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THE MAXIMUM ALLOWABLE DEPTH OF APPLES IN A

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YOOSEF SHAHABASI

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# EXPERIMENTAL AND NUMERICAL TECHNIQUES RELATED TO THE MAXIMUM ALLOWABLE DEPTH OF APPLES IN A BULK STORAGE

By

Yoosef Shahabasi

### A DISSERTATION

Submitted to Michigan State University in partial fulfillment of the requirements for the degree of

DOCTOR OF PHILOSOPHY

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#### ABSTRACT

### EXPERIMENTAL AND NUMERICAL TECHNIQUES RELATED TO THE MAXIMUM ALLOWABLE DEPTH OF APPLES IN A BULK STORAGE

By

Yoosef Shahabasi

The objective of this study was to develop experimental and analytical techniques to determine the maximum safe depth for apples in a bulk storage. The problem was perceived to consist of two components. One component involved the determination of the contact forces in a bulk bin while the other component related to the determination of whether a specific loading would produce a bruise.

The mechanical properties of Jonathan apples before storage and two periods during storage were determined experimentally by compressing cylindrical specimens until a failure occurred. The modules of elasticity changed significantly between October 1 and November 15. A very small change occurred between November 15 and December 31, 1978. The respective averages were  $E_{OCT} =$ 3279 Kpa,  $E_{NOV} = 2516$  Kpa and  $E_{DEC} = 2360$  Kpa. The average maximum normal strain at failure was 0.14, 0.11 and 0.12 for three dates. The average normal stress at failure decreased from 444 Kpa on October 1 to 252 Kpa on November 15 and 235 Kpa on December 31. One hundred fifty apples were sampled on each date with four samples being removed from each apple.

The distributions of the elastic modulus and failure strain were used in a computer model to predict bruising for a particular load. The model was based on the assumption that a bruise occurs when the maximum normal strain exceeds a specified value. The load which produced a bruise was converted to a depth by assuming a single column stack for the apples.

Apple-to-apple contact was found to govern the allowable depth. The October 1 apples could be piled 5.14 meters without brusing but the November 15 and December 31 apples could be piled only about 1.8 meters. The significant decrease was attributed to the decrease in the modulus of elasticity between October 1 and November 15.

As existing finite element type computer model which had been used to model the contact forces between small diameter steel balls was modified for use with large diameter low modulus materials such as apples. The validity of the model was established by experimentally measuring the contact forces between 6 cm diameter rubber

balls which were stacked in a rhombohedral fashion. There were seven layers with either four or five balls in each horizontal layer. All of the contact forces differed by less than 20 percent from the values calculated using the computer model.

Approved: ofessor

Department Chairman

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### I. INTRODUCTION

Apples utilized by the processing industry are usually stored in stacks of bulk bins in the plant yard. Apples received early in the harvest season are subjected to moderate daily temperatures that hastens ripening which increases the shrink due to weight loss and spoilage. Apples harvested late in the season may suffer from freezing damage around the stacks despite some protection from straw or plastic covering.

Cold storage is being used with increasing frequency despite its cost, particularly by slice processors. The higher cost of storage and a trend to fewer and larger plants necessitates reducing the overhead by operating the plant for a longer period. One method of reducing the overhead is to construct less expensive types of storage which will protect the apples from excessive heat or freezing conditions.

The USDA agricultural research group at Michigan State University have been developing a totally integrated system of equipment for mechanically harvesting, transporting, and storing apples. A description of the bulk storage silo and the procedure for its loading and

unloading has been reported by Burton and Tennes (1977), and Tennes, et al. (1978). The system consists of:

a. A quick acting mechanical shaker designed to shake the trunks of trees spaced as close as 2.4 meters (8 feet) apart, while moving at 1.6 Km/h (1 mile/h).

b. A horizontally positioned reinforced fiber glass tank mounted on a heavy duty trailer with a fillingwell at the top center of the tank. The tank holds 8300 liters of water and is used to transport the apples, Figure 1.1.

c. A bulk silo storage system consisting of a storage facility, a reservoir that can provide a large amount of water and a conveyer and handling apparatus for the loading and unloading of fruit from the storage, Figure 1.2.

Recommended management practices for bulk storage includes the avoidance of loading warm fruit into the storage (this can be done by hydrocooling of fruits before storage) and controlling the temperature during storage by proper insulation and ventilation using cool night air. The depth of apples should also be such that little bruising occurs within the stack. The determination of the maximum safe depth for apples stored in bulk is a factor that has not been determined. A study of the



Figure 1.1 View of the Handling Tank System with a Conventional Apple Harvester.

loads acting on apples and the allowable depth for safe storage is needed.

The primary objective of this study was to develop some experimental and analytical techniques in determining the maximum depth for safe storage of Jonathan apples in a bulk bin. Specific objectives were:

- To study the changes in the mechanical properties of Jonathan apples during refrigerated storage and use these in a simulation model to predict the allowable storage depth.
- To study the contact force distribution and transmission in the bulk storage of large diameter spheres.



Figure 1.2.--Silo Storage Being Filled (Tennes, 1977).

#### II. REVIEW OF LITERATURE

## 2.1 Mechanical Injury

Bruising and injury to agricultural commodities during mechanical handling operations has been a problem of interest to agricultural engineers for several years. As a result, many investigations have been conducted to determine the mechanical behavior of agricultural products when they were subjected to various types of external forces. The increase in use of mechanical harvesting for agricultural products has generated a need for basic information on material properties.

Gaston and Levin (1951) reported an extensive study of causes of apple bruising in handling operations. Their study included the loading of apples under both impact and dead load conditions. For the impact test, apple samples were dropped from heights up to 24 inches onto various types of surfaces. Apples were subjected to an increasing dead load until the desired load of bruising has been achieved. They reported that a dead load of 8.5 pounds was required to produce a 3/8 inch diameter bruise in a 2.5 inch diameter McIntosh apple. No bruise occurred below 8.5 pounds. Above this load, the amount of bruising was proportional to the load applied. Their test included a very large number of apple samples and

the data reported were aimed more toward demonstrating the importance of minimizing loads on apples.

Mohsenin and Göhlich (1962) carried out a more intensive study to determine some of the important engineering parameters involved in mechanical damage. They studied apple sections under impact and static loads. The static test was conducted by applying various dead loads on the apples for 100 hours at 34°F. The energy required to bruise under an impact load was found to be roughly twice that required under a static load.

Fletcher et al. (1965) studied the effect of variable loading rates to determine trends of the rate characteristics, rather than the properties at isolated rates of loading. They wanted to correlate the mechanical properties at one rate of loading with those at another. Their study brought out some important relationships between slow and fast rates of loading but they did not investigate maximum allowable loads.

Hammerle and Mohsenin (1966) used a vertical drop tester for dynamic impact loading. The main objective of their study was to develop the apparatus and the method of testing.

Bittner et al. (1967) developed the concept of using a simple pendulum to simulate free fall of the fruit specimens. They used the energy balance theory to

evaluate the effectiveness of various cushioning materials based on rebound energy, energy absorbed by the cushion and energy absorbed by the apple.

Fridley and Adrian (1966) worked on resistance to mechanical injuries of apples. Their results, given in terms of compression yield force and impact yield energy, showed that in comparison with peaches, pears and apricots, apples had the least potential for mechanical harvesting.

Mattus et al. (1960) showed that drop heights more than six inches onto a hard surface produces internal bruise in pears which developed brown spots in the flesh of the fruit.

Location of the bruise has suggested that maximum shear stress can be a possible failure parameter (Fridley and Adrian, 1966). A dynamic triaxial compression test was conducted by Miles and Rehkugler (1971) at varying levels of compression stress, shear stress and axial strain rates. These investigators reported that shear stress was the most significant failure parameter.

Dal Fabbro (1979) concluded that the maximum normal strain is the primary factor causing the failure of apple flesh.

### 2.2 Stress Analysis in Fruits Under Loading

Knowledge of the stress distribution in fruits under static and impact load is limited because of the difficulty involved in determining material properties and the lack of analytical solutions valid for the irregular shapes involved. The load could be exerted from a flat body to the fruits or from one fruit to another.

Finney (1963) reported a significant difference existing between certain potato varieties in their response to applied surface pressure. Mohsenin and Gölich (1962) applied the same technique to apples, potatoes, pears and tomatoes concluding that the compression test appeared to offer the most promise of evaluation of mechanical behavior as related to bruising.

The most common type of loading that fruits are subjected to is the contact load which can produce a bruise. Contact forces occur in harvesting, handling and storage. Contact stresses are caused by the pressure of two bodies having a point (small area) contact: common type of contacts are the sphere and a plane, Figure 2.1 or two spheres, Figure 2.2. Boussineq (1885) solved the problem of concentrated forces acting on the boundary of a semi-infinite body. Timoshenko and Goodier (1970) discuss the contact problem as solved by Hertz. The maximum



Figure 2.1.--Pressure Distribution on the Contact Surface of Sphere as Subjected to a Flat Plate Under Load.



Figure 2.2.--Hertz Theory of Contact for Two Spherical Bodies in Contact.

pressure on two spherical bodies in contact is 1.5 times the average pressure on the surface of contact or

$$q_0 = \frac{3P}{2\pi a^2}$$
 (2.1)

Assuming that both balls have the same elastic properties and taking  $\mu$  = 0.3 the corresponding maximum pressure is

$$q_0 = 3/2 \frac{P}{\pi a^2} = 0.388 \sqrt[3]{PE^2 \frac{(R_1 + R_2)^2}{R_1^2 \cdot R_2^2}}$$
 (2.2)

| where | ٩ <sub>0</sub> | is | maximum pressure on the surface of contact |
|-------|----------------|----|--|
|       | а              | is | the radius of the surface of contact       |
|       | Р              | is | the applied load                           |
|       | E              | is | the modulus of elasticity of spheres       |
|       | <sup>R</sup> 1 | is | the radius of sphere one                   |
|       | $R_2$          | is | the radius of sphere two                   |
|       |                |    |  |

The first known application of Hertz solution for contact stresses in agricultural products is reported by Shpolyanskaya (1952) for determination of modulus of deformability of the wheat grain compressed between two parallel plates.

