 5.49 m (17-18 ft) below the ground surface.

Although information regarding the numerical simulation of the system is not sufficient, it is modeled to show the capability of the proposed FNT. A vertical cross section of the site is chosen. The constant head boundary is assumed to be 244 m away from the center of the pond as shown in Figure 8-13a. Smaller elements are used in the recharge zone and its vicinity. Initially, the water table is horizontal at 22.9 m from the datum. A small time step equal to 0.01 day is chosen and increased gradually until it reaches 0.2 day, and is kept constant until 25 days have elapsed. Taking this cross section implies that the pond is rectangular and its longitudinal length is long enough to make the two-dimensional assumption valid. However, this assumption does not really represent the actual field situation.







Computed

⊙ Observed

Figure 8-13.--(a) Sketch of the Numerical Model for a Field Problem, and (b) Location of Water Table for Different Times.



The numerical results obtained for this model after 5.15 and 25 days are shown in Figure 8-13b. The results compare favorably with observed data in the pond area, but deviate further away from the pond. This deviation might have been caused by a number of factors. Some of the apparent reasons for the discrepancy between the numerical results and field observation data are as follows:

Aquifer parameters: Only a single measurement for hydraulic conductivity and specific yield is available. The specific yield is dependent upon moisture content, degree of saturation, and temperature; thus its value will differ in and outside the pond with respect to position and time. This subject is discussed in detail by Bear [1972] and has also been recognized by Bianchi and Haskell [1975]. The value of hydraulic conductivity has to be known in the recharging zone as well as outside the pond in order to obtain a reasonable estimate for the rise of the water table. The hydraulic conductivity appears in Equations (6.3.8) and (6.3.26), where it participates in locating the phreatic surface and defining the new Dirichlet boundary condition for piezometric heads, respectively. The value of hydraulic conductivity is highly dependent upon moisture content. It follows that in order to predict the location of the phreatic surface it is logical to replace K_{ij} by $K_{ij}K_r$, where K_r



- $(0 \le K_r \le 1)$ is the relative hydraulic conductivity [Neuman 1973] and its variation with respect to moisture content for a typical situation is given in Figure 8-14.
- 2. Observed data: At the time that was taken to be t = 0, the water table was not horizontal and a rise of 0.04 to 0.39 m at the wells was recorded. The field data shown in Figure 8-13 are the average values of head rise of four wells which are located an equal distance from the center of the pond. The recorded values for each distance have some fluctuation.
- 3. Other factors: The boundary conditions, absence of instantaneous uniform recharge, and size of the elements might also be considered as additional

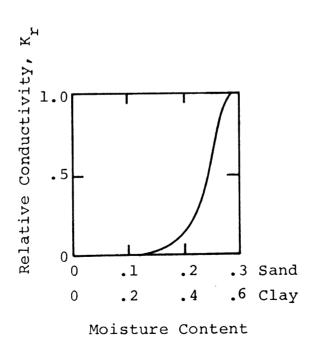
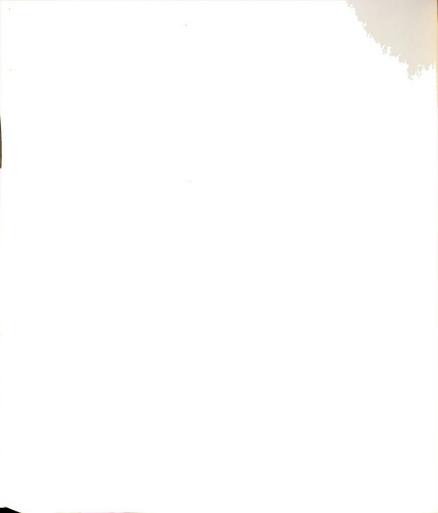


Figure 8-14.--Variation of the Relative Conductivity With Moisture Content for Two Soils [After Neuman 1973].



reasons for the discrepancy of numerical results from observed data.

In summary, although some deviation exists between observed field data and simulated results in the region away from the recharge pond, the rise of the water table beneath the pond compares favorably with recorded data. It is believed that the fixed node technique is capable of predicting the rise of the free surface due to accretion in unconfined aquifers, provided sufficient information concerning the field parameters is available.

8.3.3 Steady-State Solution

In Section 6.3.4 it was stated that the system will reach the steady-state condition when the velocity normal to the phreatic surface approaches zero. To show this condition, a vertical cross section of an unconfined aquifer is taken (Figure 8-15). The height of the phreatic surface was kept constant at 126~m (413~ft) away from the center of the recharge zone in order to obtain a rapid steady-state condition. The computer run started with a time step equal to 0.01 day and gradually increased to DTMAX = 0.2 day. When the average shifting distance \bar{d} (Section 6.3.4) was less than 5.0E-6 m the steady-state condition was assumed to be satisfied. This condition was reached after 19 days. The location of the phreatic surface at this state is shown in Figure 8-16a.



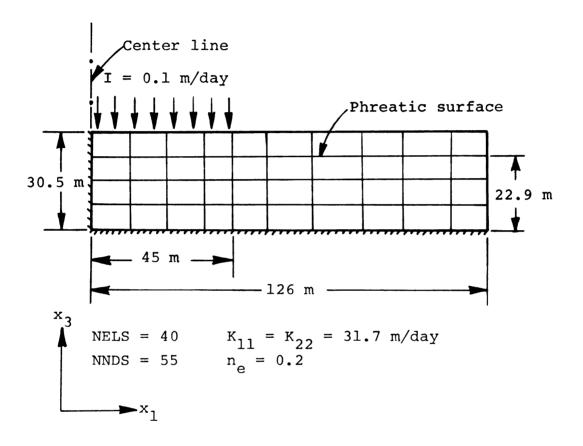
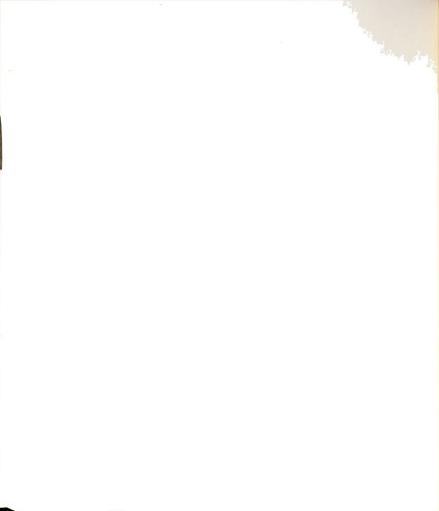
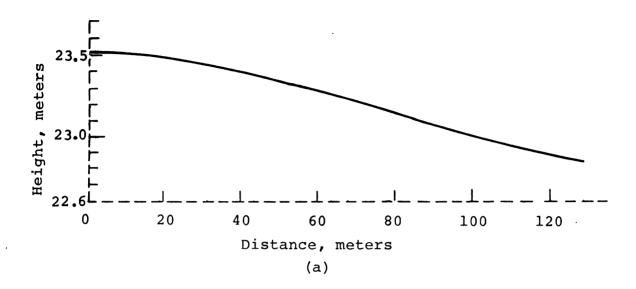


Figure 8-15.-- Representative Sketch of a Vertical Cross Section of a Phreatic Aquifer and Grid System.





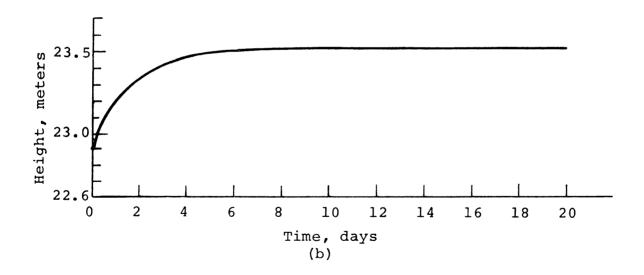


Figure 8-16.--(a) Location of the Phreatic Surface at the Steady-State Condition, and (b) Rise of Water Table With Respect to Time at the Center Line.

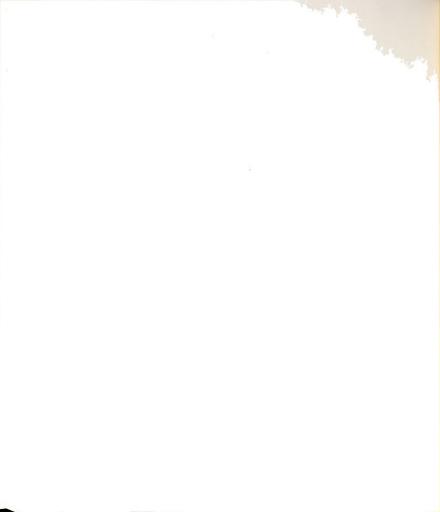


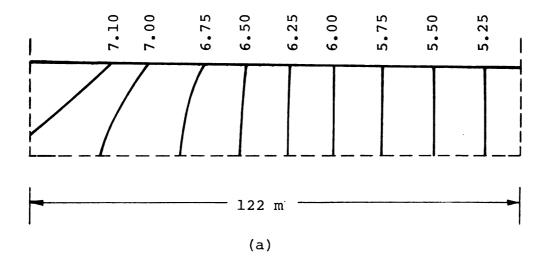
The rise of the water table at the center line with respect to time is shown in Figure 8-16b. The free surface at first rises very rapidly, and then gradually slows while approaching the steady state. In Figure 8-17 the equipotential lines and calculated velocity vectors at the nodes are shown. One observes from this figure that beneath the recharge zone and in its immediate vicinity the vertical velocities play an important role; but far from the accretion area the velocity components are almost horizontal, and thus the Dupuit approximations can be used in this region.

8.3.4 Two-Layer Aquifer

To investigate the effects of two-layer aquifers on the flow regime and location of the water table, an aquifer with the following characteristics is chosen and its vertical cross section is depicted in Figure 8-18. Initially, the water table is horizontal with a height of 20 m and located 10 m below the ground surface. The size of each element is 20 m along the x_1 - and 5 m along the x_3 -direction. At zero time the recharge with 0.1 m/day intensity is introduced, along the first 60 m of the center line of the system. A constant piezometric head is assumed at the other end.

Four different cases are examined as shown in Table 8-1. At the beginning of each run a small time





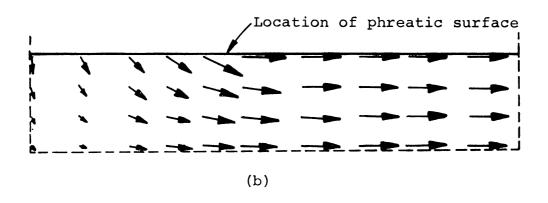


Figure 8-17.--(a) Equipotential Lines, and (b) Specific Discharge Vectors Beneath and in the Vicinity of the Recharge Zone at the Steady-State Condition. Velocity components are calculated at the nodes and the length of the arrows represents the approximate magnitude of the vectors.



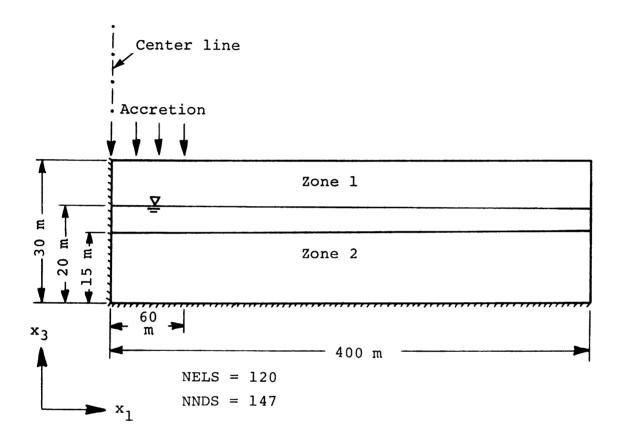


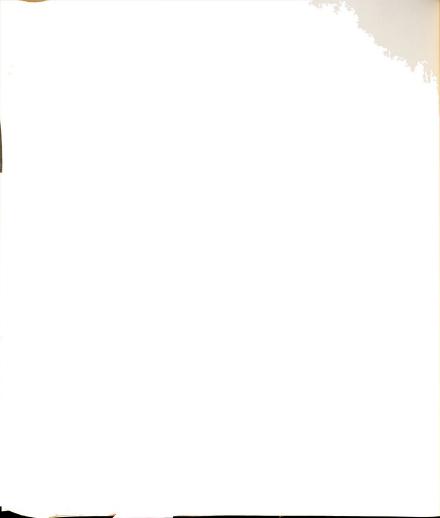
Figure 8-18.--Vertical Cross Section of a Two-Layer Phreatic Aquifer.



TABLE 8-1.--Different Combinations of Hydraulic Conductivities for Aquifer Shown in Figure 8-18 and Maximum Specified Time Step.

