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A RHEOLOGICAL MODEL OF APPLE FLESH FAILURE DURING THE MAGNESS-TAYLOR PUNCTURE TEST

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Sanghyup Jeong

A THESIS

Submitted to
Michigan State University
in partial fulfillment of the requirements
for the degree of

MASTER OF SCIENCE

Department of Agricultural Engineering

1997

ABSTRACT

A RHEOLOGICAL MODEL OF APPLE FLESH FAILURE DURING THE MAGNESS-TAYLOR PUNCTURE TEST

By

Sanghyup Jeong

In an attempt to assess firmness, rheological behavior of apple flesh was studied by developing a laminated cell layer model which can be used to emulate Magness-Taylor puncture test.

Each layer of the model consisted of a compressive and a shear-tensile element.

Compressive component consisted of a Maxwell model with a fracture element. The shear-tensile component consisted of an elastic element. The annular pumping effect was also included in model. Layers were connected in series and each shear-tensile component was grounded. Simulation was performed on this model with element parameters and the postulated deformation sequence. The model depicted fairly well the Magness-Taylor test results both quantitatively as well as qualitatively.

Nondestructive techniques using Hertz contact stress theory and acoustic impulse technique (Armstrong, 1990) were used for estimating model parameters. These methods were fairly accurate in predicting the model parameters. Also, a strong correlation with the Magness-Taylor firmness reading was found.

ACKNOWLEDGMENTS

I would like to thank Dr. Ajit K. Srivastava for his direction, support, encouragement, and patience. I also would like to thank the other members of my committee, Dr. Dan Guyer and Dr. Randolph M. Beaudry for their suggestions and assistance. I thank Mr. Dick Wolthius for his technical advice.

I also give my thanks to my parents for their financial and spiritual support for me.

All of the Paul Harris fellows of the Rotary Foundation deserve my heartfelt thanks for their financial support for student like me.

I would like to thank all the colleagues of graduate office 9.

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Chapter 1

INTRODUCTION

1.1 Justification

Fresh market apples are sorted by color, size, weight and firmness. Apples classed by their physical properties and texture are processed for the various apple products. The texture of apples is an important quality attribute. Apple firmness is an indicator of textural qualities. Bourne (1982) defined firmness as a texture characteristic during mastication that displays moderate resistance to breaking. The Magness-Taylor pressure test developed in 1925 to measure firmness, has been accepted as an industry standard. The Magness-Taylor pressure test consists of forcing a cylindrical metal tip into the apple flesh without skin and measuring the maximum force. The penetration depth is about 8 mm (5/16 in) and the diameter of the tip is approximately 11 mm (7/16 in) or 8 mm (5/16 in). The plunger tip is round and the penetration depth is marked on the plunger.

The destructive nature of the Magness-Taylor pressure test results in fruit loss which contributes to economic loss for the apple industry. The firmness assessment is based on sampling. Export markets require a better quality assessment of the product. A non-destructive technique that can make it possible to test individual fruit would be beneficial. The Magness-Taylor method relies on only one indicator to describe the textural quality of the apple, the maximum force to masticate apple flesh. Other parameters such as

1

An in-depth understanding of the Magness-Taylor test would be very helpful in developing a nondestructive technique assessing fruit firmness to meet the needs of high quality produce.

1.2 Objectives

The objectives of the research were

- to develop a rheological model of apple flesh and to examine the capability of the model for predicting the Magness-Taylor puncture test result.
- to evaluate the potential of a rubber spherical indentor as a means of estimating the compression and shear failure stresses of apple flesh.
- to investigate the capability of the spherical indentor as a non-destructive technique to measure apple firmness.

Chapter 2

REVIEW OF LITERATURE

2.1 Rheological Models Representing the Textural Characteristics of Food Materials

Much effort have been devoted to developing rheological models to describe complex texture qualities of food materials. New rheological elements have been suggested along with a combination of the conventional rheological elements e.g. spring and dashpot. These models have been subjected to different types of loading conditions and their responses have been compared to the experimental results. Maxwell and Kelvin models which are the classical models in rheology have already been studied and evaluated under compression, extension, creep, and relaxation tests. Maxwell and Kelvin models are composed of a spring and a dashpot in a serial and in a parallel pattern, respectively.

Peleg (1976) proposed a model system which was basically a parallel array of generalized Maxwell body. He added two fracture elements which were suggested by Drake (1971) to the model. The general elements of this model had an elastic stage and a Maxwellian stage when the model worked as a Maxwell model until fracture. The number of each stage and the parameters of the elements determine the pattern of rheological behavior of a given material. This model can describe various rheological behavior qualitatively.

Chen and Rosenberg (1977) suggested a nonlinear viscoelastic model which consisted

of Maxwell and Kelvin models containing an yield element. The model exhibited

Maxwellian behavior and then it showed the behavior of a four element Burgers model
which was a serial combination of a Maxwell and a Kelvin model after the yield element
was activated. He indicated that the model was able to represent the behavior of

American cheese by adjusting model parameters.

Dickinson and Goulding (1980) tested various types of cheese and found that the model suggested by Chen and Rosenberg (1977) could be used to fit individual compression curve but the model was unable to predict general load deformation behavior.

Johnson and et al. (1981) modified the model suggested by Peleg (1976) by incorporation of a contact and a fracture element. In this study the contact and the fracture element were functions of time and temperature to describe the rheological behavior of raw and cooked fish meat. They demonstrated that the model worked well to describe the rheological behavior qualitatively.

McLaughlin (1987) had a different approach to develop a model describing apple tissue failure phenomenon. Yield stress data for 800 cylindrical samples was studied. Statistical test failed to reject (p=0.85) the hypothesis that the data followed a three parameter Weibull distribution. This statistical model was suggested as an adequate probability model for predicting the failure stress of apple flesh.

During the past decade most of the efforts were concentrated on any developing property measurement techniques rather than developing a model having any physical meaning.

2.2 Rheological Considerations of Apple

Much research has been conducted about the phenomenological and morphological characteristics of the failure of fruits. Mechanical and structural characteristics of fruits were observed and discussed. An understanding the nature of failure will help develop appropriate models having physical meaning.

Peleg and et al. (1976) studied compressive failure patterns of several juicy fruits.

Compression tests were performed to the cylindrical samples by using an Instron

Universal Testing Machine. Fruits showed their characteristic failure patterns. Effect of
the juiciness and dimension of the sample were studied. In this research it was found that
structural characteristics, liquid content, and geometry of the sample affect the forcedeformation relationship.

Mohsenin (1977) studied the failure of food materials from the point view of solid mechanics. Working with apple tissues he found that the shear resistance was a result of the cementing agent (pectic substance) in the middle lamella holding together the cells in the parenchyma tissue. He also studied the potential usage of the Hertz contact stress theory to find failure parameters from the force-deformation relationship.

Pitt (1982) considered two different types of failure, cell wall rupture and cell debonding. Sample was considered as the sum of a large number of cell layers normal to the direction of the applied force. In case of constant rate strain due to the rupture of the weakest cell of a layer, neighboring cells had higher stress level than before. If the stress is higher than cell strength then failure occurs and this process propagates through the layers of the sample. Due to this fractured transverse plane enzymatic browning occurred

in that cross section. This catastrophic failure was observed in the fresh apple tissue.

Lin and Pitt (1986) reported that turgor pressure and the strain rate affect failure mode of fruit and vegetables and tissue strength.

Khan and Vincent (1990) observed that cells of apple flesh are arranged in radial column and the cell size and shape changes along with the radial direction by using scanning electron microscope, indicating that apple flesh should be considered as an anisotropic material in the radial direction.

Khan and Vincent (1993) observed that if a radial cylindrical apple flesh sample was subjected to a compressive force then it generally fractured by collapse of a single layer of cells at right angle to the direction of force. Tangential sample failed in shear. This phenomenon is due to the anisotropic property of apple parenchyma which was observed by Khan and Vincent (1990).

Chapter 3

THEORY

3.1 Modeling Principles and General Assumptions

To develop a model representing Magness-Taylor puncture test it is necessary to observe and analyze the mastication phenomenon. Basically Magness-Taylor puncture test involves three major factors which are compression, shear-tension and annular pumping effects. These factors will be combined into one system representing the apple flesh after being studied and modeled individually. Throughout this research several simplifying assumptions were made. Even though the actual Magness-Taylor firmness instrument has round tip the radius of curvature is large enough so that the plunger tip was considered as flat. The second assumption was about the cylindrical apple flesh sample. Actually the layer of the sample is curved because of its radial arrangement of cells. However, the curvature was relatively large enough to be treated as flat because the sample diameter was considerably less than that of the whole apple.

3.2 Modeling Compression Component

If several ideal conditions of the compression test are satisfied the saw-tooth shaped force deformation relationship as shown in Figure 1 could be obtained. The first ideal condition is to make perfect cylindrical apple flesh sample which means the flat cutting

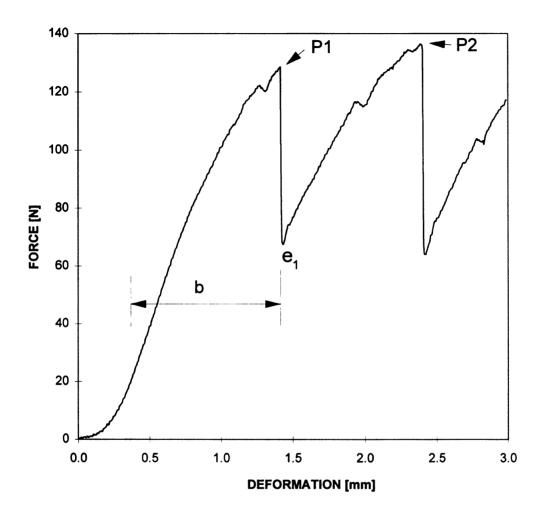


FIGURE 1 - Force-deformation relationship of compression test for cylindrical apple flesh

surface are normal to the direction of the compressive force. The second ideal condition is lack of friction between cutting surface and the pressing die. The last ideal condition is satisfied if the sample is obtained in the radial direction from the whole apple because of the anisotropic characteristics of apple parenchyma studied by Khan and Vincent (1990). In these ideal conditions sample shows two or three fractures caused by collapse of a single layer of cells at right angles to the force (Figure 1). A simple model representing this phenomenon was developed. This model was the building block of the final model.

The first compression region (b, Figure 1) was represented by a Maxwell model and a fracture element having σ_c as its yield stress (Figure 2). The model was used to represent a single layer of cells.

Mathematical expression for the model at constant strain rate is,

$$\sigma = \eta V (1 - e^{-\frac{E}{\eta}}) \quad , \quad 0 \le \sigma < \sigma_c \label{eq:sigma}$$
 [1]

where

 $\sigma = Stress$

 $\eta = Viscosity$

E = Elastic modulus

 σ_c = Critical yielding stress

t = Time

V = Constant strain rate

Based on the research of Khan (1993) a cylindrical apple sample can be considered as a system of stacks of a single cell layer. By connecting the model in a series pattern the system of a cylindrical sample can be depicted physically. The model is shown in Figure 3. If apple flesh is assumed to be an ideal isotropic material in the radial direction σ_c would be the same for each layer. However, it is expected that a limited variation would exist from layer to layer in real life. Thus, if the model is subjected to compressive

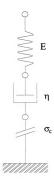


FIGURE 2 - Compression model consisted of a Maxwell and a fracture element

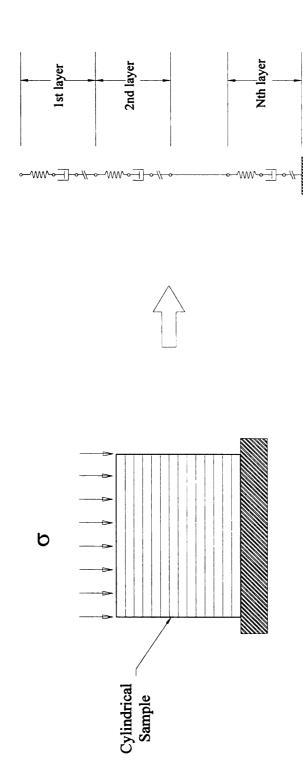


FIGURE 3 - Modeling concept of cylidrical apple flesh sample

force a layer containing the lowest yield stress will collapse first and so on. The fact that the P2 is greater than the P1 in Figure 1 supports the above reasoning.

If the cylindrical apple flesh sample is subjected to a stress σ (Figure 3) the total deformation will be distributed to each cell layer. The layer having the weakest yield stress will collapse first. The cylindrical apple flesh sample which has multi-layer of cells can be represented by stacking the unit compression model (Figure 2). Elasticity and viscosity of each layer were assumed to be the same. However each layer had different critical yield stress.

At time t, the strain and the stress of every layer were equal because the compression model in Figure 3 was a serial system until the stress reached the lowest critical yield stress. The strain of unit layer was equal to the total strain divided by the number of cell layers.

Thus,

$$\sigma_1 = \sigma_2 = \cdots = \sigma_N = \sigma_u = \sigma$$

$$\varepsilon_1 = \varepsilon_2 = \dots = \varepsilon_N = \varepsilon_u = \frac{\varepsilon}{N}$$

[2]

where $\sigma_u = \text{Stress of unit layer}$

 σ = Total stress

 $\varepsilon_u = Strain of unit layer$

 ε = Total strain

N = Layer index

If the cylindrical apple flesh sample was considered as a simple combination of a Maxwell model and a fracture element then,

$$\dot{\varepsilon} = \sum_{i=1}^{N} \dot{\varepsilon}_{i} = N \dot{\varepsilon}_{u} = V$$
 [3]

Eq. [3] can be expressed as follows by using eq. [1]

$$N\frac{\sigma_u}{\eta_u(1-e^{\frac{\epsilon}{\eta_u}})} = \frac{\sigma}{\eta(1-e^{\frac{\epsilon}{\eta_l}})}$$
 [4]

From eq. [4] the following relationships were obtained.

$$E_u\!\!=\!\!NE$$
 $\eta_u\!\!=\!\!N\eta$ [5]

Therefore the properties of unit layer, E_u and η_u , can be calculated from eq. [5] if the E and η were known. E and η can be found from the force deformation relation of the compression test by curve fitting using eq. [1]. These unit parameters will be used in the final allied model.

After the first collapse system is undergoing inequilibrium state for a short period until the system gets to equilibrium state. When the system gets to the equilibrium the total number of layers is reduced by one. Because the transition period is too short, total strain can be assumed to remain unchanged. Thus,

$$\varepsilon_{u,\text{before}} = \frac{\sigma}{E}\Big|_{\text{before}}$$

$$\varepsilon_{u,\text{after}} = \frac{\sigma}{E}\Big|_{\text{utter}}$$
[6]

The total strains, before and after collapse of a layer, are assumed to be same due to the short transition. Therefore,

$$\mathbf{N} \cdot \mathbf{\varepsilon}_{u \text{ hefore}} = (\mathbf{N} - 1) \cdot \mathbf{\varepsilon}_{u \text{ after}}$$
 [7]

The strain after the fracture is expressed as follows

$$\epsilon_{\text{u,after}} = \frac{\ell - \left\{ (\ell - \ell \epsilon_{\text{before}}) + \frac{(\ell - \ell \epsilon_{\text{before}})}{N - 1} \right\}}{\ell} = \frac{N \epsilon_{\text{before}} - 1}{N - 1}$$
 [8]

where ℓ = thickness of a unit layer at time t

By using eq. [6], [7] and [8] the following equation is found

$$\sigma_{\text{ufter}} = \sigma_{\text{before}} - \frac{E_{\text{u,before}} - \sigma_{\text{before}}}{N - 1}$$
 [9]

By using eq. [9], e_1 in Figure 1 can be predicted. This fact supports that the suggested compression model is reasonable.

3.3 Modeling Shear-tensile Component

Figure 4 shows the cross sectional deformation pattern of the apple flesh subjected to compression by a cylindrical plunger. The strain of each layer decreased as the depth increased. In Figure 4, layer shaped like a circular plate in a real situation modeled as a two dimensional elastic element with the presence of a fracture element. A fracture element represents the yield stress of a layer. In this case yielding involves tensile and shear simultaneously. It is difficult to perform an ideal shearing test because of the ductile characteristic of food material and the clearance between the shearing plunger and guiding wall. Thus, the yield stress obtained from shearing test naturally contains tensile and shear contribution. The yield stress from shearing can be used as the critical stress of the fracture element. Also the angle α is important to determine the deformation rate of shear-tensile component. The angle will vary within 0 to 90 degrees as the compression goes on. If the angle is 90 degrees the deformation of shear-tensile component will be the same as the sum of deformation rate of each layer below that shear-tensile component. However, in a real situation the angle is less than 90 degrees and also it changes with time. In this research the angle is assumed constant and the deformation rate of the shear-



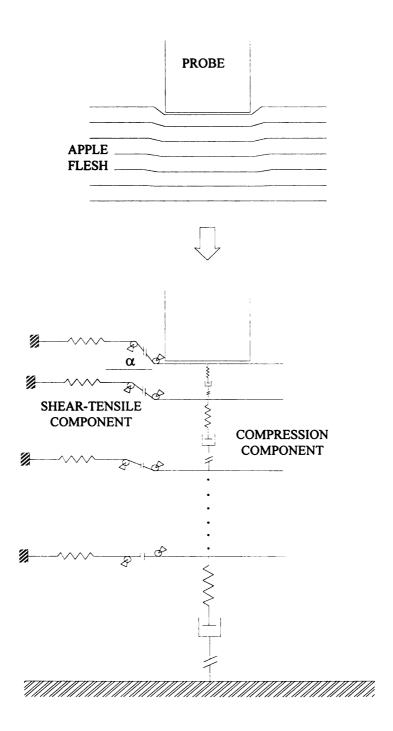


FIGURE 4 - Modeling concept of shear-tensile component

tensile component to the compression component was assumed 0.7.

3.4 Modeling Annular Pumping Component

Apple flesh contains about 80% of water by weight. Therefore, the squeezing action of cylindrical probe produces juice. If the speed is high, then the resistance created by juice is significant. The juice is expelled through the clearance in between probe and apple flesh wall in Figure 5. The produced juice is pumped by pressure, P_o, and this pressure pushes the apple flesh wall to make clearance, c. in Figure 5.

A mechanical model was devised to imitate the phenomenon as depicted in Figure 6. The model consisted of a piston and a cylinder which represent probe and apple flesh wall respectively. A sliding block and a spring represents apple flesh wall pushed by pressure P₀. The cylinder is filled with apple juice entrapped air. As the piston moves forward, pressure, P₀, is produced which pushes the fluid through the clearance created by the sliding block moving backward compressing the spring with a force F. Apple juice flows at a high velocity through the clearance.

From the model displaced liquid inside the cylinder discharged through the clearance.

The steady flow-mass-conservation relation can be written as

If the juice is assumed as an incompressible liquid, then the inlet and the outlet volume should be the same so that eq. [10] can be replaced with

$$\dot{Vol}_{in} = \dot{Vol}_{out}$$
 [11]



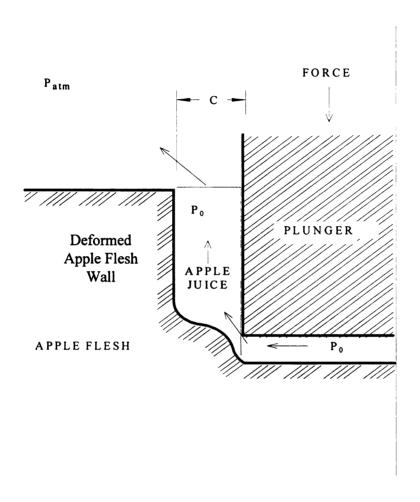


FIGURE 5 - Annular pumping effect during mastication



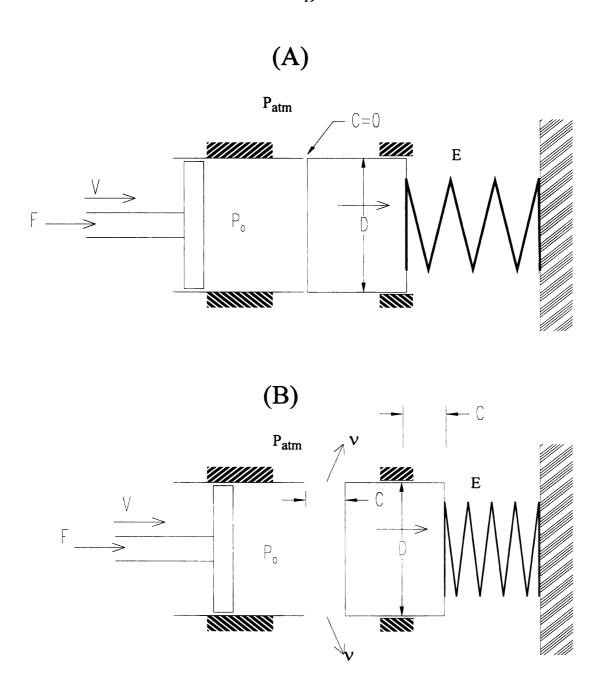


FIGURE 6 - Model representing annular effect

If the piston moves with a constant velocity, the speed of the juice flowing through

the clearance can be calculated as follows.

$$v = V \frac{A_m}{A_c} = \frac{VD}{4c}$$
 [12]

where v = Velocity of exiting apple juice

V = Velocity of the moving piston

c = Clearance

 $A_m = Cross$ sectional area of the piston

 $A_c = Area of exit (= c\pi D)$

D = Diameter of the piston

Bernoulli's equation is used to determine pressure inside the cylinder.

$$P_{o}(t) - P_{a} = \frac{1}{2}\rho v^{2}$$
 [13]

where $P_o(t)$ = Pressure inside the cylinder

P_a = Atmospheric pressure

 ρ = Fluid density inside the cylinder

υ = Exiting speed of the liquid

The pressure or the stress acting on the moving block is,

$$\mathbf{P}_0 = \mathbf{\sigma} = \mathbf{\varepsilon} \mathbf{E} = \frac{\mathbf{c}}{\ell} \mathbf{E} \tag{14}$$

where ℓ is the original length

The clearance, c, can be expressed as following,

$$c = \frac{\ell P_0}{E}$$
 [15]

Using eq. [12], [13], and [15] following equation is obtained.

$$P_{o}(t)^{3} - P_{a}P_{o}^{2} - \frac{\rho V^{2}D^{2}E^{2}}{32} = 0$$
 [16]

Eq. [16] can be used to calculate P_{o} .

3.5 Postulated Deformation Sequence

In compression test the displacement of a layer is equal to the total displacement divided by the total number of layers because each layer is free to move. Therefore, the displacement of each layer is all the same. For the Magness-Taylor test, however, total displacement cannot be divided equally due to the restraint of each layer. Shear-tensile resistance plays a role of those restraints. Thus, the distribution of the deformation of the entire sample should be considered.

In this study it is assumed that the distribution of deformation of each layer follows the geometric ratio. Therefore, the total displacement is the sum of the deformation of each layer. It is expressed as,

$$D(t) = d_1(t) + d_2(t) + \dots + d_n(t)$$
 [17]

where D(t) = Total deformation at time tand $d_n(t) = Displacement of the n th layer at time t$

Because the deformation of each layer follows geometric series, eq. [17] can be represented as follows,

$$D(t) = a_1 t + a_2 t + a_3 t + \dots + a_n t$$

= a,t + a,rt + a,r²t + \dots + a,rⁿ⁻¹t

where $a_n = Deformation rate of n th layer$ t = Timer = Geometric ratio(0 < r < 1)

If D(t) is constant strain rate then eq. [18] is

$$D(t) = Vt = (a_1 + a_1r + a_1r^2 + \dots + a_1r^{n-1})t$$
 [19]

where V = Speed of the cross-head

From eq. [19], a1 is expressed as



$$\mathbf{a}_1 = \left(\frac{1-\mathbf{r}}{1-\mathbf{r}^n}\right)\mathbf{V} \tag{20}$$

By using a_1 and r the deformation rate of each layer can be determined. In addition, deformation history with time should be considered. Figure 7 shows the deformation sequence based on the mastication phenomenon. The slope of the line OG represents the speed of the cross head. Each layer is assumed to be deformed linearly according to its deformation rate until the deformation gets to d_c which is the critical deformation for the mechanical system failure of a layer. During the period, from t_1 to t_2 , deformation rate of the first layer is increased rapidly due to collapse of mechanical structure until the deformation of the first layer reaches d_t which is the deformation of a layer showing no more compression. At the same period (t_1-t_2) the other layers experience rapid relaxation and they recover some amount of deformation to compensate the rapid compression of the first layer. The period (t_1-t_2) can be considered as a non-equilibrium state. However, at time t_2 system turns to an equilibrium state. In the practical point of view, time span, t_1 and t_2 , can be neglected because the phenomenon of this period is so rapid.

After t₂ the total number of layers is reduced by one because the first layer has already been destroyed. Therefore, the deformation rate of each layer should be determined again.

At t₃ the deformation of the second layer reaches d_c and it is deformed to d_t rapidly until the time gets to t₄. The other layers get into rapid relaxation.

This discrete pattern of deformation occurs throughout the destruction phenomenon of apple flesh in the ideal condition. However this pattern of deformation is found three to four times in the real life situation.

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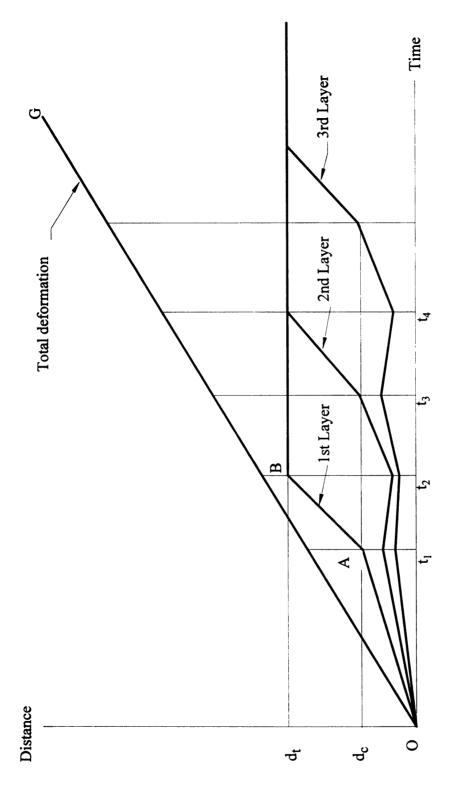


FIGURE 7 - Postulated deformation sequence of Magness-Taylor puncture test

3.6 Compression, Shear-tensile and Annular Pumping Component allied Model

Models of each component were combined to generate the combined force deformation response. Figure 8 shows the allied model composed of the three models. A in Figure 8 represents the compressive component which consisted of a Maxwell model and a fracture element. B represents a shear-tensile component and every component is grounded. C is an annular pumping component. During the early period of compression flow of juice is not fully developed to produce resistance. S delays the activation timing of the annular pumping component for a while. Then the postulated deformation sequence is applied to the allied model to get the force response.

Castigliano's theorem is used to calculate the force response of the system. This is the simple method for the deflection problems. Castigliano's theorem is that when forces act on elastic systems subjected to small displacements the displacement corresponding to any force, collinear with the force, is equal to the partial derivative of the total strain energy with respect to that force. Using Castigliano's theorem, energy of a system consisting of non-linear elements is expressed as following.

$$\mathbf{E} = \int_{0}^{\delta} \mathbf{P} \cdot \mathbf{d}\delta = \sum_{i=1}^{n} \int_{0}^{\mathbf{e}_{i}} \mathbf{F}_{i} \cdot \mathbf{d}\mathbf{e}_{i}$$
 [21]

where

P = Resultant force

 $d\delta$ = Differential total deformation

 F_i = Force of i th element

 e_i = Differential deformation of i th element

i = Layer index

n = Total number of layers

E = Energy

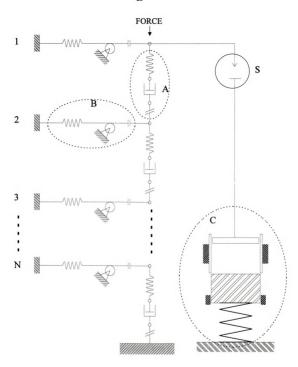


FIGURE 8 - Final model of apple flesh for Magness-Taylor puncture test

Each cell layer is elastic until it gets to failure and the each layer is subjected to very small deflection. Thus the Castigliano's theorem has no problem in the application for this approach. Because the deformation is already known the force response of each layer can be determined. Therefore the strain energy which is the force times of deformation can be calculated for each layer at a certain time

The total strain energy calculated by the theorem is differentiated numerically with total deformation to obtain the resultant force response which is the predicted Magness-Taylor test result.