Finney (1963) used the Boussinesq solution for concentrated forces acting through a rigid die for potato, apple, portions of corn kernels, peaches and pears. The Hertz and Boussinesq techniques have also been applied to apples (Morrow and Mohsenin, 1966). McIntosh apples were subjected to constant load (creep test) or constant deformation (stress relation test) by a flat rigid plate. In the case of McIntosh apples loaded with a 1/8 inch cylindrical rigid die, they concluded that the stress was approximately zero at a depth of two inches below the surface of the fruit.

Fridley, et al., (1968) applied Hertz and Boussinesq theories to obtain force-deformation curves for peaches, pears and apples. They showed that the bruise in an apple usually occurs under the center of the area of contact at a small distance beneath the surface of the fruit. A flat plate loading was used in the study.

Agricultural products are generally viscoelastic. Viscoelasticity comprises an irreversible energy transformation where the relation between stress and strain is governed by time effect. During the early experiments on mechanical behavior of fruits and vegetables, it was observed that force deformation relations includes the time effect (Finney, 1963; Mohsenin, 1963; Timbers et al., 1966). Experiments conducted on McIntosh apples showed that apple flesh behaves as alinear viscoelastic material (Morrow and Mohsenin, 1966). Latter Chappell and Hamann (1970) studied the viscoelastic behavior of apple flesh,

but they found the material properties to be somewhat stress dependent and thus could not be characterized as linear. Hamann (1967 and 1970) also noted the nonlinear properties in apple flesh but he solved the apple impact problem for stress at the surface and in the interior of the apple considering apple flesh as a linear-viscoelastic material. In the later studies, the finite element method was used to determine stresses in apples resulting from contact with a flat plate.

Apaclla (1973) considered the apple as an elastic material and used the finite element method. Rumsey and Fridley (1974) used the finite element method assuming a linear viscoelastic shear modulus and an elastic bulk modulus for the material. De Baerdemaeker (1975) considered a material with time dependent bulk modulus and shear modulus to obtain the creep deformation and the stress distribution of a sphere in a contact with a flat rigid plate using the finite element method. He concluded that apples subjected to contact creep loads experience maximum stresses at the initial application of the force. Sherif (1976) solved the quasi-static contact problem for nearly incompressible agricultural products using finite element method.

Other theoretical studies involving vegetative materials includes Gustofson (1974) who obtained a

numerical solution to the axisymmetric boundary value problem for a gas-solid-liquid medium and Murase (1977) who developed stress-strain constitutive equations containing parameters necessary to describe the mechanical behavior of vegetative material including the water potential term.

#### 2.3 Criteria for Maximum Allowable Load

One of the major reasons for studying the mechanical properties of fruits and vegetables has been to determine the maximum allowable load to which these materials can be subjected without causing objectionable damage.

Limited work has been done to understand the mechanics of bruising in fruits and vegetables. A recent work using a compression test on tissue specimens reported that shear stress is the most important failure parameter (Miles, 1971). The question of how the maximum allowable shear stress can be determined for a whole fruit was not answered in this work. The concept of "Bioyield Point," indicating initial cell rupture in whole fruits such as apples and pears, has been proposed as the criterion for maximum allowable load that fruit can sustain without showing any visible surface bruising (Mohsenin, 1962 and 1965). In the case of apples the bruise is immediately below the skin and in most cases can be removed in the peeling process. In fruits such as peaches, bruising and tissue failure can occur some distance below the skin and the damaged protion can be seen only in canned products by the consumers. In these cases, shear strength has been taken as the maximum allowable load (Horsfield, 1970).

In recent studies the application of the theory of elasticity and importance of the modulus of elasticity of the fruit in single and multiple impacts suggested that the failure was due to excessive internal shear stresses. It is shown that maximum shear stress is proportional to (a) the energy of fall, (b) the moduli of elasticity of the fruit, and (c) the radii of the fruit and the impact surface (Horsfield, 1970). Nelson and Mohsenin (1968) have determined a relation between bruise volume and load. They report that bruises caused by dynamic loads are larger than those caused by equivalent quasi-static loads. Fletcher et al. (1965) reported that energy force and deformation to repture first decreases with increasing loading rate, then increases. Mohsenin (1971) reported that, the significance of certain viscoelastic properties of fruits and vegetables may not be understood but we should be ready to use such data as the maximum allowable load that these products can sustain under impact, dead loads, and vibration in designing handling systems. He also mentioned that these data should be available in the form which can be used by engineers.

Dal Fabbro (1979) studied the strain failure of apple material. Cylindrical and cubic apple specimens were subjected to uniaxial, biaxial and triaxial state of stress. Linear elastic and viscoelastic material properties were used to calculate the stress and strain components within the apple flesh.

He reported that in the uniaxial loading of cylindrical specimens the normal stress at failure varied for different strain rates. Triaxial loading of cylindrical specimens indicated that maximum shear stress and normal stress at failure vary for different levels of cylindrical stress. He also reported that the uniaxial, biaxial and rigid die loading of cubic and cylindrical specimens discards the maximum normal stress failure criteria.

Experimental results from his studies indicate that the maximum normal strain at failure remains relatively constant for all the loading situations. The most significant conclusion of his research was that apple material fails when the normal strain reaches a critical value.

## 2.4 Mechanical Properties of Granular Systems

Granular systems consist of cohesionless particles where the individual grains are independent of each other except for frictional interaction and geometric

constraints resulting from the particular type of packing. The two most important properties of granular materials are their strength and compressibility characteristics. The component particles in a granular system may be of any size from the smallest diameter of  $10\mu$  (like powder) to pebbles, cobbles, or even boulders (several inches in diameter) (Brown, 1970; Farouki, 1964).

Many studies of the packing of solid particles have been based on spherical or near spherical particles. Graton and Fraser (1935) discussed the geometry of various assemblages of discrete, ideal spheres. Also systematic arrangement of spheres in connection with the flow of water through soil was first studied by Slichter (1899).

# 2.4.1 The Arrangement of the Particles in a Stack

The stacking arrangement of the particles in a mass of material determines the points of contact between the particles and the direction of the normal at contact points; this essentially establishes the force system which acts between the particles.

### 2.4.1.1 Systematic arrangement of uniform

<u>spheres</u>. Ideal spheres may be packed in ordered layers of various types definable by the angle of intersection of the set of rows in the layer. Layers in which sets of rows have angles of intersection of any value between the limiting values of 60° and 90° are possible. Since square and simple rhombic layers (Figure 2.3) represent the limiting types of systematic packing, only these two are considered.



Figure 2.3.--The Angle of Intersection of the Sets of Rows in the Layer.

Three different systems may be formed by stacking square horizontal layers one above another.

- Case 1. A simple rectangular system; each sphere has its center vertically above that of the sphere below (Figure 2.4).
- Case 2. The orthorhombic system results when the center of the upper sphere is offset a distance R in the direction of one of the rows (R is the radius of the spheres) (Figure 2.5).

SQUARE LAYER



Figure 2.4.--A Simple Rectangular System.



Figure 2.5.--The Orthorhombic System.
To get the vertical distance, d, between the center of a sphere in a row whith the one above or below it, consider triangle  $C_1^0 C_3^-$  (Figure 2.5) which can be written:

- $R^{2} + (d + R)^{2} = 4R^{2}$  $d^{2} + 2dR 2R^{2} = 0$  $d = \frac{-2R + 2\sqrt{3}R}{2} \text{ or } d = -R + \sqrt{3}R = 0.73R \quad (2.3)$
- Case 3. Rhombohedral system. In this case each sphere is in contact with four spheres below, four above, and four in the same layer (Figure 2.6).



PLAN

END ELEV.

Figure 2.6.--Rhombohedral System.

To get the projectional distance between two spheres in two layers, consider the pyramid of  $C_1 C_2 C_3 C_4 C_5$ with the trianglar side of  $C_1 C_3 C_5$  (Figure 2.7).

Knowing y in this triangle which is:

$$y^{2} = 4R^{2} - R^{2}$$
$$y = \sqrt{3} R$$

will give x as

or

$$X^2 = 3R^2 - R$$
 or  $X = \sqrt{2} R$  (2.4)



Figure 2.7.--Pyramid and Its Triangler Side.

The rhombohedral system is the most important theoretically, and usually is the basis for calculations. It is also the most important from a practical viewpoint, because it gives the densest state.

The three packing configurations discussed below may be formed by stacking simple rhombic layers one above another.

- Case 4. When in the orthorbombic system, the spheres of the next rehombic layer are placed in such a way that the center of each sphere lies vertically above the sphere below it (Figure 2.8).
- Case 5. There is a rhombic layer at the bottom and each sphere in the next rhombic layer rests in the cusp between two spheres in the layer below (Figure 2.9).

The distance d is obtained by considering the triangle  $C_1C_2C_3$ , giving d = R  $\sqrt{3}$ .

Case 6: This system is similar to case three except each layer from the top is in rhombic form (Figure 2.10).

To find distance d consider pyramid  $C_1 C_2 C_3 C_4$ , distance y can be calculated by considering triangle  $C_4 C_3 C_2$ , then y would be  $y = \sqrt{3} R$ .

Having y, d can be calculated easily from triangle  $\rm C_4OH$  as:

$$d^{2} = y^{2} - \frac{1}{9}y^{2}$$
 or  $d = \frac{2\sqrt{2} R}{\sqrt{3}}$ 

The angle which the side wall makes with the bottom influences the packing formation (Faruki and Winterkorn, 1964).



Figure 2.8.--A Rehombic Layer above Another.



Figure 2.9.--A Rhombic Layer above the Cusp of Another.



Figure 2.10.--A Rhombohedral System with a Rhombic Form on the Top.

A 90° angle will favor cases 1, 2, 4, and 5, a 60° or 120° angle favors cases 2 and 3. The packing is also influenced by the angel which the side walls make with each other. A 90° angle favors formation of square pattern and hence cases 1, 2, and 3. Intersection of the side walls at 60° with themselves and at 90° with the bottom favors cases 4 and 6.