Case No.	Hydraulic Conductivity (m/day)		DTMAX
	Zone l	Zone 2	(day)
1	30	30	0.2
2	75	75	0.1
3	75	30	0.0625
4	30	75	0.1

step equal to 0.01 day was chosen. Depending on the kinds of materials, the time step was gradually increased to a maximum specified time step as presented in Table 8-1. In all of these cases the effective porosity is assumed constant and equal to 0.3. The rise of the phreatic surface corresponding to these conditions after 5.10 days is shown in Figure 8-19. From this figure, it can be observed that with the same effective porosity the rise of the water table at the recharge site and in its immediate vicinity is greater for less permeable aquifers. At early times, in the area away from the recharge zone the water table rises faster in the highly conductive than in the less conductive soil. As time increases this process reverses (not shown in Figure 8-19). At any time the phreatic surface remains higher for smaller K at the recharge zone. The equipotential



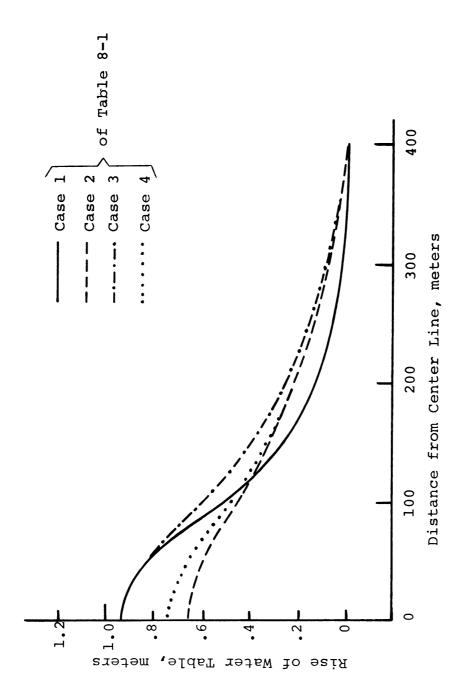
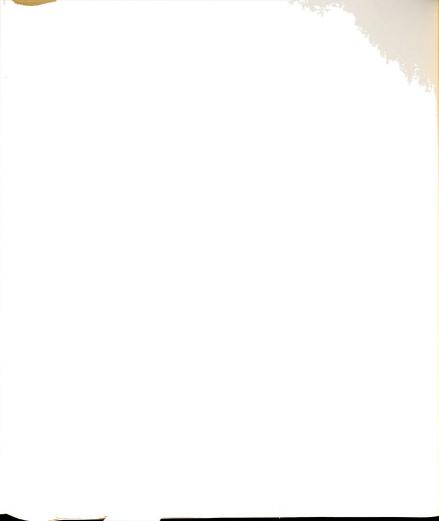


Figure 8-19.--Effects of Two-Layer Aquifer on the Water Table Profile After 5.1 Days.



lines and discharge vectors for a two-layer aquifer (Case No. 4) are shown in Figure 8-20. As expected, the discharge vectors are greater in the zone possessing high hydraulic conductivity.

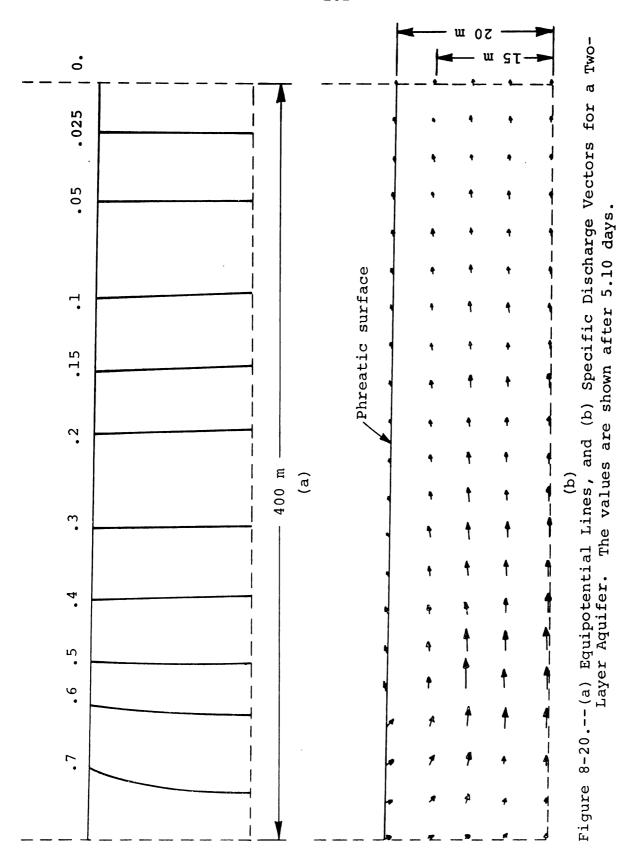
8.3.5 Change of the Effective Porosity

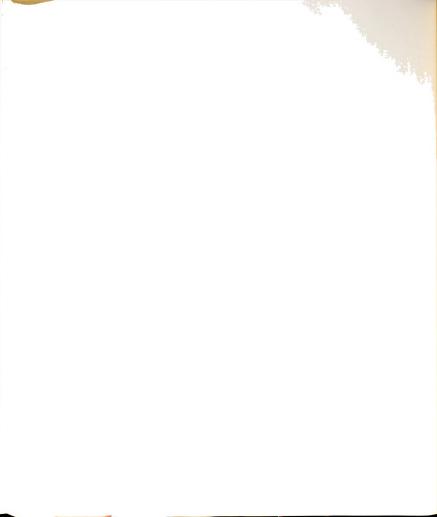
To show the influence of the effective porosity on the rise of the water table due to the recharge, an aquifer similar to the one described in Section 8.3.4 is chosen. Case No. 4 of Table 8-1 represents the hydraulic conductivities employed in this section. Three different effective porosities, namely, $n_e = 0.20$, 0.25, and 0.3, were examined. The rise of the water table for these effective porosities is shown in Figure 8-21. With similar conditions, the height of the water table is greater for the aquifer with a small effective porosity. The obvious reason for this occurrence can be seen by examining Equation (6.3.5).

8.3.6 Effect of the Depth of the Water Table

The depth of the water table also has a major influence on the location of the phreatic surface. In order to observe this effect an aquifer similar to the one described in Section 8.3.4 is selected, except that the height of the initial phreatic surface is lowered to 10 m. The rise of the water table after 10 days for two







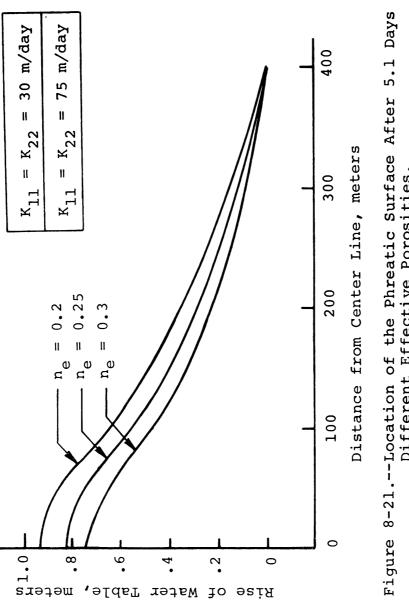


Figure 8-21. -- Location of the Phreatic Surface After 5.1 Days for Different Effective Porosities.



different initial depths is given in Figure 8-22. As one might have expected, the mound in the recharge zone and its vicinity will be more pronounced for a shallow aquifer than for a deep one.

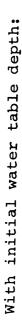
8.3.7 Decay of a Groundwater Mound

The investigation of the decay of a mound is an interesting subject in groundwater hydrology. To show the capability of the proposed technique of handling such cases, an aquifer similar to Figure 8-18 is taken and recharge is applied up to 4.9 days before being halted. The rise and fall of the water table and the location of the phreatic surface at the center line, up to 9.9 days, are shown in Figures 8-23 and 8-24, respectively. In this example the hysteresis and variation of the specific yield with time are neglected. It can be observed that more time is required for the water table to decay to the initial steady-state condition than the time necessary to build the mound.

8.3.8 Maximum Applicable Time Interval

Specifying the time interval is an important task in solving transient phreatic problems. There is no particular rule for selecting the time step. A general procedure which is useful for defining Δt is given in Section 6.3.5. In all the different examples illustrated





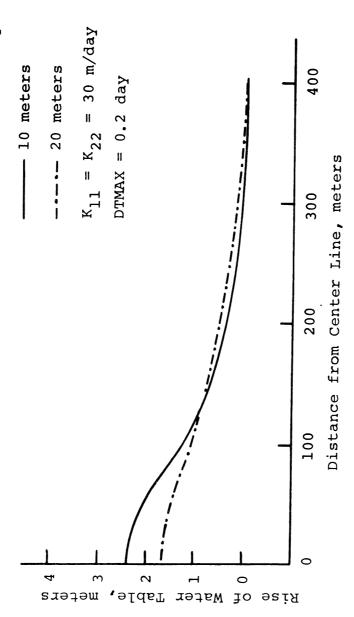


Figure 8-22. -- Rise of Water Table for Different Initial Depths Medium is of the Free Surface After 10.0 Days. homogeneous.



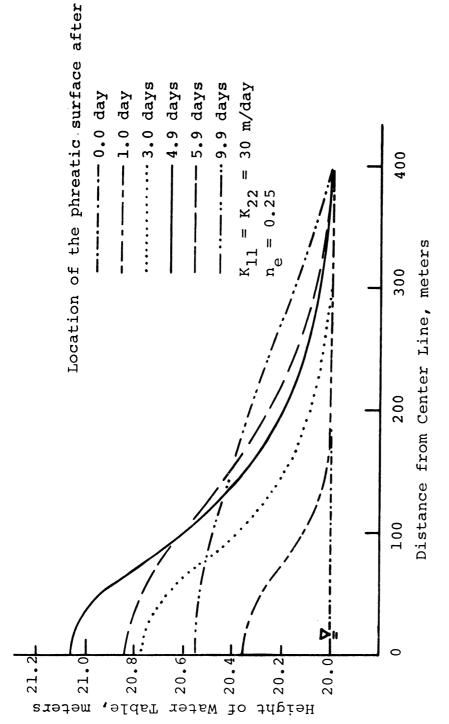
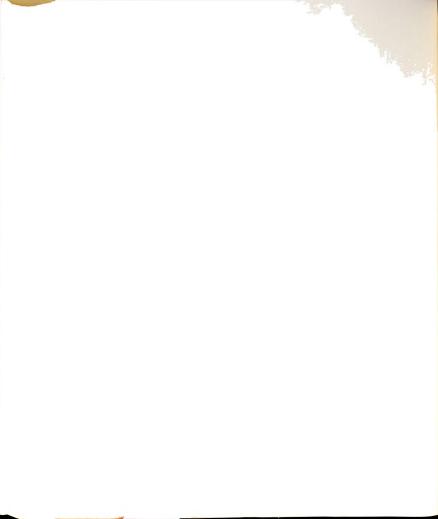


Figure 8-23. -- Rise and Fall of Water Table in a Phreatic Aquifer. Medium is homogeneous.



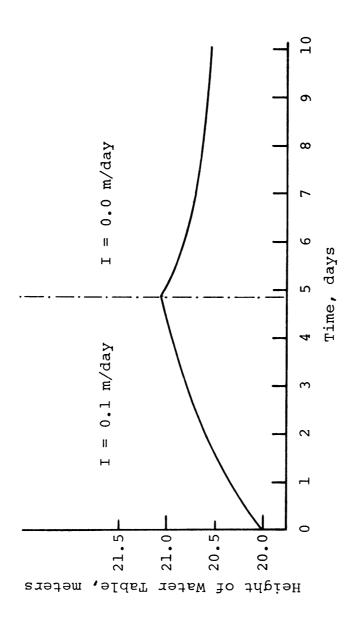


Figure 8-24.--Rise and Fall of Water Table at the Center Line of a Recharge Zone.



in Section 8.4 a small time step was chosen, and using Equation (6.3.28) it was increased to a maximum specified time step and maintained constant thereafter. Defining the maximum time step requires some experience which might be obtained after a few runs for a specific problem. In general, for highly conductive porous media or nonhomogeneous aquifers, a smaller maximum time interval should be employed. Usually, the rise of the water table follows a trend similar to that depicted in Figure 8-16b. When it starts to oscillate with respect to time, the maximum time step should be decreased.

For example, consider a phreatic aquifer represented by Figure 8-15, in which $K_{11} = K_{22} = 31.7 \text{ m/day}$. In Section 8.3.3 it is stated that 0.2 day was used to obtain steady-state conditions for this specific problem. To show the sensitivity of the numerical solution to the time step, the hydraulic conductivities were increased 2.5 times. For the first time, DTMAX equal to 0.2 day was used to find the location of the phreatic surface at the steady-state condition. As expected, this value was too large and the results were not correct. Then DTMAX was reduced to 0.1 day. Although the numerical results were realistic for early times, after 4 days the values of the piezometric heads started to fluctuate. oscillation increased with time as shown in Figure 8-25. By further reduction of DTMAX to 0.05 day, the oscillation



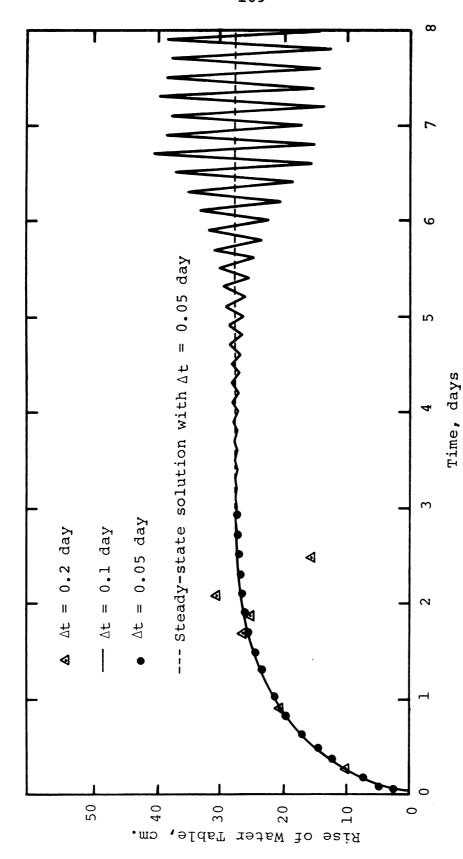


Figure 8-25.--Oscillation of Water Table Due to a Large Time Step. Rise of the water table at the center line is shown in this figure.



was eliminated. It can be concluded that 0.05 day is an appropriate DTMAX for this specific example.