3.7 Parameter Determination Using Rubber "Thumb Test"

Stresses occur at the contact area when two bodies having curved surfaces are pressed together. Hertz contact stress theory has been developed to describe the above situation. Figure 9 shows two spheres contacting each other. By Hertz contact theory the radius of contact area is

$$\mathbf{a} = \sqrt[3]{\frac{3\mathbf{F}}{8} \cdot \frac{(1 - v_1^2) / \mathbf{E}_1 + (1 - v_2^2) / \mathbf{E}_2}{1 / d_1 + 1 / d_2}}$$
 [22]

where d_1, d_2 = Diameters of body 1 and 2

F = Force applied

 v_1 , v_2 = Poisson ratio of body 1 and 2

 E_1 , E_2 = Modulus of elasticity of body 1 and 2

Radius of contact area

The maximum normal stress at the center of the contact area is

$$\sigma_{z, \max} = P_{\max} = \frac{3F}{2pa^2}$$
 [23]

Stresses in direction of x and y can be represented as functions of z.

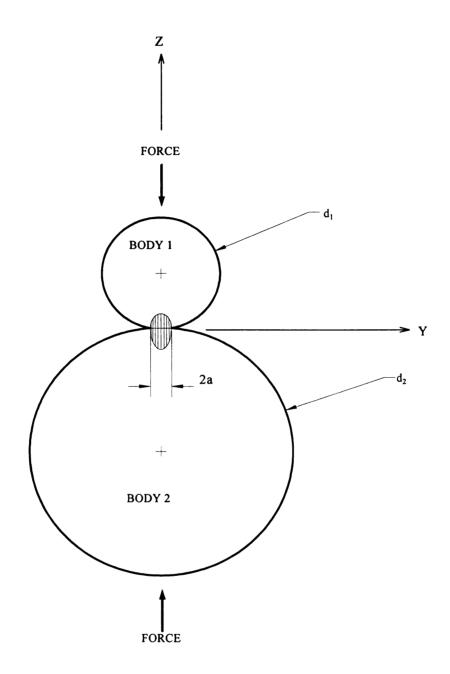


FIGURE 9 - Hertz contact stress of two spheres

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$$\sigma_{x} = \sigma_{y} = -P_{\text{max}} \left[\left(1 - \frac{z}{a} \tan^{-1} \frac{1}{z} \right) (1+m) - \frac{1}{2\left(1 + \frac{z^{2}}{a^{2}}\right)} \right]$$
 [24]

$$\sigma_{\mathbf{z}} = \frac{-P_{\text{max}}}{1 + \frac{z^2}{a^2}} \tag{25}$$

where
$$m = \frac{3K - E}{6K}$$

K = Bulk modulus or incompressibility

And the shear stress can be calculated by using σ_x and σ_z as

$$\tau_{xy} = \frac{\sigma_x - \sigma_z}{2} \tag{26}$$

where τ_{xy} = Shear stress

Figure 10 shows the stress change for the distance below the surface *i.e.* along z direction. By using this relationship maximum compressive stress and the shear stress can be calculated. The maximum shear stress is approximately 1/3 of the maximum normal stress.

If body 1 is rubber sphere and the body 2 is apple then by pressing the two objects until the apple goes over the elastic region the yield compressive and shear stress can be predicted.

3.7 Parameter Determination Using Acoustic Impulse Technique

Elasticity of a resonating elastic sphere is defined as

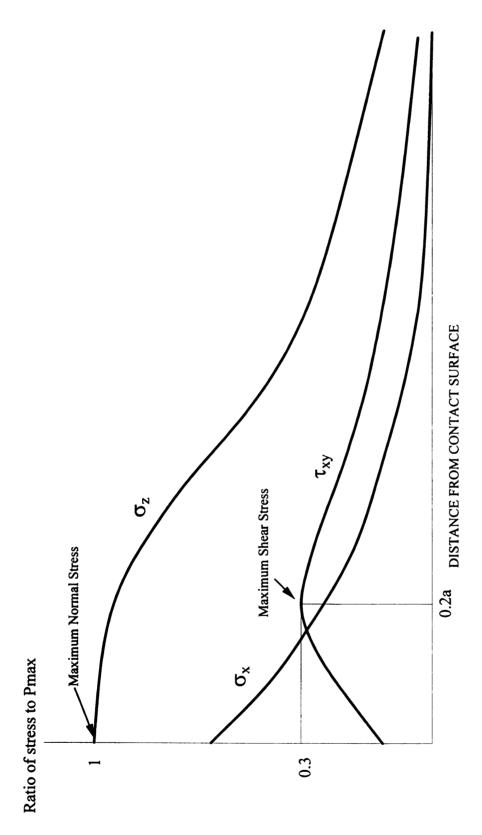


FIGURE 10 - Stress distribution of Hertz contact problem.

$$E = \left[\frac{\rho^{1/3} (6\pi^2)^{2/3} 2(1+\nu)}{\Omega^2} \right] f^2 m^{2/3}$$
 [27]

where $\rho = Density$

v = Poisson's ratio

 Ω = Normalized frequency

E = Modulus of elasticity

m = Mass

f = Frequency

The model includes apple mass, apple density, Poisson ratio, and normalized frequency. There are three different kind of vibrating modes. The compressional mode vibrating in the radial direction was assumed as the most predominate resonant frequency by Armstrong (1989).

4



Chapter 4

EXPERIMENTAL METHODS AND PROCEDURES

4.1 Methods

4.1.1 Compression Test

Cylindrical apple flesh samples 20 mm in diameter and 12 mm in length were prepared for the compression test. Cylindrical specimen was cut from fresh apples by using a borer. The borer was pushed into the direction of center around equator in the radial direction (Figure 11). Then the cylindrical sample was placed on the V-shaped cutting guide (Figure 12) to get the sample having desired length and made ends flat and normal to the direction of compression. A razor blade was used to cut the sample ends.

The sample was placed on platform of the Instron Universal Testing Machine and was pressed with a flat die. The force signal was transmitted to the data acquisition system. Cross-head speed was 0.00423 m/s (10 in/min). Mineral oil was added to both ends of the cylindrical apple flesh specimen to reduce friction between the die and the sample. The sampling rate of the data acquisition system was set to 100 Hz.

Actual force-deformation result was fitted using equation [1] and an optimization program to determine elastic modulus and viscosity of the viscous element in the model. The critical stress, σ_c , for the fracture element is determined by P1 (Figure 1) divided by the area of die. Elastic modulus, viscosity, and the critical stress will be used as parameters of compression model.



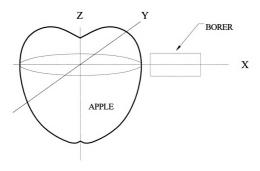


FIGURE 11 - Sampling direction of apple flesh for compression test



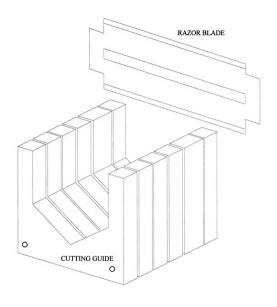


FIGURE 12 - Cutting guide for preparing cylindrical apple flesh

4.1.2 Shear test

Slices of apple flesh were prepared by using a slicer. A shearing device suggested by Mohsenin and Goehlich in 1962 to determine shear strength of apple skin was used for the test (Figure 13). The apparatus was installed on the platform of the Instron Universal Testing Machine. The shear strength will be used as a critical stress of the fracture element of shear-tensile model.

Strain rate was 0.00423 m/s (10 in/min). Force signal from the load cell was recorded by data acquisition system at a sampling rate of 100 Hz.

4.1.3 Tensile Test

Slice of apple was obtained from the cheek of whole apple. Tensile specimen was prepared from the slice of apple flesh (Figure 14). Length, L, and the width, W, were 20 mm and 9 mm, respectively. Thickness of the specimen, t, was measured using calipers. The specimen was installed on the Instron Universal Testing Machine and tensile force was applied until it failed. Tensile elasticity was calculated from the stress-strain relationship. The elasticity will be used as a parameter of shear-tensile model.

4.1.4 Magness-Taylor Puncture Test

An apparatus as shown in Figure 15 was used to make the cutting surface of apple normal to the direction of the force applied. Also this device was used to hold the apple.

The apparatus has cutting blade and guide to remove the skin of apple and to make the surface flat and even. This device was placed on the platform of the Instron Universal

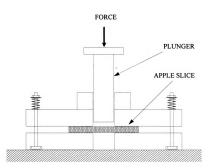


FIGURE 13 - Apparatus for shear test

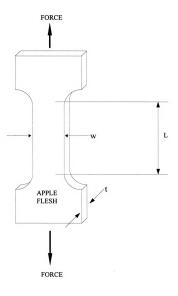


FIGURE 14 - Specimen for tensile test

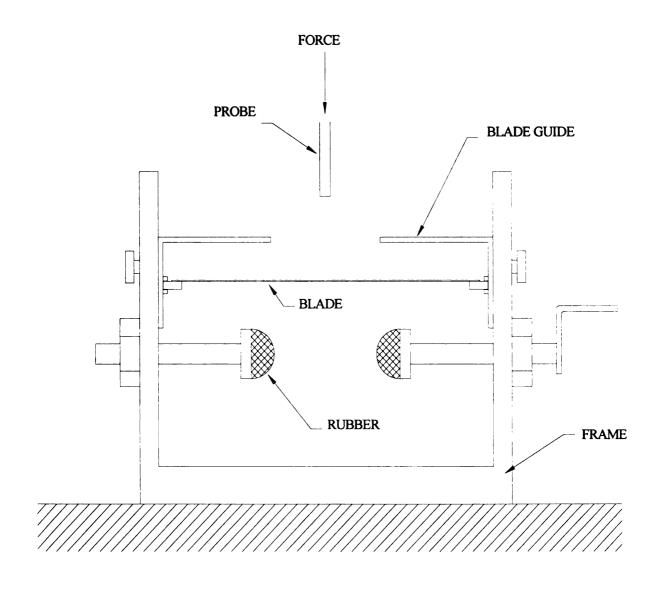


FIGURE 15 - Apple holding apparatus

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Testing Machine. A 11.11 mm (7/16 in) diameter probe with flat end was used to push down into apple flesh. Cross-head speed was 0.0042 m/s (10 in/min).

The probe was pushed to a depth of 15 mm. The location for this test was approximately on the equator as the probe was inserted in the radial direction (Figure 16). The probe was lubricated with mineral oil to reduce friction.

4.1.5 Nondestructive Tests

Two nondestructive tests were studied in this research. One is rubber thumb test which is developed in this study to predict the critical yield strengths of compression and shear for the apple flesh. The other one is acoustic method which has been already developed and evaluated by Armstrong in 1989. Through acoustic method elasticity of apple flesh was estimated.

4.1.5.1 Contact Pressure Measurement using Rubber "Thumb Test"

A rubber indentor was used to measure contact stress using Hertz contact stress theory. The rubber diameter of 26.8 mm diameter had an elastic modulus of 3.2 MPa and a Poisson's ratio of 0.62. The indentor was attached to a steel plunger using adhesive (Figure 17). The pressure between the ball and apple was measured using a paper thin Uniforce pressure sensor. This sensor had a range of 0 to 222 N (50 lbf) and a circular sensing area having 6.35 mm (0.25 in) diameter. The location for this test was on the equator of apple. The radius of curvature of this point was measured in two directions. One is in the direction of equator and the other is in the longitudinal direction. The average value of the two measurements was used as the radius of curvature at that point.

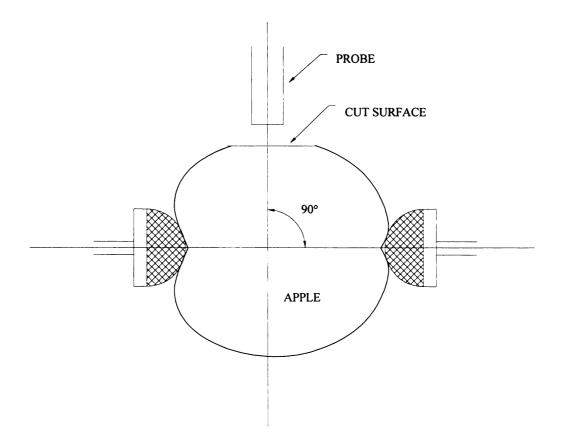


FIGURE 16 - Location of the Magness-Taylor puncture test

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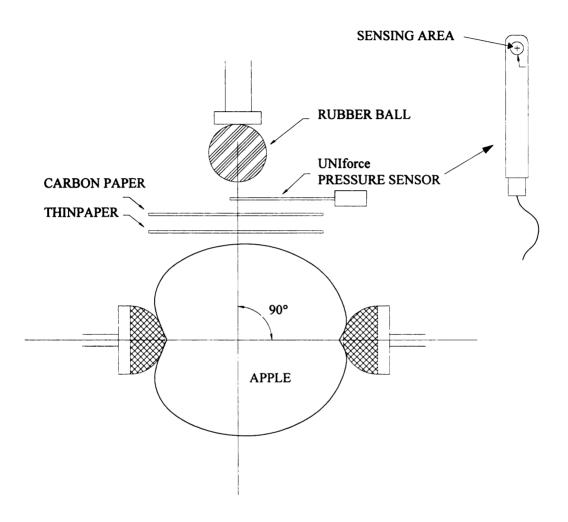


FIGURE 17 - Rubber thumb test

To measure the radii of curvature of convex body, an apparatus composed of a dial indicator and two sharpened rods was used (Figure 18). Radii were calculated as following

Radius =
$$\frac{(AC)^2}{8(BD)} + \frac{(BD)}{2}$$
 [28]

Apple was held by a fixture as shown in Figure 15. Carbon paper and a sheet of paper were used to get the contact area when the first "give" occurred. The term "give" means a change of force response. The probe, pressure sensor, carbon paper, paper and apple were arranged in the order as shown in Figure 17. The probe speed was 0.0021 m/s (5 in/min). The Instron Universal Testing Machine was manually stopped as quickly as possible when the first failure of tissue was detected by hand. The "give" produced vibration enough to be sensed by hand and it usually accompanied an audible sound.

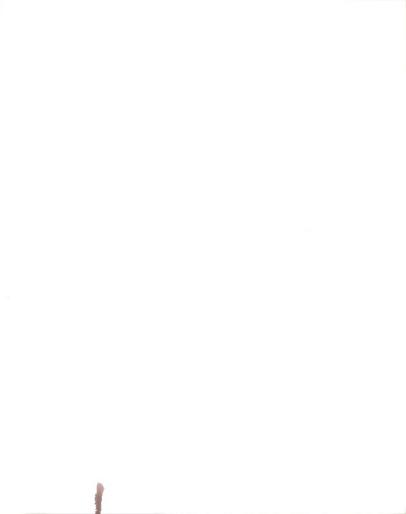
4.1.5.2 Elastic Modulus Measurement using the Acoustic Impulse Response

An apparatus which was designed by Armstrong, et al. (1992) was used to determine the elastic modulus of apple. The apparatus was composed of mass measuring sensor, an actuator to swing an impulse hammer, and a microphone to measure the acoustic signal (Figure 19). The device was connected to a data acquisition system and the signal was converted automatically to the elastic modulus.

Every apple was tested using this device before they were tested by any destructive testing methods. Impulse location was approximately at the equator of the apple.

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4.2 Materials



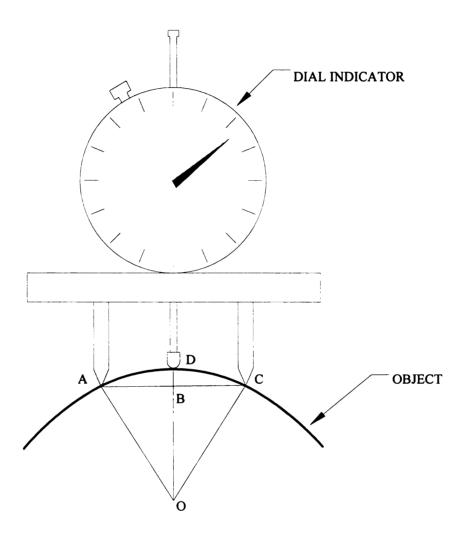


FIGURE 18 - Device for measuring the radius of curvature (Radius of Curvature Meter)

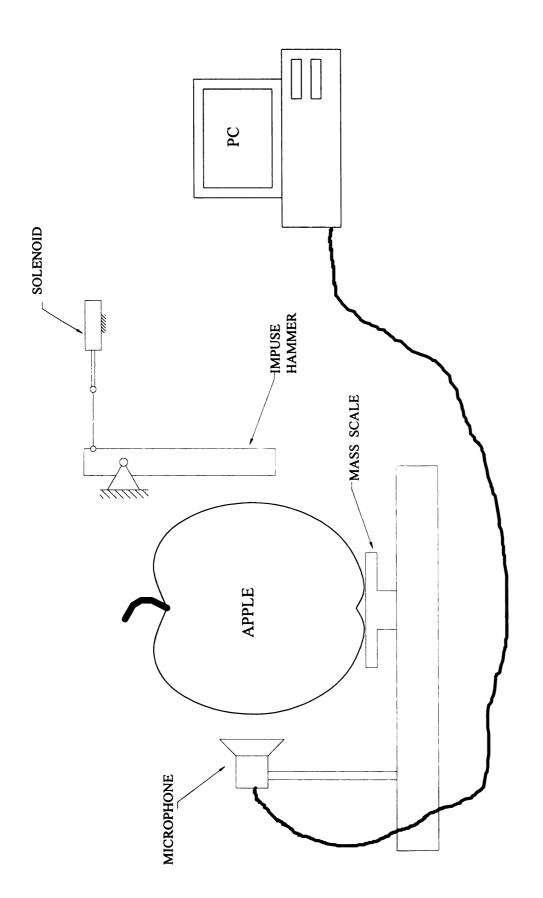


FIGURE 19 - Device for measuring acoustic response of apple.



To have wide range of firmness, nine different cultivars were selected. Cultivars were Red Delicious, Red Rome, Fuji, Granny Smith, Golden Delicious, Braeburn, Ida Red, Gala, and Jona Gold. A total of 45 apples were used for the experiment. Apples were purchased from a local supermarket on the day of experiment and they were stored in a refrigerator at 14° C.

Weight measurement, acoustic impulse method, and "rubber thumb" test were performed on each apple prior to the destructive tests. Magness-Taylor puncture test, compression, shear, and tensile test were carried out after non-destructive tests were conducted. Locations for tests were different from test to test.

4.3 Computer Simulation of Laminated Cellular Model

A computer program for simulation was written in C language. The major parts of the program were the laminated cellular model, deformation sequence generator, compressive and shear-tensile force calculator, and the Castigliano's theorem. The block diagram of the simulation program structure is in Figure 20.

The model parameters of this simulation come from the compression, shear, and tensile test results. In this simulation 100 layer model represented a thickness, approximately 20 mm of apple flesh sample. Geometric ratio, r, was set to 0.997. The speed of testing tip was 0.00423 m/s (10 in/min) which was the same as that of the experiment. During the infinitesimal time, deformation generator produced deformations for every layer and the force of the components of each layer was calculated based on the deformation information. Force was multiplied by infinitesimal deformation to obtain infinitesimal energy and the energy was summed up throughout the simulation. After 5

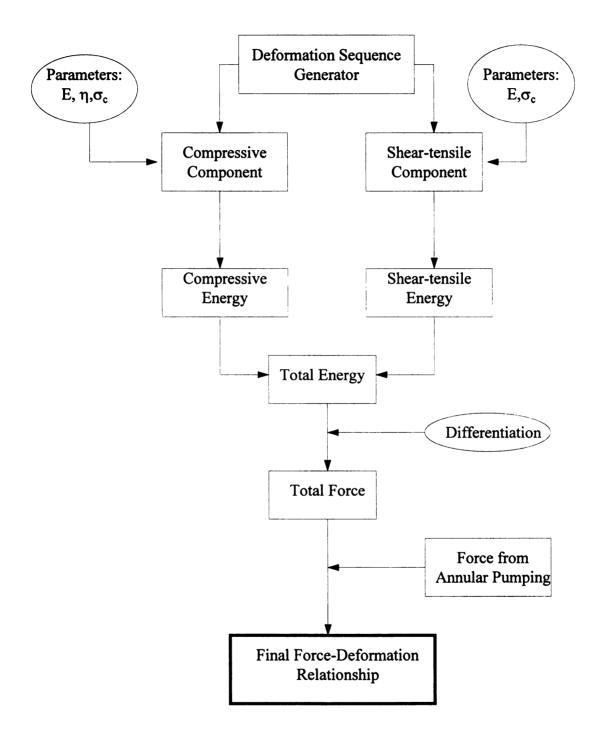


FIGURE 20 - Structure of the simulation program

seconds of simulation, numerical differentiation of energy for the deformation was performed and it produced force-deformation relationship. To the force-deformation result, annular pumping resistance was added gradually after the second peak force.

Chapter 5

RESULTS AND DISCUSSION

5.1 Results

5.1.1 Model Prediction And Magness-Taylor Puncture Test

From the force-deformation relationship of the Magness-Taylor puncture test the first peak force was determined at a point showing significant drop of force. Figure 21 illustrates the result of the comparison between predicted and measured first peak force of Magness-Taylor test. The correlation coefficient was 0.76.

The measured second peak force of Magness-Taylor puncture test was compared with that of model prediction. Figure 22 shows the correlation coefficient of 0.79 which is a little higher than that of the first peak force shown in Figure 21. In some cases, the second peak force could not be observed and those cases were not considered.

Even though the first peak forces are same different stiffness indicates different texture of apple flesh. Figure 23 illustrates that the correlation coefficient was 0.83. Stiffness constant was defined as the slope of the secant drawn from the origin to the first peak of the force-deformation curve. Figure 24 is an example of the simulation result.

Magness-Taylor firmness predicted by the model was compared with that measured by experiment. Figure 25 shows that the correlation coefficient is 0.82.

Even though the juice expelling effect produced constant resistance of 9.67 N which was calculated by using the eq. [16] it took time until this resistance was fully developed.

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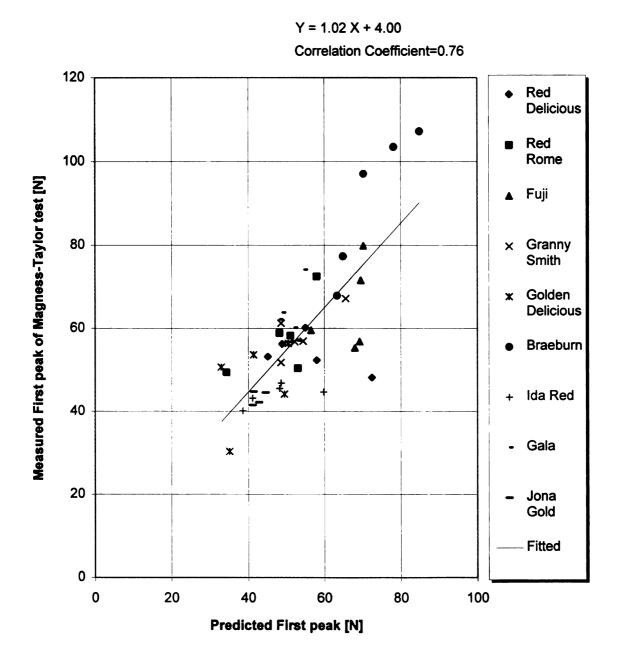


FIGURE 21 - Predicted versus measured first peak force



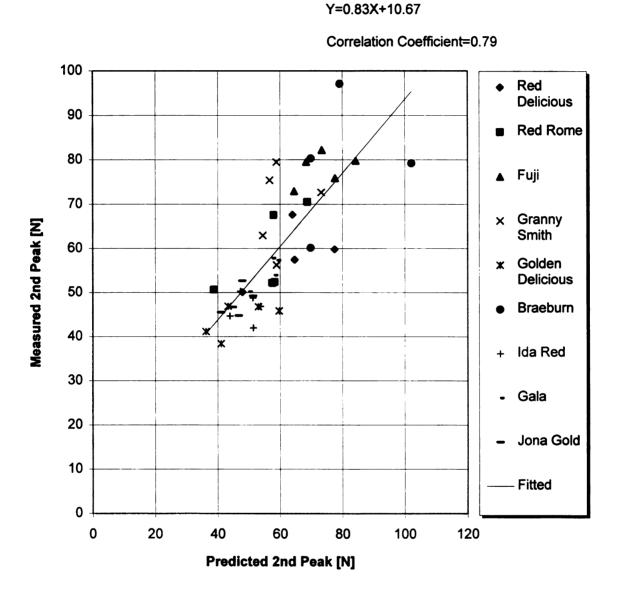


FIGURE 22 - Predicted versus measured second peak force



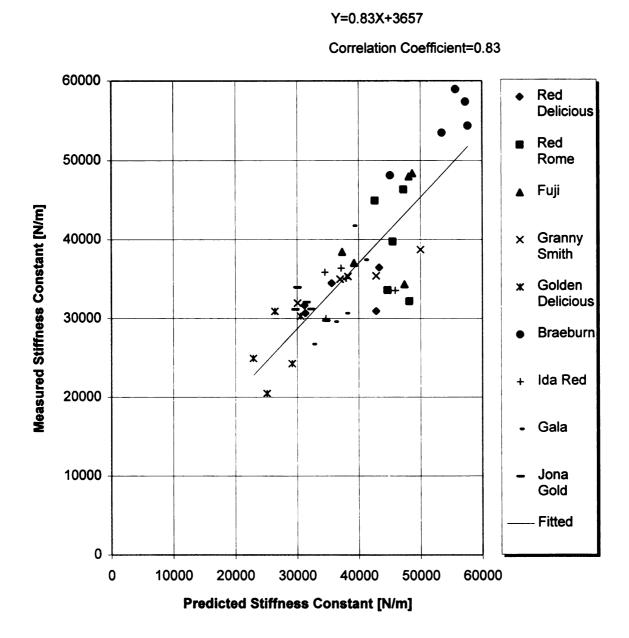


FIGURE 23 - Predicted versus measured stiffness constant

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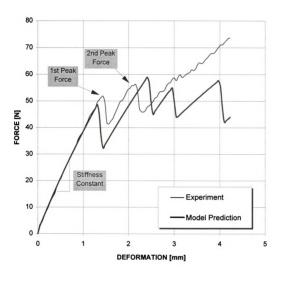


FIGURE 24 - Comparision of model prediction with experiment

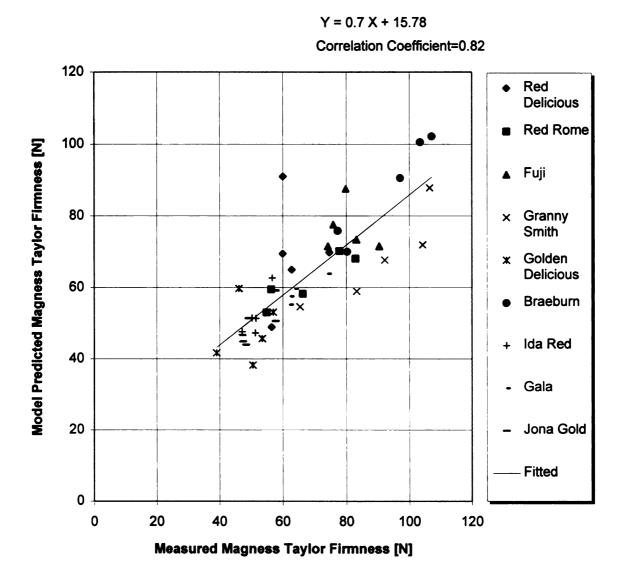


FIGURE 25 - Predicted versus measured Magness Taylor Firmness

The resistance increased linearly after the first peak until it reached a value of 9.67 N and then it was fixed at 9.67 N. Because the viscous element was connected in parallel to the model the resistance was just added to the result.

5.1.2 Performance of Rubber "Thumb Test"

The maximum normal stress at the center of the contact area between rubber ball and the apple was calculated by using eq. [23]. Speed of the rubber thumb test was 0.00212 m/s (5 in/min). Figure 26 shows the correlation coefficient of 0.76.

Figure 27 illustrates the correlation coefficient of 0.64 between measured shear failure stress and the predicted maximum shear stress. According to the Hertz contact stress theory the maximum shear stress occurred below the contact surface and it was approximately 1/3 of the maximum normal stress.

Magness-Taylor firmness was compared directly with the failure force which was observed when the contact pressure dropped significantly. In this research this moment of "give" was detected and the testing machine was stopped at that moment. Figure 28 shows a correlation coefficient of 0.9. In rare cases, no significant drop of the pressure was observed due to tissue failure.

The first peak force showed a much higher correlation coefficient of 0.90 with the failure force than that of the Magness-Taylor firmness (Figure 29). The high correlation coefficient implies that the failure force represents the first peak force.

The relationship between the first peak force and the Magness-Taylor firmness was observed in Figure 30. More than a half of the data is almost identical in the range of error.