The walls of the container give rise to a wall effect which causes the porosity in the vicinity of the wall to be greater than that in the body of the packing. This has been studied by Furnas (1929) who obtained an expression for the voids,  $V_W$ , present in a ring at the wall of area  $\pi d \cdot D/2$ 

$$V_{W} = \{V + K (1 - V)\} (\frac{1 + 2 Kd}{D}) - \frac{2Kd}{D}$$
(2.5)

where d is the diameter of the particle.

D is the diameter of container.

V is the voids present in the interior.

K is an experimental factor found to be 0.3.

The wall effect increases as the ratio d/D decreases.

## 2.4.2 Systematic Packing of Spheres of Different Sizes

In 1934 Horsfield calculated the decrease in porosity resulting from the insertion into the voids of the rhimbohedral system of successive spheres just large enough to fill the voids. The spheres filling the concave cube voids are termed secondary spheres, whereas those filling the concave-tetrahedron voids are tertiary spheres. The spheres which are inserted in the largest voids left after the secondary and tertiary spheres are called the quanternary spheres. Table 2.1 shows the type of spheres, their radius and porosity for each sphere.

TABLE 2.1.--Effect on Porosity of Spheres Inserted in Voids of Rhombohedral System (Horsfield, 1934)

| Sphere Type | Radius <sup>a</sup> | Number of Spheres | Porosity |  |
|-------------|---------------------|-------------------|----------|--|
|             | 1                   |                   | 25.05    |  |
| Primary     | 1                   | l                 | 25.95    |  |
| Secondary   | 0.4142              | 2                 | 20.69    |  |
| Tertiary    | 0.2247              | 2                 | 19.01    |  |
| Quaternary  | 0.1766              | 8                 | 15.74    |  |
| Quinary     | 0.1163              | 8                 | 14.81    |  |
| Filler      | 0.000               | -                 | 3.84     |  |

<sup>a</sup>Primary radius = 1.

It should be noted that it is practically impossible to attain a system packed in such a manner.

Hudson (1949) imagined the voids of the rhombohedral system filled with S spheres of equal radii, r, arranged in cubic symmetry. The densest state was obtained when each concave-cube void contained 21 spheres with r = 0.1782R (in which R is the radius of the primary spheres).

### 2.5 Contact Theory

Most of the theoretical considerations reported in the soil mechanics literature regarding the effect of particle size in granular systems are based on Hertz's contact theory. Since this theory has also been used to evaluate the stresses in fruits and vegetables, it is appropriate to discuss the primary results of this theory.

There are two types of contacts which can occur between apples, apple-to-apple contact and an apple in contact with a flat surface, Figure 2.11. Hertz's equations are different for these two situations.

Heinrich Hertz (1896) proposed a solution for contact stresses in two elastic isotropic bodies touching each other. He attempted to find answers to such questions as the distribution and magnitude of the surface of pressure, and the approach of the center of the bodies under the pressure.

Hertz started by assuming that the contacting solids are isotropic and linearly elastic, and also that the representative dimensions of the contact area are very



Figure 2.11.--Two Cases of Exerted Load on the Bottom Spheres.

small compared to the radii of curvature of the deformed bodies. This leads naturally to two further simplifying assumptions. Near the contact zone, it is considered sufficiently accurate to represent the actual shape of the two nearly flat bodies by general surfaces of the second degree

$$Z = Ax^{2} + By^{2} + Hxy$$
 (2.6)

in which A, B and H are constants depending on the magnitudes of the principal curvature of the surfaces in contact. Hertz also takes the bodies to be "flat" enough, in the neighborhood of the contact surface to be treated by the powerful analytical methods available for the semi-infinite solid, or body bounded by plane. The results obtained by Hertz are summarized below and are found in Shigley (1977).

When two solid circular spheres of diameter  $D_1$  and  $D_2$  are pressed together with a force F, a circular area of contact of radius a is obtained.

Specifying  $E_1, \mu_1$ , and  $E_2, \mu_2$  as the respective elastic constants of the two spheres, the radius a is given by

$$a = \left[\frac{3F}{8} \frac{\left[(1 - \mu_1^2)/E_1\right] + \left[(1 - \mu_2^2)/E_2\right]}{(1/D_1) + (1/D_2)}\right]^{1/3}$$
(2.7)

The maximum pressure occurs at the center of the contact area and is

$$q_0 = -\frac{3F}{2\pi a^2}$$
 (2.8)

Equations (2.7) and (2.8) are perfectly general and can be applied to contact with a plane surface by setting either  $D_1$  or  $D_2$  equal to infinity.

The stress components for any point along the Z axis, the axis of symmetry, are

$$\sigma_{rr} = \sigma_{\theta\theta} = q_0 [-(1 + \mu)[1 - \Psi Tan^{-1}(\frac{1}{\Psi})] + \frac{1}{2} (1 + \Psi^2)^{-1}]$$

$$\sigma_{zz} = -q_0 (1 + \Psi^2)^{-1}$$
 (2.9)

where  $\Psi = z/a$ 

These normal stress components are the principal stresses since the shear stress components are always zero along an axis of symmetry. The maximum shear stress is given by

$$\tau_{\max} = \left| \frac{\sigma_{\max} - \sigma_{\min}}{2} \right|$$
(2.10)

It occurs below the surface and is approximately 0.31  $q_0$ when  $\mu = 0.3$ . The ratio of the stress components  $\sigma_{rr}$ ,  $\sigma_{\theta\theta}$ ,  $\sigma_{zz}$  and  $\tau_{max}$  to  $q_0$  as related to the distance from the surface of contact is shown in Figure 2.12.



Figure 2.12.--Stress Distribution Within an Elastic Sphere with Poissan's Ratio = 0.3 Compressed with a Flat Plate.

#### 2.6 Contact Forces Between Granular Particles

The load distribution in a granular material has been studied by many people but only a few have suggested equations for determining the contact forces between individual particles. Those studies which appeared to apply to this study are discussed briefly herein.

Ross and Isaacs (1961) formulated a theoretical approach to estimate the forces acting on individual particles in a particle stack. The particles were assumed to be perfect inelastic spheres represented by a certain density and characteristic diameter. They used a simple rhombic stacking model and analyzed the forces acting in such a stack. They generalized equations to predict horizontal and vertical forces on any particle in a stack which were independent of the coefficient of friction of the material, Figures 2.13 and 2.14. They found the total vertical forces exerted by a particle on its support to be

$$F_{tv-n} = \overline{L}_{\theta} W/\overline{d}$$
 (2.11)

where  $\overline{L}_{\theta}$  is the average length of the four axis. W the weight of each particle.  $\overline{d}$  the characteristic diameter.



Figure 2.13. Section View of a Model Particle Stack (Ross-Isaccs)



Figure 2.14.- Forces Acting on a Particle in the Particle Stack (Ross-Isaccs)

Marsal (1963) proposed the use of the following semi-empirical equation to calculate the contact forces between the particles:

$$\overline{P} = \left(\frac{\pi r_v (1 + e)}{3^{1/4} nc}\right)^{2/3} d^2 \sigma \qquad (2.12)$$

where r, is the shape factor.

- e is the void ratio.
- nc is the average number of contacts per particle.
- d is the grain diameter.
- $\sigma$  is the confining pressure.

Keck and Goss (1965) suggested the shape factor as the ratio of geometric mean diameter to the diameter of equivalent sphere.

A study was made by Davis (1974) on the compressibility of and force transmission in a granular material, modeled as a two-dimensional random packing of spheres in elastic contact. He assumed this problem analogous to that of a stochastic planar graph, the nodes of which represent centers of the spheres and the branches, contacts between adjacent spheres. Davis considered the planar graph as an elastic structure and solved the force transmission in a two-dimensional array of randomly packed small steel spheres.

## III. ANALYSIS OF THE PROBLEM

The interest in the bulk storage of apples for processing has raised the question of the allowable height to which apples can be piled before causing objectionable damage. The objective of this dissertation is to answer this question.

It has been established by Dal Fabbro (1979) that apple flesh fails when the maximum normal strain exceeds a specific value. The maximum normal strain in a spherical body can be calculated using Hertz's contact theory. This theory shows that the maximum strain is a function of the modulus of elasticity of the body.

There is no reason to believe that the modulus of elasticity of apples remains constant with time. The first part of this study is concerned with determining whether, and if so, how the elastic modulus varies during the storage period. The change in the failure strain and the failure stress must also be studied. Once the distributions of the elastic modulus and failure strain are known, it will be possible to predict whether a force of a specific magnitude will cause a bruise.

The other aspect of the bulk storage problem is the contact force which occurs between the apples. A bulk bin contains too many apples to analyze all of the contact forces even though a computer model for such analysis does exist. This model, however, was developed for small diameter steel balls. The model should be modified to accommodate large diameter, low modulus balls and checked to determine whether it is still valid. The second part of this study is concerned with this modification and check.

# IV. CHANGES IN THE MECHANICAL PROPERTIES OF APPLE FLESH DURING COLD STORAGE

## 4.1 General Remarks

Fruits and vegetables are living organisms which undergo all physiological and pathological processes associated with life. To sustain physiological activities, they draw energy from the food reserves stored within them prior to harvest. This process causes deterioration in the commodities. Deterioration of fresh produce also results from other things, including physiological break down, physical injury to tissue, moisture loss, and invasion by microorganisms. Some, if not all, of these processes can be reduced by placing the produce in a cold storage with proper temperature and humidity conditions. The optimum storage temperature for most varieties is -1°C to 0°C (30°F to 32°F) at 90 percent relative humidity. The normal storage period for the Jonathan variety is two to three months with a miximum of five to six months.

No information is available which shows how the mechanical properties, particularly the elastic modulus and the failure strain, varies with the time period in

cold storage. A study to obtian this information is reported in this section.