8.4 Summary

In this chapter it was shown that the finite element technique is capable of solving flow problems, both in confined and unconfined aquifers. Under transient conditions the fixed node technique (FNT) was tested and it was verified that this method can predict the location of the phreatic surface and can yield piezometric heads and velocity vectors with a reasonable degree of accuracy. It was shown that the simultaneous solution of velocity functions at the nodes is continuous and acceptable. Finally, the effects of the order of time approximation on the accuracy of the predicted results were examined.



CHAPTER IX

NUMERICAL RESULTS FOR PREDICTION OF CONCENTRATION OF A TRACER

The computer model developed in this study solves a set of partial differential equations. One equation is the combined equation of motion and continuity of flow which describes the piezometric head distribution of the aquifer; in turn, the velocity components and hence the dispersion coefficients are computed. Then the second partial differential equation, the mass-transport equation (convective-dispersion equation), is solved to yield the concentration distribution in the flow domain.

The numerical results for the flow equation were discussed in Chapter VIII. In this chapter the finite element model for simulating mass transport is verified by comparing the numerical results with several existing analytical solutions. If the results of a known analytical solution can be approximated, a great deal of confidence in the numerical simulation can be gained. The feasibility of the model to predict the concentration of a tracer in field problems is shown. In this chapter concentration is used as a synonym for dimensionless



concentration, i.e., $\mathrm{C/C_O}$, $\mathrm{C_O}$ being the initial tracer concentration. The $\mathrm{D_L}$ and $\mathrm{D_T}$ are longitudinal and transversal dispersion coefficients, respectively, and are assumed constant for the porous medium independent of velocity.

9.1 Longitudinal Dispersion in Steady Uniform One-Dimensional Flow

A semi-infinite cross section of a homogeneous and isotropic porous medium with a plane source maintained at x_1 = 0 is shown in Figure 9-1.

Figure 9-1.--Cross Section of a Homogeneous and Isotropic Porous Medium With a Plane Source Maintained at $x_1 = 0$.

The flow is maintained at a constant flux q_1 in the x_1 -direction. For an isotropic medium, the axis of the dispersivity tensor is assumed to coincide with the velocity vector. Equation (3.5.1) reduces to

$$D_{L} \frac{\partial^{2}C}{\partial x_{1}^{2}} - V_{1} \frac{\partial C}{\partial x_{1}} = \frac{\partial C}{\partial t}$$
 (9.1.1)



where D_L is the longitudinal dispersion coefficient, and V_1 is the velocity vector along the x_1 -direction = q_1/n_e . The solution of Equation (9.1.1) with the following initial and boundary conditions,

$$C(0, t) = C_0$$
 $t \ge 0$
 $C(x_1, 0) = 0$ $x_1 \ge 0$
 $C(\infty, t) = 0$ $t \ge 0$

is given [Bear 1972] as:

$$\frac{C}{C_{O}} = \frac{1}{2} \left[\text{erfc} \left(\frac{x_{1}^{-V} + v_{1}^{t}}{2\sqrt{D_{L}t}} \right) + \exp \left(\frac{v_{1}^{x_{1}}}{D_{L}} \right) \right] + \exp \left(\frac{v_{1}^{x_{1}}}{2\sqrt{D_{L}t}} \right) \right]$$
(9.1.2)

where erfc(u) = 1 - erf(u).

erf(u) =
$$\frac{2}{\sqrt{\pi}} \int_{0}^{\infty} e^{-\xi^{2}} d\xi$$
 (9.1.3)

The progress of a concentration front in an infinite column of a porous medium is modeled numerically on the computer, as shown in Figure 9-2. A constant source is maintained at x=0 and the following parameters are employed in the finite element model: $\Delta x_1=0.4$ cm, $V_1=0.1$ cm/sec, L=10 cm, $\Delta t=2$ sec, and $D_L=0.01$ cm²/sec.



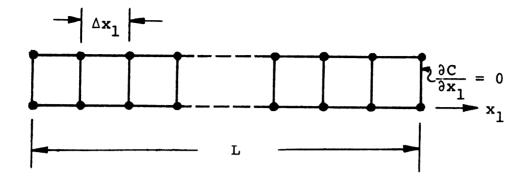


Figure 9-2.--Finite Element Model to Simulate One-Dimensional Longitudinal Dispersion.

The results of runs using the above data are shown in Figure 9-3. The numerical results compare favorably with the analytical solution; however, in some instances they are slightly higher. Reddell and Sunada [1971] have solved one-dimensional longitudinal dispersion with the method of characteristics using 2 and 4 moving points per grid. Since the finite element technique used in this study does not employ moving points per grid and since the Δx in this study is greater than the one used by Reddell and Sunada, it is difficult to compare the degree of accuracy of the two methods.

The order of time approximation has a great effect on the accuracy of the numerical solution. A second order approximation, introduced in Chapter V, was used to obtain Figure 9-3. The first order time approximation provides poor results as shown in Figure 9-4. In Figure 9-4 for $\theta=1$ (implicit method), the numerical



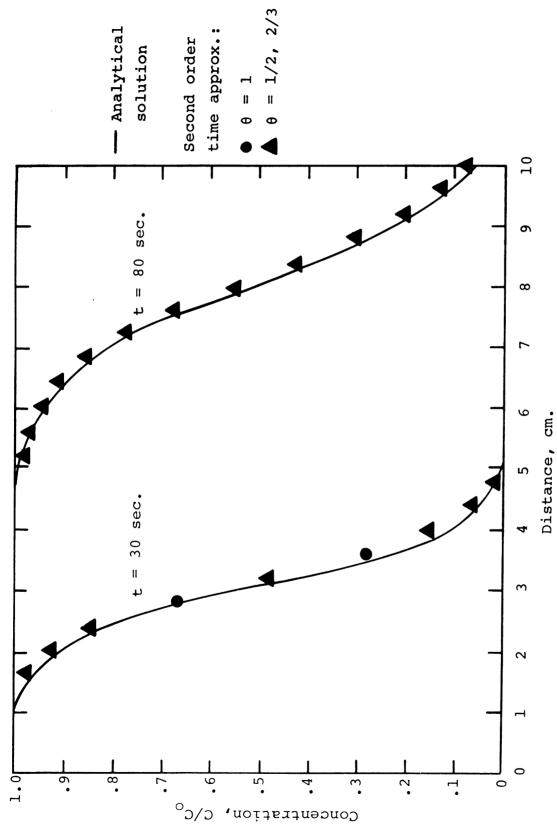


Figure 9-3.--Numerical Solution of the Longitudinal Dispersion at t = 30 Sec. and t = 80 Sec. Using Second Order Time Approximation and Comparison With the Analytical Results.



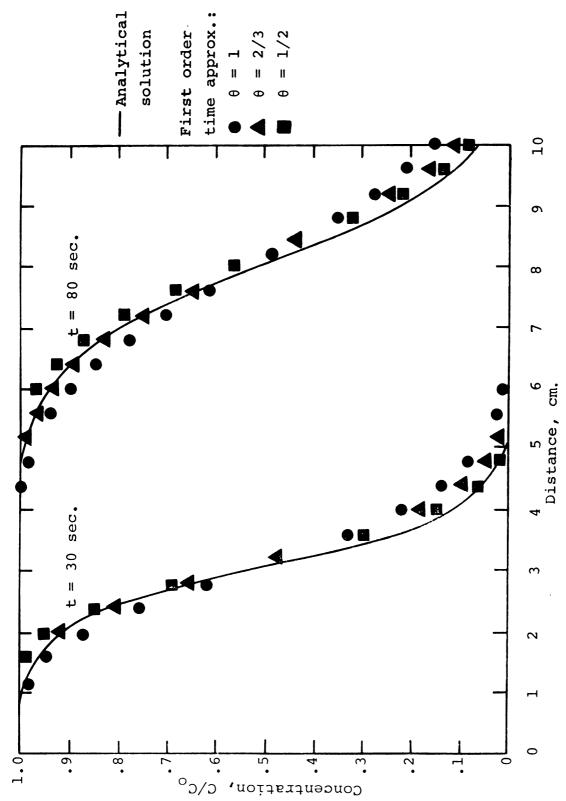
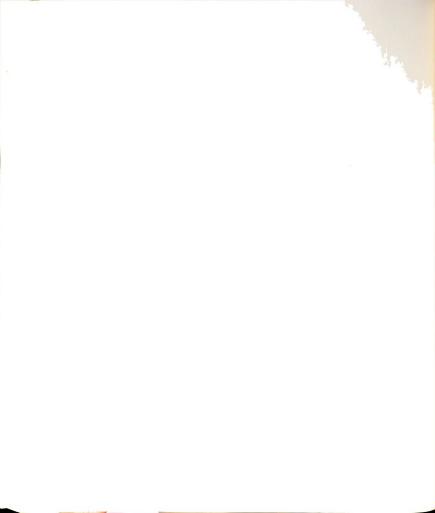


Figure 9-4.--Numerical Solution of the Longitudinal Dispersion at t = 30 Sec. and t = 80 Sec. Using First Order Time Approximation and Comparison With the Analytical Results.



results fall below the curve close to the point source and then move above the curve as the mass travels further from the origin; $\theta = 2/3$ gives good results at high concentrations, but diverges later; $\theta = 1/2$ yields a higher concentration than the analytical solution at every point. The second order approximation provides a better estimate for tracer concentration compared to first order. Pinder [1974] examines the stability of the first order time approximation by Fourier analysis of the mass-transport equation. He concludes that $\theta = 0$ provides unstable results for any spatial distance, but $\theta = 0.5$ and $\theta = 1.0$ give stable numerical results for any reasonable space mesh. Cheng [1973] mentions using second and third order implicit approximations of the time derivative in the solution of the convective-dispersion equation. Although third order approximations were derived in Section 5.5, they were not examined in this study.

9.2 Sensitivity Analysis for Time Step and Grid Size

To investigate the effects of the types of θ on the convergence of the numerical solution using larger time steps, several Δt were chosen and the above data was employed to solve the concentration distribution for a porous medium 14 cm in length. Let the residual be the difference between the analytical and numerical solutions for a given x_1 . Then define ZT as the averaged sum of the squares of residuals, i.e.,



$$\sum_{\text{T}}^{\text{NT}} (C^* - C)^2$$

$$\text{T} = \frac{n=1}{NT}$$
(9.2.1)

where NT is the number of available data from time zero to 90 seconds, and C* and C are the analytical and numerical values of concentration at \mathbf{x}_1 = 2 cm, respectively. The variations of ZT with respect to Δt for different second order θ values, namely, 1/2, 2/3, and 1, are given in Figure 9-5. For small time steps up to Δt = 4 sec, θ = 1 gives better results compared with the other two θ 's. For greater Δt , θ = 2/3 is superior and the best results for this θ are obtained at Δt = 5 sec. The value of ZT for θ = 1/2 is always the highest, and for Δt greater than 4 sec it increases very rapidly. This means that for θ = 1/2 at larger Δt the results diverge and are not correct.

To see this effect very clearly the variation of concentration using $\Delta t = 7$ sec at $\mathbf{x}_1 = 2$ cm with different θ is given in Figure 9-6. It can be observed from this figure that poor results are obtained using $\theta = 1/2$. The probable conclusion that one can make from Figures 9-5 and 9-6 is that for larger time steps $\theta = 1/2$ diverges and the results oscillate, and as Δt increases the error increases gradually for both $\theta = 2/3$ and $\theta = 1$, with $\theta = 2/3$ showing the least error.



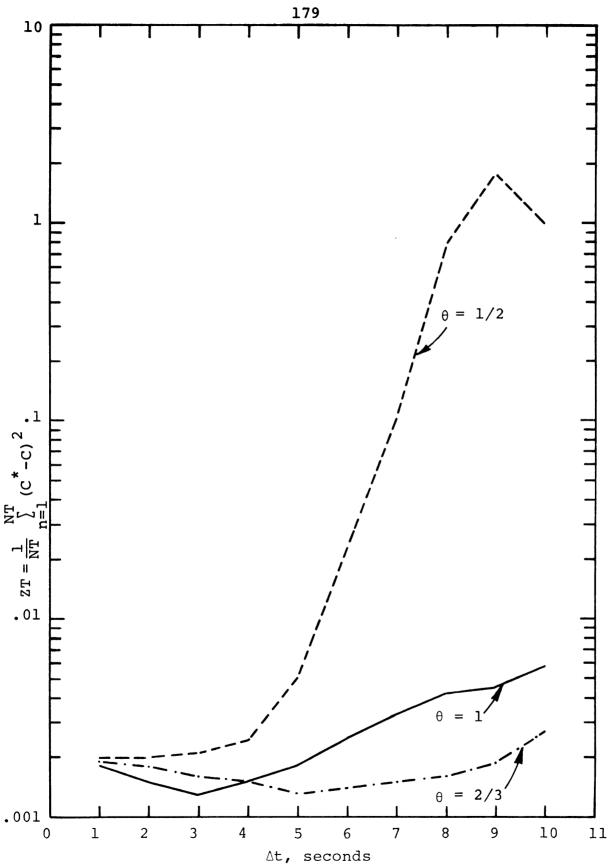


Figure 9-5.--Variation of ZT With Respect to Δt for Different $\theta.$ Second order time approximation is used.