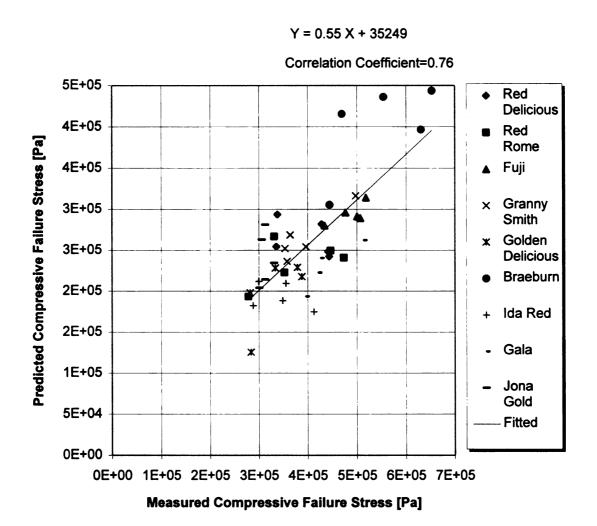
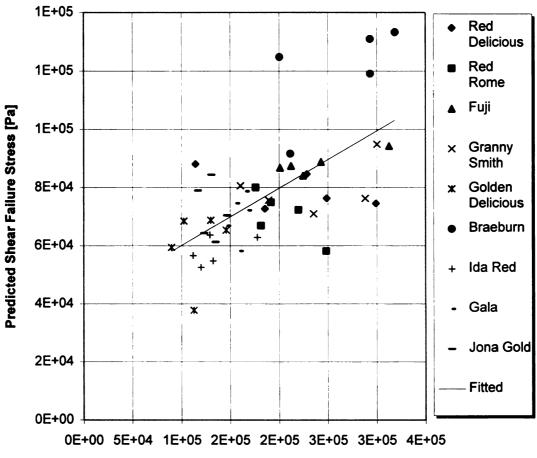


FIGURE 26 - Predicted versus measured compressive failure stress

Y = 0.2 X + 40408

Correlation Coefficient=0.64



Measured Shear Failure Stress [Pa]

FIGURE 27 - Predicted versus measured shear failure stress

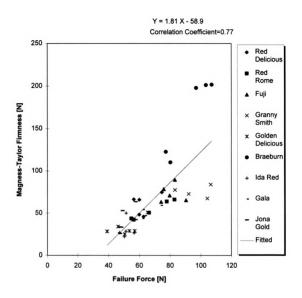


FIGURE 28 - Magness-Taylor firmness versus failure force of rubber thumb test

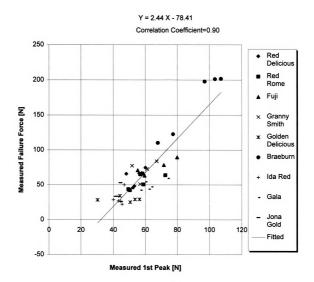


FIGURE 29- First peak force versus Filure Force

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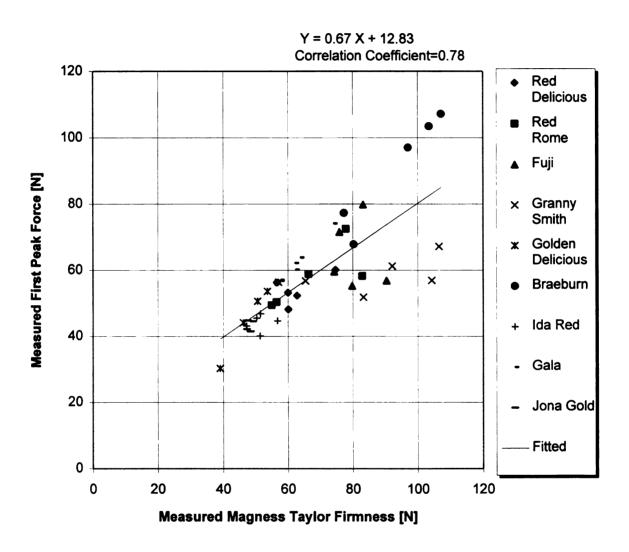


FIGURE 30 - Measured first peak force versus Magness-Taylor Firmness

5.1.3 Performance of Acoustic Impulse Technique

Modulus of elasticity of apple tissue was predicted using eq. [27]. In the compression mode, a Poisson's ratio of 0.3 and the density of 800 kg/m³ were assumed for the resonant sphere model. Measured modulus of elasticity was obtained from the compression test of cylindrical apple flesh. From the compression test results initial tangent modulus of elasticity was found. In the compression test cylindrical apple flesh sample was bored in the radial direction which was different from the sampling direction used by Armstrong (1989). Figure 31 illustrates 0.7 as the correlation coefficient between the measured and predicted modulii of elasticity.

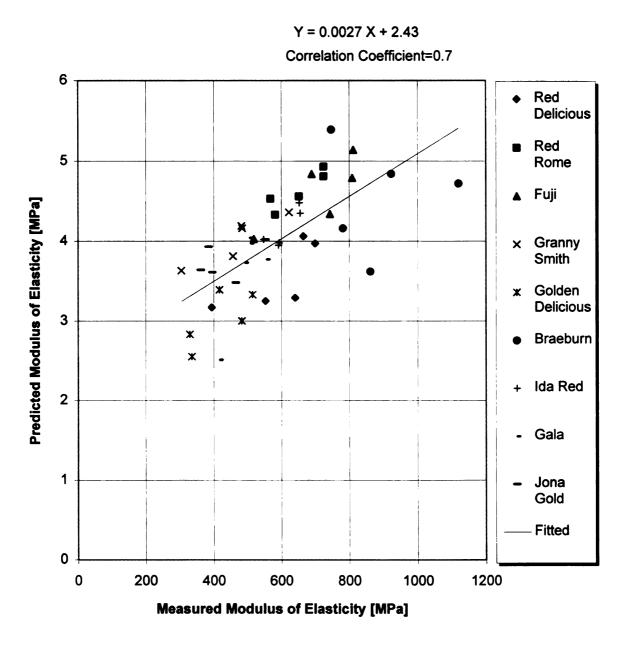


FIGURE 31 - Measured versus predicted modulus of elasticity by using acoustic impulse

5.2 DISCUSSION

5.2.1 Quantitative Aspects of The Model Prediction

The laminated cell layer model showed fairly good correlation for the first peak force and the second peak force prediction. The correlation coefficient of the second peak (0.79) was little higher than that of the first peak force (0.76). Thus, the model demonstrated its prediction capability for the first two peak forces. In addition, Figure 23 showed a strong correlation (0.83) in predicting stiffness constant which meant that the model worked not just for the peak forces but also for the first slope of the force-deformation relationship.

After the second peak it was hard to compare because unidentified disorder occurred as the mastication progressed. However, the information about the early period of the mastication period was enough to determine the texture of apple flesh.

In Figure 32, it was found that the percentages of the sample having less than 20% error with its measured values were 51% for the first peak prediction, 78% for the second peak prediction, 73% for the stiffness prediction, and 84% for the Magness-Taylor firmness prediction. From the results explained above the laminated cell layer model was considered as a good means to depict the Magness-Taylor puncture test quantitatively.

5.2.2 Qualitative Aspects of The Model Prediction

The characteristics of the force-deformation relationship was a saw-tooth shape which inferred the significant change of force or the catastrophic destruction of material. The model represented well this characteristics of the force-deformation relationship. Figure 23 shows the correlation coefficient of 0.83 for predicting the stiffness which indicates

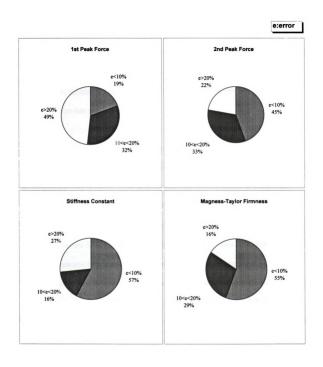


FIGURE 32 - Precision of model prediction

that the model has the capability to describe the different textural quality of two apples having the same first peak force.

Until the second peak the juices expelling effect did not contribute significantly to the total trend of force-deformation relationship because juices were not produced enough to induce resistance and also because the penetration depth was so small.

Therefore, the model was considered to represent the early period of the Magness-Taylor puncture test qualitatively.

5.2.3 Evaluation of Nondestructive Method

Rubber thumb test was considered a good method to provide model parameters such as compressive, shear failure stress nondestructively. Figure 26 and 27 support the possibility of the rubber thumb test. In addition, the method could be used to replace Magness-Taylor puncture test. Figure 28 indicates that there was 0.77 of correlation between the failure force and the Magness-Taylor firmness. This new method was not perfectly nondestructive. Actually it made small bruise on the apple surface but it was not significant. If the diameter of rubber ball was minimized up to some point the size of bruise would be negligible. The first peak force was correlated to the failure force to identify what the failure force in fact measured. Figure 29 shows correlation coefficient of 0.9 which means that the rubber thumb test measures the first peak force efficiently rather than the Magness-Taylor firmness.

The contact pressure was also measured using Uniforce pressure sensor and the theoretations with Magness-Taylor firmness and the failure force were investigated.

Ideally, contact pressure was expected to have good correlation with the failure force but in fact it was not so. The reason for this was the characteristics of the pressure sensor. If the film surface of pressure sensor was bent then it produced undesirable output. In the experiment the contact surface was not flat so that the sensor output interfered with true value. If a proper sensor for this purpose were available then the contact pressure would also be expected to have good correlation with the failure force.

Acoustic impulse technique showed fairly good correlation of predicting the initial modulus of elasticity. This technique was also considered to have the potential for providing model parameters.

CONCLUSIONS

Considering the results of this research following conclusions are drawn:

- 1. The laminated cell layer model developed in this research was considered to have reasonable capability for depicting the Magness-Taylor puncture test results quantitatively and also qualitatively. Correlation coefficients for predicting the first peak was 0.76 and for the second peak it was 0.79. Also, the model predicted well the stiffness constant with correlation coefficient of 0.83.
- 2. Rubber thumb test may be used to estimate model parameters. In this research this method showed 0.76 and 0.64 of correlation for predicting the compressive and shear failure stress, respectively. Moreover, rubber thumb test was found to have strong correlation of 0.9, with Magness-Taylor firmness.
- 3. Acoustic impulse technique may be used to estimate model parameters. This technique showed correlation coefficient of 0.7 with the measured initial modulus of elasticity.



SUGGESTIONS FOR FUTURE RESEARCH

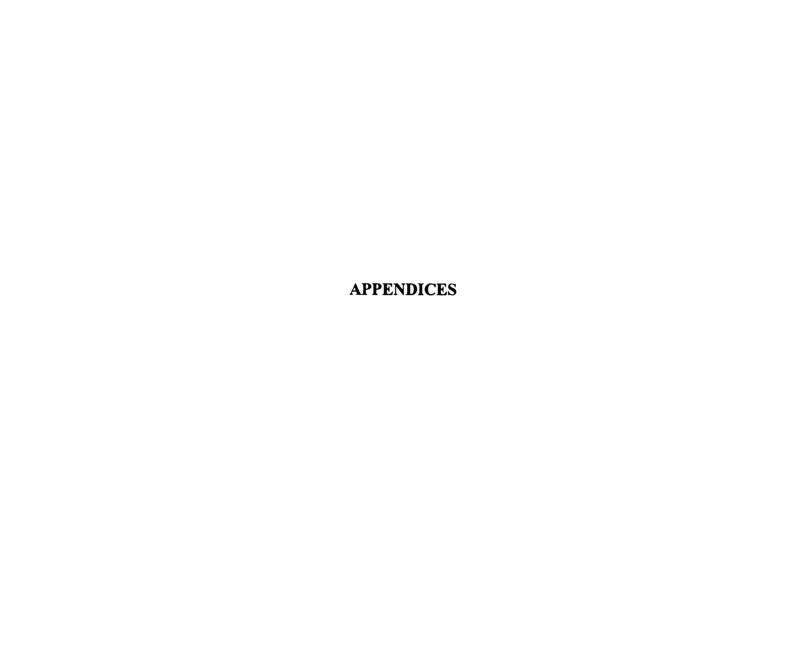
For future research the following ideas are suggested:

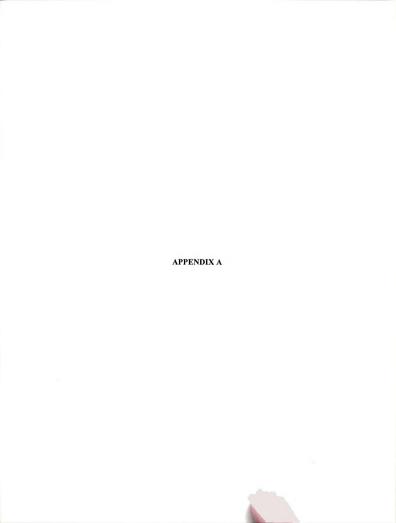
- 1. In preparing the cylindrical sample for compression test it is very critical to ensure that the sample have two flat ends normal to the direction of the compression force. If those ends are not normal to the compression force the shear failure comes first rather than the collapse of the layer. It is also difficult to obtain correct mechanical properties from the compression test if it is not flat.
- 2. There are several supplemental constants such as number of layers, geometric ratio, ratio of stress recovery, ratio of increase of contact area and the initial length for calculating the strain of shear-tensile component. Effects of these constants are worthy of future study for better performance of the laminated cell layer model.
- 3. Bruise can be minimized by reducing the size of the rubber ball. The effect of size reduction for the performance of the rubber thumb technique can be studied.
- 4. As the rubber ball was pressed onto the apple, sudden change of contact pressure occurred which produced audible sound and vibration. By incorporating accelerometer on the surface of apple this vibration can be sensed and testing machine

can be stopped. Thus, it is possible to measure the contact diameter at the moment of flesh failure.

 The elasticity and the viscosity obtained by compression test can be studied to find a relationship of these parameters to a textural quality.







APPENDIX A

Table A1 - Magness-Taylor test results

	1st Pea	k Force	2nd Per	k Force	Initia	Slope	Magnes	-Taylor
	Obs.	Predic.	Obs.	Predic.	Obs.	Predic.	Obs.	Predic.
No.	[N]	[N]	[N]	[N]	[N/m]	[N/m]	[N]	[N]
1	56.23	48.9	50.06	47.9	31652	31209	56.49	48.88
2	48.12	72.5	59.80	77.5	34469	35660	60.06	91.01
3	52.29	58.1	57.41	64.7	30906	42853	62.80	64.94
4	53.16	45.1	•	54.4	30655	31359	59.99	69.44
5	60.12	55.1	67.61	64.0	36445	43365	74.70	69.78
6	58.24	51.1	67.56	57.9	33580	44690	82.92	68.06
7	72.50	58.0	70.53	68.8	46324	47261	77.88	70.19
8	50.37	53.1	52.21	57.5	32184	48218	56.41	59.49
9	58.85	48.2	52,41	58.2	39748	45549	66.33	58.22
10	49.40	34.3	50.72	38.7	44921	42656	54.92	53.03
11	55.28	68.0	79.83	84.3	48400	48697	79.83	87.60
12	59.56	56.5	72.96	64.5	37055	39259	74.27	71.54
13	71.59	69.6	75.93	77.6	38467	37351	75.93	77.64
14	56.85	69.3	79.57	68.5	48001	48132	90.47	71.56
15	79.89	70.3	82.25	73.4	34339	47429	83.17	73.44
16	67.15	65.6	72.67	73.3	38721	50022	106.54	87.84
17	51.77	48.6	56.23	58.9	34967	37026	83.28	58.91
18	61.22	48.6	75.40	56.7	35300	38305	92.21	67.66
19	56.75	52.2	62.92	54.5	31940	30101	65.41	54.57
20	56.90	54.4	79.48	58.8	35399	42868	104.30	71.92
21	53.59	41.4	46.90	43.3	30903	26434	53.59	45.64
22	44.12	49.5	45.82	59.7	24255	29226	46.22	59.72
23	56.35	50.5	46.77	53.1	30277	30607	57.01	53.08
24	50.60	33.0	41.14	36.2	24920	22906	50.60	38.22
25	30.28	35.1	38.42	41.1	20454	25142	39.08	41.65
26	77.35	65.0	60.15	69.9	48120	45126	77.35	75.83
27	107.21	84.9	79.25	102.1	58944	55716	107.21	102.15
28	67.86	63.4	80.33	70.0	53474	53498	80.33	69.97
29	103.50	78.2	•	89.7	54374	57713	103.50	100.55
30	97.10	70.3	97.10	79.2	57389	57289	97.10	90.58
31	44.64	59.8	46.87	53.7	36392	37183	56.72	62.68
32	45.51	48.2	41.97	51.4	35866	34528	50.37	51.42
33	43.13	41.1	44.70	43.9	29986	34706	47.20	47.58
34	46.81	48.6	48.78	51.3	33537	45952	51.54	51.32
35	40.07	38.6	50.31	47.3	35085	38008	51.36	47.27
36	63.82	48.9	50.17	50.0	29585	36127	63.82	59.65
37	62.18	48.3	*	55.2	26725	32570	62.18	55.15
38	57.06	53.0	57.32	59.1	30655	37948	57.71	59.17
39	74.15	54.7	53.93	58.1	41736	39166	74.15	63.87
40	60.16	52.1	57.80	57.5	37426	41032	62.39	57.52
41	44.52	44.6	49.25	51.4	33953	30081	49.25	51.37
42	41.48	41.2	45.55	41.0	29715	34761	48.44	43.92
43	44.77	41.4	46.74	44.9	32075	31555	47.53	44.87
44	42.17	42.8	44.80	46.7	31155	29758	47.29	46.68
45	56.75	50.6	52.68	47.9	31198	32318	57.80	50.62

^{*} unobservable

Table A2 - Results of compression, shear, and tensile test

	Compression Test			Shear Test	Tensile Test	
	E	η	σς	σς	σ_{c}	E _t
No.	×108 [Pa]	×108 [Pa s]	[Pa]	[Pa]	[Pa]	[Pa]
1	3.94	2.65	338438	114515		2219400
2	6.41	1.83	441721	299244		2893126
3	7.00	2.42	443736	185405		2661103
4	5.53	1.61	335967	248737		2019565
5	6.65	2.7	428455	228580		2603817
6	7.25	3.07	446245	191686		2186119
7	7.25	3.22	473311	219752		2644603
8	6.52	2.09	352807	181504		3603414
9	5.68	2.53	331709	176118		3262707
10	5.82	2.61	278642	248325		2382488
11	8.09	2.75	518319	312775		3038228
12	5.19	4.29	434385	224689		2365463
13	6.90	2.09	476922	242944		2669880
14	8.12	2.56	506877	201034		3140829
15	7.44	2.64		212239		3262553
			500491			
16 17	6.23	5.17 2.76	497526	300421		3204485
	4.82		353833	189325		2495790
18	4.57	4.02	358053	235456		2556493
19	3.05	19.3	364249	160527		2106323
20	4.84	6.18	396447	288385		2878381
21	4.17	2.17	334104	102475		1488604
22	4.83	2.21	388122	145849		1718797
23	5.15	2.01	379075	130190		1910229
24	3.30	3.05	282443	89790		111363
25	3.36	3.01	284230	112771		1423046
26	8.62	1.88	444801	211617		3227850
27	9.24	3.35	631107	293539		3545828
28	7.48	3.31	469624	200621		3437427
29	11.20	3.2	653307	318823		303402
30	7.82	4.76	554547	293341		3415703
31	6.56	1.46	355620	177696		3085208
32	5.48	2.93	412565	120155		1748530
33	5.94	1.48	300310	129266		2268548
34	6.53	1.67	348930	111836		3514630
35	5.92	1.62	288335	132484		243824
36	4.92	7.24	437958	155948		168708
37	4.17	6.04	396105	159658		1794609
38	5.09	4.36	421764	146990		2176547
39	5.06	256	513378	165813		1578980
40	5.56	4.68	426326	168136		2231997
41	3.62	4.42	331139	146777		1916969
42	4.65	2.38	306164	116952		2252100
43	3.96	2.69	300994	134993		211610
44	5.53	1.44	313349	123545		1938822
45	3.85	2.09	313121	130808		2625809

Table A3 - Rubber Thumb Test Results

	Failure Force	Failure Pressure	Contact Diameter
No.	[N]	[Pa]	[mm]
1	66.3	513727	12
2	65.8	513727	13
3	46.0	405574	11
4	48.3	459651	11
5	74.8	567804	13
6	66.2	513727	13
7	63.9	540765	13
8	42.3	405574	11
9	50.7	486689	11
10	43.8	405574	12
11	71.0	999740	12
12	63.4	896319	12
13	78.5	965266	13
14	65.4	965266	12
15	89.6	965266	14
16	83.9	999740	13
17	77.5	1034214	14
18	72.7		14
19	51.0	758424 792897	11
20	67.4		13
21	29.0	965266	9
		344738	
22	34.2	379212	10
23	29.1	448159	9
24	25.2	379212	9
25	28.4	344738	12
26	122.7	720816	16
27	201.8	655002	18
28	110.3	586055	13
29	201.3	620528	17
30	197.9	689476	17
31	26.6	344738	9
32	22.2	344738	9
33	27.0	344738	9
34	50.0	379212	13
35	28.6	413686	10
36	47.2	413686	11
37	43.8	344738	12
38	42.3	344738	11
39	59.3	448159	12
40	54.3	379212	12
41	53.0	379212	12
42	33.4	344738	9
43	25.9	306434	9
44	33.6	379212	10
45	63.5	379212	12

Table A4 - Elasticity measured by acoustic impulse technique

	Elasticity
	E
No.	[MPa]
1	3.17
2	3.29
3	3.97
4	3.25
5	4.06
6 7	4.81 4.93
8	4.56
9	4.53
10	4.33
11	4.79
12	4.03
13	4.84
14	5.14
15 16	4.34 4.36
17	4.19
18	3.81
19	3.63
20	4.16
21	3.39
22	3
23	3.33
24 25	2.83 2.55
26	3.62
27	4.84
28	5.39
29	4.72
30	4.16
31	4.35
32	4.02
33 34	3.98
35	4.48 3.95
36	3.73
37	2.51
38	3.97
39	4.04
40	3.77
41	3.64
42	3.48
43 44	3.61 4.02
44 45	3.93



APPENDIX B

Force-Deformation Plots of Magness-Taylor Puncture Test

General Information:

- 1. Material: 5 apples from 9 cultivars (total=45 apples)
- 2. Explanation of file name: e.g. Y01 02

1st character (Y): No meaning 1st two digits (01): Sample No. 2nd two digits (02): Testing Code

> 02=Magness-Taylor Test 03=Compression test 04=Shear Test 05=Tensile Test

3. Numbering:

01 - 05: Red Delicious

06 - 10: Red Rome

11 - 15: Fuji

16 - 20: Granny Smith

21 - 25: Golden Delicious

26 - 30: Braeburn

31 - 35: Ida Red

36 - 40: Gala

41 - 45: Jona Gold

- * The above setups were applied to Magness-Taylor, compression, shear, and tensile test.
- * In compression test, Maxwell model was fitted to the points in between A and B of Figure C1.

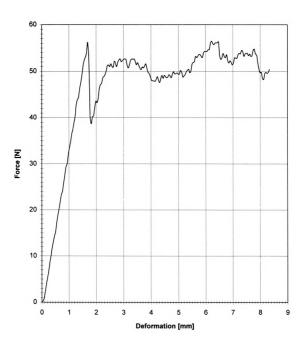


FIGURE B1 - Magness-Taylor puncture test result of $Y01_02$

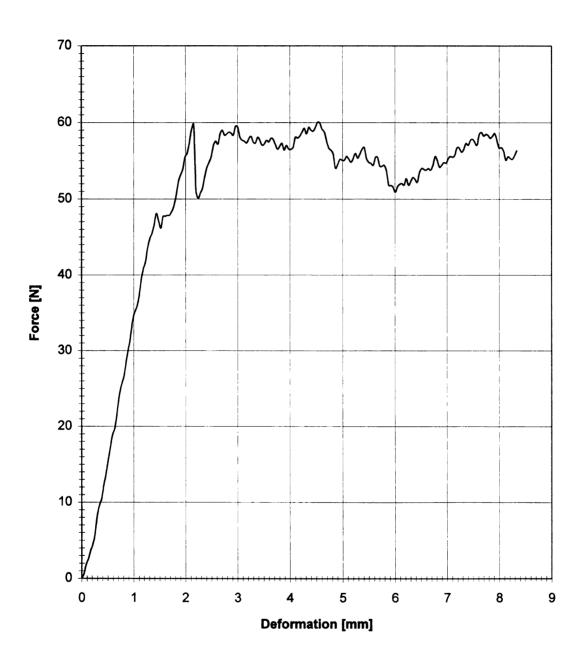


FIGURE B2 - Magness-Taylor puncture test result of $\ensuremath{\,^{\circ}} Y02_02$



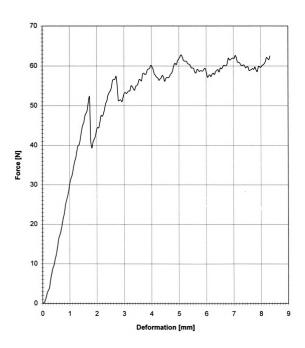


FIGURE B3 - Magness-Taylor puncture test result of $Y03_02$



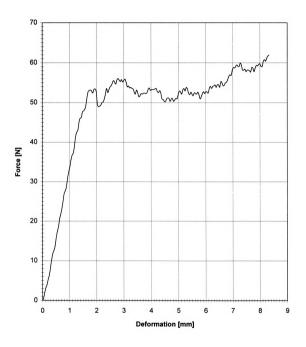


FIGURE B4 - Magness-Taylor puncture test result of Y04_02

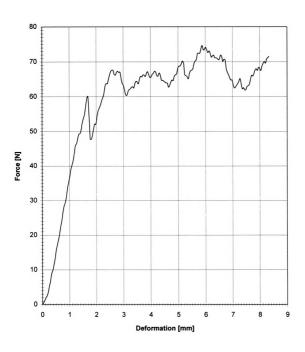


FIGURE B5 - Magness-Taylor puncture test result of Y05_02

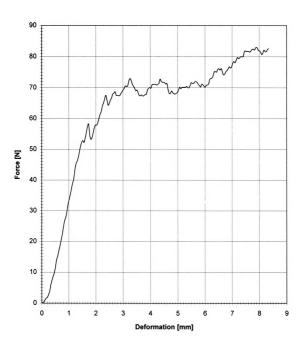


FIGURE B6 - Magness-Taylor puncture test result of Y06_02

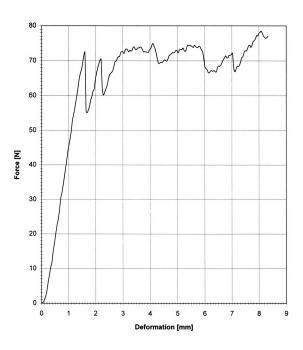


FIGURE B7 - Magness-Taylor puncture test result of Y07_02



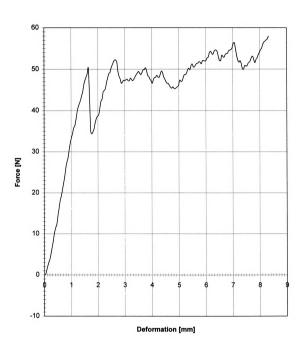


FIGURE B8 - Magness-Taylor puncture test result of Y08_02

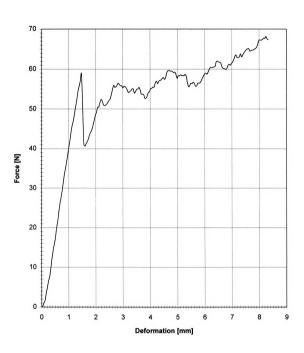


FIGURE B9 - Magness-Taylor puncture test result of Y09_02

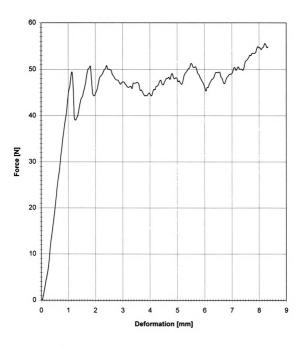


FIGURE B10 - Magness-Taylor puncture test result of $\ Y10_02$

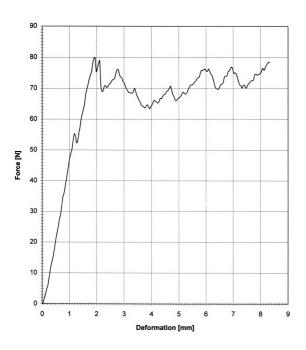


FIGURE B11 - Magness-Taylor puncture test result of Y11_02

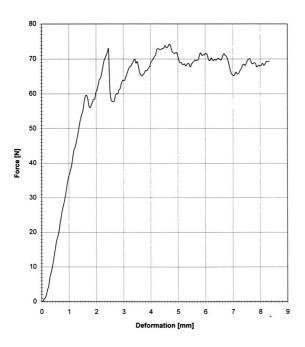


FIGURE B12 - Magness-Taylor puncture test result of Y12_02

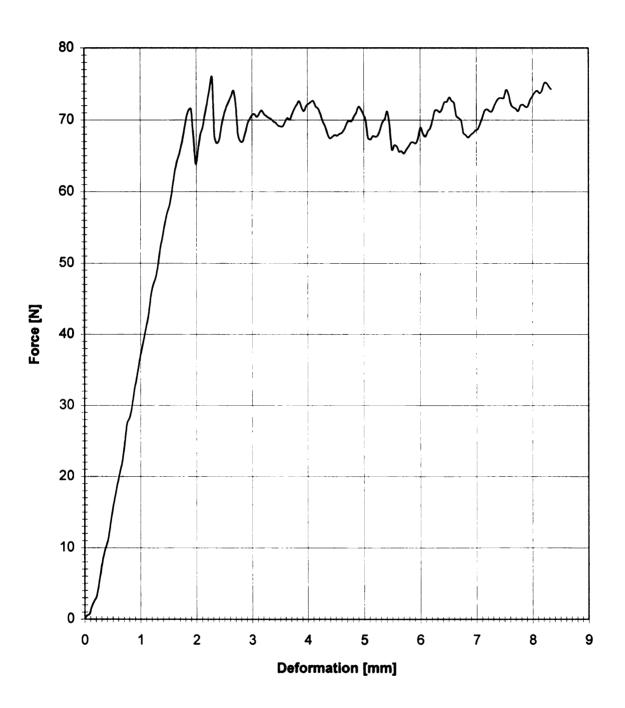


FIGURE B13 - Magness-Taylor puncture test result of Y13_02

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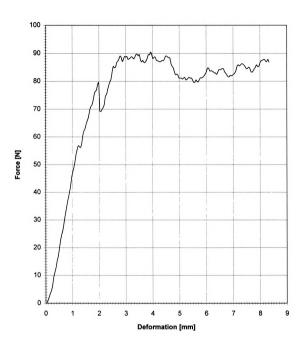