# 4.2 Experimental Study

#### 4.2.1 The Test Fruit

Jonathan apples grown during the 1978 harvest season at the Horticultural Farm on the Michigan State University campus were used. The apples were picked on September 29. Eight bushels of 6.35-6.73 cm apples were obtained. Six bushels of these were placed in plastic bags and then in wooden crates and stored in a refrigerated storage. A storage temperature of 1.6-2.2°C (35-36°F) was provided. The other two bushels of apples were taken to the laboratory for immediate testing.

## 4.2.2 Specimen Preparation

Cylindrical specimens with a height of 1.27 cm (0.5 inch) and a diameter of 1.27 cm and cross-sectional area of 1.266 cm<sup>2</sup> (0.196 in<sup>2</sup>) were prepared by driving a corkborer into the apple parallel to the stem calyx axis. The specimen was then put in a cylindrical hole in a plexiglass bar and the ends were cut parallel to the face of the bar by using a sharp blade. All of the apples which were stored at 2.2°C were removed from storage 24 hours prior to testing and allowed to come to room temperature before preparation.

## 4.2.3 Uniaxial Loading of Cylinderical Specimens

The prepared cylindrical specimens were placed at the center of the load cell of an Instrom TM model testing machine and compressed using a deformation rate of 1.27 cm per minute until a failure occurred. Failure was defined as a discontinuity of the deformation curve as shown in Figure 4.1.



Deformation

Figure 4.1.--The Failure Point on the Load Deformation Diagram.

Compression tests were performed on three different dates, October 1, November 15 and December 31. These were denoted as Groups I, II and III. Each group consisted of 150 apples and four samples were removed from each apple. The failure strain,  $\varepsilon_{\rm f}$ , the stress at failure,  $\sigma_{f}$ , and the elastic modulus, E, were determined for every sample.

## 4.3 Experimental Results and Discussion

Tables Al, A2, and A3 of Appendix A give the mean values of the three different parameters ( $\varepsilon_{\rm f}$ ,  $\sigma_{\rm f}$  and E) for each of the 150 apples tested for deformation rate on each date. Table 4.1 summarizes the mean value and standard deviation of these parameters for three different groups of Jonathan apples.

TABLE 4.1.--Mean Values and Standard Deviation of the Mean of Three Different Elastic Parameters for 150 Jonathan Apples (1978)

| Group                             | Octo | ober 1   | Nove | ember 15 | Dece | ember 31 |
|-----------------------------------|------|----------|------|----------|------|----------|
| Strain at<br>Failure              | 0.14 | (±0.005) | 0.11 | (±0.009) | 0.12 | (±0.008) |
| Stress at<br>Failure<br>(Kpa)     | 444  | (±30)    | 252  | (±26)    | 235  | (±14)    |
| Modulus of<br>Elasticity<br>(Kpa) | 3279 | (±260)   | 2516 | (±188)   | 2360 | (±262)   |

( ) Indicates Standard Deviation of the mean of each apple.

The average modulus of elasticity value for the three storage periods is shown in Figure 4.2. The modulus





of elasticity changed significantly between October 1 and November 15. The decrease in the modulus value was not as much between November 15 and December 31.

Figure 4.3 shows the failure stress had a very sharp decrease in the first 1.5 months, but did not decrease with the same rate for the next 1.5 months.

The failure strain as shown in Figure 4.4 decreased in the first 1.5 months and increased slightly in the next 1.5 months. A statistical investigation  $(\chi^2 \text{ test})$  indicated that the modulus of elasticity and the failure stress depend on the storage period. Failure strain did depend on the storage period for the first 1.5 months but not for the second 1.5 months.

The distribution of the elastic modulus, failure stress and failure strain during the storage period is shown in Figure 4.5 through 4.7. The higher modulus of elasticity and the higher failure strain values for the freshly picked apples indicates a higher resistance to bruising. The changing of the modulus and failure strain values toward smaller values as storage time increases shows a lower resistance of apple material to bruising.



















### V. BRUISE MODEL; ALLOWABLE DEPTH

### 5.1 Maximum Normal Strain

The bruise model used here to predict the allowable depth for bulk stored apples is based on the failure strain criteria developed by Dal Fabbro (1979). It states that apple flesh fails when a maximum normal strain exceeds a critical value.

The stress components for any point along the z axis were given in Chapter II (2.9). Since failure occurs because of the maximum normal strain, an equation for the maximum normal strain,  $\varepsilon_{zz}$  is needed. Assuming a homogeneous, isotropic material, Hooke's law gives

$$\varepsilon_{zz} = \frac{1}{E} \left[ \sigma_{zz} - \mu \left( \sigma_{rr} + \sigma_{\theta\theta} \right) \right]$$
 (5.1)

Substituting (2.9) produces

$$\varepsilon_{zz} = \frac{(1 + \mu)q_0}{E} [2\mu[1 - \Psi Tan^{-1}(\frac{1}{\Psi})] + (1 + \Psi^2)^{-1}]$$
(5.2)

The most common contact which occurs in a bulk storage is apple-to-apple contact which can be modeled as the contact between two spheres. To simplify the problem, it was assumed that all apples have the same diameter, 0.0625 m, and a Poisson's ratio of 0.35. The modulus of elasticity and failure strain were treated as random variables.

Using the assumptions on diameter and Poisson's ratio, the radius of contact, (2.7) become

$$a = 0.21745 F^{1/3} \left[ \frac{E_1 + E_2}{E_1 + E_2} \right]^{1/3}$$
(5.3)

- - -

while 
$$q_0 = 10.097 F^{1/3} \left[ \frac{E_1 + E_2}{E_1 E_2} \right]^{-2/3}$$
 (5.4)

The modulus of elasticity values  $E_1$  and  $E_2$  in (5.3) and (5.4) are for the two apples in contact.

Equation (5.2) gives the maximum strain as a function of  $\Psi$  which is equal to Z/a. Given a load F and the modulus of elasticity of each apple, the radius a, (5.3), can be calculated as well as  $q_0$ , (5.4). Knowing a and  $q_0$ ,  $\varepsilon_{zz}$  can be calculated for any value of z.

Theoretically the location of the maximum value of  $\varepsilon_{zz}$  can be obtained by differentiating (5.2) with respect to z, setting the resulting equation to zero and solving for z. The resulting equation, however, is not easily solved and it was decided to calculate  $\varepsilon_{zz}$  for several loadings and moduli values to see how it behaved (see

Appendix D). Examples of some curves for  $\varepsilon_{zz}$  are shown in Figure 5.1. The average location of the maximum value was 3.65 mm (3.5 - 3.8 mm) for the October 1 data and 3.0 mm (2.8 - 3.2 mm) for the other two sets of data. These values of z were used in the simulation model discussed in the next section because the value of  $\varepsilon_{zz}$  at 3.65 mm or 3.0 mm differs very little from the maximum value when these values of z are not right at the location of  $\varepsilon_{zz}$  maximum.

The apples at the bottom of the bin are in contact with a flat hard surface which could be either steel or concrete. In this case  $E_1$  for the steel (or concrete) is much greater than  $E_2$  while  $R_1$  is infinity. These properties modify the equations for a and  $q_0$  to

$$a = \left[\frac{F(1715278 + 8.6 E_2)}{82740000E_2}\right]^{1/3}$$
(5.5)

and 
$$q_0 = 0.4774 F^{1/3} \left[ \frac{1715278 + 8.6 E_2}{82740000E_2} \right]^{-2/3}$$
 (5.6)

The location of  $\varepsilon_{zz}$  maximum for this type of contact was determined in the same manner as for appleto-apple contact. The location of the maximum values were at z = 3.88 mm (3.55 - 4.21 mm) for the October 1 data and 3.0 mm (2.88 - 3.12 mm) for the other two dates.



# 5.2 Calculation of the Allowable Storage Depth

The calculation of the allowable storage depth was carried out using (5.2) while treating  $E_1$ ,  $E_2$  and the failure strain of the pair of apples (or apple in the case of flat plate contact) as a random variable. A normal load F was selected. A random generator was used to select the values of  $E_1$ ,  $E_2$ ,  $\varepsilon_{f1}$ , and  $\varepsilon_{f2}$  from the data discussed in Chapter IV. The maximum strain in each apple was calculated. If this strain exceeded the failure strain for the apple, the apple was said to be bruised. This calculation was repeated 500 times for each loading. A percent of apples bruised was then calculated. The normal force was increased and the process repeated. The equivalent depth was calculated assuming a single column stack where each apple weighed 0.85 N (Appendix B).

## 5.3 Results and Discussion

The percent of apples with bruise for each storage group are shown in Figures 5.2 and 5.3. It is immediately obvious that fresh apples can be piled much deeper than apples which have been in storage 1.5 to 3 months. The fresh apples have a significantly larger modulus of elasticity value which means that it takes more force to produce a fixed amount of deformation. There is not an appreciable difference for the results for apple-to-apple



Figure 5.2.--Height and Percent Bruise Relationship for Jonathan Apples in a Bulk Storage (Case I).



Figure 5.3.--Height and Percent Bruise Relationship for Jonathan Apples in a Bulk Storage (Case II).

contact and the results for apple-to-flat surface contact.

In Figures 5.2 and 5.3 it can be seen that curve for Group III apples is above that of Group II because the value of strain at failure as the failure criterion for Group II was lower than Group III (Figure 4.4).

The curves in Figures 5.2 are used as follows. If an individual wants to store Jonathan apples for three months and is willing to accept 10 percent bruising, these apples can be stored to a depth of about 3 m or 10 feet. This is a conservative estimate, however, because single column contact is not what occurs within a stack and the contact forces on the apple are smaller, allowing a greater depth.

## VI. CONTACT FORCE MODEL

## 6.1 Introduction

The allowable depth for apples stored in bulk that was calculated in the previous chapter assumed a single column stack. This is not what occurs in the actual pile. One apple will contact several others and the actual contact force probably is less. A computer model for calculating the contact force between spherical bodies is presented in this Chapter. An experimental verification using rubber balls is discussed in the next chapter.

The computer model developed here is based on the model developed by Davis (1974). His model is basically a two-dimensional truss analysis where the center of each sphere is considered as the node and the members connecting the nodes have the nonlinear property of two spheres in contact. Davis used small diameter approximately onehalf inch diameter, steel balls in his study. Six centimeter diameter rubber balls were used in this study. The sphere is much larger and softer.