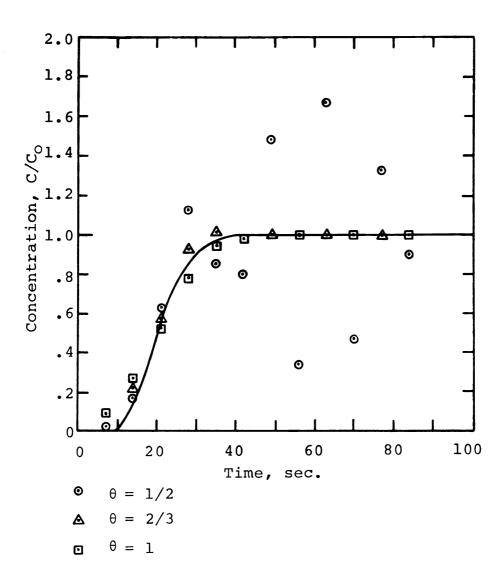
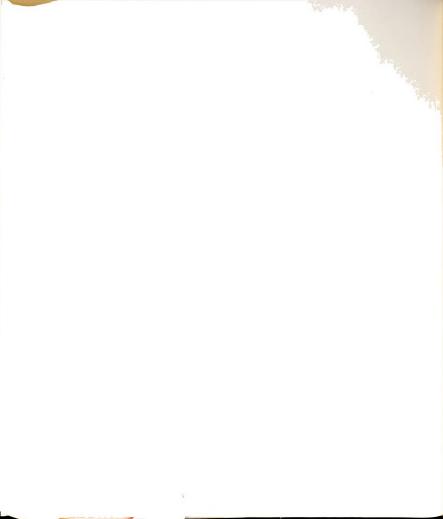


Figure 9-6.--Effects of Different θ on the Accuracy of the Concentration Distribution at x=2 Cm. With $\Delta t=7$ Sec. Second order time approximation is used.



To show the effect of the grid size on the accuracy of the numerical solution for the convective-dispersion equation obtained by the finite element method, one can define ZX as the average sum of the squares of residuals, i.e.,

$$\sum_{X}^{NX} (C^* - C)^2$$

$$ZX = \frac{n=1}{NX}$$
(9.2.2)

In Equation (9.1.5) C^{\star} and C are the analytical and numerical values of the concentration, respectively, at a specific time, and NX is the number of available data between $\mathbf{x}_1 = 0$ and $\mathbf{x}_1 = \mathbf{L}$. Different grid sizes from $\Delta \mathbf{x}_1 = 0.2$ to $\Delta \mathbf{x}_1 = 2$ cm are employed, and second order time approximation with $\theta = 2/3$ is used. The variation of ZX with respect to $\Delta \mathbf{x}_1$ for t = 80 sec is given in Figure 9-7. It can be observed that the error is reduced as the size of the elements becomes smaller.

9.3 Two-Dimensional Dispersion With Uniform Flow

For steady and uniform flow parallel to the x-axis in an isotropic and homogeneous porous medium, the convective-dispersion equation can be written as

$$\frac{\partial C}{\partial t} = D_L \frac{\partial^2 C}{\partial x_1^2} + D_T \frac{\partial^2 C}{\partial x_2^2} - V_1 \frac{\partial C}{\partial x_1}$$
 (9.3.1)



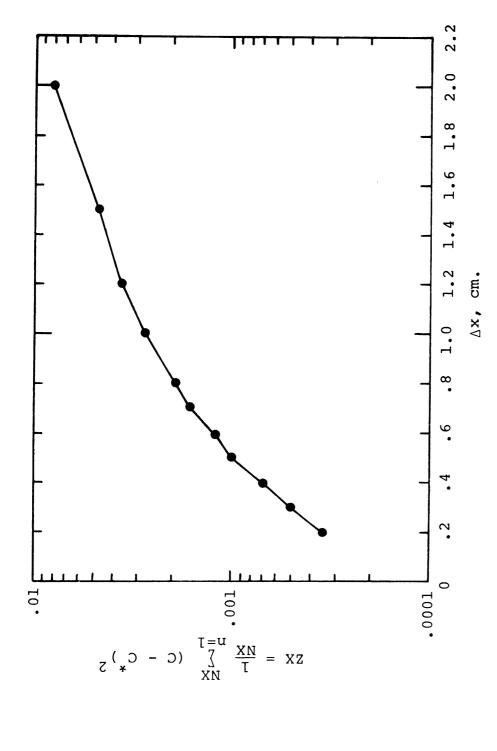


Figure 9-7.--Variation of ZX With Respect to Δx at t = 80 Sec.



where $\mathbf{D_L}$ and $\mathbf{D_T}$ are the longitudinal and transversal dispersivities, respectively. Consider a porous medium as shown in Figure 9-8. The seepage velocity $\mathbf{V_l}$ is uniform and constant throughout the medium.

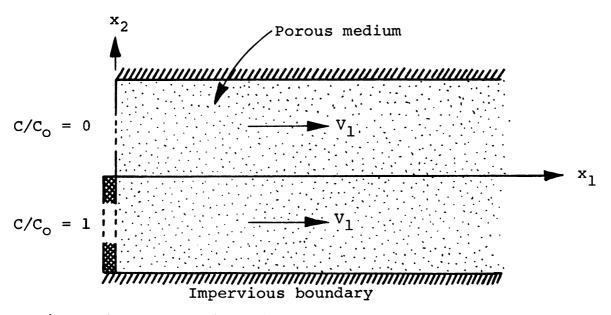
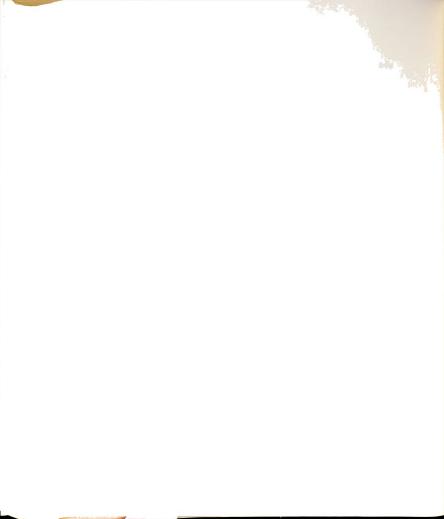


Figure 9-8.--Two-Dimensional Dispersion With One-Dimensional Flow.

At \mathbf{x}_1 = 0, the concentration is held constant and equal to C_0 for \mathbf{x}_2 < 0 and equal to zero for \mathbf{x}_2 > 0. The medium is confined by two impervious boundaries far apart from each other, such that the concentration remains C_0 and zero at the bottom and top of the boundaries, respectively, for all \mathbf{x}_1 . The following boundary conditions can be stated for steady-state distribution of the tracer concentration:



$$C(0, x_2) = C_0$$
 $-\infty < x_2 \le 0$
 $C(0, x_2) = 0$ $0 < x_2 < \infty$
 $\frac{\partial C}{\partial x_2} = 0$ $x_2 = \pm \infty \text{ for all } x_1$
 $(9.3.2)$

For this special case Equation (9.3.1) reduces to

$$v_1 \frac{\partial C}{\partial x_1} = D_T \frac{\partial^2 C}{\partial x_2^2}$$
 (9.3.3)

The solution of Equation (9.3.3) with the indicated boundary conditions is given by Harleman and Rumer [1963]. The solution is

$$\frac{C}{C_O} = \frac{1}{2} \operatorname{erfc} \left[\frac{x_2}{2\sqrt{D_T x_1/V_1}} \right]$$
 (9.3.4)

In simplifying Equation (9.3.1) to Equation (9.3.3), Harleman and Rumer assumed and experimentally justified that $\partial^2 C/\partial x_1^2 << \partial^2 C_2/\partial x_2^2$.

The movement of a tracer in two dimensions with one-dimensional uniform flow is modeled numerically, as shown in Figure 9-9a. The following parameters are employed: $V_1 = 0.1$ cm/sec, $D_L = 1 \times 10^{-7}$ cm²/sec, $D_T = 0.01$ cm²/sec, $\Delta x_1 = 1.0$ cm, and $\Delta x_3 = 0.5$ cm. The steady-state solution of the problem by the finite element technique is obtained using the coefficients given in Table 5-3.



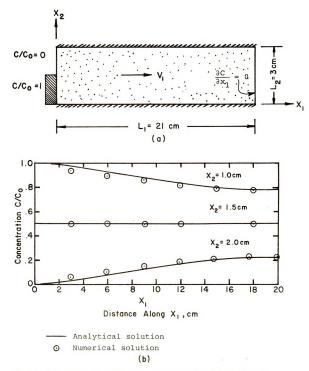


Figure 9-9.--Steady State Solution of Two-Dimensional Dispersion With One-Dimensional Flow.

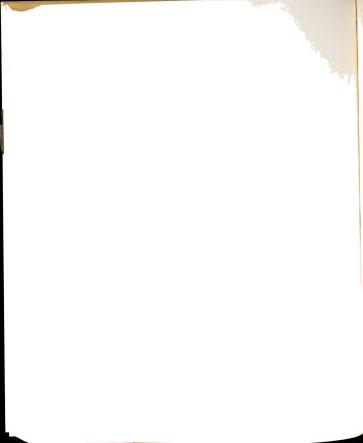
(a) Representative sketch, (b) concentration at different x₂. Cubic quadrilateral elements with NELS = 14 and NNDS = 98 are used.

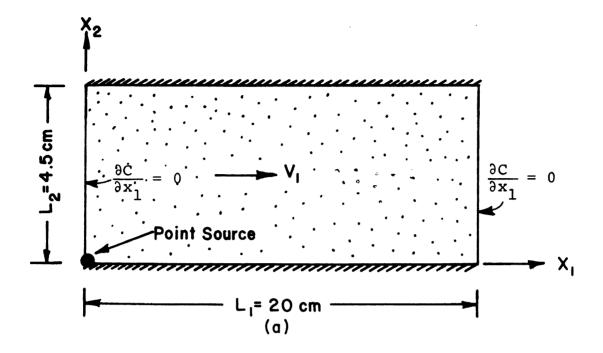


The numerical results using cubic elements are compared with the analytical solution in Figure 9-9b. The results compare quite well with the analytical solution except in the vicinity of $\mathbf{x}_1=0$, where the results are slightly lower than those obtained by the approximate analytical solution. This slight deviation from the analytical solution for small x might be due to neglecting $\partial^2 \mathbf{C}/\partial \mathbf{x}_1^2$ in Equation (9.3.3). The type and size of the elements may be considered other reasons for this small discrepancy.

9.4 Point Source With Uniform Flow

In this example a source is maintained at a constant concentration for t > 0 at point A in a porous medium, as shown in Figure 9-10a. The parameters used in this example are: $D_L = 0.01 \text{ cm}^2/\text{sec}$, $D_T = 0.001 \text{ cm}^2/\text{sec}$, $V_1 = 0.1 \text{ cm/sec}$, $V_2 = 0$, $\Delta t = 5 \text{ sec}$, $\Delta x_1 = 1 \text{ cm}$, and $\Delta x_2 = 0.5 \text{ cm}$. The concentration distribution of the tracer at $x_2 = 0.0$, $x_2 = 0.5$, and $x_2 = 1.0 \text{ cm}$ after 180 seconds is depicted in Figure 9-10b. Although V_2 in this example is zero and D_T is smaller than D_L , tracer movement along the x_2 -direction is evident. This implies that velocity components facilitate the spreading of chemical substances, but in the absence of a velocity component the tracer will move due to the dispersion phenomenon.





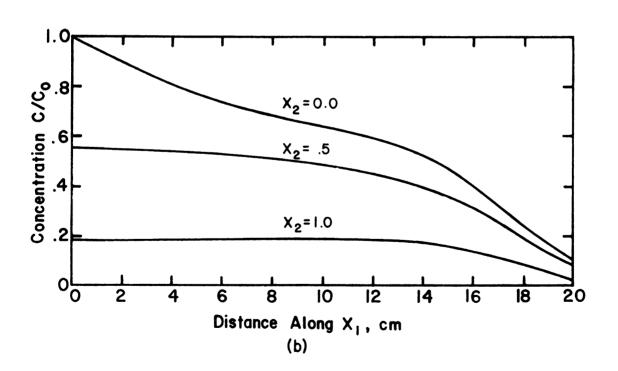


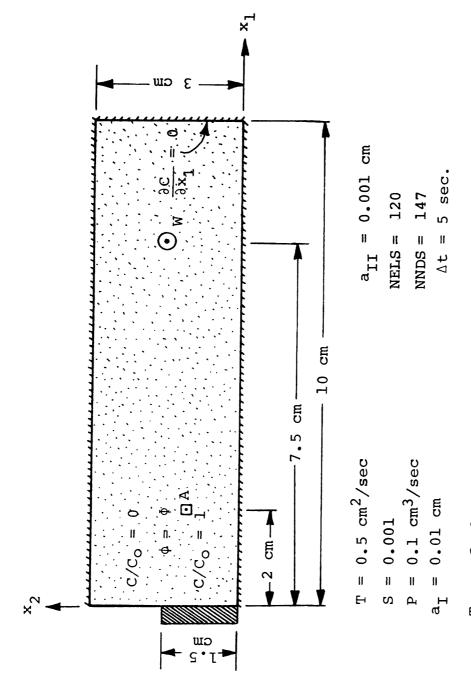
Figure 9-10.--Two-Dimensional Dispersion With a Point Source. (a) Sketch of the system, (b) concentration distribution at time 180 sec. along \mathbf{x}_1 at different \mathbf{x}_2 . Linear quadrilateral elements with NELS = 120 and NNDS = 147 are used.