FIGURE B14 - Magness-Taylor puncture test result of Y14_02

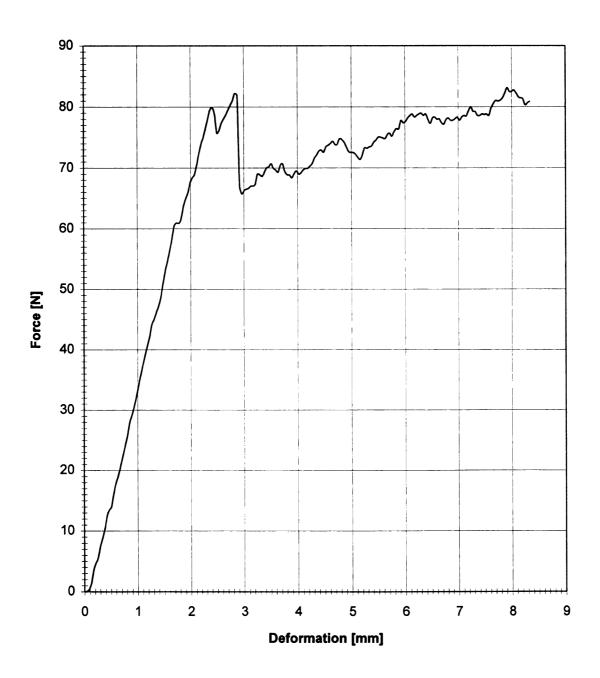


FIGURE B15 - Magness-Taylor puncture test result of $Y15_02$

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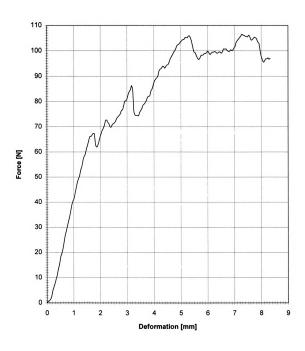


FIGURE B16 - Magness-Taylor puncture test result of Y16_02

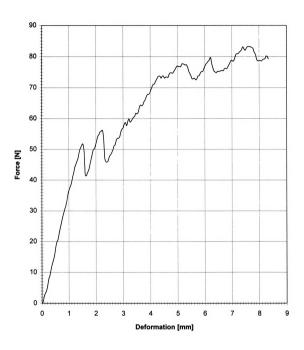


FIGURE B17 - Magness-Taylor puncture test result of Y17_02

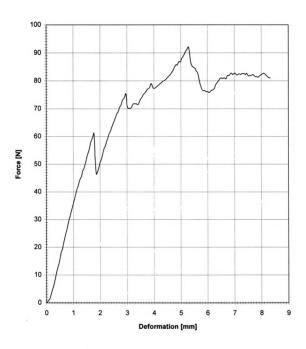


FIGURE B18 - Magness-Taylor puncture test result of Y18_02

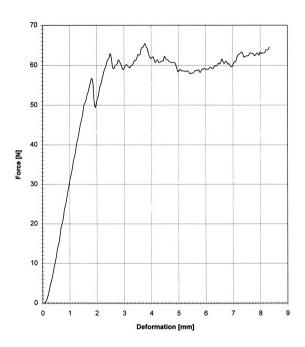


FIGURE B19 - Magness-Taylor puncture test result of Y19_02

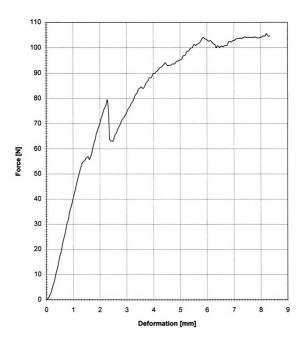


FIGURE B20 - Magness-Taylor puncture test result of Y20_02

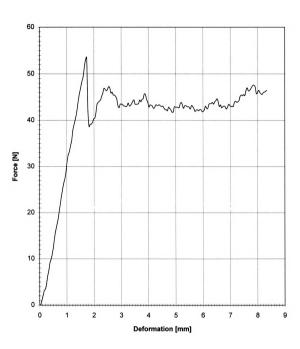


FIGURE B21 - Magness-Taylor puncture test result of Y21_02

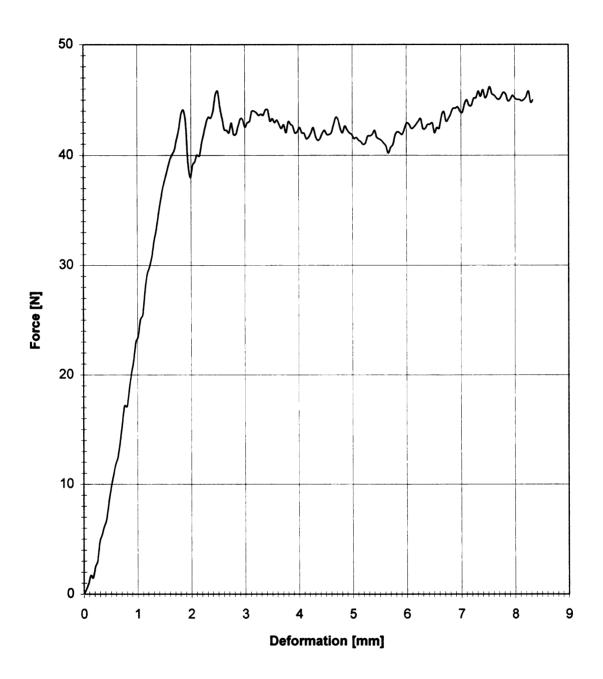


FIGURE B22 - Magness-Taylor puncture test result of Y22_02



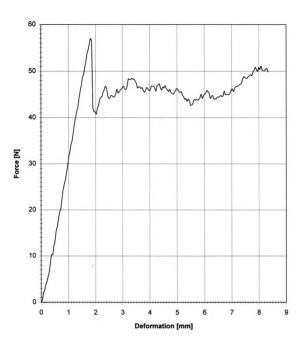


FIGURE B23 - Magness-Taylor puncture test result of Y23_02

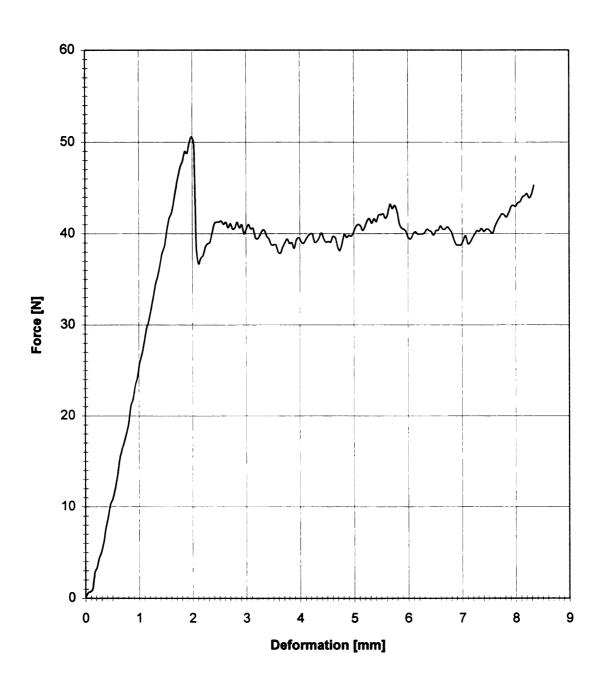


FIGURE B24 - Magness-Taylor puncture test result of $\ Y24_02$

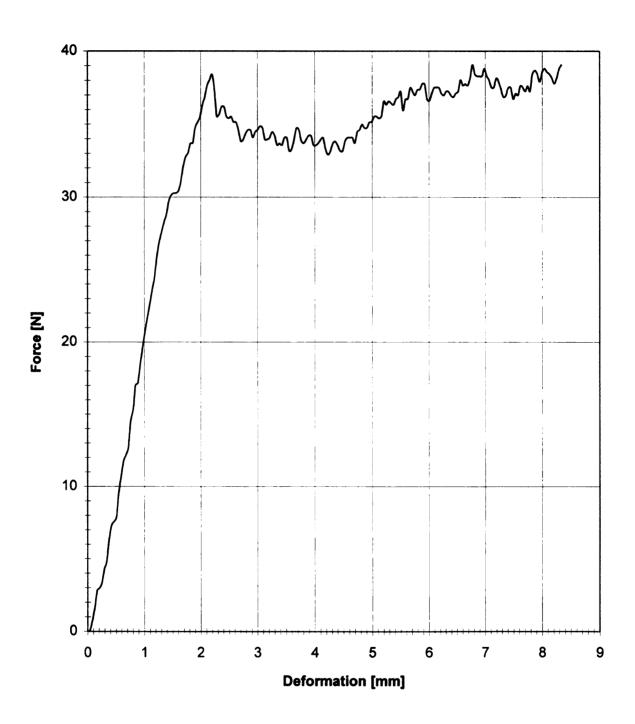


FIGURE B25 - Magness-Taylor puncture test result of Y25_02

A.

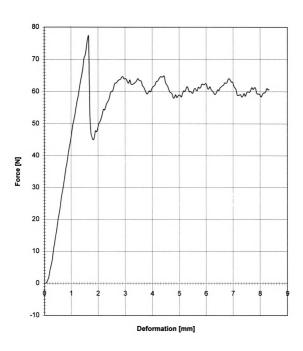


FIGURE B26 - Magness-Taylor puncture test result of Y26_02

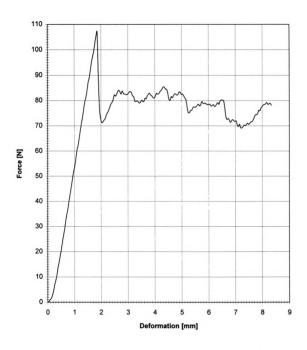


FIGURE B27 - Magness-Taylor puncture test result of Y27_02



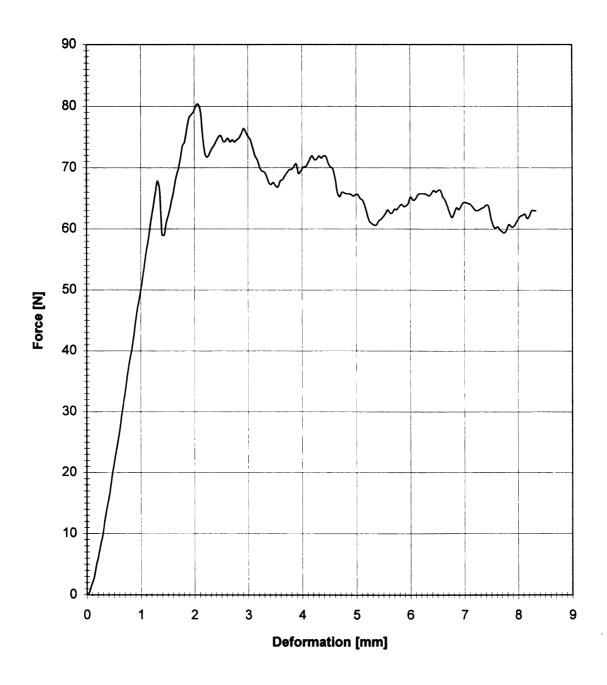


FIGURE B28 - Magness-Taylor puncture test result of $\ensuremath{\,^{\circ}} Y28_02$

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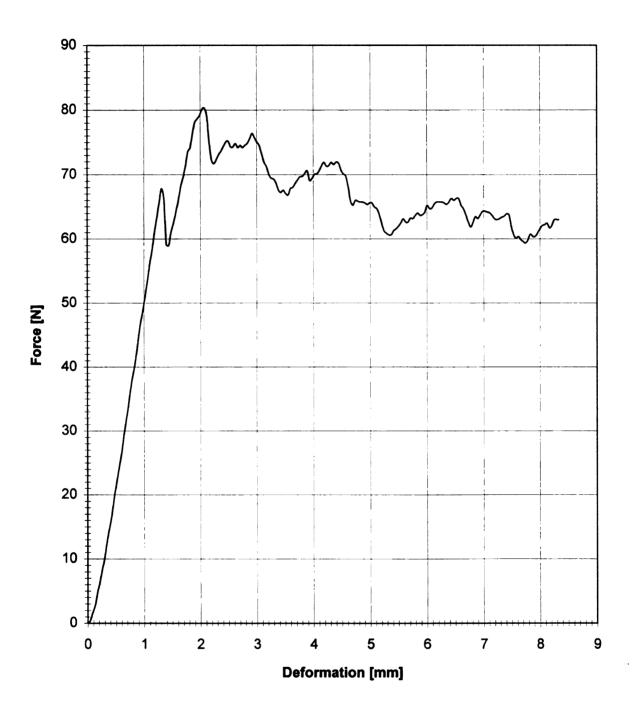


FIGURE B28 - Magness-Taylor puncture test result of Y28_02

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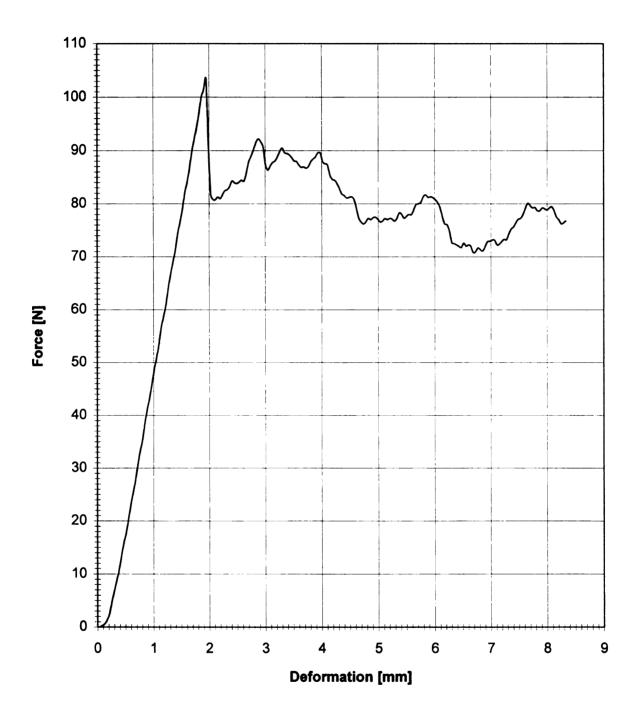


FIGURE B29 - Magness-Taylor puncture test result of Y29_02

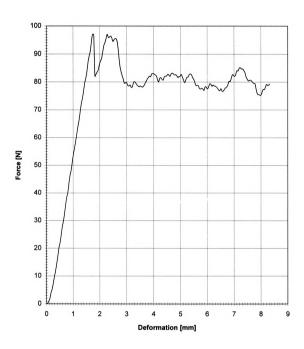


FIGURE B30 - Magness-Taylor puncture test result of Y30_02

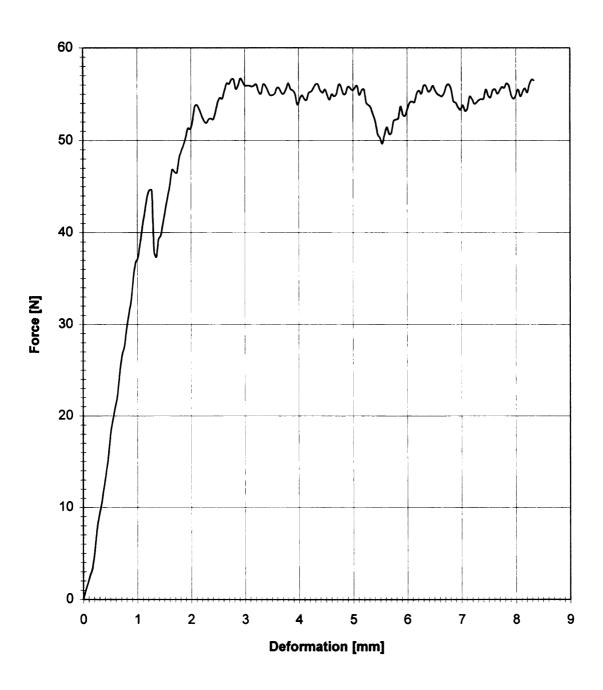


FIGURE B31 - Magness-Taylor puncture test result of Y31_02

<u>*</u>

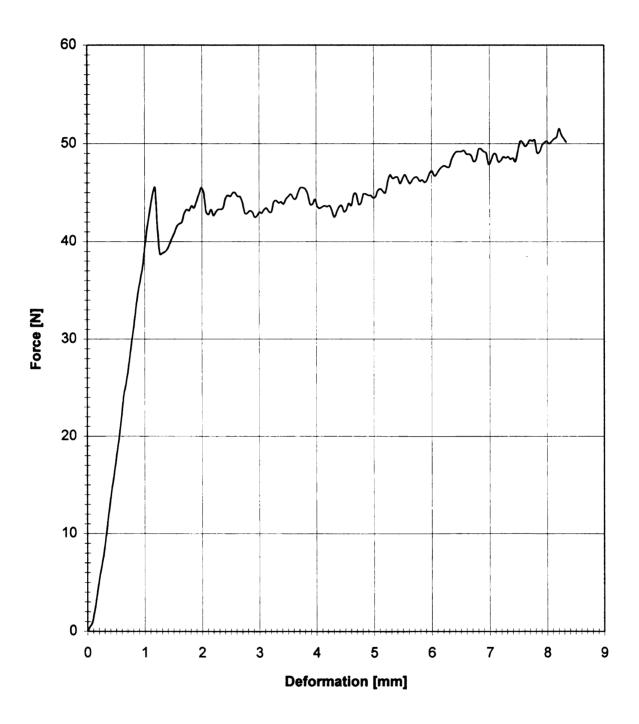


FIGURE B32 - Magness-Taylor puncture test result of Y32_02

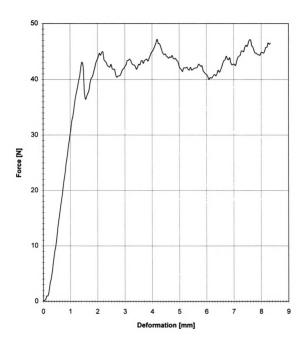


FIGURE B33 - Magness-Taylor puncture test result of Y33_02

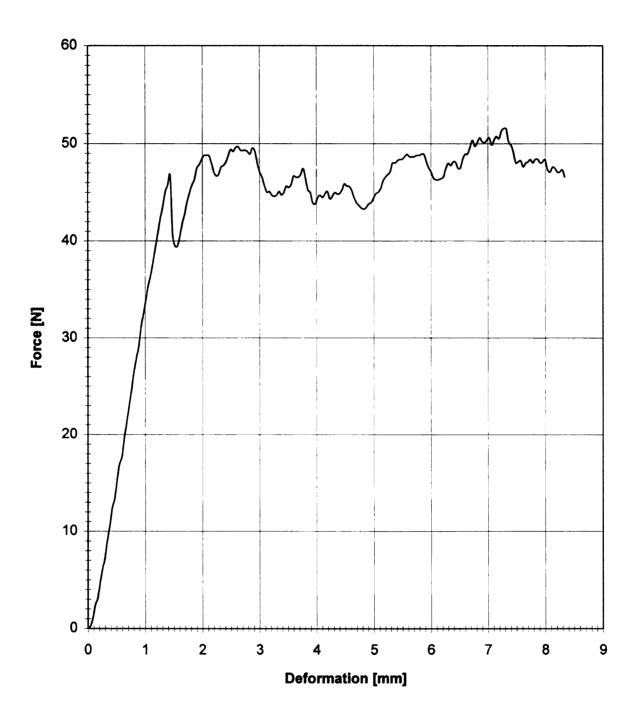


FIGURE B34 - Magness-Taylor puncture test result of Y34_02

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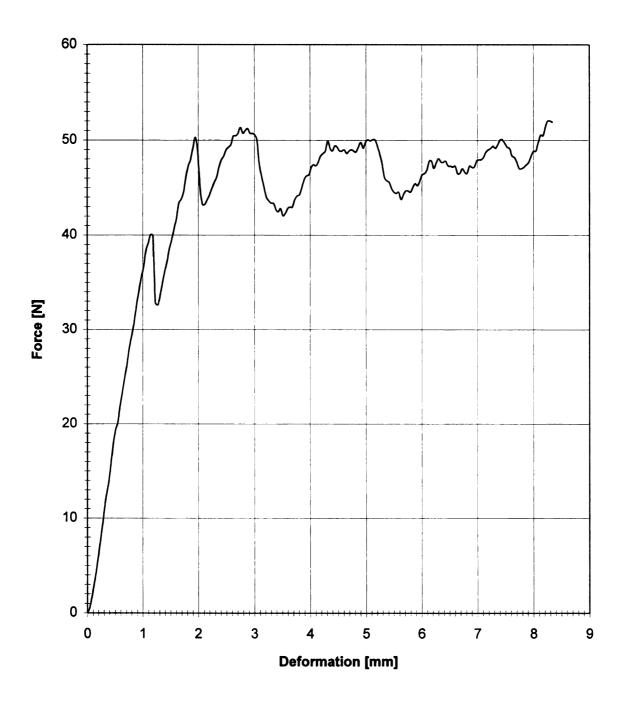


FIGURE B35 - Magness-Taylor puncture test result of Y35_02

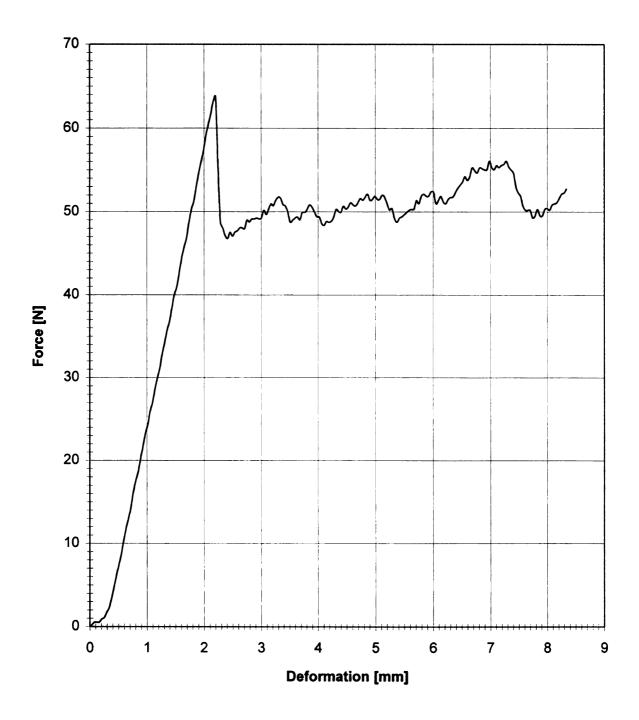


FIGURE B36 - Magness-Taylor puncture test result of Y36_02

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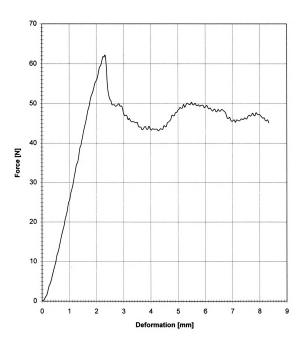


FIGURE B37 - Magness-Taylor puncture test result of Y37_02



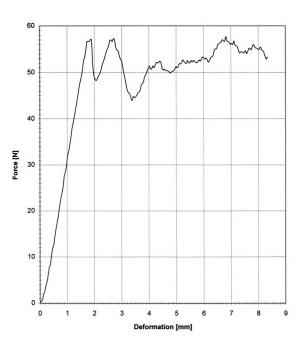


FIGURE B38 - Magness-Taylor puncture test result of $Y38_02$

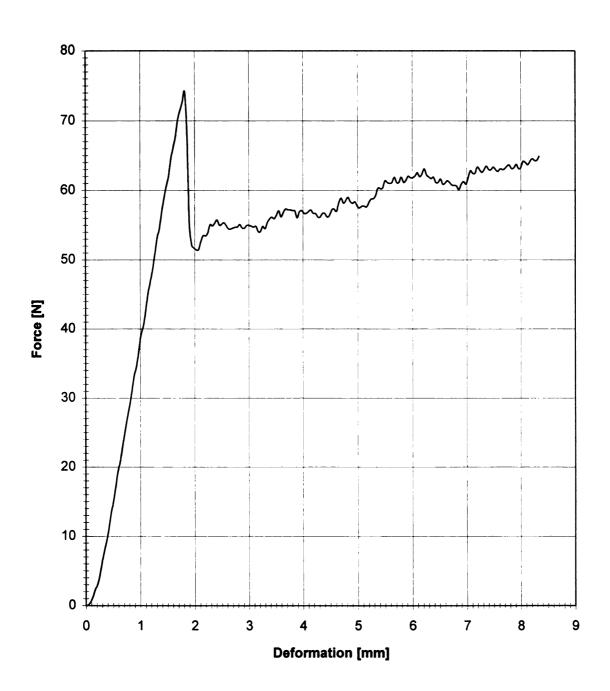


FIGURE B39 - Magness-Taylor puncture test result of Y39_02

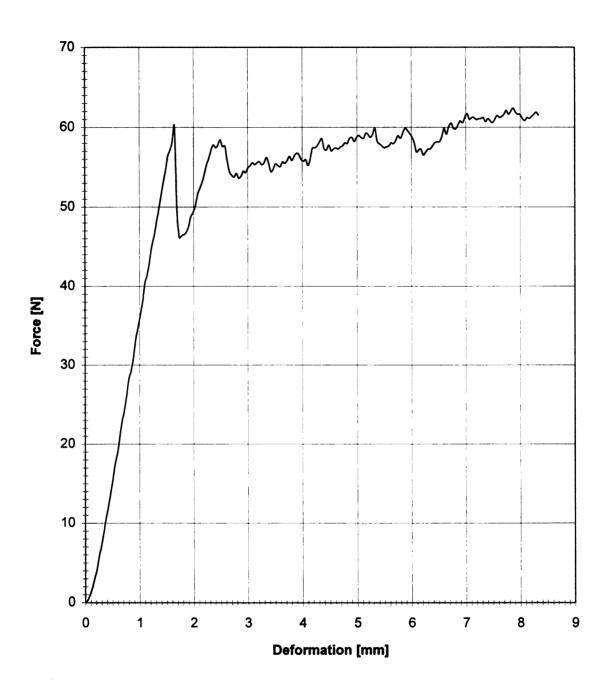


FIGURE B40 - Magness-Taylor puncture test result of Y40_02

-

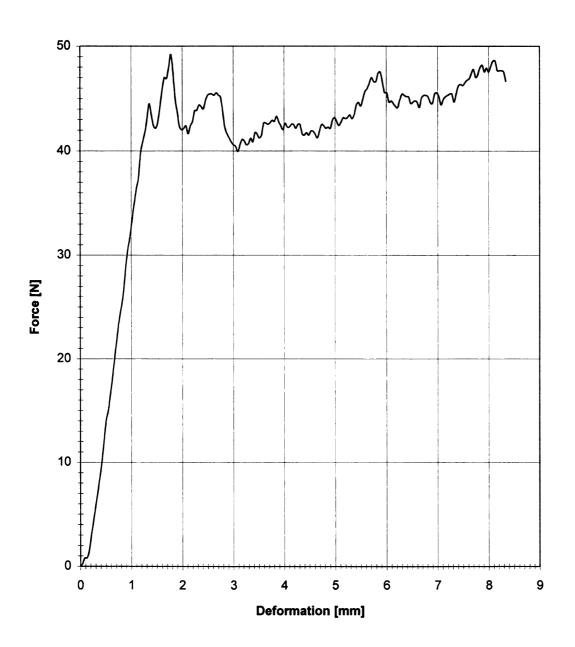


FIGURE B41 - Magness-Taylor puncture test result of $Y41_02$

*

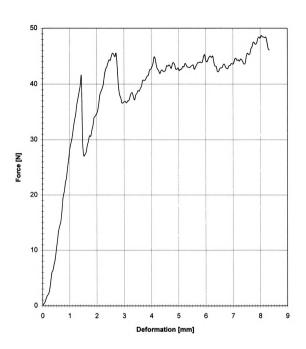


FIGURE B42 - Magness-Taylor puncture test result of Y42_02

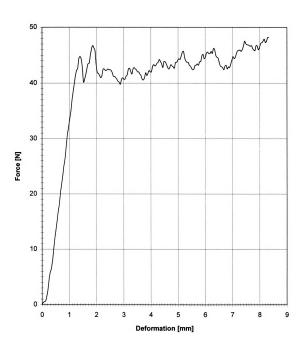


FIGURE B43 - Magness-Taylor puncture test result of Y43_02

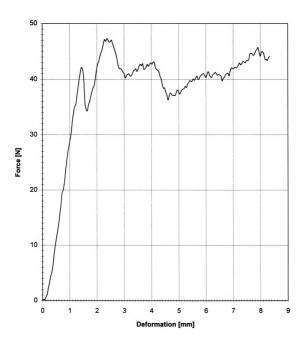


FIGURE B44 - Magness-Taylor puncture test result of Y44_02



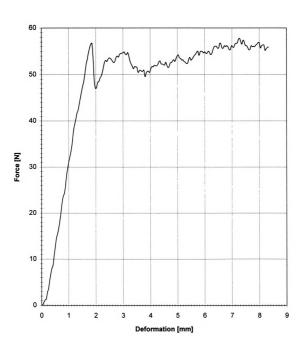


FIGURE B45 - Magness-Taylor puncture test result of Y45_02

APPENDIX C

APPENDIX C

Force-Deformation Plots of Compression Test and Curve Fitting



Elasticity 3.36E+08 [Pa]
Viscosity 3.01E+08 [Pa*s]
Yield stress 284230 [Pa]