The following simplifications were made.

a. The spheres are assumed to be identical and only two-dimensional packing problem is considered, i.e., each sphere in the assemblage of spheres has its center on a common plane.

b. Only normal forces at the sphere contacts are considered (shear forces are neglected due to small magnitudes) and the Hertz theory is used to relate the magnitudes of such forces to the corresponding sphere compression.

c. Arrangement of spheres in a sample is assumed to be in a geometrically stable configuration. The initial configuration for this analysis, therefore, must be taken so that no gross movement or rearrangement of the spheres will occur during uniform pressure loading.

e. The change in the contact area is due to the small compressions at the sphere contacts rather than the gross change in the packing geometry.

#### 6.2 Model Formulation

Considering a small stable sample of identical spheres arranged in two-dimensional rhombohedral system and loaded at the top by a uniform pressure, Figure 6.1. The information in Figure 6.1 is shown in Figure 6.2 in the form of a planar graph which is obtained by connecting the center of the spheres. The nodes of the graph in


Figure 6.1.--Two Dimensional Rhombohedral Packing of Identical Spheres Subjected to Uniform Pressure Loading at the Top



Figure 6.2.--Planar Graph Representation of Figure 6.1 and Corresponding Contact Forces.

Figure 6.2 represent the sphere center and branches of the graph correspond to the contacts between adjacent Since shear components of the contact forces spheres. are neglected, forces are transmitted throughout the packing along the branches of the planar graph. The task is to find the forces at each of these branches and study their distribution in the bin where they are located. As mentioned before, since the initial configuration of the assemblage is geometrically stable and no sphere will be moved or dislocated, only a small compression at the contact points of the spheres will occur during the load-In structural analysis, members (branches) are ing. idealized as lines which meet at points (nodes) which are called joints, so the problem formulation can be employed and the displacement of the sphere centers are introduced as unknowns.

Assuming that the rigid-body degrees for the assemblage have been removed by the introduction of appropriate supports, the equilibrium equations for each movable joint of the assemblage may be written as:

$$[N] \{F\} = \{P\}$$
(6.1)

where {F} is the (B x 1) matrix of contact force magnitude and it is customary to assume the contact forces positive when they are in compression (B is the number of bars).

- {P} represents (2J x 1) matrix of applied node forces where J is the number of joints or nodes.
- [N] is a generalized branch node incidence matrix or simply an incidence matrix.

To understand the construction of [N] consider Figure 6.3.



Figure 6.3.--Representation of Member i.

Figure 6.3 shows a typical truss bar and the associated with which is the bar force  $F_i$  and the bar length change  $\Delta_i$ , chosen so that positive  $F_i$  and  $\Delta_i$  corresponds to tension or stretching within the bar and a unit vector  $n_i$ . Knowing the displacement of the ends of a bar, it is possible to compute the change in length of the bar by the following relationship

$$\Delta_{i} = n_{i} \left( \delta_{A} - \delta_{C} \right) \tag{6.2}$$

which involves projecting the joint displacement vector along the original position of the bar. Equation 6.2 can be written for the entire structure as

$$[\Delta] = [N] \{\delta\} \tag{6.3}$$

where [N] is a  $(B \times J) = (row \times column)$  matrix whose

B and J are the number of bars and joints, respectively.  $\{\delta\}$  is the node displacement matrix.

In the circular assemblage of Figure 6.1, B is the number of contact points and J is the number of movable nodes. From elementary mechanics of solids it is known that the bar forces and displacement are related through Hooke's law, in which

$$F_{i} = K_{i} \Delta_{i}$$
(6.4)

The Hertz contact theory applied to the ith contact yields (Davis, 1974):

$$F_{i} = \begin{cases} k \Delta_{i}^{3/2} , & \Delta_{i} \ge 0 \\ 0 & & \Delta_{i} < 0 \end{cases}$$
(6.5)

where  $K = 2G \sqrt{2R} / 3 (1-\mu)$ 

- k is a constant which depends on the radius and elastic properties of spheres
- G and  $\mu$  are shear modulus and Poison's ratio, respectively.

Equations (6.1), (6.3) and (6.5) can be combined in the usual manner for the node formulation and written (Zienkiewicz, 1971)

$$[N]^{T}[K(\delta)][N]\{\delta\} = \{P\}$$
(6.6)

where [N] is (B x B) transformation matrix [K( $\delta$ )] is the (B x B) diagonal Hooke's law matrix { $\delta$ } is (2J x 1) deflection vector {P} is (2J x 1) external load vector

Equation (6.6) represents a set of 2J simultaneous, nonlinear equation for the unknowns  $\{\delta\}$ .

## 6.3 Calculations

The objective here is to use the formulation of the previous section to calculate the joint deflections and then the forces between the uniform spheres arranged in the two-dimensional rhombohedral system shown in Figure 6.1. The displacements for all nodes are not the same. Nodes 1, 5, 10 and 14 can move only in the y-direction while nodes 19, 20, 21, 22 and 23 cannot move in either direction. Each of the rest of the nodes can move in both the x and y directions.

Using the appropriate degrees of freedom for each node, there are 32 force equilibrium equations. Since there are 50 contact forces between spheres, the transformation matrix has a dimension of (32 x 50) and can easily be constructed from the force equilibrium equations. The transformation matrix is in fact the direction cosines of the branches in the assemblage which for rhombohedral system would be either cos 60° or cos 30° depending on the location of the nodes and branches.

It is necessary to have values for k and  $\Delta_i$  to calculate  $F_i$  in (6.5). A value for  $\Delta_i$  is necessary because (6.5) is nonlinear in  $\Delta_i$ . An arbitrary value of -0.254 cm was assumed for each component of { $\delta$ }. Equation (6.3) was then used to calculate each  $\Delta_i$ . Assuming Cos 30° for each component of transformation matrix in first iteration, the value of  $\Delta_i$  will be 0.254 cm.

To determine the value of k in (6.5) a single rubber ball, 6.09 cm diameter, was marked with a small drop of black ink. A sheet of white paper was placed between the ink spot and the rigid flat circular head of

the Instron machine. After exerting a load of 72 Newtons the radius of the contact area of the ball on the white paper was measured to be a = 1.2 cm.

Timoshenko gives the radius of contact area as

$$a = \sqrt{\frac{3}{\frac{3\pi F(K_1 + K_2) R_1 R_2}{4(R_1 + R_2)}}}$$
(6.7)

where 
$$K_1 = \frac{1 - \mu_1^2}{\pi E_1}$$
 and  $K_2 = \frac{1 - \mu_2^2}{\pi E_2}$ 

- F is the applied load R<sub>1</sub> and R<sub>2</sub> are the radius of the two spheres in contact.
- E and  $\mu$  are modulus of elasticity and Poisson's ratio.

Assuming the rigid plate of the base of Instron has  $E = \infty$  when compared with rubber ball and its radius is equal to infinity, (6.7) reduces to

$$a = \sqrt{\frac{3}{4} \frac{3\pi F K_1 R_1}{4}}$$
(6.8)

Substituting the appropriate values in (6.8)  $K_1$  would be

$$K_1 = 0.003345 \text{ cm}^2/\text{N}$$

To find the value for  $\mu_1$ , a cubical specimen of one centimeter in dimension was carefully cut from the rubber ball, and was loaded in the Instron machine and the strain perpendicular to the load and parallel to the load were measured. Poisson's ratio was determined to be  $\mu =$ 0.13.

Using  $\mu = 0.13$  and  $K_1 = 0.003345$ ,  $E_1$  was calculated as  $E_1 = 93.55$  N/cm<sup>2</sup> and the shear modules becomes

$$G = \frac{E}{2(1+\mu)} = 41.39 \text{ N/cm}^2$$

Since  $k = 2G \sqrt{2R} / [3(1 - \mu)]$ 

Substitution gives  $k = 78.26 \text{ N/cm}^{3/2}$ 

The diagnonal terms of  $[K(\delta)]$  in (6.6) are each given by

$$k_{ii} = k \ (\Delta_i^{1/2})$$

or  $k_{ii} = -39.44 \text{ N/Cm}$ 

Since (6.6) is a nonlinear equation, it must be solved using iterations until the calculated displacements on two successive iterations differ by less than a specified amount. Convergence to the solution was obtained in about 18 iterations for the problems discussed in the last section (see Appendix C).

# 6.4 Results

Table 6.1 gives the values of the nodal deflections for the configuration of Figure 6.2 with a concentrated load of 45 N applied at each of the upper nodes. These values are also shown in Figure 6.4. The contact forces are shown in Figure 6.5. The contact forces for other loading situations are given in Figures 6.6, 6.7, and 6.8.

Table 6.1. Deflection of nodes in x and y direction of a two dimensional rhombohedral assemblage for 45 N of load at each node (18 iterations)

| Node<br>No. | Deflection in<br>X direction<br>(mm) | Deflection in<br>Y direction<br>(mm) | Node<br>No. | Deflection in<br>X direction<br>(mm) | Deflection in<br>Y direction<br>(mm) |
|-------------|--------------------------------------|--------------------------------------|-------------|--------------------------------------|--------------------------------------|
|             | 0                                    | -31.82                               | 13          | -1.34                                | -13.61                               |
| 7           | -0.304                               | -25.04                               | 14          | Ο                                    | -13.10                               |
| ы           | 0                                    | -23.90                               | 15          | -0.15                                | - 6.47                               |
| 4           | 0.304                                | -25.04                               | 16          | -0.43                                | - 6.78                               |
| Ŋ           | 0                                    | -31.82                               | 17          | -0.43                                | - 6.80                               |
| 9           | -2.64                                | -21.18                               | 18          | 0.15                                 | - 6.47                               |
| 7           | -0.68                                | -18.56                               | 19          | 0                                    | 0                                    |
| 8           | 0.68                                 | -18.56                               | 20          | 0                                    | ο                                    |
| ŋ           | 2.64                                 | -21.18                               | 21          | 0                                    | 0                                    |
| 10          | 0                                    | -13.10                               | 22          | 0                                    | 0                                    |
| 11          | 1.34                                 | -13.61                               | 23          | 0                                    | Ο.                                   |
| 12          | 0                                    | 12.77                                |             |                                      |                                      |



Figure 6.4.-- Planar Graph Representation of Deflection of Nodes in x and y direction when 225 N Load was exerted



Figure 6.5.--Planar Graph Representation of Calculated Contact Forces for a 225 N Load. All Values in Newtons.