9.5 Two-Dimensional Dispersion With Transient Flow

To show the capability of the numerical model in solving the convective-dispersion problems with transient flow conditions, two examples are given in this section. In the first example a porous medium is taken with boundary conditions and employed parameters shown in Figure 9-11. Water is withdrawn at a constant rate at point W to produce the piezometric gradient. The variation of the piezometric head, magnitude of the velocity vector, and concentration for point A with time is given in Figure 9-12, and the concentration distribution for the system after 100 seconds is depicted in Figure 9-13. An important observation which can be made (see Figure 9-12) is that the simultaneous solution of the velocity vectors is sensitive to the fluctuation of the piezometric heads. Even so, the concentration distribution is a smooth curve. In this study the first order time approximation was used to solve the flow equation, while the second order time approximation was used for dispersion. It is believed that the employment of a higher order time approximation helps in the smoothing process and it might have been used in the hydrodynamic problem. However, this idea needs to be investigated further before any concrete conclusion can be drawn.





Type of elements: Linear quadrilateral

Figure 9-11.--Representative Domain of a Porous Medium With Time Variable Velocity and Dispersion.



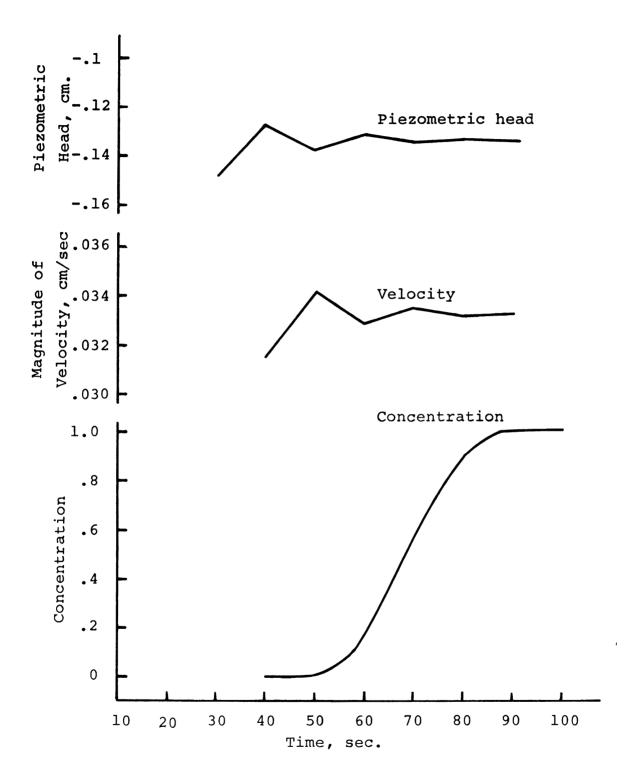


Figure 9-12.--Variation of the Piezometric Head, Velocity, and Concentration With Time at Point A.



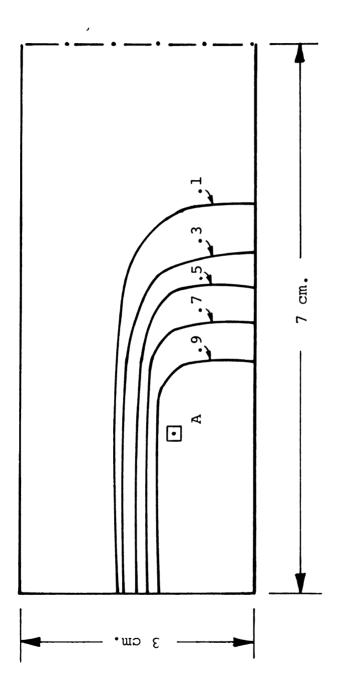
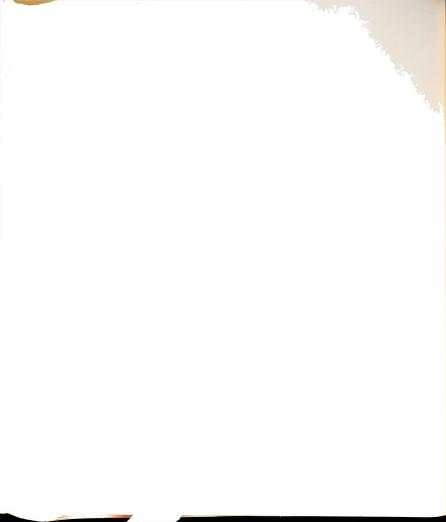


Figure 9-13.--Iso-Concentration Lines for a Medium With a Transient Flow Field After 100 Seconds.



In the second example a confined aquifer, $80 \times$ 80 m, is chosen. The concentration and the piezometric head at point C are held constant, and a well is located at point W, of Figure 9-14. A high pumpage value is chosen to produce a steep gradient in order to obtain high velocity vectors. This in turn will cause a tracer buildup at a lesser time. Initially, the system is at zero potential, and after three days of pumping a constant source of tracer is introduced at point C. The variations of concentration at points A and B, 10.00 and 14.14 meters away from the point source, respectively, are shown in Figure 9-15. The concentration distribution at the aguifer after 10 and 20 days of initial pumping is shown in Figure 9-16. Since dispersivity along the x_1 -direction is greater than along the x_2 -direction, the tracer will advance more in the x_1 - than in the x_2 -direction. curves are non-symmetric as one might have anticipated. As time increases the asymmetry becomes more pronounced.

As far as it is known by the author, no published results are available to compare with the obtained numerical results. The only conclusion that can be made at present is that the simultaneous calculation of the velocity vectors and the use of higher order time approximations can be considered to be a step in predicting tracer concentration in transient groundwater flow problems.



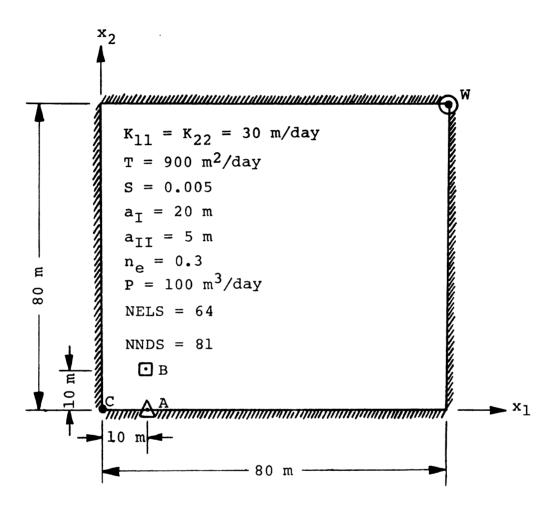


Figure 9-14.--Domain of Two-Dimensional Dispersion With a Transient Flow Field.



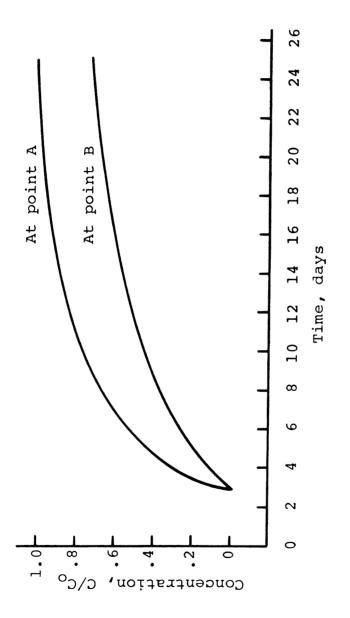


Figure 9-15.--Variation of Tracer Concentration With Time at Points A and B.



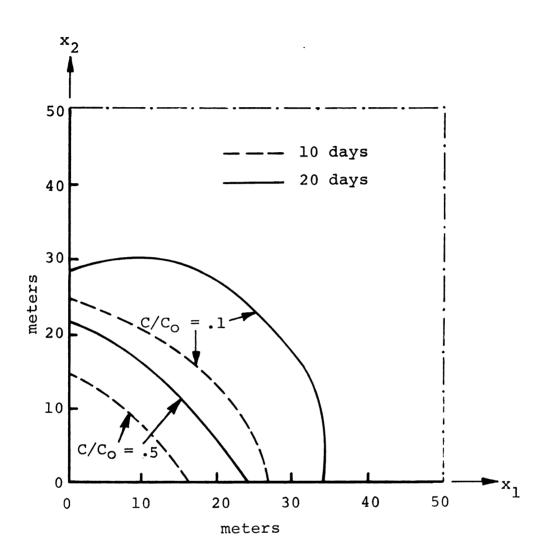


Figure 9-16.--Iso-Concentration Lines After 10 and 20 Days.



9.6 Dispersion in a Phreatic Aquifer With Accretion

To investigate the capability of FNT (fixed node technique) in solving the convective-dispersion equation, a portion of a phreatic aquifer was taken. A sketch of a vertical section of the chosen model is depicted in Figure 9-17. Initially, the water table is maintained horizontal at 20 m, and both ends of the system are kept at a constant piezometric head. It is assumed that the recharge site is a continuous source of a tracer with constant concentration, and the tracer maintains the same concentration until it reaches the water table. The reason for choosing this model is to develop the conditions such that piezometric heads and velocity components vary considerably within the period of interest and trends of velocity components will differ in the system.

The procedure outlined in Section 6.3 was used to locate the phreatic surface and define the velocity vectors at the nodes above the free surface. It was assumed that recharge directly reaches the phreatic surface. In calculating the location of the phreatic surface the primary interest was to find the velocity vectors on the surface, and the above assumption was perfectly applicable. In the dispersion process the primary source of the pollutants is located on the ground surface and the material is washed and carried



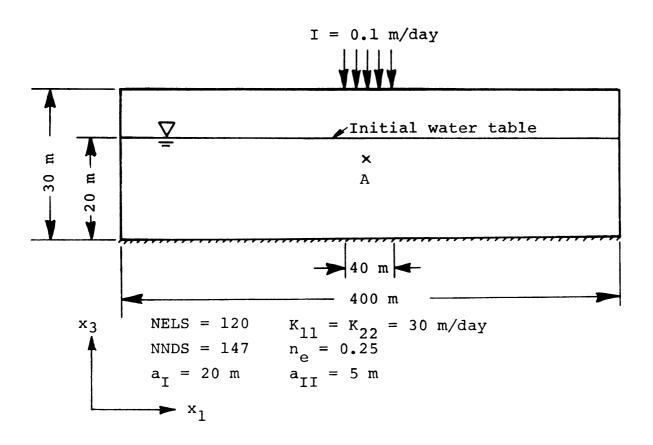


Figure 9-17.--Representative Model Used in Simulating Tracer Movement in a Phreatic Aquifer.



downward by infiltration. This event suggests that in the dispersion model at the accretion zone, the value of V_3 has to be equal to the recharge rate. This modification of the assumption is realistic for the dispersion phenomena and improves the numerical model.

The following parameters are used in the system: NELS = 120, NNDS = 147, $K_{11} = K_{22} = 30.0 \text{ m/day}$, $n_e = 0.25$, I = 0.1 m/day, $a_I = 20 \text{ m}$, $a_{II} = 5 \text{ m}$, and DTMAX = 0.2 days. As outlined in detail in Chapter VII, velocity vectors and hence dispersion coefficients are calculated and then used in the convective-dispersion equation. A constant concentration of a tracer is introduced at the end of the second day. The reason for choosing the second day is that usually at most sites prior to recharge a potential gradient exists in the porous media, and secondly, it takes some time for a tracer to travel from the ground surface to the water table beneath the recharge site. However, the starting time for computing the dispersion of a tracer can be any time, including t = 0.

The discharge vectors at t=2 and t=8 days and the iso-concentration lines at t=4, 8, and 12 days are depicted in Figures 9-18 and 9-19, respectively. From the consideration of physical aspects, a solute cannot move in a completely dry soil. This condition in turn implies that the concentration of a tracer at the nodes



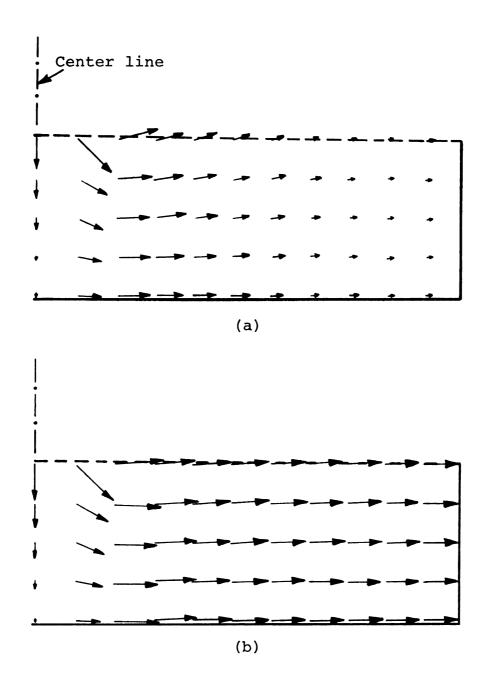


Figure 9-18.--Discharge Vectors in a Phreatic Aquifer With Recharge After (a) 2 and (b) 8 Days.