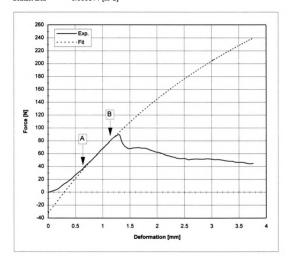


FIGURE C1 - Compression test and curve fit for Y01_03

Elasticity 6.41E+08 [Pa]
Viscosity 1.83E+08 [Pa*s]
Yield stress 441721 [Pa]

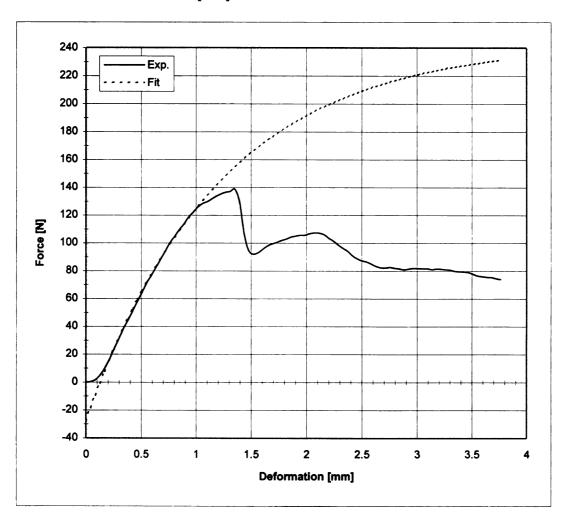


FIGURE C2 - Compression test and curve fit for Y02_03

| Elasticity | 7.00E+08 [Pa] | Viscosity | 2.42E+08 [Pa*s] | Yield stress | 443736 [Pa] | Velocity | 0.00423 [m/s]

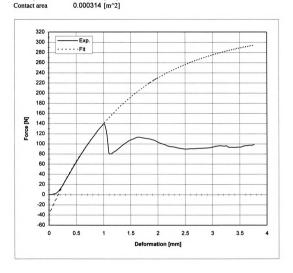


FIGURE C3 - Compression test and curve fit for Y03_03

Elasticity 5.53E+08 [Pa]
Viscosity 1.61E+08 [Pa*s]
Yield stress 335967 [Pa]

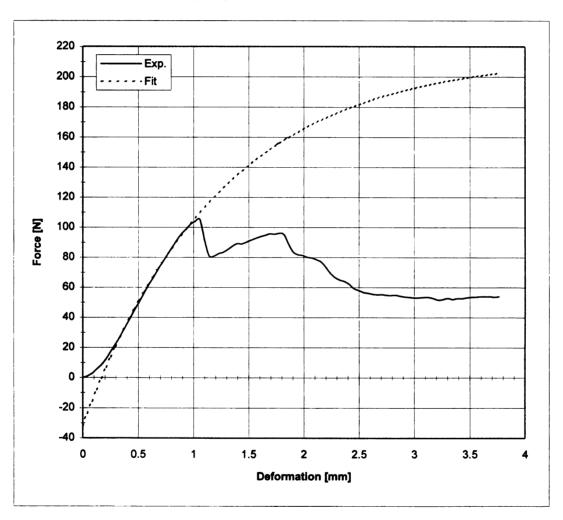


FIGURE C4 - Compression test and curve fit for Y04_03

Elasticity 6.65E+08 [Pa]
Viscosity 2.70E+08 [Pa*s]
Yield stress 428455 [Pa]

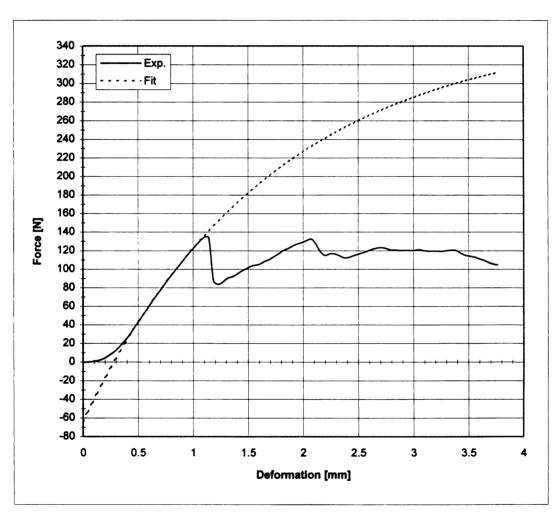


FIGURE C5 - Compression test and curve fit for Y05_03



 Elasticity
 7.25E+08 [Pa]

 Viscosity
 3.07E+08 [Pa*s]

 Yield stress
 446245 [Pa]

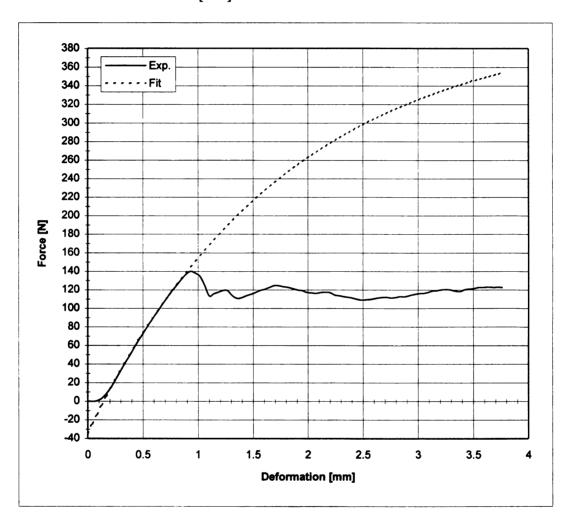


FIGURE C6 - Compression test and curve fit for Y06_03

Elasticity 7.25E+08 [Pa]
Viscosity 3.22E+08 [Pa*s]
Yield stress 473311 [Pa]

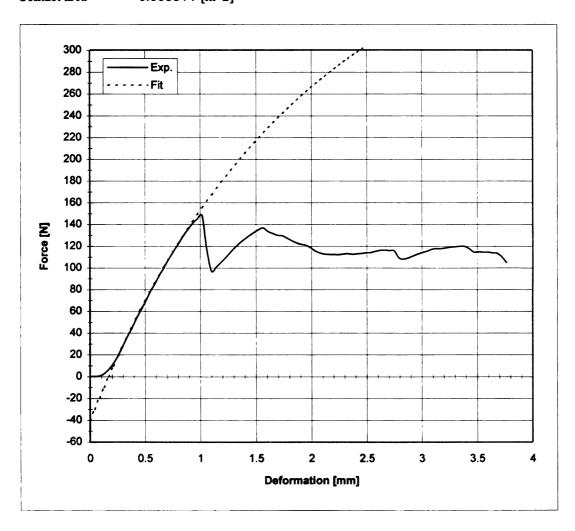


FIGURE C7 - Compression test and curve fit for Y07_03

Elasticity Viscosity Yield stress 6.52E+08 [Pa] 2.09E+08 [Pa*s] 352807 [Pa]

Velocity Contact area 0.00423 [m/s] 0.000314 [m^2]

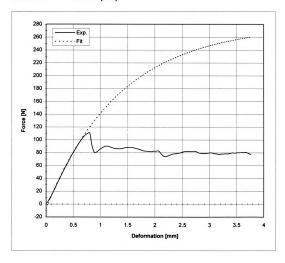


FIGURE C8 - Compression test and curve fit for Y08_03



Elasticity 5.68E+08 [Pa]
Viscosity 2.53E+08 [Pa*s]
Yield stress 331709 [Pa]

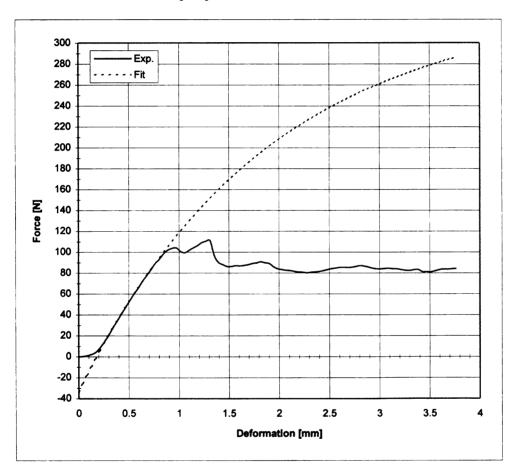
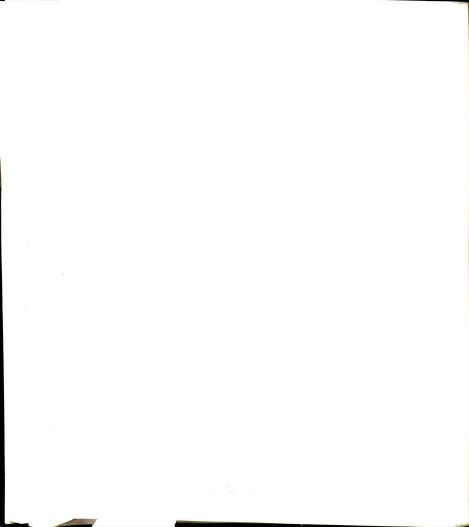


FIGURE C9 - Compression test and curve fit for Y09_03



Elasticity 5.82E+08 [Pa]
Viscosity 2.61E+08 [Pa*s]
Vield stress 278642 [Pa]
Velocity 0.00423 [m/s]

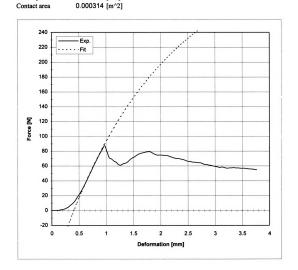
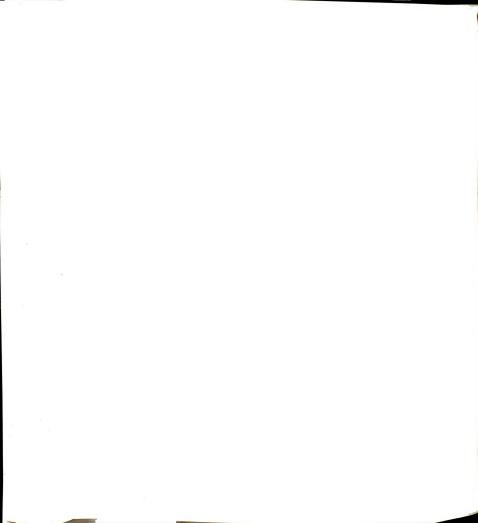


FIGURE C10 - Compression test and curve fit for Y10_03



 Elasticity
 8.09E+08 [Pa]

 Viscosity
 2.75E+08 [Pa*s]

 Yield stress
 518319 [Pa]

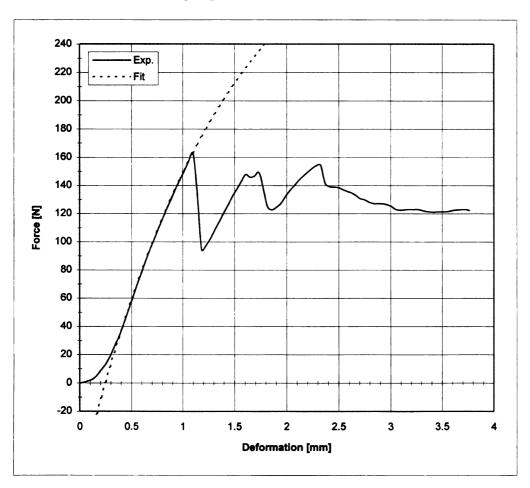


FIGURE C11 - Compression test and curve fit for Y11_03

lasticity 5.19E+08 [Pa]
(iscosity 4.29E+08 [Pa*s]
(ield stress 434385 [Pa]

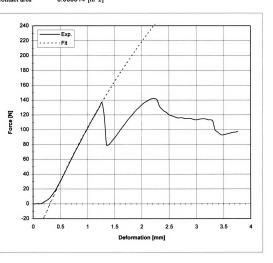
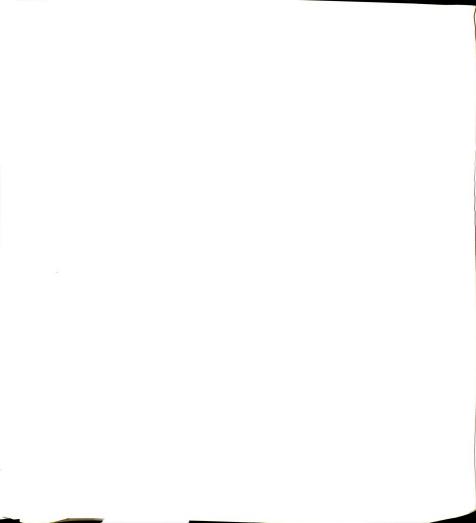


FIGURE C12 - Compression test and curve fit for Y12_03



Elasticity 6.90E+08 [Pa]
/iscosity 2.09E+08 [Pa*s]
/ield stress 476922 [Pa]

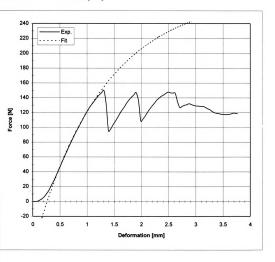
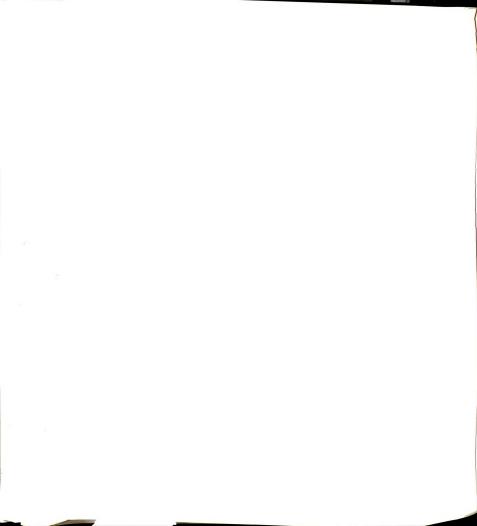


FIGURE C13 - Compression test and curve fit for Y13 $_03$



Elasticity 8.12E+08 [Pa] Viscosity 2.56E+08 [Pa*s] Yield stress 506877 [Pa]

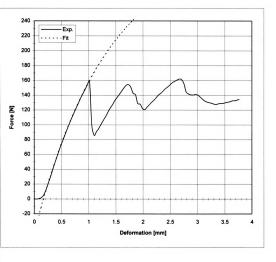


FIGURE C14 - Compression test and curve fit for Y14_03

lasticity iscosity ield stress 7.44E+08 [Pa] 2.64E+08 [Pa*s] 500491 [Pa]

elocity ontact area 0.00423 [m/s] 0.000314 [m^2]

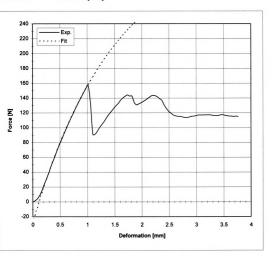


FIGURE C15 - Compression test and curve fit for Y15_03



Elasticity 6.23E+08 [Pa] Viscosity 5.17E+08 [Pa*s] Yield stress 497526 [Pa]

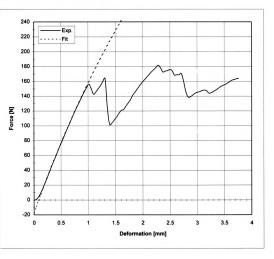


FIGURE C16 - Compression test and curve fit for Y16_03



Elasticity 4.82E+08 [Pa] Viscosity 2.76E+08 [Pa*s] Vield stress 353833 [Pa]

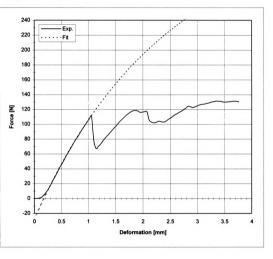
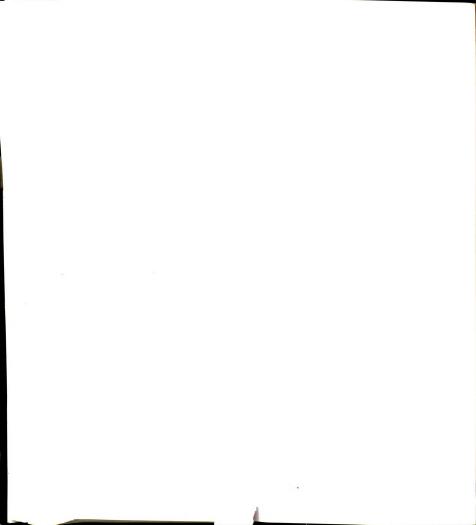


FIGURE C17 - Compression test and curve fit for Y17 03



Elasticity	4.57E+08 [Pa]
Viscosity	4.02E+08 [Pa*s]
Yield stress	358053 [Pa]

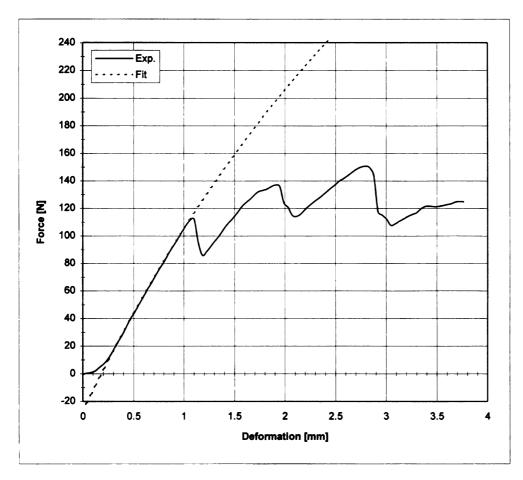


FIGURE C18 - Compression test and curve fit for Y18_03

Elasticity 3.05E+08 [Pa]
Viscosity 1.93E+09 [Pa*s]
Yield stress 364249 [Pa]
Velocity 0.00423 [m/s]

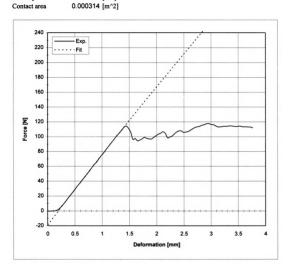


FIGURE C19 - Compression test and curve fit for Y19_03



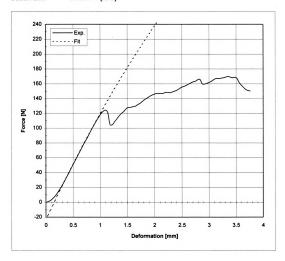
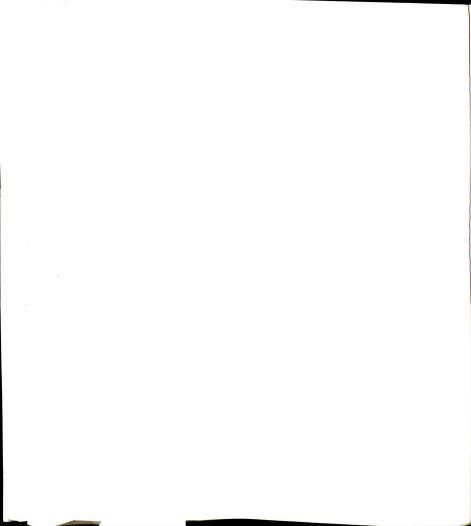


FIGURE C20 - Compression test and curve fit for Y20_03



Elasticity 4.17E+08 [Pa]
Viscosity 2.17E+08 [Pa*s]
Yield stress 334104 [Pa]

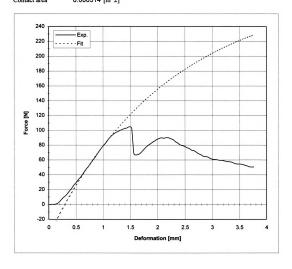
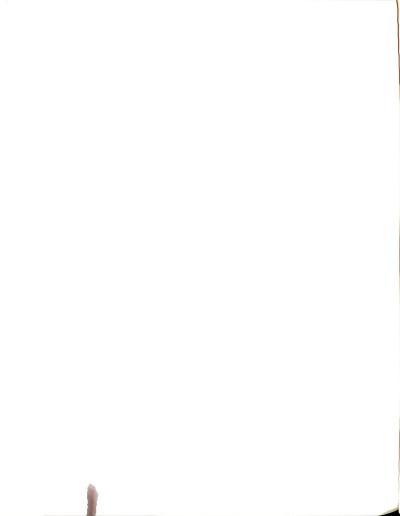


FIGURE C21 - Compression test and curve fit for Y21 03



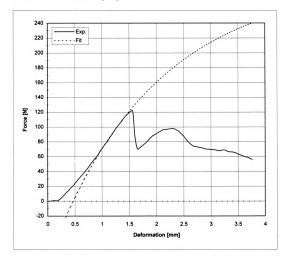
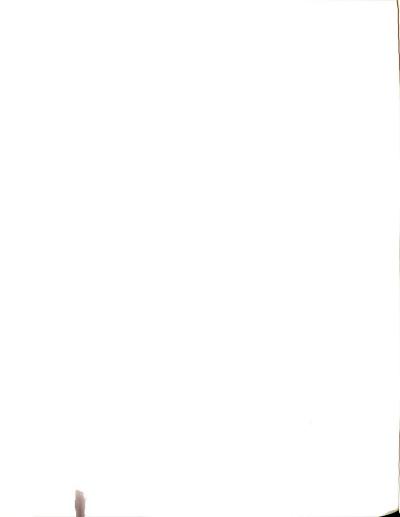


FIGURE C22 - Compression test and curve fit for Y22_03



Elasticity 5.15E+08 [Pa]
Viscosity 2.01E+08 [Pa*s]
Yield stress 379075 [Pa]

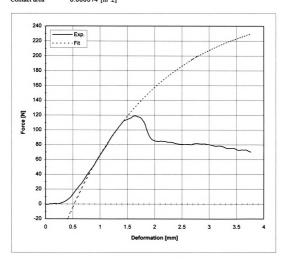


FIGURE C23 - Compression test and curve fit for Y23_03



Elasticity 3.30E+08 [Pa] Viscosity 3.05E+08 [Pa*s] Yield stress 282443 [Pa]

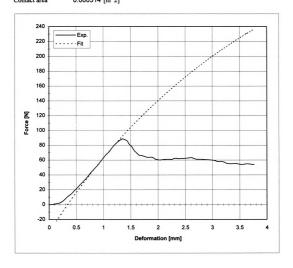


FIGURE C24 - Compression test and curve fit for Y24_03

 Elasticity
 3.36E+08 [Pa]

 Viscosity
 3.01E+08 [Pa*s]

 Yield stress
 284230 [Pa]

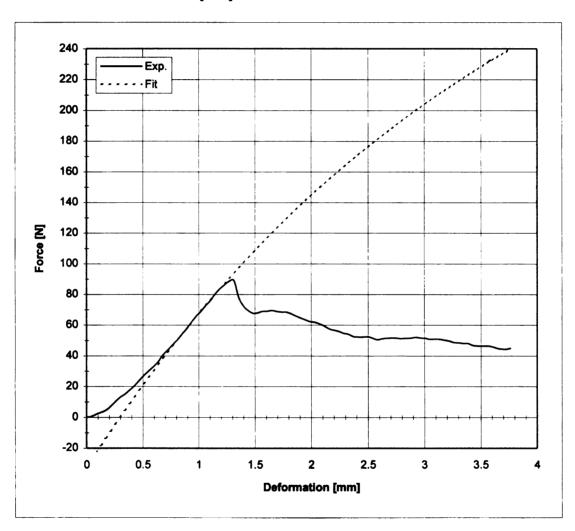


FIGURE C25 - Compression test and curve fit for Y25_03

Elasticity 8.62E+08 [Pa]
Viscosity 1.88E+08 [Pa*s]
Yield stress 444801 [Pa]

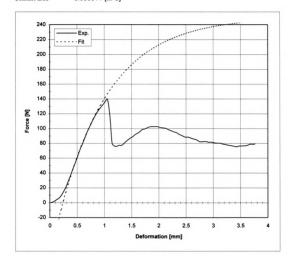


FIGURE C26 - Compression test and curve fit for Y26_03



Elasticity 9.24E+08 [Pa]
Viscosity 3.35E+08 [Pa*s]
Yield stress 631107 [Pa]

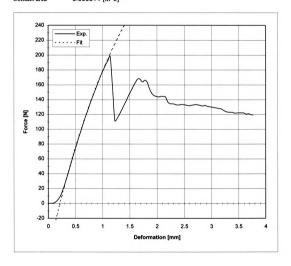


FIGURE C27 - Compression test and curve fit for Y27_03



Elasticity 7.48E+08 [Pa]
Viscosity 3.31E+08 [Pa*s]
Yield stress 469624 [Pa]

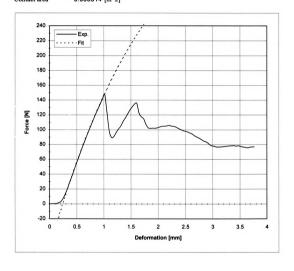


FIGURE C28 - Compression test and curve fit for Y28 03

Elasticity 1.12E+09 [Pa]
Viscosity 3.20E+08 [Pa*s]
Yield stress 653307 [Pa]

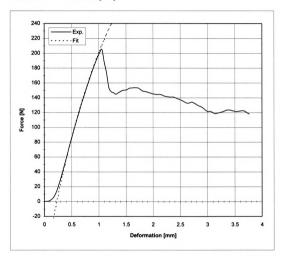


FIGURE C29 - Compression test and curve fit for $Y29_03$

Elasticity 7.82E+08 [Pa]
Viscosity 4.76E+08 [Pa*s]
Yield stress 554547 [Pa]

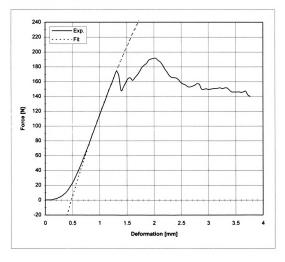


FIGURE C30 - Compression test and curve fit for Y30_03

Elasticity 6.56E+08 [Pa]
Viscosity 1.46E+08 [Pa*s]
Yield stress 355620 [Pa]