Figure 6.6.--Planar Graph Representation of Calculated Contact Forces for a 157.5 N Load. All Values in Newtons.



Figure 6.7.-- Planar Graph Representation of Calculated Contact Forces for a 180 N Load. All values in Newtons



Figure 6.8.--Planar Graph Representation of Calculated Contact Forces for a 202.5 N Load. All the Values in Newtons.

# VII. EXPERIMENTAL INVESTIGATION OF THE CONTACT MODEL

#### 7.1 General Remarks

To simulate the force distribution between bodies in contact in a bulk storage of apples, a series of tests were conducted in the laboratory. In order to have some uniformity in the granular material, it was decided to use rubber balls with an average diameter of 6.25 to 6.85 cm which is close to that of Jonathan apples. A simple case of two dimensional packing with a rhombohedral assemblage was used. The total of twenty-three balls was used with five balls in a row and five rows in a rhombic manner.

In order to insure that the rubber balls were similar in size and stiffness, the balls were sorted by diameter and then by stiffness. The stiffness was measured using a flat plate test on the Instron Testing machine. The stiffness criteria was that the balls should require between 54-63 N to produce a 1.7 cm deflection. Several portions on each ball were tested.

#### 7.2 Equipment

1. Test Box: A wooden box 53 cm high, 32 cm wide and 6.8 cm thick was constructed. The box was made

with these dimensions so that it could be loaded using an Instron testing machine and could hold five rubber balls in each layer with a little gap for the installation of the pressure transducers. The perforated plexiglass window of the box was easily assembled or removed so that the connecting wires from the pressure transducers could extend from the box to the strain indicator, Figure 7.1.

2. Loading Piston: To exert the load in a uniform manner from the head of the Instron to the balls in the testing box, a loading pistion was constructed, Figure 7.2.

The piston consisted of a metal strip to which the supporting bars were connected. A wooden layer (1.5 cm thick) and a foam layer (2.5 cm thick) were attached on the bottom side of the metal strip. The piston had rectangular cross-section with dimension of (31.5 cm x 6.5 cm) which would fit into the top of the testing box.

3. Pressure Transducer: A special pressure transducer was designed for the experiment and is shown in Figure 7.3. The pressure transducer consisted of a 3.2 x 2 cm upper plate 0.5 mm thick. It was made from spring steel. A lower plate, with the same dimensions except it was thicker (1 mm), was made from hard steel. Two millimeter diameter rollers, two cm long, were glued to the bottom plate. The upper plate rested on the



Figure 7.1.--Test Box with Plexiglass Window and Connecting Wires.



Figure 7.2.--Different Parts of Loading Piston.



Figure 7.3.--Dimensions and Assemblage of the Pressure Transducers.

rollers. A strain gage, (Micro Measurement EA-06-250BG-120) was attached to the lower surface of the upper plate. The upper plate acts as a simply supported beam and was used to determine the magnitude of a load once it was calibrated.

4. Multi-Channel Digital Strain Indicator 161mini-system: The strain gage transducer was attached to a model 161 (B & F Instrument, Inc.) digital strain indicator. This apparatus had ten channels and a terminal box where the pressure transducers and their compensating gages were connected, Figure 7.4 and 7.5.

5. A model OD-1014 printer was connected to the multi-channel Digital strain indicator. Total of twenty pressure transducers were made, of which eight were used compensating gages.

# 7.3 Calibration of Gages of Pressure Transducers

Each pressure transducer was assembled and connected in a singel active arm bridge form to the strain indicator and calibrated. Calibration was done by gluing one of the rubber balls on the head of the Instron machine and locating the transducer beneath the ball on the load cell of the Instron, Figure 7.6. Different levels of load were exerted on the transducer and the corresponding strain was read from the strain indicator and a



Figure 7.4.--Terminal Box of Multichannel Digital Strain Indicator.



Figure 7.5. Multi-Channel Digital Strain Indicator with Model OD-1014 Printer



Figure 7.6. Calibration of Pressure Transducer

calibration curve for each gage was obtained. An example of such a curve is given in Figure 7.7. This process was repeated at least five times for each gage in order to obtain an average curve. The average calibration curve had a variation of  $\pm$  3% from the other curves at a maxi= mum load of 90 N (20 lbs.). Example force and corresponding strain data are given in Table 7.1.

#### 7.4 Experimental Procedure

First step was to connect the eight pressure transducers and the eight corresponding compensating gages to the digital strain indicator. Digital strain indicator was adjusted for  $R_{cal} = 1474 \ \Omega$  from the table of calibration set points based on 120  $\Omega$  single active arms input and gage factor of 2.03. Since there was just one active gage in the pressure transducer, a compensating gage was placed on the steel plate, outside the text box and a two arm-bridge hook-up was used.

# 7.4.1 Pressure Transducer Placement

The objective was to measure the contact forces at as many contact points in the assemblage as reasonably possible. The eight pressure transducers allowed the contact force at eight different points to be measured at a time. The attachment of the transducers on the exact contact points in the assemblage was very important.



Figure 7.7.--Calibration Curve for Gage Six.

| Load F (N) | Strain | Load F (N) | Strain |
|------------|--------|------------|--------|
| 4.5        | 308    | 49.5       | 2390   |
| 9          | 606    | 54         | 2552   |
| 13.5       | 872    | 58.5       | 2686   |
| 18         | 1110   | 63         | 2804   |
| 22.5       | 1344   | 67.5       | 2920   |
| 27         | 1540   | 72         | 3018   |
| 31.5       | 1724   | 76.5       | 3118   |
| 36         | 1892   | 81         | 3210   |
| 40.5       | 2072   | 85.5       | 3310   |
| 45         | 2232   | 90         | 3410   |
|            |        |            |        |

TABLE 7.1.--Values of Strain (ε) at Different Values of Load (N) for Gage Number 6

Transducers were located at the proper points and were glued on the surface of the balls at the designated contact points. The contact points should seat exactly in the middle of the upper-plates of the transducers in order to obtain a correct force value. Three loading replications were made at each contact point considered. This process was completed for eight different transducer positions in the assemblage, Figure 7.8 a-h. The two top rows, 1 and 2, are added to the arrangement to create a uniform loading effect on the balls of row three and below. To locate the pressure transducer in Figure 7.8b, for example, the balls of the bottom row were carefully placed in the test box while the plexiglass window was removed. Then transducers were positioned and glued on the surface of the balls of the bottom row (layer 7) on the lower plate side of the transducer. Balls in the other rows were placed in a manner that it was made sure the balls were in contact at the right points. As the balls of the rows were placed and arranged from bottom to top, the plexiglass window was slided in piece by piece To reduce the friction between the balls and the sides of the test box, a lubricant (vaseline) was used.

#### 7.4.2 Readjusting Digital Strain Indicator

Once the balls and the transducers for a particle arrangement were in palce, the test box was placed on the



Figure 7.8.--Pressure Transducer Placement in Different Locations of the Assemblage.

load cell of the Instron and the initial strain resulting from the weight of the balls were removed by setting the strain indicator to zero.

## 7.4.3 Loading and Strain Reading

The load was applied using the Instron testing machine located in the Wood Technology Laboratory of the Forestry Department.

The Instron was zero balanced and the load was applied using a head speed of 0.508 cm/min (0.2 in/min). The Instron chart recorder was moving at 1.225 cm/min (0.5 in/min). Each location required about 12 minutes to print all the strain values. A maximum load of 202.5 N was applied. Strain values were printed every one-half minute and a mark was made on the loading curve of the Instron strip-chart recorder to indicate the corresponding load at that moment. The strain indicator printed one channel at the time with a two second time interval (20 seconds for all channels), Figure 7.9.

# 7.5 Results and Discussion

The measurement of the contact forces was replicated at least three times for each location and the average of these results are reported here. The contact forces were obtained for axial loads of 157.5, 180 and 202.5 N (35, 40 and 45 lbs). The applied forces and



Figure 7.9-- A Complete Set-Up of the Simulated Force Distribution Experiment of Two Dimensional Rhombohedral Arrangement of the Rubber Balls

correspoding strain values are given in Tables 7.2 through 7.4. Loads greater than 202.5 N were not applied because the pressure transducers were not as reliable in this loading range.

Corresponding planar graph of the applied loads are shown in Figures 7.10 through 7.12. For comparison between calculated and measured values of contact forces the planar graph representation of calculated contact forces are repeated in this section and are shown in Figures 7.13 through 7.15.

Comparing planar graph representation of calculated contact forces with the measured ones for three different loadings (157.5, 180 and 202.5 N) the following was concluded.