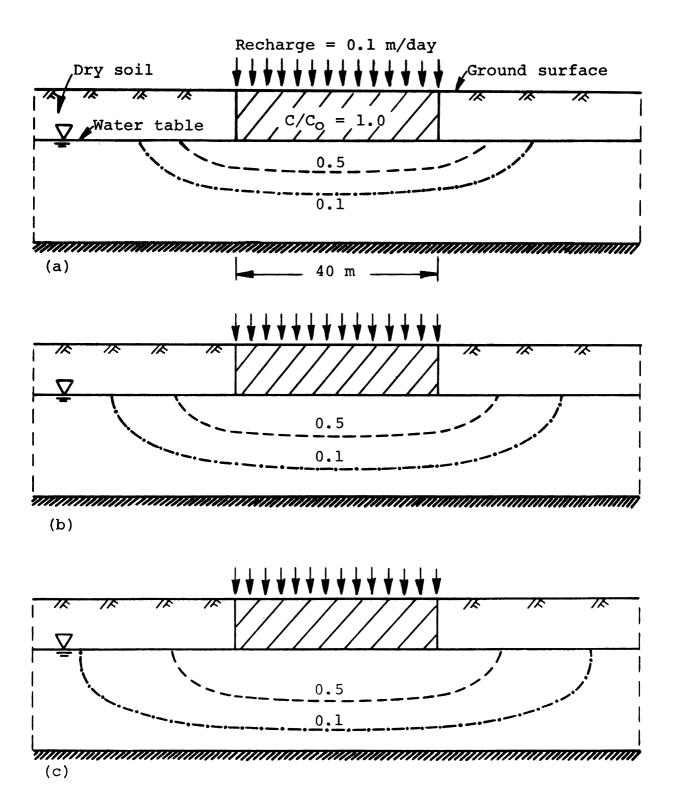


Figure 9-19.--Iso-Concentration Lines in a Phreatic Aquifer After (a) 4 Days, (b) 8 Days, and (c) 12 Days.



above the phreatic surface should remain constant. investigate the capability of the proposed numerical technique for handling such conditions, two different computer runs are examined. In both tests all conditions are kept the same, except that in one run the concentration is set equal to zero at the nodes above the phreatic surface, while in the second run the computer is allowed to calculate the concentration at these nodes. The numerical results for concentration distribution at the interior nodes are the same for both runs (see Figure 6-10 for definitions). But it is observed that a slight concentration of the tracer appears at the nodes above the phreatic surface for the second run. numerical technique might account for this slight deviation. At each element adjacent to the recharge zone the concentration is kept equal to 1 at two nodes, so the numerical solution will compute a small concentration at the other two nodes. Since the computing of the concentration distribution beneath the phreatic surface is the major concern, the deviation discussed above will not contribute any error. As far as it is known by the investigator, no other results exist to compare with the computed velocity vectors and tracer distribution. However, the results are promising and seem realistic.



9.7 Reasonable Time Step for Calculating Tracer Concentration in a Phreatic Aquifer

As discussed in Section 9.2, the numerical solution of the mass transfer equation is sensitive to the time step. Usually, an allowable time step for the solution of the convective-dispersion equation in confined aquifers is smaller than that for flow. In the process of calculating the location of the phreatic surface, choosing the proper time step is very important and instability and oscillation occur for larger time intervals (see Section 8.3.8). Experience shows that in phreatic aquifer problems, the same time step can be used for the solution of both flow and mass transfer phenomena. For example, it was found that DTMAX = 0.2days is adequate for obtaining the location of the phreatic surface in the study of the previous section, and the same time step can be used for the solution of the mass transfer equation. Reduction of the time step to 0.1 day does not have any effect on the results, as shown in Figure 9-20. In this figure the concentration distribution at point A (of Figure 9-17) versus time for two different time steps is given and the results are identi-It is obvious that the time step for the solution cal. of the convective-dispersion equation should not exceed the DTMAX.



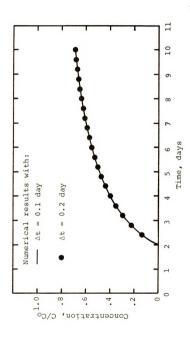


Figure 9-20. -- Variation of Concentration at Point A With Time Using Different Time Steps.



9.8 Summary

In this chapter numerical examples are presented to show the applicability of the employed techniques in predicting tracer movement, both in confined and phreatic aquifers with a transient flow domain. The sensitivity of the numerical results to time steps and orders of time approximation was examined. Primary results reveal that the fixed node technique is capable of solving the convective-dispersion equation in an unconfined phreatic aquifer.



CHAPTER X

SUMMARY AND CONCLUSIONS

In this study the movement of a tracer in an aquifer is investigated. The tracer may be introduced as a constituent of artificial recharge, for example, a chloride ion present in treated sewage water. It is assumed that the tracer remains unaltered in the aquifer. Both two-dimensional regional (horizontal) and two-dimensional local (vertical) flows are considered.

In order to accomplish the movement of the free surface within the grid system without repositioning the nodal coordinates of the elements, a procedure for locating aphreatic boundary of an unconfined aquifer is adapted. In order to obtain continuous flow across elements and at the nodes, the Galerkin formulation of the Darcy law is constructed and velocity vectors are calculated simultaneously at the nodes. These transient velocities are subsequently used in shifting the phreatic surface as well as computing dispersion coefficients and convective terms of the mass-transport equation. Finite element formulation of the flow and convective-dispersion equations leads to a set of first order partial differential relations. Using the finite element concept,



higher order time approximations for the system of equations are derived. The validity of the proposed techniques is established by first comparing the numerical flow results with existing analytical, experimental, and field data. Upon verification of the solution of the flow equation, the prediction of the movement of a tracer in an unconfined aquifer with a transient phreatic boundary and in a confined aquifer with a transient flow condition is conducted. Numerical examples have been presented to demonstrate the capability of the proposed techniques.

It is shown that the Galerkin finite element method can be used to solve the flow and convective-dispersion equations, both in confined and phreatic aquifers under time-variable flow conditions. On the basis of the present study the following conclusions can be made:

1. The fixed node technique is capable of locating the transient phreatic aquifer due to accretion. The numerical results compare favorably with experimental and analytical solutions. The major advantage of this method is that the solution of the convective-dispersion equation for an unconfined aquifer with a movable free surface is possible. The primary results of predicting the tracer movement seem realistic. Much work yet



remains to be done in order to complete this investigation and cover all related aspects of the problems.

- 2. The Galerkin formulation of the Darcy law provides continuous velocity vectors across element boundaries. Calculation of the transient velocity vectors based on known piezometric heads becomes straightforward.
- 3. The time approximation of equations describing the transient behavior of the field problems is an important factor on the stability and convergence of the numerical results. A second order time approximation gives more accurate results for convective-dispersion problems than a first order one.
- 4. The finite element numerical technique provides facility for solving field problems related to flow and mass-transport situations. It yields accurate and realistic results provided that the physical behavior of the phenomena under investigation is well understood, and related parameters and initial and boundary conditions are properly specified.



CHAPTER XI

RECOMMENDATIONS FOR FUTURE STUDIES

- 1. This study is the first step in predicting the tracer movement in an unconfined aquifer with a transient phreatic surface. The fixed node technique is used to find the location of the phreatic surface. The flow solution of this technique agrees favorably with experimental and analytical results and field data, and is believed to provide reasonable results particularly when the rise of the water table is caused by accretion. The calculated tracer distributions appear to be realistic. Still there is more to be done, especially in the areas outlined below:
 - a. Equation (6.3.26) was used to obtain the value of the piezometric head at the phreatic nodes (see Figure 6-10) with constant effective porosity and hydraulic conductivity. In Section 8.5 it was stated that one of the possible ways to reduce the error is to modify the Darcy law for unsaturated flow. This means that in order to improve the technique, especially for the area away from the recharge zone, it is required to



extend the work for saturated and unsaturated porous media.

- b. Assigning the value of the piezometric heads at nodes above the phreatic nodes needs to be further pursued until more realistic conditions can be obtained. As was already discussed, this task becomes more important when the velocity vectors are calculated simultaneously and used in shifting the phreatic nodes, calculating the dispersion coefficients, and evaluating the convective term of the dispersion equation.
- c. Extending this work to three dimensions will be a significant contribution and will reduce the errors that are associated with two-dimensional assumptions.
- d. Obtaining field and laboratory data regarding the movement of dissolved chemical substances in unconfined aquifers will help to verify the numerical models, so that increased confidence will be gained in more sophisticated problems dealing with convective-dispersion phenomena.
- 2. It was shown that the second order time approximation provides more accurate results than the first order approximation. Investigating the effects of

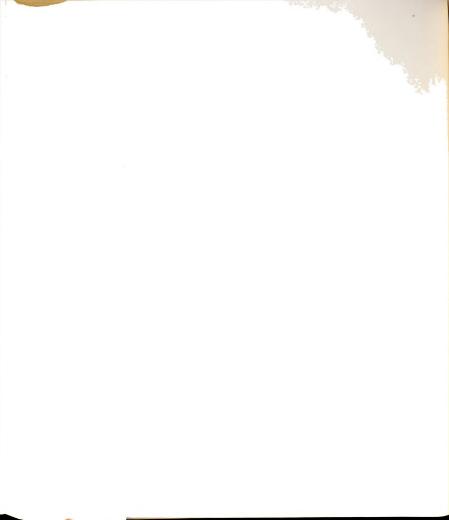


second and third order time approximations on flow and third order time approximation on dispersion is highly recommended.

3. It was observed that the accuracy of the numerical results depends on the type and size of elements. Further investigations regarding both size and type of elements will be very useful.



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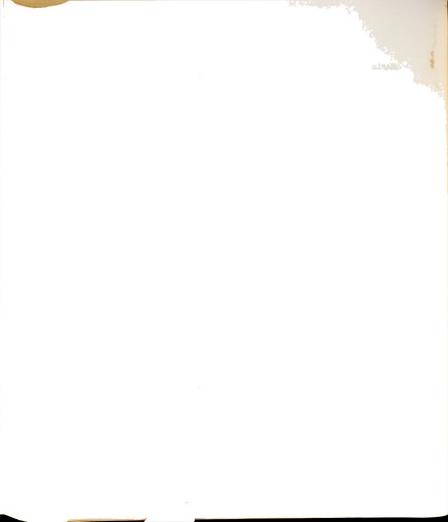
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APPENDICES



APPENDIX I

FINITE ELEMENT DEVELOPMENT



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FINITE ELEMENT DEVELOPMENT

I.l Introduction

In the finite element technique, a continuum is divided into a finite number of subdomains which are called "elements." Each element is designated by "nodes." It is possible to define a functional such that it will describe uniquely the state of a parameter within an element based on its values at the nodes. Polynomials are most commonly used in deriving such functionals, which are termed "shape functions" [Segerlind, in press]. A detailed formulation of the finite element method is given in the literature, e.g., Zienkiewicz [1971], Norrie and de Vries [1973].

Let the dependent variable C in the domain $D^{\mathbf{e}}$ be approximated by

$$C = [N]^{e} \{C\}^{e} = \sum_{n=1}^{M} N_{n}^{e} C_{n}^{e}$$
 (I.1.1)

where N_n are the appropriate shape functions defined piecewise, element by element; C_n are the nodal values of C in the discretized domain; and M is the number of



the degree of freedom (number of nodes in each element). For example, for an element with three nodes

$$C = \begin{bmatrix} N_1 & N_2 & N_3 \end{bmatrix}^e \begin{pmatrix} C_1 \\ C_2 \\ C_3 \end{pmatrix}^e$$

where []^e and {}^e denote a row matrix and a column vector respectively, which contain properties of the three nodes associated with one element.

In this section the shape functions for the different types of elements used in this study are described, and the numerical integration of element matrices for isoparametric elements is discussed. The integrated element matrices for one-dimensional quadratic and two-dimensional triangular elements are given in Appendix II. Finally the procedure for allocation of a constant line source to the boundary nodes is given.

I.2 Types of Finite Elements and Their Shape Functions

One cannot subdivide a continuum into elements without first knowing what general shapes are permissible. In this part some of the more common finite elements which are used in the analysis of flow and dispersion phenomena are given. For the derivation of the shape functions, the reader is referred to Zienkiewicz [1971].



I.2.1 One-Dimensional Element

a. Simple element. -- The simplest one-dimensional element has two nodes, one at each end (Figure I-1).

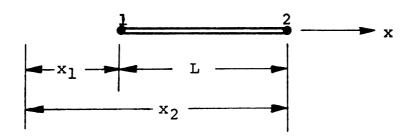


Figure I-1.--Simple One-Dimensional Finite Element.

The shape functions for this element are

$$N_1 = 1 - \left(\frac{x}{L}\right)$$
 and $N_2 = \left(\frac{x}{L}\right)$ (1.2.1)

b. Quadratic element. -- This element has three nodes, two nodes at each end and one in the center of the element (Figure I-2).

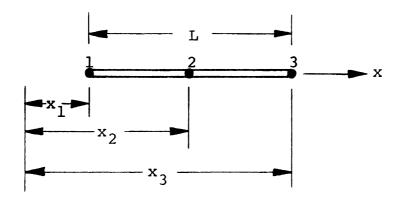


Figure I-2.--One-Dimensional Quadratic Element.



The shape functions for a quadratic element are

$$N_{1} = \left(1 - \frac{2x}{L}\right) \left(1 - \frac{x}{L}\right)$$

$$N_{2} = \frac{4x}{L} \left(1 - \frac{x}{L}\right)$$

$$N_{3} = -\frac{x}{L} \left(1 - \frac{2x}{L}\right)$$
(I.2.2)

c. Quadratic element for axisymmetric case.-This element also has three nodes, but the shape functions are written with respect to the origin of the global coordinate (Figure I-3).

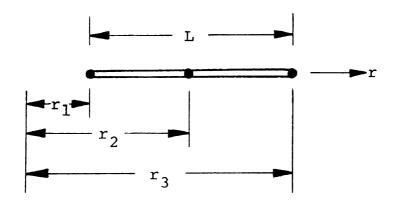


Figure I-3.--Quadratic Element for One-Dimensional Axisymmetric.

The shape functions can be written

$$N_{1} = 2(r - r_{2})(r - r_{3})/L^{2}$$

$$N_{2} = 4(r - r_{1})(r - r_{3})/L^{2}$$

$$N_{3} = 2(r - r_{1})(r - r_{2})/L^{2}$$
(I.2.3)



I.2.2 Two-Dimensional Simplex Element

The two-dimensional simplex element is the triangle shown in Figure I-4. The triangular element has rightly become more popular due to the ease with which the subdivision can be graded and the boundary shapes approximated. The evaluation of the element matrices is simple.