Velocity 0.00423 [m/s] Contact area 0.000314 [m^2]

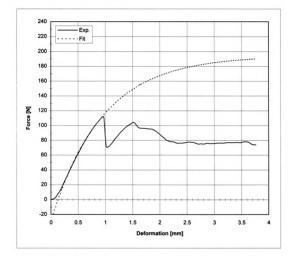


FIGURE C31 - Compression test and curve fit for Y31_03

-



Elasticity 5.48E+08 [Pa]
Viscosity 2.93E+08 [Pa*s]
Yield stress 412565 [Pa]

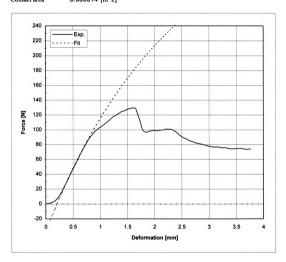
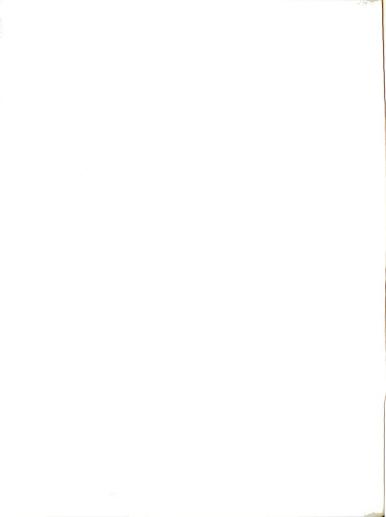


FIGURE C32 - Compression test and curve fit for Y32_03



Elasticity 5.94E+08 [Pa]
Viscosity 1.48E+08 [Pa*s]
Yield stress 300310 [Pa]

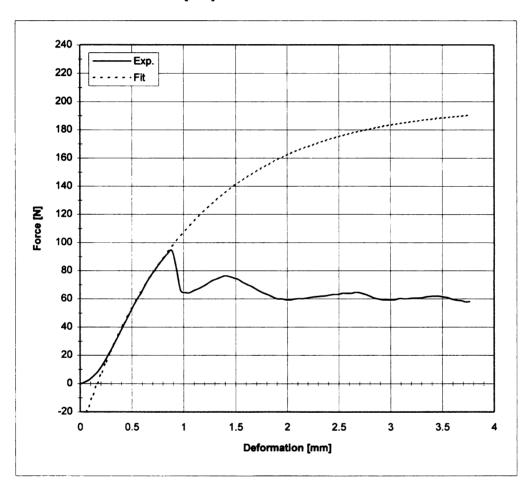
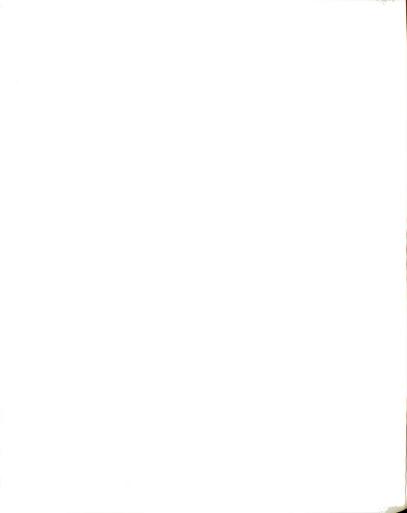


FIGURE C33 - Compression test and curve fit for Y33_03



Elasticity 6.53E+08 [Pa]
Viscosity 1.67E+08 [Pa*s]
Yield stress 348930 [Pa]

Velocity 0.00423 [m/s]
Contact area 0.000314 [m^2]

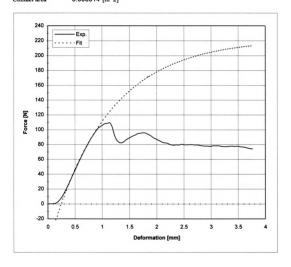


FIGURE C34 - Compression test and curve fit for Y34_03

 Elasticity
 5.92E+08 [Pa]

 Viscosity
 1.62E+08 [Pa*s]

 Yield stress
 288335 [Pa]

 Velocity
 0.00423 [m/s]

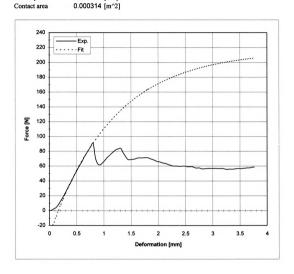


FIGURE C35 - Compression test and curve fit for Y35_03



Elasticity	4.92E+08 [Pa]	
Viscosity	7.24E+08 [Pa*s	j
Yield stress	437958 [Pa]	
Velocity	0.00423 [m/s]	

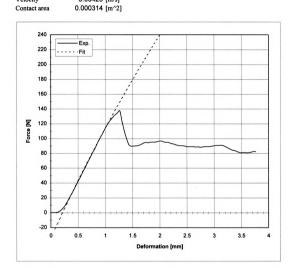


FIGURE C36 - Compression test and curve fit for Y36_03

Elasticity 4.17E+08 [Pa]
Viscosity 6.04E+08 [Pa*s]
Yield stress 396105 [Pa]
Velocity 0.00423 [m/s]

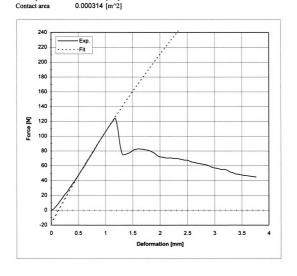


FIGURE C37 - Compression test and curve fit for Y37_03



Elasticity
Viscosity
Yield stress

5.09E+08 [Pa] 4.36E+08 [Pa*s] 421764 [Pa]

Velocity Contact area 0.00423 [m/s] 0.000314 [m^2]

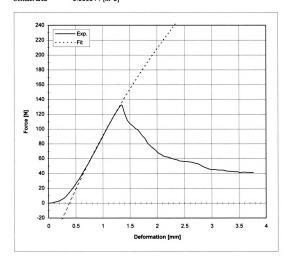


FIGURE C38 - Compression test and curve fit for Y38_03



Elasticity 5.06E+08 [Pa]
Viscosity 2.56E+10 [Pa*s]
Vield stress 513378 [Pa]
Velocity 0.00423 [m/s]

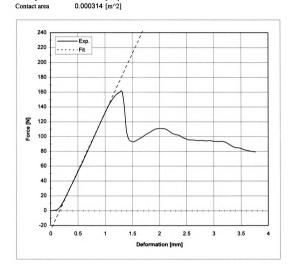
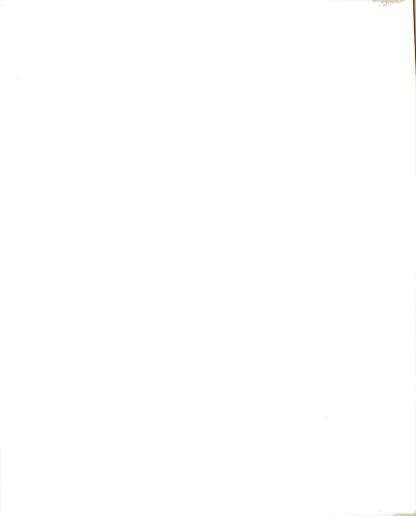


FIGURE C39 - Compression test and curve fit for Y39 03



Elasticity 5.56E+08 [Pa]
Viscosity 4.68E+08 [Pa*s]
Yield stress 426326 [Pa]

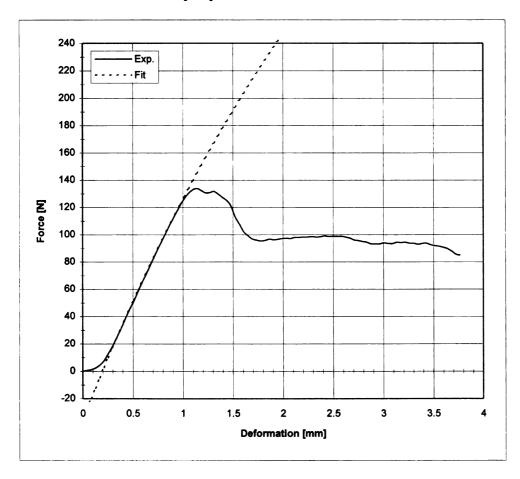


FIGURE C40 - Compression test and curve fit for Y40_03



Elasticity 3.62E+08 [Pa]
Viscosity 4.42E+08 [Pa*s]
Yield stress 331139 [Pa]

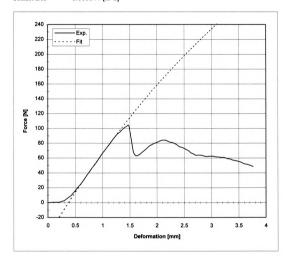


FIGURE C41 - Compression test and curve fit for Y41_03



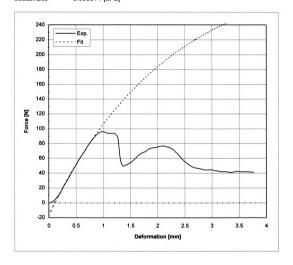


FIGURE C42 - Compression test and curve fit for Y42_03

Elasticity 3.96E+08 [Pa] Viscosity 2.69E+08 [Pa*s] Yield stress 300994 [Pa]

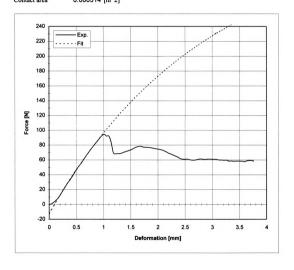


FIGURE C43 - Compression test and curve fit for Y43_03

Elasticity 5.53E+08 [Pa]
Viscosity 1.44E+08 [Pa*s]
Yield stress 313349 [Pa]

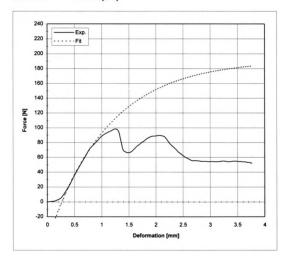


FIGURE C44 - Compression test and curve fit for Y44 03

Elasticity 3.85E+08 [Pa]
Viscosity 2.09E+08 [Pa*s]
Yield stress 313121 [Pa]

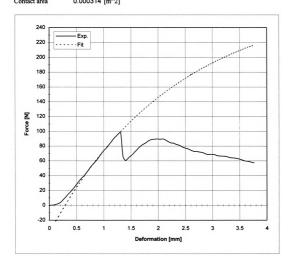
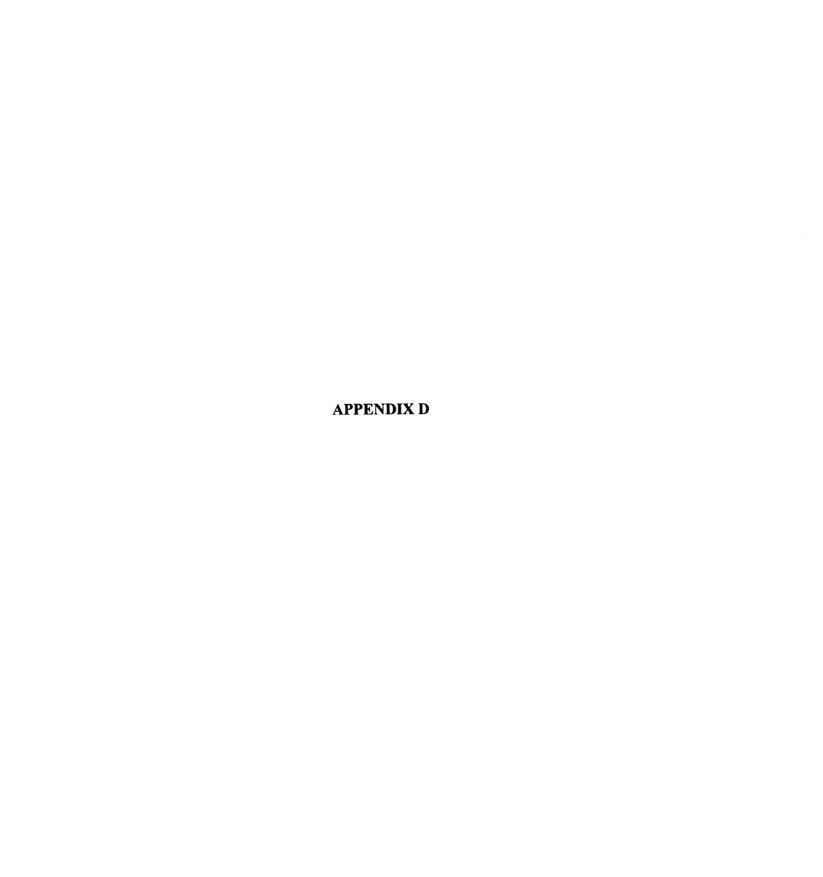


FIGURE C45 - Compression test and curve fit for Y45_03





APPENDIX D

Force-Deformation Plots of Shear Test



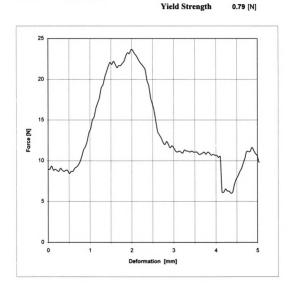
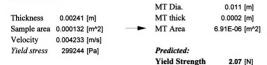


FIGURE D1 - Shear test result for Y01 04





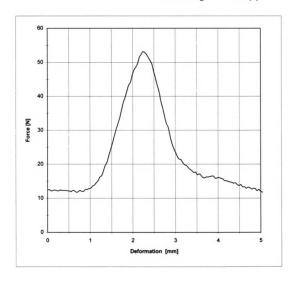


FIGURE D2 - Shear test result for Y02 04



Yield stress 185405 [Pa] Predicted:

Predicted:
Yield Strengt 1.28 [N]

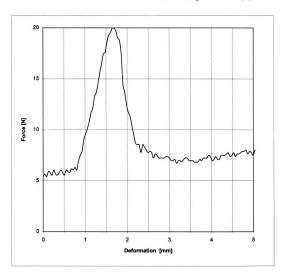
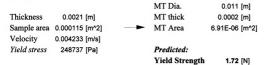


FIGURE D3 - Shear test result for Y03_04





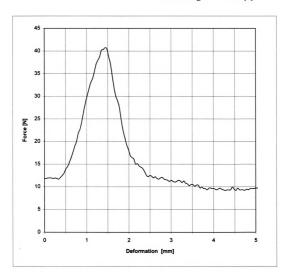


FIGURE D4 - Shear test result for Y04_04





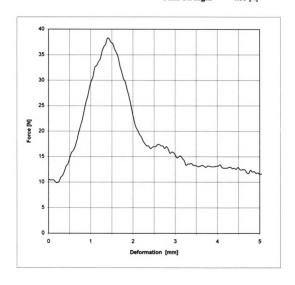


FIGURE D5 - Shear test result for Y05 04



MT Dia. 0.011 [m] MT thick **Thickness** 0.00193 [m] 0.0002 [m] Sample area 0.000106 [m^2] MT Area 6.91E-06 [m^2] Velocity 0.004233 [m/s] Yield stress 191686 [Pa]

Predicted:

Yield Strength 1.32 [N]

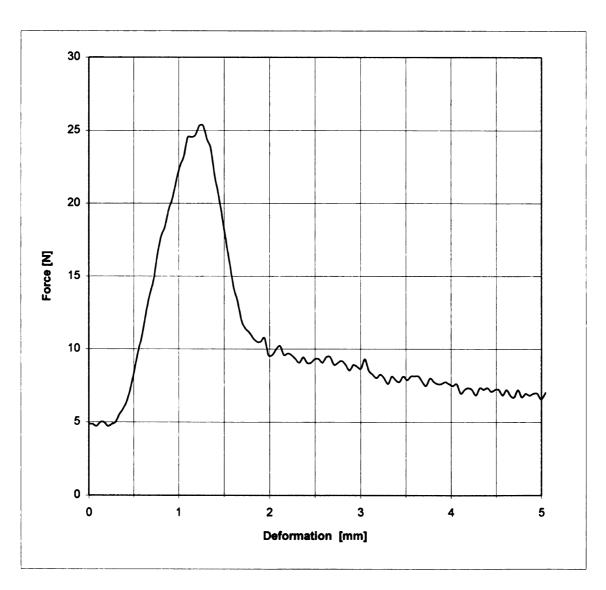
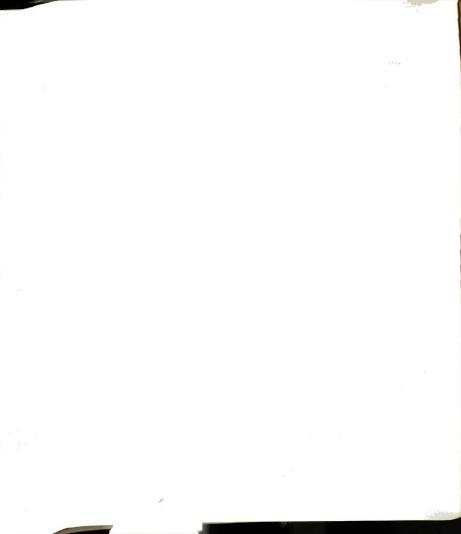
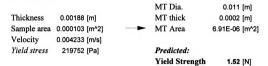


FIGURE D6 - Shear test result for Y06_04





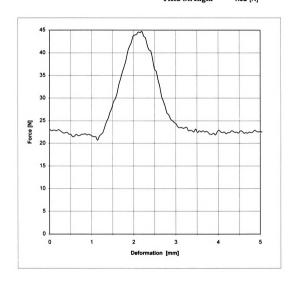
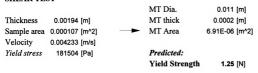


FIGURE D7 - Shear test result for Y07_04





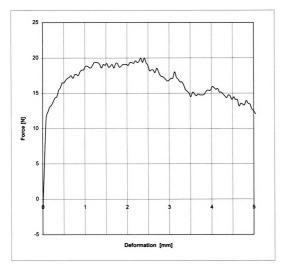
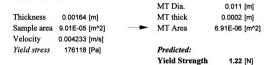


FIGURE D8 - Shear test result for Y08_04





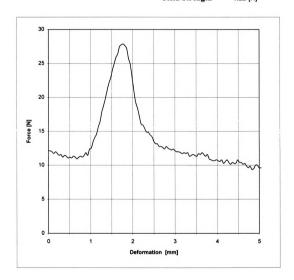
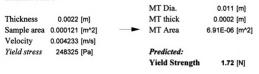


FIGURE D9 - Shear test result for Y09_04





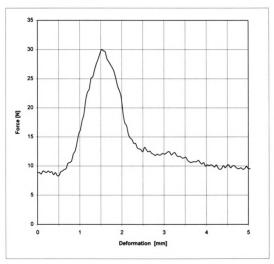
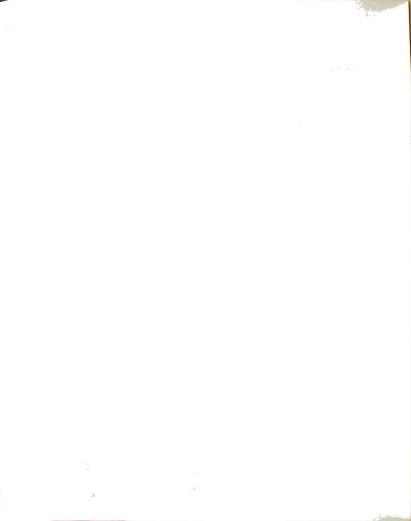
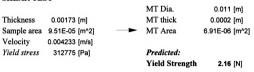


FIGURE D10 - Shear test result for Y10_04





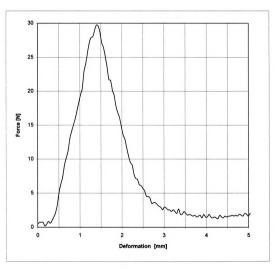
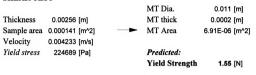


FIGURE D11 - Shear test result for Y11_04





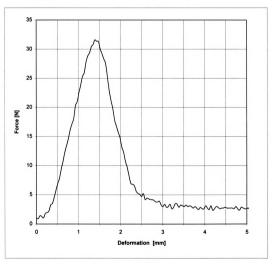
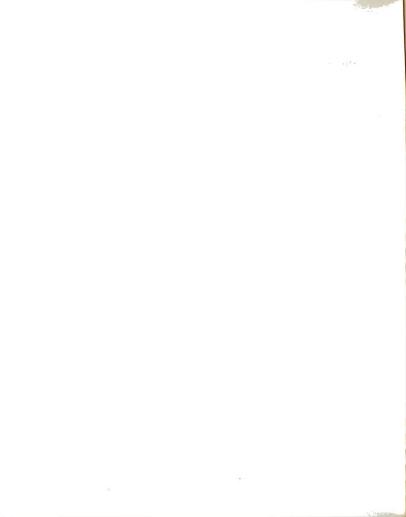


FIGURE D12 - Shear test result for Y12_04





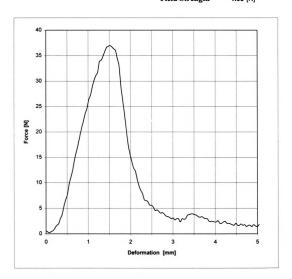
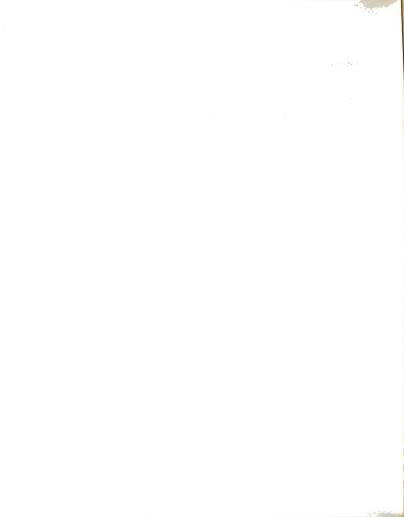


FIGURE D13 - Shear test result for Y13 04





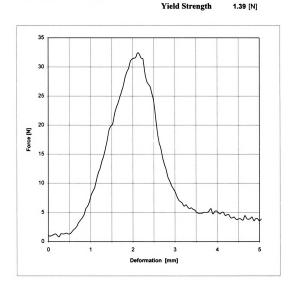
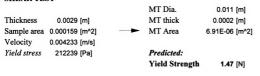


FIGURE D14 - Shear test result for Y14 04







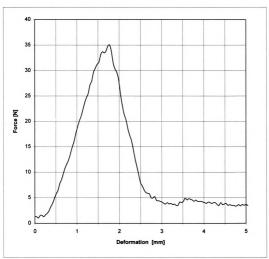
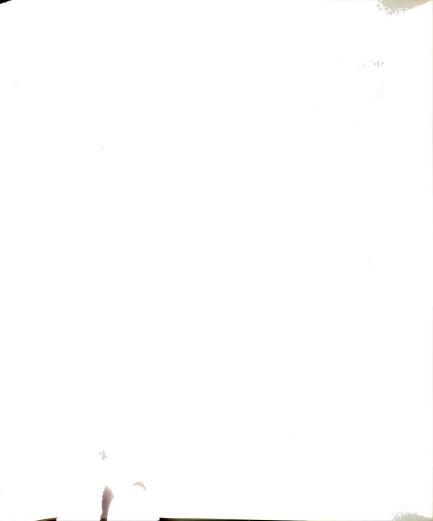
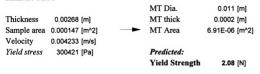


FIGURE D15 - Shear test result for Y15_04





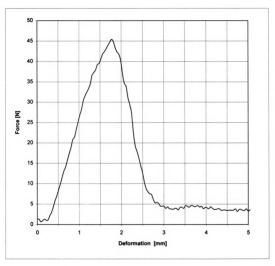
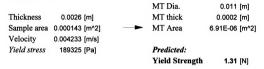


FIGURE D16 - Shear test result for Y16 04





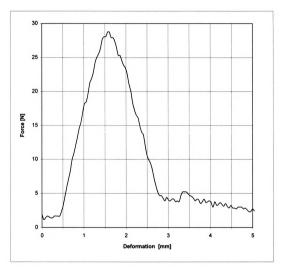


FIGURE D17 - Shear test result for Y17 04



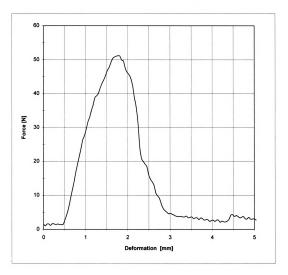


FIGURE D18 - Shear test result for Y18_04





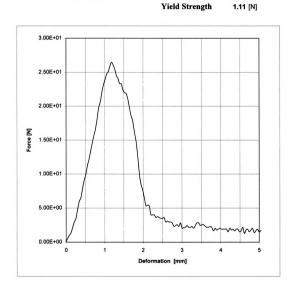
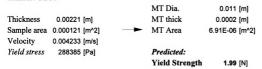


FIGURE D19 - Shear test result for Y19_04





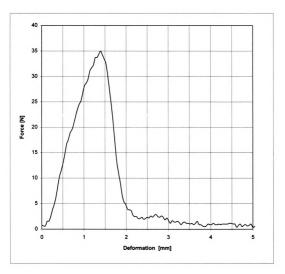
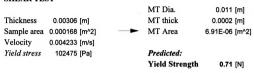


FIGURE D20 - Shear test result for Y20_04





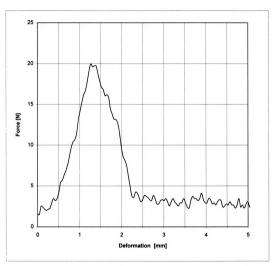


FIGURE D21 - Shear test result for Y21 04







Yield stress 145849 [Pa] Predicted:

Yield Strength 1.01 [N]

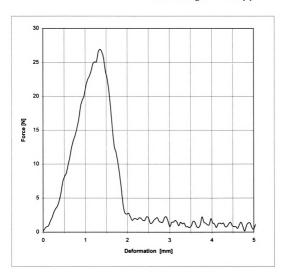


FIGURE D22 - Shear test result for Y22_04





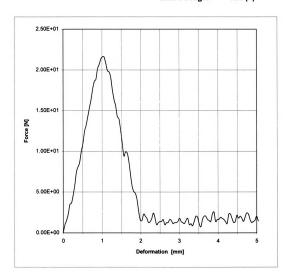


FIGURE D23 - Shear test result for Y23_04







Yield stress 89790 [Pa] **Predicted:**

Yield Strength 0.62 [N]

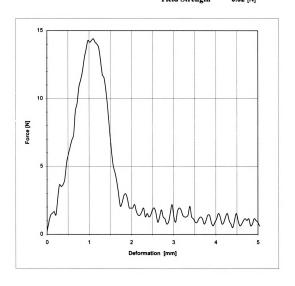


FIGURE D24 - Shear test result for Y24_04





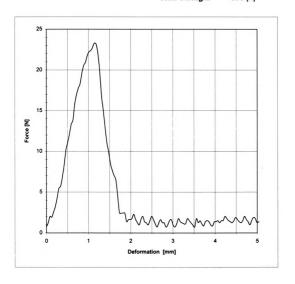


FIGURE D25 - Shear test result for Y25 04



MT Dia. 0.011 [m]
Thickness 0.00248 [m] MT thick 0.0002 [m]
Sample area 0.000136 [m^2] → MT Area 6.91E-06 [m^2]
Velocity 0.004233 [m/s]

Yield stress 211617 [Pa] Predicted:

Yield Strength 1.46 [N]

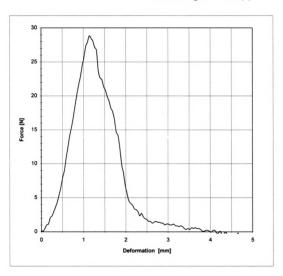
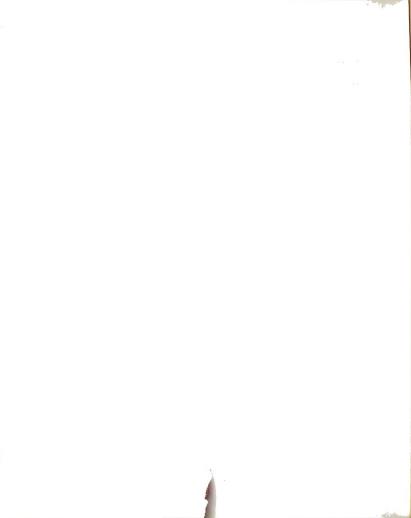


FIGURE D26 - Shear test result for Y26_04







 Velocity
 0.004233 [m/s]

 Yield stress
 293539 [Pa]
 Predicted:

Yield Strength 2.03 [N]

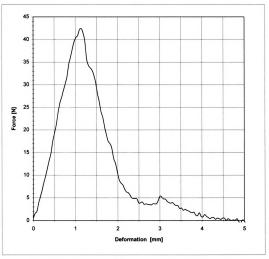
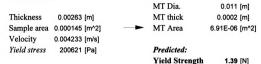