All the measured contact forces are within 20 percent of the calculated ones. A possible reason for this is that in the theory which was used to calculate the contact forces, the frictional forces were ignored. Actually there is friction between adjacent spheres and between the spheres and the walls of the test box. There is an approximate constant ratio of 1.14 between measured and calculated contact forces. If the frictional forces in the model were taken under consideration, then this constant ratio would be very small. In other words, the measured contact force values would be very close to the

| 157.5 Newtons of Applied Load (31.5 Newtons on each Node) |                                      |  |             |
|---|--------------------------------------|--|-------------|
| Channel No.   | Ave. Observed                        | Ave. Measured Contact                        | Location of |
|   | Strain <sub>x</sub> 10 <sup>0</sup>  | Forces (N)                                   | Transducers |
| 0<br>5<br>7<br>8<br>9                                     | 1420<br>1560<br>1582<br>1668<br>1520 | 27.9<br>27.9<br>27.9<br>27.9<br>27.9<br>27.9 |             |
| 7   | 1150                                 | 18   |             |
| 0   | 1040                                 | 18.45  |             |
| 3   | 1050                                 | 18.45  |             |
| 9   | 1200                                 | 20.7   |             |
| 5   | 1230                                 | 20.25  |             |
| 8   | 1200                                 | 18   |             |
| 4   | 1150                                 | 18   |             |
| 6   | 1100                                 | 17.55  |             |
| 3   | 1050                                 | 18.45  |             |
| 7   | 1140                                 | 17.55  |             |
| 8   | 1380                                 | 21.6   |             |
| 5   | 1050                                 | 16.65  |             |
| 4   | 1070                                 | 17.1   |             |
| 0   | 1180                                 | 21.6   |             |
| 6   | 1100                                 | 17.55  |             |
| 9   | 1100                                 | 18.45  |             |
| 7   | 1170                                 | 18.45  |             |
| 3   | 1390                                 | 26.1   |             |
| 5   | 870                                  | 13.27  |             |
| 4   | 1070                                 | 17.1   |             |
| 0   | 960                                  | 16.65  |             |
| 8   | 960                                  | 13.95  |             |
| 9   | 1430                                 | 25.65  |             |
| 6   | 1100                                 | 17.55  |             |
| 7   | 1665                                 | 30.15  |             |
| 3   | 910                                  | 15.3   |             |
| 5   | 980                                  | 15.5   |             |
| 4   | 980                                  | 15.52  |             |
| 0   | 885                                  | 15.07  |             |
| 8   | 1040                                 | 15.3   |             |
| 9   | 910                                  | 14.62  |             |
| 6   | 1635                                 | 29.25  |             |
| 0   | 9                                    | 0.22   |             |
| 9   | 8                                    | 0.23   |             |
| 3   | 4                                    | 0.21   |             |
| 7   | 200                                  | 2.44   |             |
| 3   | 160                                  | 2.38   |             |
| 4   | 215                                  | 2.38   |             |
| 5   | 195                                  | 2.43   |             |
| 0   | 290                                  | 3.94   |             |
| 9   | 170                                  | 2.3  |             |
| 3   | 265                                  | 3.92   |             |

Table 7.2--Average measured contact forces for 157.5 N of load

| 180 Newtons of Applied Load (36 Newtons on each Node) |                          |                       |             |
|---|--------------------------|-----------------------|-------------|
| Channel No.   | Ave. Observed            | Ave. Measured Contact | Location of |
|   | Strain x 10 <sup>6</sup> | Forces (N)            | Transducers |
| 0   | 1530                     | 30.6                  |             |
| 5   | 1680                     | 31.05                 |             |
| 8   | 1792                     | 31.05                 |             |
| 9   | 1523                     | 27.90                 |             |
| 7   | 1512                     | 30.6                  |             |
| 7   | 1280                     | 20.7                  |             |
| 0   | 1175                     | 21.37                 |             |
| 3   | 1150                     | 20.7                  |             |
| 9   | 1430                     | 23.4                  |             |
| 5   | 1390                     | 23.85                 |             |
| 8   | 1340                     | 20.7                  |             |
| 4   | 1250                     | 21.15                 |             |
| 6   | 1245                     | 20.25                 |             |
| 3   | 1155                     | 20.92                 |             |
| 7   | 1280                     | 20.7                  |             |
| 8   | 1550                     | 25.2                  |             |
| 5   | 1210                     | 19.80                 |             |
| 4   | 1180                     | 19.35                 |             |
| 0   | 1270                     | 23.85                 |             |
| 6   | 1245                     | 20.25                 |             |
| 9   | 1230                     | 21.15                 |             |
| 7   | 1280                     | 20.7                  |             |
| 3   | 1525                     | 29.47                 |             |
| 5   | 1020                     | 16.2                  |             |
| 8   | 1290                     | 19.8                  |             |
| 0   | 1080                     | 19.35                 |             |
| 4   | 990                      | 15.52                 |             |
| 9   | 1610                     | 30.15                 |             |
| 6   | 1280                     | 21.15                 |             |
| 7   | 1805                     | 34.2                  |             |
| 3   | 950                      | 16.2                  |             |
| 5   | 1130                     | 18.22                 |             |
| 4   | 1100                     | 17.55                 |             |
| 0   | 980                      | 17.1                  |             |
| 8   | 1185                     | 17.55                 |             |
| 9   | 1000                     | 16.2                  |             |
| 6   | 1815                     | 33.75                 |             |
| 0   | 12                       | 0.27                  |             |
| 9   | 105                      | 1.38                  |             |
| 3   | 4                        | 0.24                  |             |
| 7   | 215                      | 3                     |             |
| 3   | 180                      | 2.99                  |             |
| 4   | 205                      | 3.01                  |             |
| 5   | 210                      | 2.92                  |             |
| 0   | 340                      | 4.98                  |             |
| 9   | 240                      | 3.09                  |             |
| 3   | 330                      | 4.89                  |             |

Table 7.3--Average measured contact forces for 180 N of load

| 202.5 Newtons of Applied Load (40-5 Newtons on each Node) |  |  |             |  |
|---|--|--|-------------|--|
| Channel No.   | Ave. Observed  | Ave. Measured Contact  | Location of |  |
|   | Strain x 10 <sup>6</sup>                                     | Forces (N)   | Transducers |  |
| 0   | 1574   | 31.95  |             |  |
| 5   | 1822   | 35.1   |             |  |
| 8   | 1911   | 34.2   |             |  |
| 9   | 1750   | 33.75  |             |  |
| 7   | 1849   | 35.55  |             |  |
| 7<br>0<br>3<br>9<br>5<br>8<br>4<br>6                      | 1360<br>1270<br>1310<br>1450<br>1500<br>1490<br>1370<br>1330 | 22.5<br>23.85<br>24.3<br>26.1<br>26.55<br>23.85<br>23.85<br>23.85<br>22.05 |             |  |
| 3   | 1290   | 23.85  |             |  |
| 7   | 1370   | 22.95  |             |  |
| 8   | 1685   | 28.35  |             |  |
| 5   | 1295   | 21.6   |             |  |
| 4   | 1280   | 21.6   |             |  |
| 0   | 1420   | 27.90  |             |  |
| 6   | 1365   | 22.95  |             |  |
| 9   | 1350   | 23.85  |             |  |
| 7   | 1400   | 23.4   |             |  |
| 3   | 1715   | 35.1   |             |  |
| 5   | 1080   | 17.1   |             |  |
| 8   | 1400   | 22.05  |             |  |
| 0   | 1180   | 21.6   |             |  |
| 4   | 1075   | 17.1   |             |  |
| 9   | 1725   | 35.32  |             |  |
| 6   | 1390   | 23.4   |             |  |
| 7   | 1965   | 31.15  |             |  |
| 3   | 1070   | 18.90  |             |  |
| 5   | 1250   | 20.7   |             |  |
| 4   | 1280   | 21.6   |             |  |
| 0   | 1180   | 21.6   |             |  |
| 8   | 1310   | 20.25  |             |  |
| 9   | 1200   | 20.7   |             |  |
| 6   | 2000   | 38.7   |             |  |
| 0   | 11   | 0.29   |             |  |
| 9   | 115  | 1.68   |             |  |
| 3   | 6  | 0.31   |             |  |
| 7   | 270  | -3.26  |             |  |
| 3   | 220  | 3.3  |             |  |
| 4   | 250  | 3.3  |             |  |
| 5   | 240  | 3.28   |             |  |
| 0   | 400  | 5.83   |             |  |
| 9   | 235  | 3.48   |             |  |
| 3   | 370  | 5.46   |             |  |

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# Table 7.4--Average measured contact forces for 202.5 N of load

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Figure 7.10.--Planar Graph Representation of Measured Contact Forces for a 157.5 N Load.



Figure 7.11.--Planar Graph Representation of Measured Contact Forces for a 180 N Load.

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Figure 7.12.--Planar Graph Representation of Measured Contact Forces for a 202.5 N Load.



Figure 7.13.--Planar Graph Representation of Calculated Contact Forces for a 157.5 N Load. All Values in Newtons.



Figure 7.14.--Planar Graph Representation of Calculated Contact Forces for a 180 N Load. All Values in Newtons.



Figure 7.15.--Planar Graph Representation of Calculated Contact Forces for a 202.5 Load. All Values in Newtons.

calculated ones and error would be quite small. This indicates that the presented model could be more accurate than what was concluded earlier, if the frictional forces were considered.

The distribution of the contact forces in the members both in calculated and measured ones were very similar.

These will conclude that there is an agreement between the calculated values and the measured ones.

#### VIII. SUMMARY AND CONCLUSIONS

Development of experimental and analytical techniques to determine the maximum safe depth for apples in a bulk storage was the primary goal in this study. The problem was studied in two different components. One component involved the determination of the contact forces in a bulk bin while the other component related to the determination of whether a specific loading would produce a bruise.

Cylindrical specimens with a height of 1.27 cm and a diameter of 1.27 cm and a cross-sectional area of 1.266 cm<sup>2</sup> were prepared. These specimens were then compressed using a deformation rate of 1.27 cm per minute until a failure occurred. Tests were conducted on three different dates, October 1, November 15 and December 31. These were denoted as Groups I, II and III. Each group consisted of 150 apples and four samples were removed from each apple. The failure strain,  $\varepsilon_f$ , the stress at the failure,  $\sigma_f$ , and the elastic modulus, E, were determined for every sample. The modulus of elasticity changed significantly between October 1 and November 15. A much smaller change occurred between November 15 and

December 31. The respective averages were  $E_{OCt} = 3279$ Kpa,  $E_{Nov} = 2516$  Kpa and  $E_{Dec} = 2360$  Kpa. The strain at failure showed a small decrease between October 1 and November 15. The reduction rate was lower between November 15 and December 31. The average maximum normal strain at failure was 0.14, 0.11 and 0.12 for the three dates. The average normal stress at failure decreased from 444 Kpa on October 1 to 252 Kpa on November 15 and 235 Kpa on December 31.