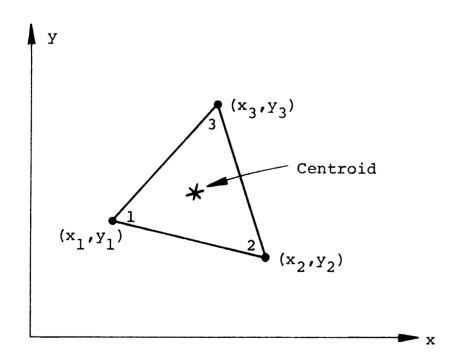


Figure I-4.--Simple Triangular Element.

The shape functions for the triangular element are given below:

$$N_1 = \frac{1}{2\Delta} (a_1 + b_1 x + c_1 y)$$



$$N_{2} = \frac{1}{2} (a_{2} + b_{2}x + c_{2}y)$$

$$N_{3} = \frac{1}{2} (a_{3} + b_{3}x + c_{3}y)$$

$$a_{1} = x_{2}y_{3} - x_{3}y_{2} ; b_{1} = y_{2} - y_{3} ; c_{1} = x_{3} - x_{2}$$

$$a_{2} = x_{3}y_{1} - y_{3}x_{1} ; b_{2} = y_{3} - y_{1} ; c_{2} = x_{1} - x_{3}$$

$$a_{3} = x_{1}y_{2} - x_{2}y_{1} ; b_{3} = y_{1} - y_{2} ; c_{3} = x_{2} - x_{1} (I.2.5)$$

$$\Delta = \frac{1}{2} (b_{1}x_{1} + b_{2}x_{2} + b_{3}x_{3})$$

$$= \text{area of triangle}$$

$$(I.2.6)$$

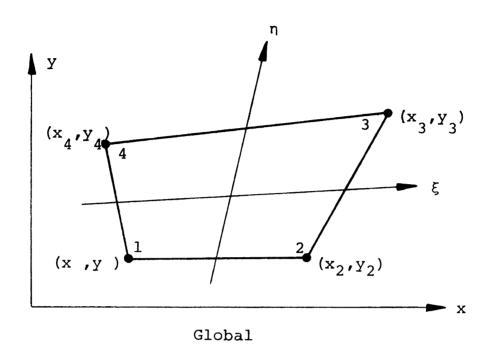
I.2.3 Two-Dimensional Isoparametric Elements

The use of a curvilinear coordinate system has definite advantages when considering two- and three-dimensional elements, because it allows the boundaries of these elements to be distorted.

a. Linear quadrilateral isoparametric element.-- Consider Figure I-5, and define ξ , η such that $-1 \le \xi \le 1$ and $-1 \le \eta \le 1$. The shape functions for the linear quadrilateral element are

$$N_1 = \frac{1}{4} (1 - \xi) (1 - \eta)$$





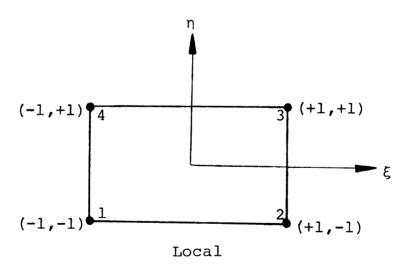


Figure I-5.--Linear Quadrilateral Isoparametric Finite Element on Global and Local Planes.



$$N_{2} = \frac{1}{4} (1 + \xi) (1 - \eta)$$

$$N_{3} = \frac{1}{4} (1 + \xi) (1 + \eta)$$

$$N_{4} = \frac{1}{4} (1 - \xi) (1 + \eta)$$
(I.2.7)

b. Quadratic quadrilateral element. -- The shape functions for the quadratic quadrilateral element (Figure I-6) are given [Zienkiewicz 1971, p. 109]:

$$N_1 = -\frac{1}{4} (1-\xi) (1-\eta) (\xi+\eta+1) ; N_2 = \frac{1}{2} (1-\xi^2) (1-\eta)$$

$$N_3 = \frac{1}{4} (1+\xi) (1-\eta) (\xi-\eta-1)$$
; $N_4 = \frac{1}{2} (1-\eta^2) (1+\xi)$

$$N_5 = \frac{1}{4} (1+\xi) (1+\eta) (\xi+\eta-1)$$
; $N_6 = \frac{1}{2} (1-\xi^2) (1+\eta)$

$$N_7 = \frac{1}{4} (1-\xi) (1+\eta) (-\xi+\eta-1) ; N_8 = \frac{1}{2} (1-\eta^2) (1-\xi) (1.2.8)$$

c. Cubic quadrilateral element. -- The shape functions for the cubic quadrilateral element (Figure I-7) are given [Zienkiewicz 1971, p. 109]:

$$N_1 = \frac{1}{32} (1 - \xi) (1 - \eta) [-10 + 9 (\xi^2 + \eta^2)]$$

$$N_2 = \frac{9}{32} (1 - \eta) (1 - \xi^2) (1 - 3\xi)$$

$$N_3 = \frac{9}{32} (1 - \eta) (1 - \xi^2) (1 + 3\xi)$$



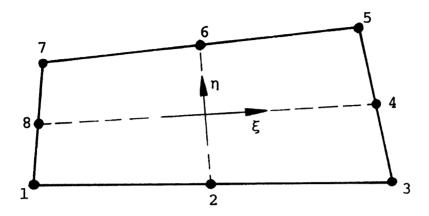


Figure I-6.--Quadratic Quadrilateral Element.

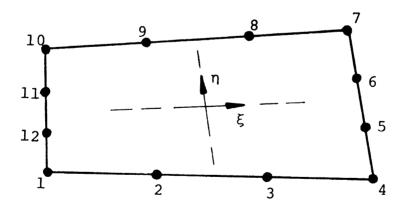


Figure I-7.--Cubic Quadrilateral Element.



$$\begin{split} N_4 &= \frac{1}{32} \; (1 + \xi) \; (1 - \eta) \; [-10 + 9 \; (\xi^2 + \eta^2)] \\ N_5 &= \frac{9}{32} \; (1 + \xi) \; (1 - \eta^2) \; (1 - 3\eta) \\ N_6 &= \frac{9}{32} \; (1 + \xi) \; (1 - \eta^2) \; (1 + 3\eta) \\ N_7 &= \frac{1}{32} \; (1 + \xi) \; (1 + \eta) \; [-10 + 9 \; (\xi^2 + \eta^2)] \\ N_8 &= \frac{9}{32} \; (1 + \eta) \; (1 - \xi^2) \; (1 + 3\xi) \\ N_9 &= \frac{9}{32} \; (1 + \eta) \; (1 - \xi^2) \; (1 - 3\xi) \\ N_{10} &= \frac{1}{32} \; (1 - \xi) \; (1 + \eta) \; [-10 + 9 \; (\xi^2 + \eta^2)] \\ N_{11} &= \frac{9}{32} \; (1 - \xi) \; (1 - \eta^2) \; (1 + 3\eta) \end{split}$$

I.3 Numerical Integration

 $N_{1,2} = \frac{9}{22} (1 - \xi) (1 - \eta^2) (1 - 3\eta)$

Usually the element matrices are in the integral form which has to be evaluated. For example, the element matrices for fow in a two-dimensional horizontal plane are:

$$[B]^{e} = \int_{De} \left[r_{ij} \frac{\partial N_{k}}{\partial x_{i}} \frac{\partial N_{n}}{\partial x_{j}} \right] dD$$
 (4.2.7a)



$$[H]^{e} = \int_{D} e^{S} N_{k} N_{n} dD$$
 (4.2.7b)

$$\{F\}^{e} = \int_{S_{e}} N_{k} Q_{2} dS - \int_{D_{e}} P N_{k} dD$$
 (4.27c)

Integration of Equation (4.2.7) or similar equations for simple elements is straightforward, and some of the integrated forms are given in Appendix II. For isoparametric elements, shape functions are described in the local (ξ , η) coordinates, but Equation (4.2.7) is written in global (x, y) coordinates. To perform the transformation of the shape function derivatives $\partial N_k/\partial x$ and $\partial N_k/\partial y$, the following relationship is used.

$$\begin{bmatrix} \frac{\partial N_{k}}{\partial x} \\ \frac{\partial N_{k}}{\partial y} \end{bmatrix} = \begin{bmatrix} J \end{bmatrix}^{-1} \begin{bmatrix} \frac{\partial N_{k}}{\partial \xi} \\ \frac{\partial N_{k}}{\partial \eta} \end{bmatrix}$$
 (I.3.1)

In which [J] is the Jacobian matrix:

$$[J] = \begin{bmatrix} \frac{\partial x}{\partial \xi} & \frac{\partial y}{\partial \xi} \\ & & \\ \frac{\partial x}{\partial \eta} & \frac{\partial y}{\partial \eta} \end{bmatrix} = \begin{bmatrix} \frac{\partial N_1}{\partial \xi} & \frac{\partial N_2}{\partial \xi} & \cdots & \frac{\partial N_M}{\partial \xi} \\ & & & \\ \frac{\partial N_1}{\partial \eta} & \frac{\partial N_2}{\partial \eta} & \cdots & \frac{\partial N_M}{\partial \eta} \end{bmatrix} \begin{bmatrix} x_1 & y_1 \\ x_2 & y_2 \\ \vdots & \vdots \\ x_M & y_M \end{bmatrix}$$
 (I.3.2)

where x_1, x_2, \ldots, x_M and y_1, y_2, \ldots, y_M are the nodal coordinates. Another transformation required is



the replacement of the element of area, dxdy(dD), by the expression:

$$dD^{(e)} = dxdy = det [J] d\xi d\eta \qquad (I.3.3)$$

The limits of integration in the local coordinate system become -1 and +1. For example, Equation (4.2.7a) will change to

$$\mathbf{B}_{\mathbf{k}n}^{\mathbf{e}} = \int_{-1}^{+1} \int_{-1}^{+1} \left(\mathbf{T}_{\mathbf{x}\mathbf{x}} \frac{\partial \mathbf{N}_{\mathbf{k}}}{\partial \mathbf{x}} \frac{\partial \mathbf{N}_{\mathbf{n}}}{\partial \mathbf{x}} + \mathbf{T}_{\mathbf{y}\mathbf{y}} \frac{\partial \mathbf{N}_{\mathbf{k}}}{\partial \mathbf{y}} \frac{\partial \mathbf{N}_{\mathbf{n}}}{\partial \mathbf{y}} \right) \det [\mathbf{J}] d\eta d\xi \qquad (1.3.4)$$

Similar expressions are developed for the remaining terms of Equation (4.2.7), Equation (4.3.13), and Equation (4.4.7), etc.

Equation (I.3.4) and similar equations are integrated by the Gaussian quadrature integration technique [e.g., see Zienkiewicz 1971, pp. 144-149]. For a polynomial of degree 2n = 1, the number of sampling points will be n.

I.4 Parameter Definition

The parameters such as hydraulic conductivity, dispersion coefficients, storage coefficient, resistivity of aquitard, etc., can be specified either for each node or each element. Because the numerical solution requires slightly less computation time when parameters are assumed constant over an element, usually the parameters



such as storage coefficient which do not change considerably are specified for each element. Since velocity components vary within the element, dispersion coefficients are specified at each node.

I.5 Allocation of a Constant Line Source to Boundary Nodes

Dirichlet (specified head or concentration) and Neuman (specified flux) conditions are the two boundary conditions generally encountered in field problems. Introducing the Dirichlet boundary condition was discussed in Section 6.1.2. When the normal flux \mathbf{q}_n is assumed constant along an element face of length L, the integration of

$$\int_{S} e^{N_k q_s ds}$$
 (I.5.1)

will result in the constant flux to allocated at the boundary nodes. The results for elements with two, three, and four nodes along the boundary line are given in Figure (I-8). One-dimensional linear, quadratic, and cub shape functions are used in this integration.



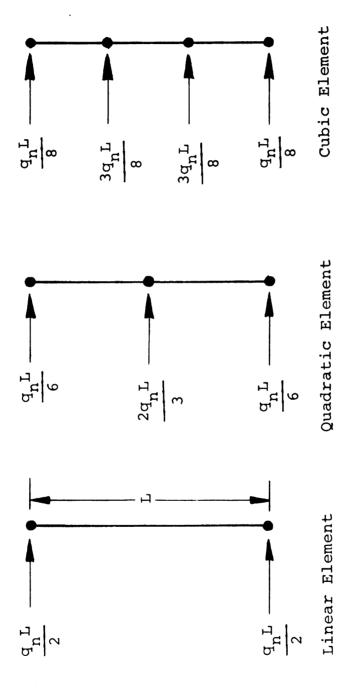


Figure I-8.--Allocation of a Constant Line Source to Boundary Nodes in Different Shapes of Elements.



APPENDIX II

INTEGRATED ELEMENT MATRICES FOR ONEDIMENSIONAL QUADRATIC AND TWODIMENSIONAL TRIANGULAR ELEMENTS



APPENDIX II

INTEGRATED ELEMENT MATRICES FOR ONEDIMENSIONAL QUADRATIC AND TWODIMENSIONAL TRIANGULAR ELEMENTS

In this section some of the integrated terms that are used in the construction of the element matrices for flow and dispersion phenomena are provided.