FIGURE D27 - Shear test result for Y27_04



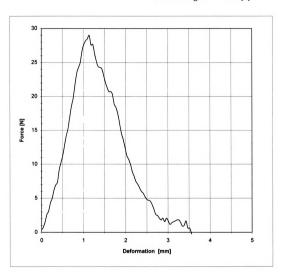


FIGURE D28 - Shear test result for Y28_04



 MT Dia.
 0.011 [m]

 Thickness
 0.00251 [m]
 MT thick
 0.0002 [m]

 Sample area
 0.000138 [m²2]
 → MT Area
 6.91E-06 [m²2]

Velocity 0.004233 [m/s] Yield stress 318823 [Pa]

318823 [Pa] *Predicted:*

Yield Strength 2.20 [N]

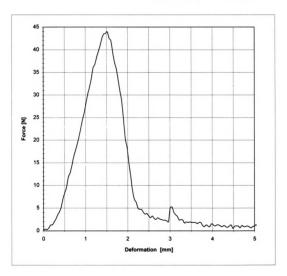


FIGURE D29 - Shear test result for Y29 04





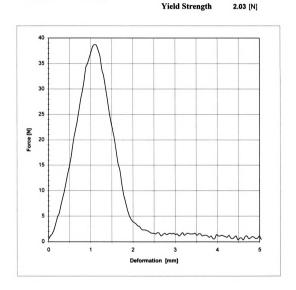


FIGURE D30 - Shear test result for Y30_04







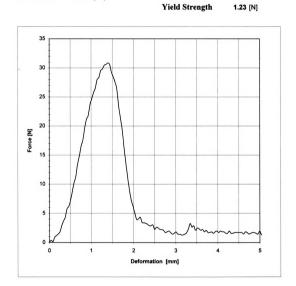
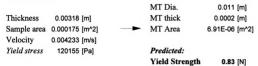


FIGURE D31 - Shear test result for Y31 04



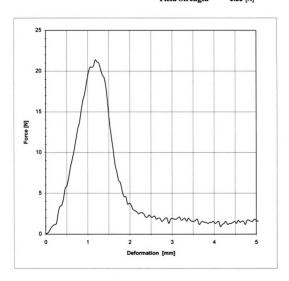
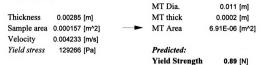


FIGURE D32 - Shear test result for Y32_04



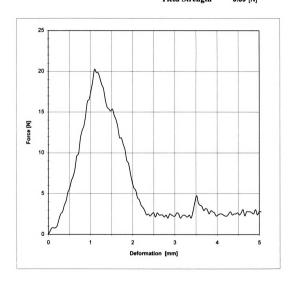
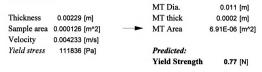


FIGURE D33 - Shear test result for Y33 04





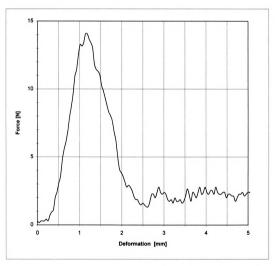


FIGURE D34 - Shear test result for Y34 04



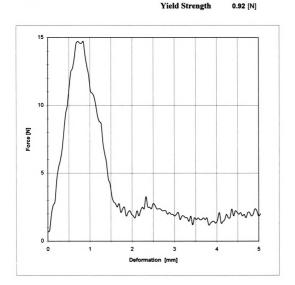


FIGURE D35 - Shear test result for Y35_04







Yield Strength 1.08 [N]

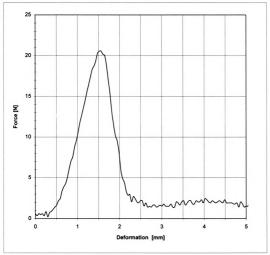
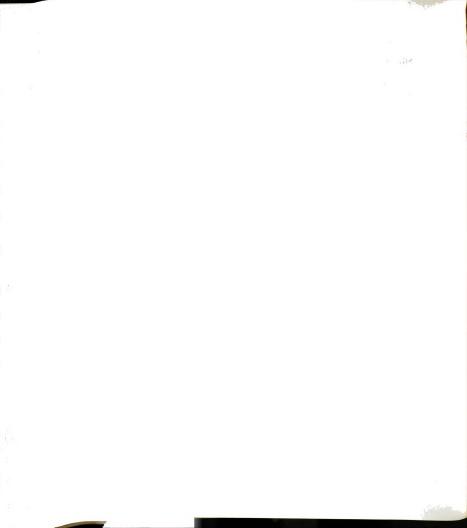


FIGURE D36 - Shear test result for Y36 04





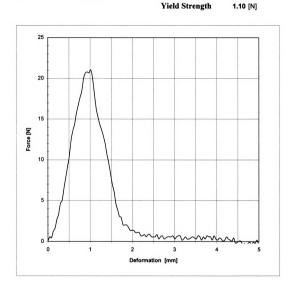
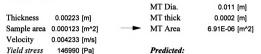


FIGURE D37 - Shear test result for Y37 04



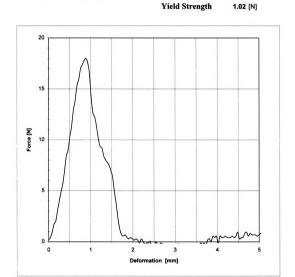


FIGURE D38 - Shear test result for Y38 04





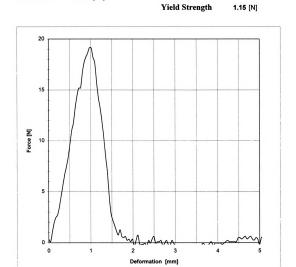
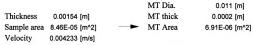


FIGURE D39 - Shear test result for Y39 04





 Yield stress
 168136 [Pa]
 Predicted:

 Yield Strength
 1.16 [N]

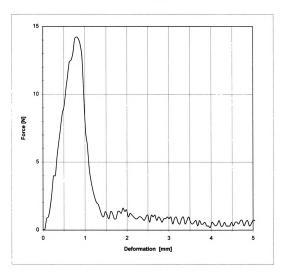


FIGURE D40 - Shear test result for Y40_04



SHEAR LETT

Thickness 0.00214 [m]
Sample area 0.000118 [m^2]
Velocity 0.004233 [m/s]

Yield stress 146777 [Pa]

MT Dia. 0.011 [m]

MT thick 0.0002 [m]

► MT Area 6.91E-06 [m^2]

Predicted:

Yield Strength 1.01 [N]

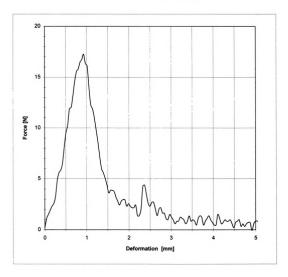
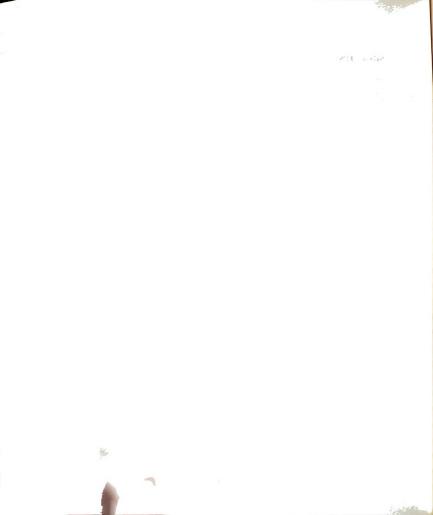


FIGURE D41 - Shear test result for Y41 04







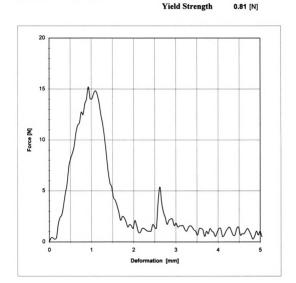


FIGURE D42 - Shear test result for Y42_04





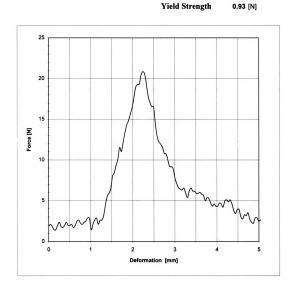
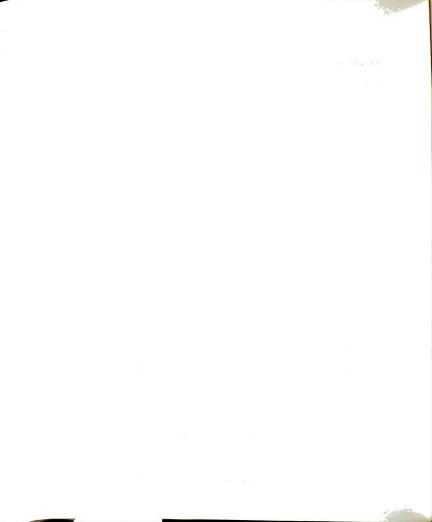


FIGURE D43 - Shear test result for Y43_04





 Thickness
 0.00256 [m]
 MT Dia.
 0.011 [m]

 Sample area
 0.000141 [m²2]
 MT Area
 6.91E-06 [m²2]

 Velocity
 0.004233 [m/s]

Velocity 0.004233 [m/s]

Yield stress 123545 [Pa] Predicted:

Yield Strength 0.85 [N]

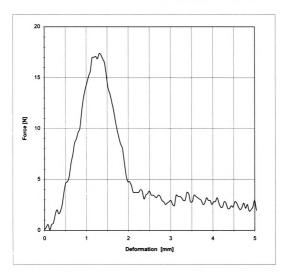
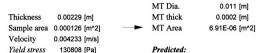


FIGURE D44 - Shear test result for Y44_04



130808 [Pa]



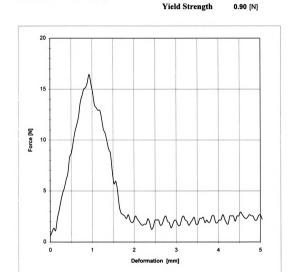
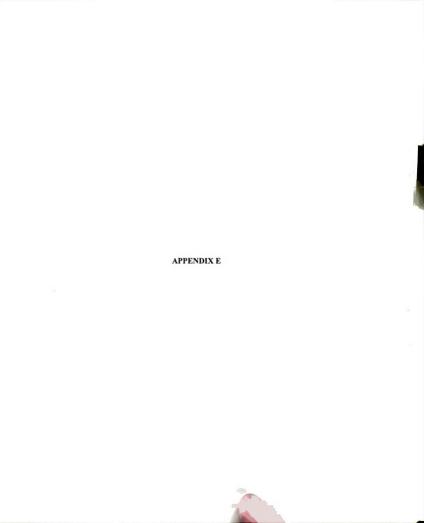


FIGURE D45 - Shear test result for Y45 04

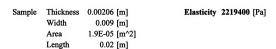




APPENDIX E

Stress-Strain Plots of Tensile Test





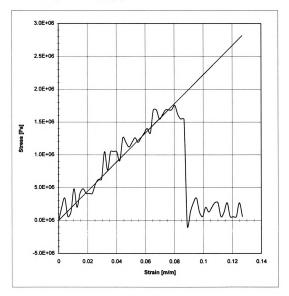


FIGURE E1 - Tensile test result for Y01_05

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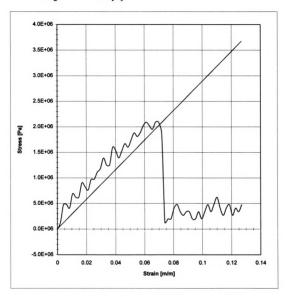
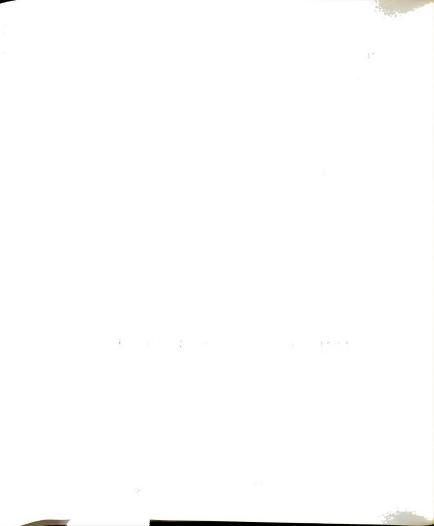


FIGURE E2 - Tensile test result for Y02_05





Sample	Thickness	0.00191 [m]	Elasticity	2661102 [Pa]
•	Width	0.009 [m]	•	
	Area	1.7E-05 [m^2]		
	T .1	0.00 5 3		

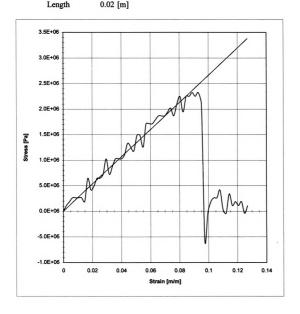
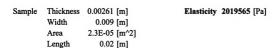


FIGURE E3 - Tensile test result for Y03 05

 $z=y=-C_1\cdots +co(x_T)$



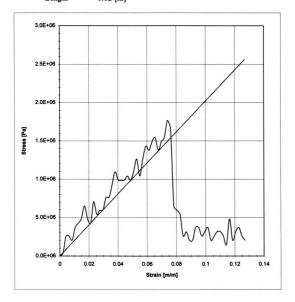
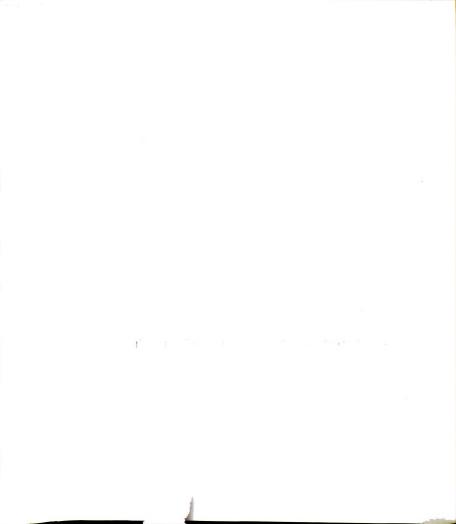


FIGURE E4 - Tensile test result for Y04_05





TENSILE TEST

Sample	Thickness	0.00178 [m]	Elasticity	2603817	[Pa]
	Width	0.009 [m]			
	Area	1.6E-05 [m^2]			
	Length	0.02 [m]			

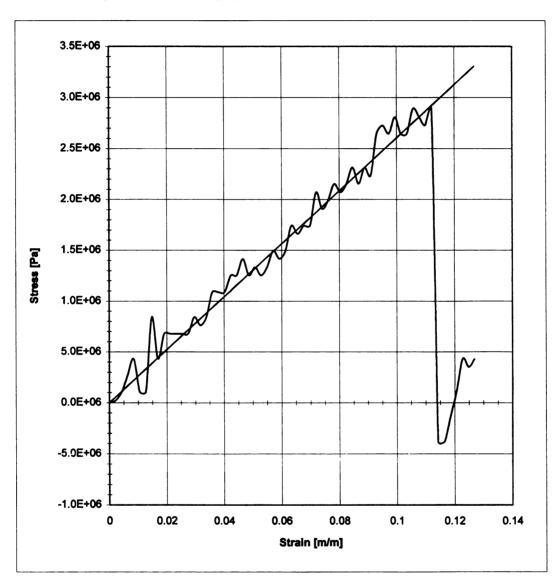


FIGURE E5 - Tensile test result for Y05_05

TENSILE TEST



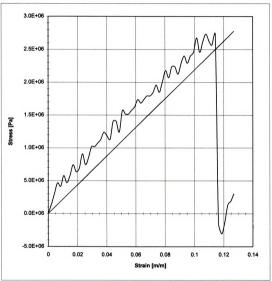


FIGURE E6 - Tensile test result for Y06_05

Sample Thickness 0.00273 [m] Elasticity 2644603 [Pa]
Width 0.009 [m]
Area 2.46E-05 [m^2]
Length 0.02 [m]

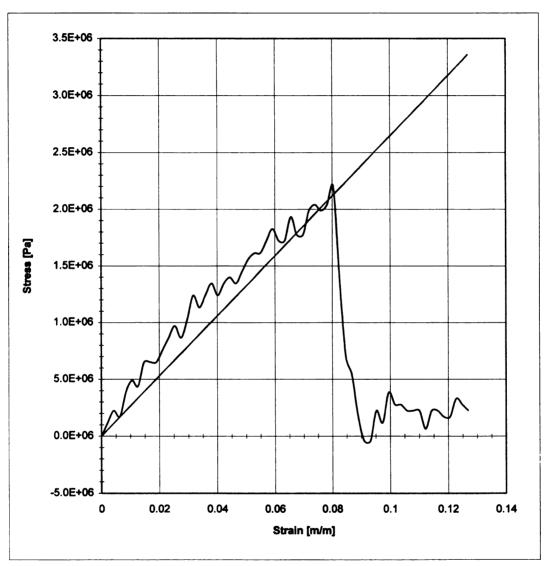


FIGURE E7 - Tensile test result for Y07_05

Sample Thickness 0.0027 [m] Elasticity 3603414 [Pa]
Width 0.009 [m]
Area 2.43E-05 [m^2]
Length 0.02 [m]

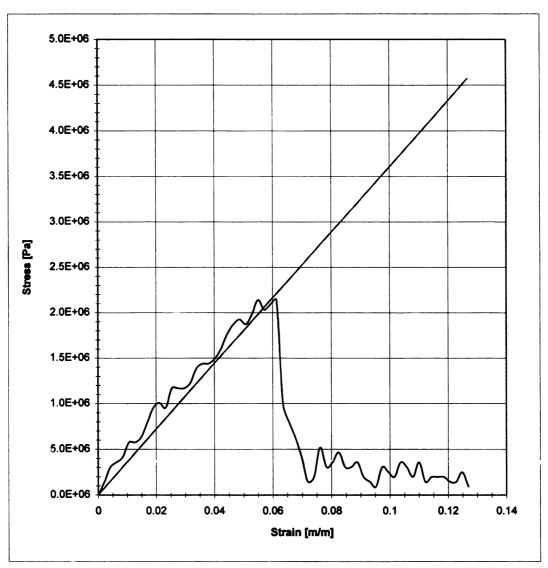


FIGURE E8 - Tensile test result for Y08_05

TENSILE TEST



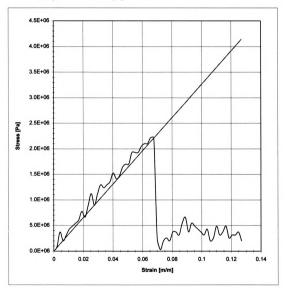


FIGURE E9 - Tensile test result for Y09_05

TENSILE TEST



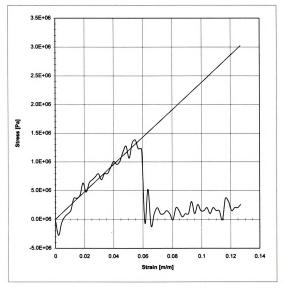
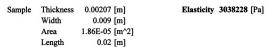


FIGURE E10 - Tensile test result for Y10_05

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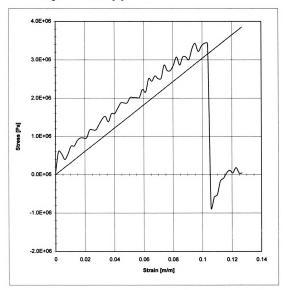


FIGURE E11 - Tensile test result for Y11_05





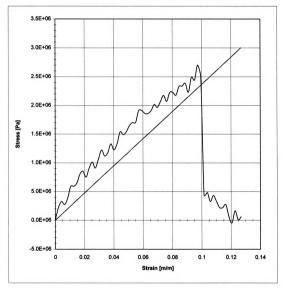
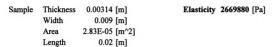


FIGURE E12 - Tensile test result for Y12 05



TENSILE TEST



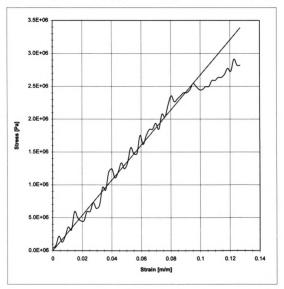


FIGURE E13 - Tensile test result for Y13_05

TENSILE TEST



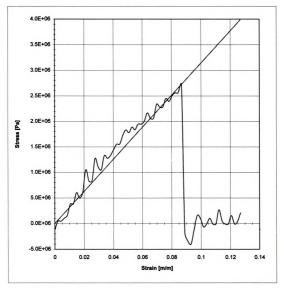


FIGURE E14 - Tensile test result for Y14 05



TENSILE TEST



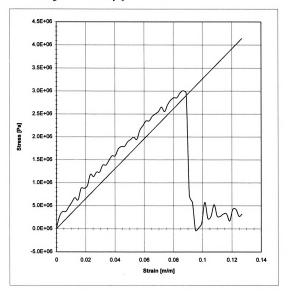


FIGURE E15 - Tensile test result for Y15_05





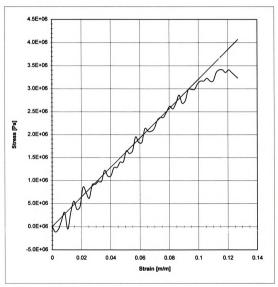


FIGURE E16 - Tensile test result for Y16 05

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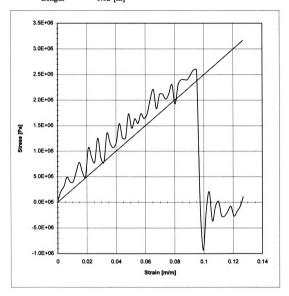
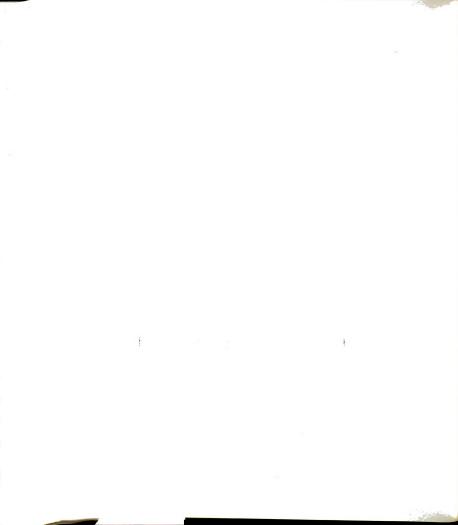


FIGURE E17 - Tensile test result for Y17_05



TENSILE TEST



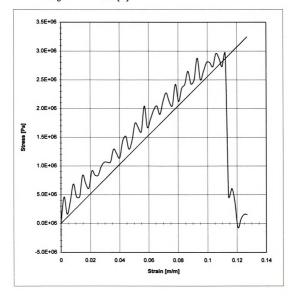
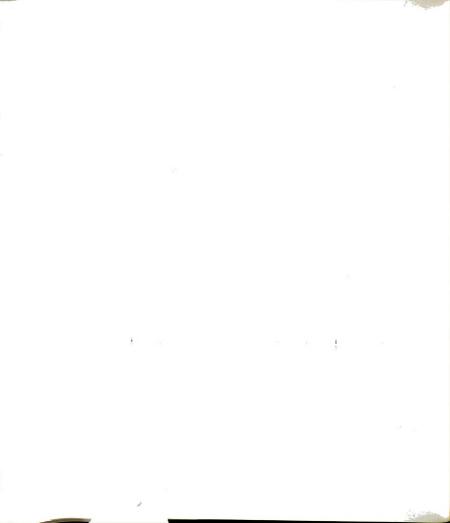


FIGURE E18 - Tensile test result for Y18_05



 Sample
 Thickness
 0.00307 [m]
 Elasticity
 2106323 [Pa]

 Width
 0.009 [m]

 Area
 2.76E-05 [m^2]

 Length
 0.02 [m]

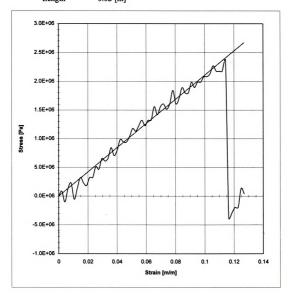
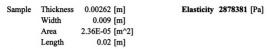


FIGURE E19 - Tensile test result for Y19_05

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TENSILE TEST



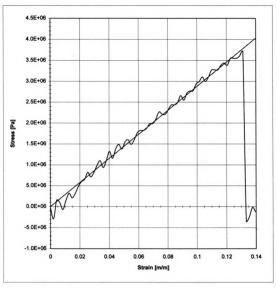
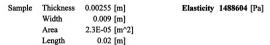


FIGURE E20 - Tensile test result for Y20_05





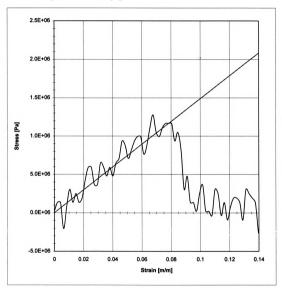
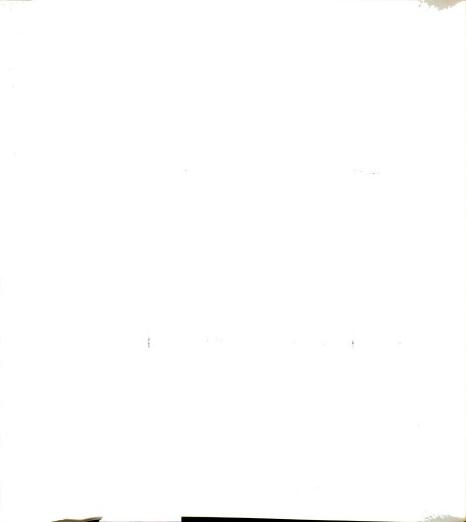
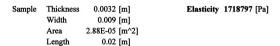


FIGURE E21 - Tensile test result for Y21_05





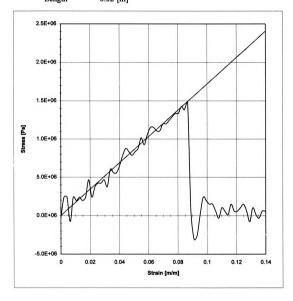
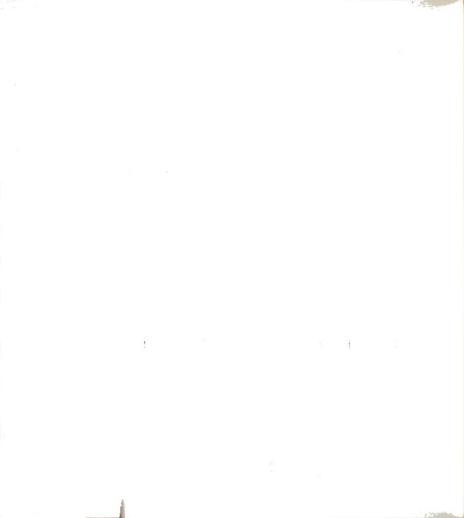


FIGURE E22 - Tensile test result for Y22 05



TENSILE TEST



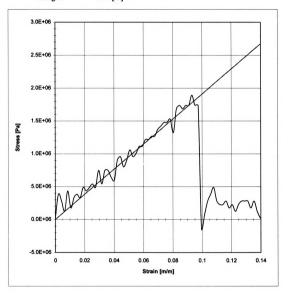


FIGURE E23 - Tensile test result for Y23_05

TENSILE TEST



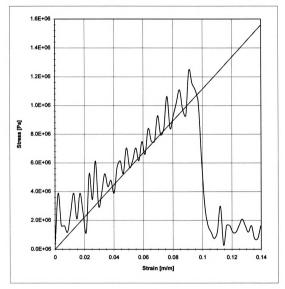


FIGURE E24 - Tensile test result for Y24 05



 Sample
 Thickness
 0.00292 [m]
 Elasticity
 1423046 [Pa]

 Width
 0.009 [m]

 Area
 2.63E-05 [m^2]

 Length
 0.02 [m]

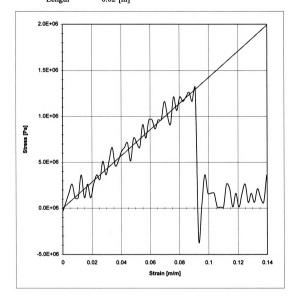
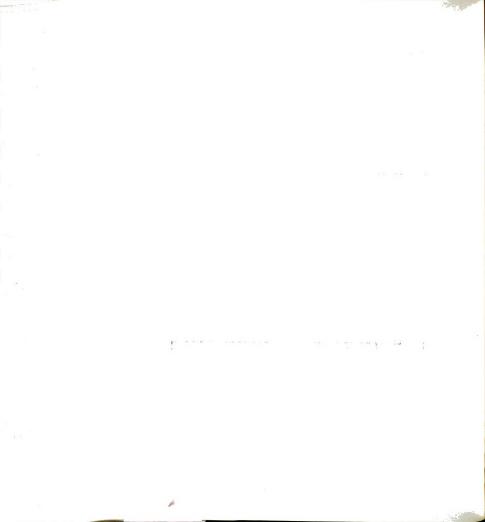