II.1 One-Dimensional Quadratic Element

The one-dimensional quadratic element is depicted in Figure I-2. Its shape functions are given by Equation (I.2.2), and can be written as

$$N_1 = 1 - \frac{3x}{L} + \frac{2x^2}{L^2}$$

$$N_2 = \frac{4x}{L} - \frac{4x^2}{L^2}$$

$$N_3 = \frac{2x^2}{L^2} - \frac{x}{L} \tag{II.1.1}$$

The first derivative of the shape functions is

$$\frac{\partial N_1}{\partial x} = \frac{4x}{L^2} - \frac{3}{L}$$



$$\frac{\partial N_2}{\partial x} = \frac{4}{L} - \frac{8x}{L^2}$$

$$\frac{\partial N_3}{\partial x} = \frac{4x}{L^2} - \frac{1}{L}$$
(II.1.2)

In Equations (II.1.1), (II.1.2), and in the following equations, L represents the length of an element.

1.
$$\int_{\mathbf{X}} \frac{\partial \mathbf{N}_{\mathbf{n}}}{\partial \mathbf{x}} \frac{\partial \mathbf{N}_{\mathbf{k}}}{\partial \mathbf{x}} d\mathbf{x} = \int_{\mathbf{X}} \begin{bmatrix} \frac{\partial \mathbf{N}_{\mathbf{1}}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N}_{\mathbf{1}}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N}_{\mathbf{1}}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N}_{\mathbf{1}}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N}_{\mathbf{2}}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N}_{\mathbf{2}}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N}_{\mathbf{2}}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N}_{\mathbf{2}}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N}_{\mathbf{3}}}{\partial \mathbf{x}} \end{bmatrix} d\mathbf{x}$$

$$= \frac{1}{3L} \begin{bmatrix} 7 & -8 & 1 \\ -8 & 16 & -8 \\ 1 & -8 & 7 \end{bmatrix}$$
(II.1.3)

2.
$$\int_{x} N_{n}N_{k} dx = \int_{x} \begin{bmatrix} N_{1}N_{1} & N_{1}N_{2} & N_{1}N_{3} \\ N_{2}N_{1} & N_{2}N_{2} & N_{2}N_{3} \\ N_{3}N_{1} & N_{3}N_{2} & N_{3}N_{3} \end{bmatrix} dx$$

$$= \frac{L}{30} \begin{bmatrix} 4 & 2 & -1 \\ 2 & 16 & 2 \\ -1 & 2 & 4 \end{bmatrix}$$
 (II.1.4)



3.
$$\int_{\mathbf{x}} \mathbf{N_k} \mathbf{N_n} \frac{\partial \mathbf{N_n}}{\partial \mathbf{x}} d\mathbf{x} = \int_{\mathbf{x}} \begin{bmatrix} \mathbf{N_1} \mathbf{N_1} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_1} \mathbf{N_2} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_1} \mathbf{N_3} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix} d\mathbf{x}$$
$$\begin{bmatrix} \mathbf{N_2} \mathbf{N_1} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_2} \mathbf{N_2} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_2} \mathbf{N_3} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix} d\mathbf{x}$$
$$\begin{bmatrix} \mathbf{N_3} \mathbf{N_1} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_3} \mathbf{N_2} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_3} \mathbf{N_3} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix}$$

$$= \frac{1}{30} \begin{bmatrix} -10 & 8 & -1 \\ -6 & 0 & 6 \\ 1 & -8 & 10 \end{bmatrix}$$
 (II.1.5)

4.
$$\int_{\mathbf{x}} \mathbf{N_k} \frac{\partial \mathbf{N_n}}{\partial \mathbf{x}} d\mathbf{x} = \int_{\mathbf{x}} \begin{bmatrix} \mathbf{N_1} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_1} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_1} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix} d\mathbf{x}$$
$$\begin{bmatrix} \mathbf{N_2} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_2} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_2} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix} d\mathbf{x}$$
$$\begin{bmatrix} \mathbf{N_3} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_3} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_3} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix}$$

$$= \frac{1}{6} \begin{bmatrix} -3 & 4 & -1 \\ -4 & 0 & 4 \\ 1 & -4 & 3 \end{bmatrix}$$
 (II.1.6)

$$5. \int_{\mathbf{x}} \mathbf{N_n} \frac{\partial \mathbf{N_k}}{\partial \mathbf{x}} \frac{\partial \mathbf{N_n}}{\partial \mathbf{x}} d\mathbf{x} = \int_{\mathbf{x}} \begin{bmatrix} \mathbf{N_1} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_2} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_3} \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix} d\mathbf{x}$$

$$\begin{bmatrix} \mathbf{N_1} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_2} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_3} \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix} d\mathbf{x}$$

$$\begin{bmatrix} \mathbf{N_1} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_1}}{\partial \mathbf{x}} & \mathbf{N_2} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_2}}{\partial \mathbf{x}} & \mathbf{N_3} \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} & \frac{\partial \mathbf{N_3}}{\partial \mathbf{x}} \end{bmatrix} d\mathbf{x}$$



$$= \frac{1}{30L} \begin{bmatrix} 37 & -32 & 7 \\ -44 & 64 & -44 \\ 7 & -32 & 37 \end{bmatrix}$$
 (II.1.7)

6.
$$\int N_k (N_n)^2 dx = \int_{x} \begin{bmatrix} N_1 N_1 N_1 & N_1 N_2 N_2 & N_1 N_3 N_3 \\ N_2 N_1 N_1 & N_2 N_2 N_2 & N_2 N_3 N_3 \end{bmatrix} dx$$

$$= \frac{L}{420} \begin{bmatrix} 39 & 16 & -3 \\ 20 & 192 & 20 \\ -3 & 16 & 39 \end{bmatrix}$$
 (II.1.8)

7.
$$\int_{X} N_{k} dx = \int_{X} \begin{Bmatrix} N_{1} \\ N_{2} \\ N_{3} \end{Bmatrix} dx = \frac{L}{6} \begin{Bmatrix} 1 \\ 4 \\ 1 \end{Bmatrix}$$
 (II.1.9)

II.2 Two-Dimensional Simplex Element

The two-dimensional simplex element shown in Figure I-4 and the shape functions and related terms are defined in Section I.2.2.

The first derivatives of the shape function with respect to \mathbf{x} and \mathbf{y} are



$$\frac{\partial N_{\ell}}{\partial x} = b_{\ell}$$

$$\frac{\partial N_{\ell}}{\partial y} = c_{\ell}$$

$$\ell = 1, 2, 3$$
(II.2.1)

1.
$$\int_{A} \frac{\partial N_{n}}{\partial x} \frac{\partial N_{k}}{\partial x} dA = \frac{1}{4\Delta} \begin{bmatrix} b_{1}b_{1} & b_{1}b_{2} & b_{1}b_{3} \\ b_{1}b_{2} & b_{2}b_{2} & b_{2}b_{3} \\ b_{1}b_{3} & b_{2}b_{3} & b_{3}b_{3} \end{bmatrix}$$
 (II.2.2)

2.
$$\int_{A} \frac{\partial N_{n}}{\partial y} \frac{\partial N_{k}}{\partial y} dA = \frac{1}{4\Delta} \begin{bmatrix} c_{1}c_{1} & c_{1}c_{2} & c_{1}c_{3} \\ c_{1}c_{2} & c_{2}c_{2} & c_{2}c_{3} \\ c_{1}c_{3} & c_{2}c_{3} & c_{3}c_{3} \end{bmatrix}$$
 (II.2.3)

3.
$$\int_{A} \frac{\partial N_{k}}{\partial x} \frac{\partial N_{n}}{\partial y} dA = \frac{1}{4\Delta} \begin{bmatrix} b_{1}c_{1} & b_{1}c_{2} & b_{1}c_{3} \\ b_{2}c_{1} & b_{2}c_{2} & b_{2}c_{3} \\ b_{3}c_{1} & b_{3}c_{2} & b_{3}c_{3} \end{bmatrix}$$
 (II.2.4)

4.
$$\int_{A} \frac{\partial N_{k}}{\partial y} \frac{\partial N_{n}}{\partial x} dA = \frac{1}{4\Delta} \begin{bmatrix} b_{1}c_{1} & b_{2}c_{1} & b_{3}c_{1} \\ b_{1}c_{2} & b_{2}c_{2} & b_{3}c_{2} \\ b_{1}c_{3} & b_{2}c_{3} & b_{3}c_{3} \end{bmatrix}$$
 (II.2.5)



5. Evaluation of $\int_A N_k \frac{\partial N_n}{\partial x} dA$:

$$\int_{A} N_{k} \frac{\partial N_{n}}{\partial x} dA = \int_{A} \begin{bmatrix} N_{1} \frac{\partial N_{1}}{\partial x} & N_{1} \frac{\partial N_{2}}{\partial x} & N_{1} \frac{\partial N_{3}}{\partial x} \\ N_{2} \frac{\partial N_{1}}{\partial x} & N_{2} \frac{\partial N_{2}}{\partial x} & N_{2} \frac{\partial N_{3}}{\partial x} \end{bmatrix} dA$$

$$\begin{bmatrix} N_{k} \frac{\partial N_{n}}{\partial x} & N_{k} \frac{\partial N_{n}}{\partial x} & N_{k} \frac{\partial N_{n}}{\partial x} & N_{k} \frac{\partial N_{n}}{\partial x} \end{bmatrix} dA$$

$$= \frac{1}{2\Delta} \int_{A} \begin{bmatrix} N_1b_1 & N_1b_2 & N_1b_3 \\ N_2b_1 & N_2b_2 & N_2b_3 \\ N_3b_1 & N_3b_2 & N_3b_3 \end{bmatrix} d\Lambda \quad (II.2.6)$$

Each term in Equation (II.2.6) can be integrated separately, e.g.,

$$\frac{1}{2\Delta} \int_{A} N_{1}b_{1} dA = \frac{1}{4\Delta^{2}} \int_{A} (a_{1} + b_{1}x + c_{1}y) b_{1} dx dy \qquad (II.2.7)$$

$$\int_{A} dx dy = \Delta = \text{area of triangle}$$

$$\int_{A} x dx dy = \int_{A} y dx dy = 0$$

Then equation (II.2.7) is reduced to



$$\frac{1}{4\Lambda^2} \int a_1 b_1 dx dy = \frac{a_1 b_1}{4\Delta}$$
 (II.2.8)

If it is assumed that the origin of the coordinates is taken at the centroid of the element, i.e.,

$$\frac{x_1 + x_2 + x_3}{3} = 0$$

$$\frac{y_1 + y_2 + y_3}{3} = 0 (II.2.9)$$

Then $a_1 = \frac{2\Delta}{3} = a_2 = a_3$ [Zienkewicz 1971, Eq. 4.8].

Substituting the value of a_1 into Equation (II.2.8) yields

$$\frac{a_1b_1}{4\Delta} = \frac{b_1}{6}$$

Thus the integral (II.2.6) is simplified as

$$\int_{A} N_{k} \frac{\partial N_{n}}{\partial x} dA = \frac{1}{6} \begin{bmatrix} b_{1} & b_{2} & b_{3} \\ b_{1} & b_{2} & b_{3} \\ b_{1} & b_{2} & b_{3} \end{bmatrix}$$
 (II.2.10)

Note that the assumption leading to Equation (II.2.9) is automatically satisfied in the computer program, regardless of whether local or global coordinates are used.



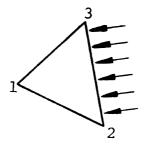
6.
$$\int_{A} N_{k} \frac{\partial N_{n}}{\partial y} dA = \frac{1}{2\Delta} \int_{A} \begin{bmatrix} N_{1}c_{1} & N_{1}c_{2} & N_{1}c_{3} \\ N_{2}c_{1} & N_{2}c_{2} & N_{2}c_{3} \\ N_{3}c_{1} & N_{3}c_{2} & N_{3}c_{3} \end{bmatrix}$$

$$= \frac{1}{6} \begin{bmatrix} c_1 & c_2 & c_3 \\ c_1 & c_2 & c_3 \\ c_1 & c_2 & c_3 \end{bmatrix}$$
 (II.2.11)

7.
$$\int_{A} N_{k} N_{n} dA = \frac{\Delta}{12} \begin{bmatrix} 2 & 1 & 1 \\ 1 & 2 & 1 \\ 1 & 1 & 2 \end{bmatrix}$$
 (II.2.12)

8. Evaluation of $\int_S N_k q_n$ ds: If q_n is the constant flux along the line 2-3, as shown in the following figure, then

$$\int_{S} N_{k} q_{n} d_{S} = \frac{Lq_{n}}{2} \begin{Bmatrix} 0 \\ 1 \\ 1 \end{Bmatrix}$$



where L is the length of line 2-3.



APPENDIX III

DEVELOPED COMPUTER PROGRAMS



APPENDIX III

DEVELOPED COMPUTER PROGRAMS

For the numerical investigation of this thesis, several computer programs have been developed for use on a CDC 6500 computer with FORTRAN IV Extended language. The finite element technique is used in formulating all of the programs. Some of these programs will be documented and available. The more pertinent programs are as follows:

- One-dimensional plane or axisymmetric medium;
 - a. flow
 - b. tracer movement with uniform or transient flow
- 2. Two-dimensional horizontal plane medium:
 - a. flow with mixed elements (triangular or quadrilateral elements)
 - b. dispersion with simplex triangular elements
 - c. dispersion with quadrilateral elements
- 3. Two-dimensional vertical plane medium with transient phreatic surface:
 - a. flow MNT quadrilateral alements
 - b. flow FNT linear quadrilateral elements
 - c. dispersion linear quadrilateral elements



In all programs, options are provided to obtain the velocity vector either directly or simultaneously. Different θ -values (see Chapter V) can be used in approximation of time-dependent functions, with a first or second order time approximation available for the convective-dispersion solution. Uniform, steady-state, or transient flow can be used in the calculation of mass-transfer phenomena.