FIGURE E25 - Tensile test result for Y25_05



TENSILE TEST



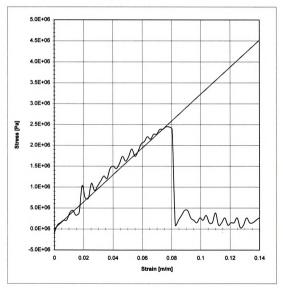
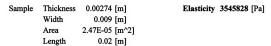


FIGURE E26 - Tensile test result for Y26_05

TENSILE TEST



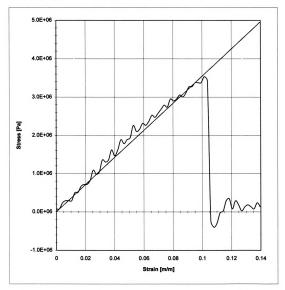
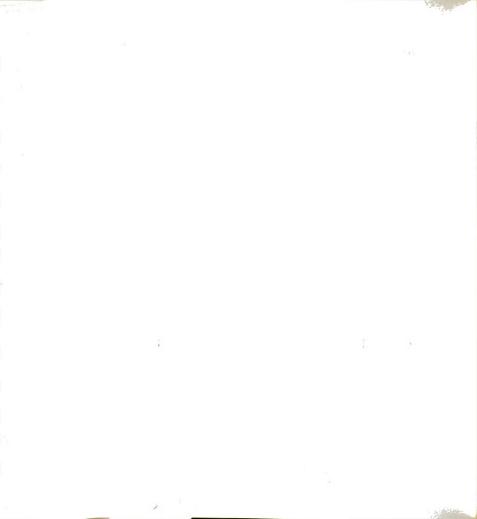


FIGURE E27 - Tensile test result for Y27_05





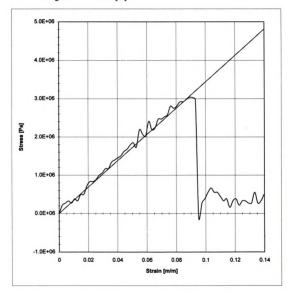


FIGURE E28 - Tensile test result for Y28 05

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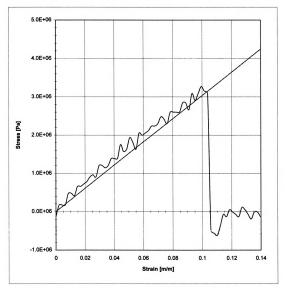


FIGURE E29 - Tensile test result for Y29 05





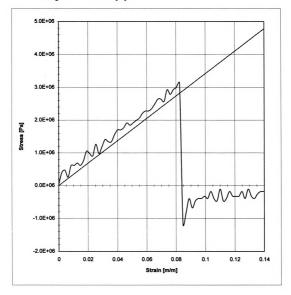
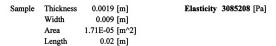


FIGURE E30 - Tensile test result for Y30 05





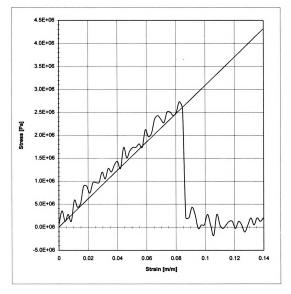
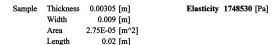


FIGURE E31 - Tensile test result for Y31 05



TENSILE TEST



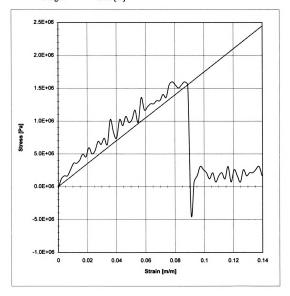


FIGURE E32 - Tensile test result for Y32_05

 Sample
 Thickness
 0.00228 [m]
 Elasticity
 2268548 [Pa]

 Width
 0.009 [m]

 Area
 2.05E-05 [m^2]

 Length
 0.02 [m]

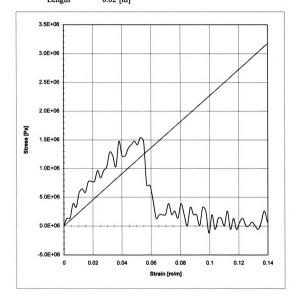
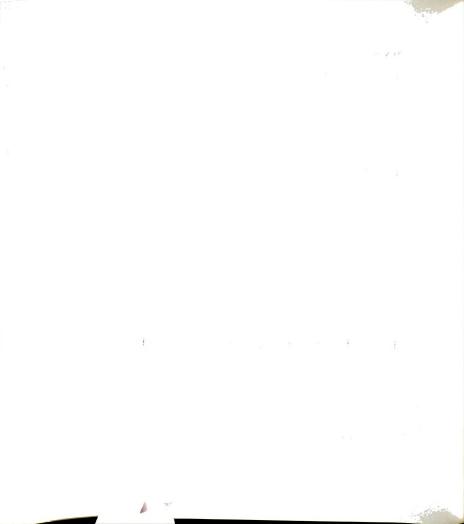


FIGURE E33 - Tensile test result for Y33 05



TENSILE TEST

Sample	Thickness	0.00267 [m]	Elasticity	3514630 [Pa]
	Width	0.009 [m]		
	Area	2.4E-05 [m^2]		
	Length	0.02 [m]		

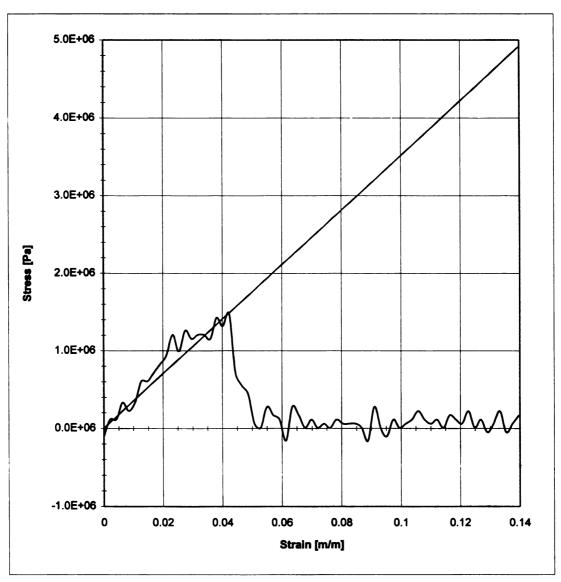


FIGURE E34 - Tensile test result for Y34_05



 Sample
 Thickness
 0.00272 [m]
 Elasticity
 2438245 [Pa]

 Width
 0.009 [m]

 Area
 2.45E-05 [m^2]

 Length
 0.02 [m]

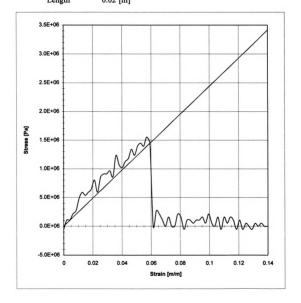


FIGURE E35 - Tensile test result for Y35 05

TENSILE TEST

Sample	Thickness	0.00209 [m]	Elasticity	1687081 [Pa]
	Width	0.009 [m]		
	Area	1.88E-05 [m^2]		
	Length	0.02 [m]		

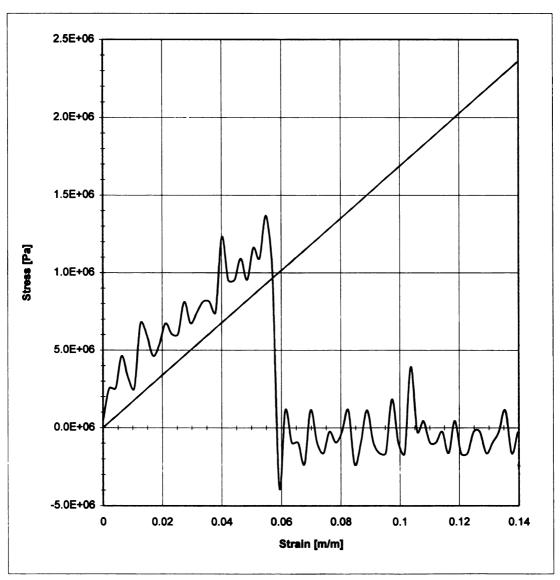
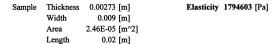


FIGURE E36 - Tensile test result for Y36_05





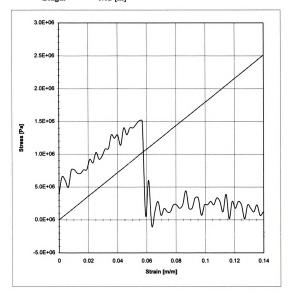


FIGURE E37 - Tensile test result for Y37_05



TENSILE TEST

Sample	Thickness	0.00282 [m]	Elasticity	2176547 [Pa]
	Width	0.009 [m]		
	Area	2.54E-05 [m^2]		
	Length	0.02 [m]		

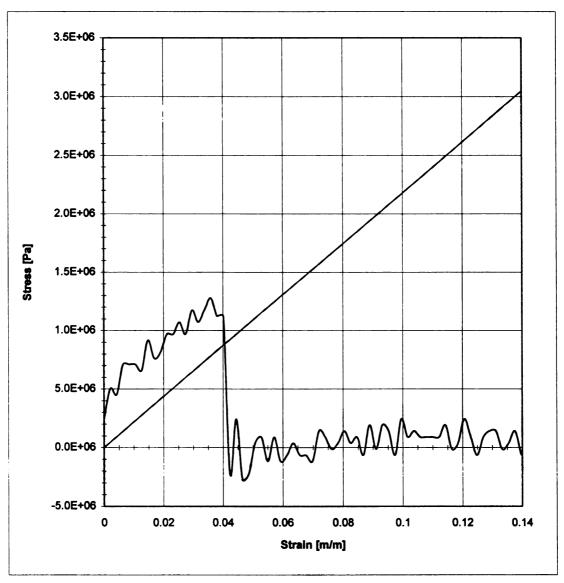


FIGURE E38 - Tensile test result for Y38_05





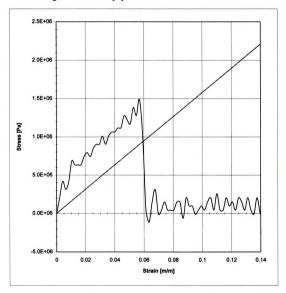


FIGURE E39 - Tensile test result for Y39 05

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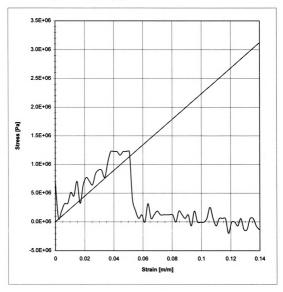


FIGURE E40 - Tensile test result for Y40_05



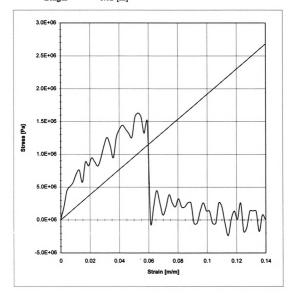


FIGURE E41 - Tensile test result for Y41_05



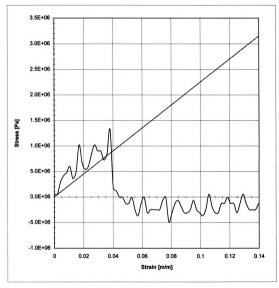
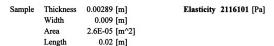


FIGURE E42 - Tensile test result for Y42 05

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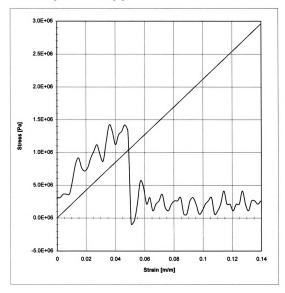


FIGURE E43 - Tensile test result for Y43_05

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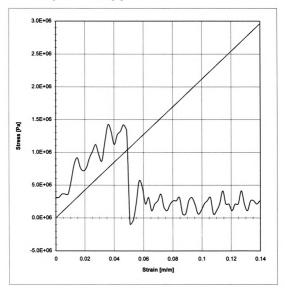


FIGURE E43 - Tensile test result for Y43_05



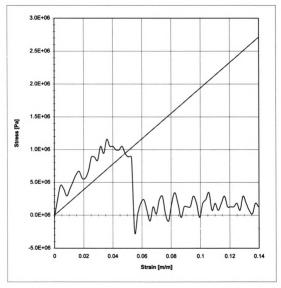


FIGURE E44 - Tensile test result for Y44 05

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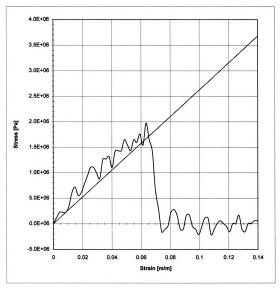
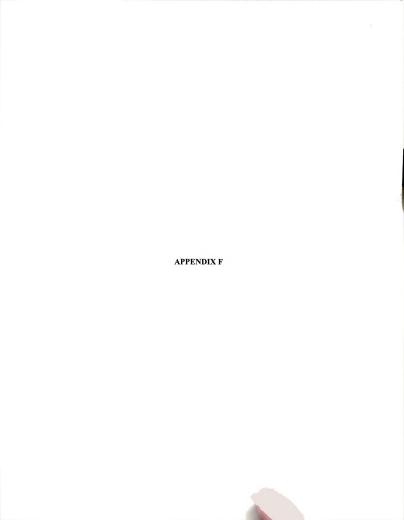


FIGURE E45 - Tensile test result for Y45_05



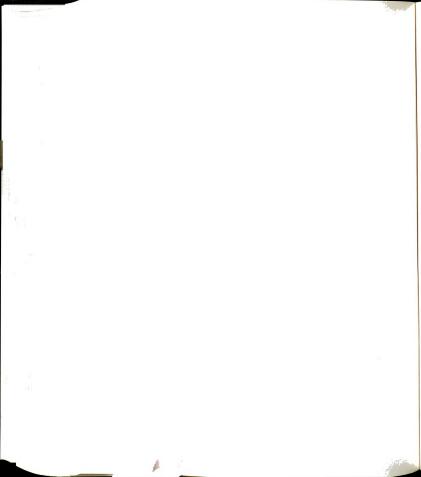




APPENDIX F

Computer Simulation Program

```
#include <stdio.h>
#include <stdlib.h>
#include <math.h>
#define V1
                 1
                           /* 1 in/min */
                           /* 10 in/min */
#define V10
                 10
#define r
                 0.997
#define N
                 100.0
       /* # of layers for grand simulation */
#define lambda
                 1
#define alpha
                 3.0
#define freq
                 100.0
                          /* sampling rate [Hz] */
#define Th
                           /* Thickness of unit layer */
                 0.0002
#define Area
                3.14*pow(0.011,2)/4 /* Area of the
plunger */
#define Area2 3.14*0.011*Th
double Vch1, Vch10;
float
      F[500].
iFC[500], iFS[500], FC[500], pFC[500], FS[500], U[500], iF[500], Fc
omp[500], Fshear[500];
float dtot[500], Frac;
float
       D max:
double d max, f max;
double kf s, kd s,pc,mo;
float
       Fvisco=9:
int
       m, mt;
 float ip[45][6]={ {2.36, 1.59, 338438, 0.79, 2219400,
0.42},
       {3.85, 1.10, 441721, 2.07, 2893126, 0.33},
       {4.20, 1.45, 443736, 1.28, 2661103, 0.40},
```



```
{3.32, 0.97, 335967, 1.72, 2019565, 0.41},
{3.99, 1.62, 428455, 1.58, 2603817, 0.39},
{4.35, 1.84, 446245, 1.32, 2186119, 0.41},
{4.35, 1.93, 473311, 1.52, 2644603, 0.37},
{3.91, 1.25, 352807, 1.25, 3603414, 0.37},
{3.41, 1.52, 331709, 1.22, 3262707, 0.35},
{3.49, 1.57, 278642, 1.72, 2382488, 0.26}.
{4.85, 1.65, 518319, 2.16, 3038228, 0.27}.
{3.11, 2.57, 434385, 1.55, 2365463, 0.38},
{4.14, 1.25, 476922, 1.68, 2669880, 0.44},
{4.87, 1.54, 506877, 1.39, 3140829, 0.28}.
{4.46, 1.58, 500491, 1.47, 3262553, 0.55},
{3.74, 3.10, 497526, 2.08, 3204485, 0.41},
{2.89, 1.66, 353833, 1.31, 2495790, 0.35}.
{2.74, 2.41, 358053, 1.63, 2556493, 0.41},
{1.83, 11.6, 364249, 1.11, 2106323, 0.42},
{2.90, 3.71, 396447, 2.00, 2878381, 0.38},
{2.50, 1.30, 334104, 0.71, 1488604, 0.41},
{2.90, 1.33, 388122, 1.01, 1718797, 0.43},
{3.09, 1.21, 379075, 0.90, 1910229, 0.44},
{1.98, 1.83, 282443, 0.62, 1113635, 0.48},
{2.02, 1.81, 284230, 0.78, 1423046, 0.35},
{5.17, 1.13, 444801, 1.46, 3227850, 0.38},
{5.54, 2.01, 631107, 2.03, 3545828, 0.43},
{4.49, 1.99, 469624, 1.39, 3437427, 0.30},
{6.72, 1.92, 653307, 2.20, 3034021, 0.45},
{4.69, 2.86, 554547, 2.03, 3415703, 0.40},
{3.94, 0.88, 355620, 1.23, 3085208, 0.29},
{3.29, 1.76, 412565, 0.83, 1748530, 0.30},
{3.56, 0.89, 300310, 0.89, 2268548, 0.34},
{3.92, 1.00, 348930, 0.77, 3514630, 0.33},
{3.55, 0.97, 288335, 0.92, 2438245, 0.27},
{2.95, 4.34, 437958, 1.08, 1687081, 0.51},
{2.50, 3.62, 396105, 1.10, 1794609, 0.55},
{3.05, 2.62, 421764, 1.02, 2176547, 0.44},
{3.04, 154, 513378, 1.15, 1578980, 0.42},
{3.34, 2.81, 426326, 1.16, 2231997, 0.38},
{2.17, 2.65, 331139, 1.01, 1916969, 0.31},
{2.79, 1.43, 306164, 0.81, 2252100, 0.33},
{2.38, 1.61, 300994, 0.93, 2116101, 0.33},
{3.32, 0.86, 313349, 0.85, 1938822, 0.32},
{2.31, 1.25, 313121, 0.90, 2625809, 0.43},
```



```
void main(void)
   int
          x, i, i, k, c, h,z,q,qc,v[30],s[3],mini;
   float pd[150],pt,A,tt,sFC,sFS,ppt,max,min,Fv;
   float a[150], as[150], t. dl. d[150];
   float ds[150], dss, dds, dx, temp, pop, force, pc;
   float Eu, Nu, SIGMAc, F max, Es1, ct;
   char
fn[45] = { "m: \m01", "m: \m02", "m: \m03", "m: \m04", "m: \m05", }
         "m:\\m06", "m:\\m07", "m:\\m08", "m:\\m09", "m:\\m10",
"m:\\m11","m:\\m12","m:\\m13","m:\\m14","m:\\m15","m:\\m16",
"m:\\m17", "m:\\m18", "m:\\m19", "m:\\m20", "m:\\m21", "m:\\m22",
"m:\\m23", "m:\\m24", \m25", \m26", \m:\\m27", \m:\\m28",
"m:\\m29", "m:\\m30", "m:\\m31", "m:\\m32", "m:\\m33", "m:\\m34",
"m:\\m35", "m:\\m36", "m:\\m37", "m:\\m38", "m:\\m39", "m:\\m40",
"m:\\m41", "m:\\m42", "m:\\m43", "m:\\m44", "m:\\m45"};
   FILE *destin, *M T;
   Vch1=V1*0.0254/60:
   Vch10=V10*0.0254/60:
   M T=fopen("m:\\MagT","wt");
for(x=0;x<=44;x++) {
  Eu=ip[x][0]*pow(10,10);
  Nu = ip[x][1] * pow(10.10):
    SIGMAc=ip[x][2];
     F max=ip[x][3];
       Es1=ip[x][4]:
        ct=ip[x][5];
   destin=fopen(fn[x],"wt");
   q=1;
   Fv=0:
a[1] = (1-r) *Vch10/(1-powl(r,N));
   for(i=0:i<=149:i++) {
```



```
ds[i]=0.0;
   Frac=SIGMAc*Area;
   dx=Vch10*(1/freq);
     ppt=0;
      pt=0;
       C=1;
       A=0;
      pc=c;
      qc=0;
   for(i=0;i<=499;i++) {
     iFC[i]=0;
     iFS[i]=0;
       U[i] = 0;
     pFC[i]=0;
   for(i=0;i<=N;i++) {
     pd[i]=0;
          }
    as[1] =a[1];
    for(j=2;j<=N;j++) {
      a[j]=a[j-1]*r;
      as[j]=a[j];
   for(i=0;i<=499;i++) {
    t=i*(1/freq); /* time */
    dtot[i] = t * Vch10;
   for(j=1;j<=N;j++) {
      d[j]=pd[j]+a[j]*(t-pt);
for(j=c;j<=N;j++) {
  /* tt=(dtot[i]-A*(c-1))/Vch10;*/
       FC[j] = Area*a[j] *Nu*(1-1/exp(Eu*((t-pt)+0.6*ppt)/Nu));
       if(j==c) FC[j] = Area*a[j] * (Nu*(1-0.06*c)) * (1-
1/exp(Eu*((t-pt)+0.6*ppt)/(Nu*(1-0.06*c))));
```



```
if(c==1) FC[j] = Area * a[j] * Nu * (1-1/exp(Eu * t/Nu));
   if(FC[i] >= (Frac*(1+0.03*(c-1)))) {
              y[c]=i;
      for(k=c+1;k<=N;k++) {
     pFC[k]=FC[k];
   pt=t;
         if(c==1) ppt=t;
   C+=1;
   FC[j]=0;
   d[i]=Th;
   A=Th:
   pd[j]=A;
   if((N-c+1)==0) goto SSS;
    a[c] = (1-r) *Vch10*(1-0.09*c)/(1-powl(r,(N-c+1)));
            for (k=(c+1); k<=N; k++) {
          a[k] = a[k-1] *r;
   for(k=c;k<=N;k++) {
       pd[k] = a[k] * (dtot[i] - A*(c-1)) / Vch10;
       d[k] =pd[k];
  goto FFF;
FFF:
     sFC=0:
  for(j=1;j<=N;j++) {
     sFC=sFC+FC[j] * (a[j] * (1/freq));
       iFC[i] = iFC[i-1] + sFC;
     if(i==0) iFC[i]=sFC:
      a[0]=Vch10;
      as[0]=a[0];
      sFS=0;
      z=1:
      for(j=q;j<=N;j++) {
```



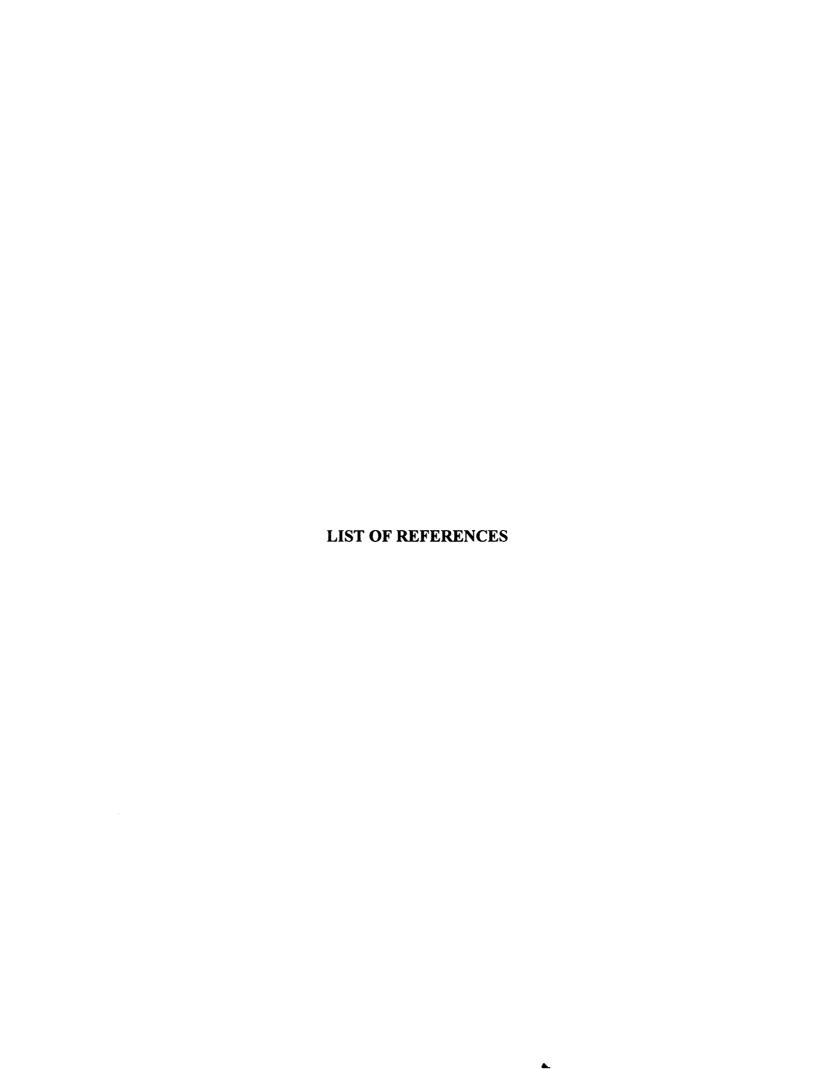
```
dss=0;
          dds=0;
  for(k=j;k<=N;k++) {
          if(i==0) z=0;
     dss+=a[k]*(1/freq)*z;
     dds+=a[k]*(1/freq)*z;
  if(qc!=q) { mo=0.6; }
  else { mo=1;}
        ds[j]=ds[j]*mo+dds;
force=(Es1*(ds[j]*0.7)/0.025)*Area2;
       if(force>F max) q+=1;
       sFS=sFS+dss*force;
pc=c;
iFS[0]=0;
if(i>0) iFS[i]=iFS[i-1]+sFS;
     U[i] = iFC[i] + iFS[i];
  SSS:
  ;
     F[0] = 0;
h=0;
      min=200;
    for (j=y[1]; j <= y[2]; j++) {
      if(F[j]<min) {</pre>
   minj=j;
   min=F[j];
  for(i=1;i<=498;i++) {
     Fcomp[i] = (iFC[i+1] - iFC[i-1]) / (2*dx);
     Fshear[i] = (iFS[i+1] - iFS[i-1]) / (2*dx);
   if(i>=minj) {
     h=1;
     Fv=h*(i-minj)*(1/freq)*Fvisco/(minj*0.01);
      if(Fv>=Fvisco) Fv=Fvisco;
```



```
F[i] = (U[i+1] - U[i-1]) / (2*dx) + Fv;
fprintf(destin, "%f %f %f %f
%f\n",i*(1/freq),iFC[i],iFS[i],U[i],F[i]);
  for(i=1;i<=2;i++) {
      \max=0:
    for (j = (y[i] - 10); j < = (y[i] + 10); j + +) {
      if(F[i]>max) {
   max=F[j];
   s[i]=j;
      \max=0;
  for(i=1; i <=189; i++) { /* 189: Depth of MT [8 mm] */
      if(F[i]>max) {
  max=F[i];
  mt=i:
  fprintf(M T, "%d %g %g %g %d
%g\n",x+1,F[s[1]],F[s[2]],F[s[1]]/(s[1]*0.01*Vch10),mt,F[mt]
) :
fclose (destin);
  printf("%2d\n",x+1);
fclose(M T);
```

^{*} This program was written by using C language.







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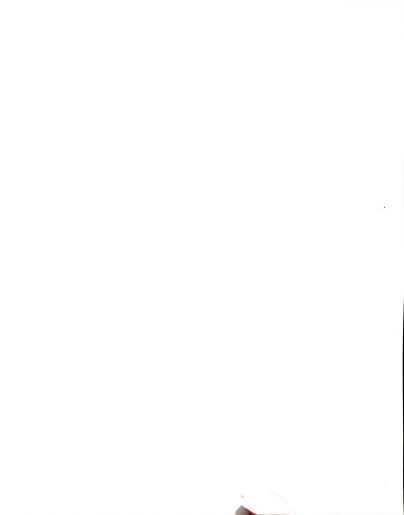
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