The distribution of the elastic modulus and failure strain were used in a computer model to predict bruising for a particular load. Since this model was based on the assumption that a bruise occurs when the maximum normal strain exceeds a specific value, an equation for the maximum normal strain was determined and used in a computer model.

The average location of the maximum normal strain for apple-to-apple contact was 3.65 mm for October first data and 3.0 mm for the other two sets of data. For apple in contact with a flat hard surface these values were 3.88 mm for October 1 data and 3.0 mm for the other two data.

To calculate the allowable storage depth, the modulus of elasticity of an apple in contact with another  $(E_1, E_2)$  and the failure strain of the pair of apples  $(\varepsilon_{F1}, \varepsilon_{F2})$  or apple in the case of flat plate contact were treated as random variables. A normal load F was selected and a random generator was used to select the values of  $E_1$ ,  $E_2$ ,  $\varepsilon_{F1}$  and  $\varepsilon_{F2}$  from the data obtained in the experimental part. The maximum strain in each apple was calculated. If this strain exceeded the failure strain for the apple, the apple was said to be bruised. The load which produced a bruise was converted to a depth by assuming a single column stack where each apple weighed 0.85 N.

Apple-to-apple contact was found to govern the allowable depth. The October 1 apples could be piled 5.14 meters without bruising but the November 15 and December 31 apples could be piled only about 1.8 meters. The significant decrease is due to the decrease in the modulus of elasticity between October 1 and November 15.

A finite element type computer model used for small diameter steel balls was modified for use with large diameter low modulus materials such as apples. This model is basically a two-dimensional truss analysis where the center of each sphere is considered as the node and the members connecting the nodes have the nonlinear property of two spheres in contact. The validity of the model was established by experimentally measuring the contact forces between 6 cm diameter rubber balls which were stacked

in a rhombohedral fashion. Specially designed pressure transducers were located at the proper points and were glued on the surface of the balls in the assemblage. There were seven layers in the assemblage with either four or five balls in each horizontal layer. A maximum load of 202.5 N was applied by Instron and strain values were printed. All of the contact forces differed by less than 20 percent from the values calculated using the computer model.

The following conclusions were drawn from this study.

1. The maximum allowable height for freshly harvested Jonathan apples in contact with each other in a mono-column arrangement is 5.14 m (17.13 ft). This value decreases as the storage period increases. It was about 1.85 m (6.1 ft) after three months of storage at 2.2°C. As for apple in contact with a flat surface the maximum allowable safe depth was around 7 m for the October 1 apples and decreased to 2.2 m by November 15.

2. The modulus of elasticity of Jonathan apples decreased 23.3 percent between October 1, 1978, and November 15, 1978. The average of 600 samples was 3279 Kpa on October 1 and 2516 Kpa on November 15. The modulus value decreased another 6.2 percent between November 15 and December 21 to 2360 Kpa.

3. The failure strain of Jonathan apples averaged 0.14 on October 1, 1978, decreased to 0.11 on November 15 and then increased slightly to 0.12 by December 31.

4. The average failure stress of Jonathan apples decreased significantly between October 1 and December 31, 1978. It averaged 444 Kpa on October 1, 1978, and decreased to 252 Kpa by November 15 and further decreased to 235 Kpa by December 31. The total decrease of 209 Kpa was 47.1 percent of the original value.

5. The structural model formulation for calculating the contact forces within an arrangement of rubber balls disagreed with the experimental by a constant ratio. The experimental results were about 0.83 of the calculated values.

6. The susceptibility of two apples in contact to bruising is higher than that of an apple in contact with a rigid flat surface because maximum allowable safe depth for any of three groups of apples in Case II (apple in contact with rigid flat surface) was always higher than Case I (two apples in contact).

#### IX. SUGGESTIONS FOR FUTURE RESEARCH

Suggestions for future research based on the experiences and results of this study include:

1. The significant variation of the mechanical properties of Jonathan apples during the first 1.5 months of storage indicates that these properties should be studied on a weekly basis or at least every two weeks.

2. Allowable depth values should be calculated for other varieties of apples which may be stored in bulk, These calculations cannot be performed without the distributions of the mechanical properties thus these properties must be studied as a function of storage time.

3. The presence of skin was neglected in this investigation and apples were assumed to be an isotropic, homogeneous mass. Rumsey and Fridley (1974) found that the presence of an elastic skin produced no significant change of the internal stress distribution. Gustafson (1974), however, showed that the restraint created by the skin can cause increased stresses in the body if the turgor pressure is accounted for. The effect of skin properties on the maximum allowable contact force from one apple to another or from the apple to a rigid flat

surface, however, has to be investigated. Consideration of apple skin and core might lead to dealing with anisotropic materials, which in reality is considering the whole apple not just one part of it.

4. The measurement of strain at failure under numerous deformation rates.

APPENDICES

## APPENDIX A

#### EXPERIMENTAL DATA

# TABLE A.1.--Mean Values of the Three Different Parameters ( $\epsilon_{f}$ , $\sigma$ , and E)--Group I

 Date
 : Oct 1, 1978 (Group 1)
 Sample Height
 : 1.2" On

 Variety
 : Jonathan
 Sample Diameter
 : 1.2" On

 Head Speed of Instron
 : 1.2" On/man
 Sample Cross-section Area
 : 1.26" Cm<sup>2</sup>

 Chart Speed
 : 25.4 Oa/min
 : 25.4 Oa/min
 : 1.26" Cm<sup>2</sup>
 : 1.26" Cm<sup>2</sup>

| Apple<br>No. | 573           | <sup>0</sup> ГА<br>(Кра) | E <sub>A</sub><br>(Rga) | No.    | <sup>0</sup> FA<br>(Kpa) | Е<br>(Лура) | Appl<br>No. | • '77  | <sup>d</sup> PA<br>(Rpa) | EA<br>(Rps)     | Apple "PA<br>No. | CPA<br>()Qpa) | х<br>(Тра) |
|--------------|---------------|--------------------------|-------------------------|--------|--------------------------|-------------|-------------|--------|--------------------------|-----------------|------------------|---------------|------------|
| 1            | 0.140         | 476.06                   | 3528.97                 | 41 0.1 | 5 453.42                 | 3144.85     | 81          | 0.148  | 417.43                   | 3000.52         | 121 0.143        | 448.00        | 3302.30    |
| 2            | 0.148         | 427.68                   | 3048.02                 | 42 0.1 | 44 455.74                | 3474.35     | 82          | 0.138  | 405.43                   | 3019.34         | 122 0.149        | 463.81        | 3678.26    |
| 3            | 0.140         | 470.83                   | 3713.83                 | 43 0.1 | 51 4"0.45                | 3312.19     | 83          | 0.138  | 428.07                   | 3182.77         | 123 0.153        | 453.78        | 3194.04    |
| 4            | 0.144         | 451.29                   | 3443.49                 | 44 0.1 | 46 492.37                | 3783.65     | 84          | 0.150  | 453.03                   | 3262.02         | 124 0.147        | 371.56        | 3111.96    |
| 5            | 0.147         | 431.16                   | 3081.01                 | 45 0.1 | 50 443.74                | 3292.57     | 85          | 0.145  | 415.10                   | 3117.08         | 125 0.160        | 526.38        | 3706.14    |
| 6            | 0.145         | 429.23                   | 3098.63                 | 46 0.1 | 50 465.78                | 3336.76     | 86          | 0.152  | 453.23                   | 3239.8          | 126 0.157        | 441.23        | 3075.39    |
| 7            | 0.134         | 434.84                   | 3453.68                 | 47 0.1 | 46 506.83                | 3900.24     | 87          | 0.146  | 442.2                    | 3452.48         | 127 0.153        | 459.61        | 3373.17    |
| 8            | 0.145         | \$26.76                  | 4259.15                 | 48 0.1 | 53 448.97                | 3115.37     | 88          | 0.152  | 436.39                   | 3451.07         | 128 0.163        | 522.45        | 3478.11    |
| 9            | 0.150         | 432.29                   | 3234.06                 | 49 0.1 | 63 416.71                | 2684.34     | 89          | 0.162  | 478.00                   | 3170.93         | 129 0.162        | 469.24        | 3224.50    |
| 10           | <b>0.1</b> 50 | 456.32                   | 3544.02                 | 50 0.1 | 36 439.10                | 3353.12     | 90          | 0.148  | 447.02                   | 3469.38         | 130 0.145        | 455.71        | 3495.63    |
| 11           | 0.147         | 447.23                   | 3492.96                 | 51 0.1 | 54 419.39                | 3063.90     | 91          | 0.151  | 409.30                   | <b>299</b> 0.99 | 131 0.148        | 412.16        | 3057.47    |
| 12           | 0.137         | 448.50                   | 3520.85                 | 52 0.1 | 38 453.62                | 3523.46     | 92          | 0.158  | 510.90                   | 3487.77         | 132 0.157        | 382.16        | 2605.50    |
| 13           | 0.159         | 439.87                   | 2842.89                 | 53 0.1 | 54 425.75                | 3141.25     | 93          | 0.148  | 464.45                   | 3273.91         | 133 0.150        | 424.75        | 3027.36    |
| 14           | 0.140         | 403.11                   | 3112.12                 | 54 0.1 | 48 484.19                | 3653.93     | 94          | 0.160  | 428.65                   | 2950.33         | 134 0.147        | 409.30        | 3008.24    |
| 15           | 0.140         | 453.23                   | 3560.57                 | 55 0.1 | 48 424.39                | 3109.03     | 95          | 0.148  | 450.50                   | 3441.33         | 135 0.162        | 480.90        | 3178.79    |
| 16           | 0.139         | 451.48                   | 3478.90                 | 56 0.1 | 45 472.77                | 3686.52     | 96          | 0.142  | 463.48                   | 3341.37         | 136 0.153        | 461.54        | 331 3. 22  |
| 17           | 0.146         | 431.55                   | 3155.89                 | 57 0.1 | 46 449.16                | 2717.82     | 97          | 0.137  | 418.39                   | 3169.96         | 137 0.160        | 443.16        | 3265.93    |
| 18           | 0.129         | 423.04                   | 3371.31                 | 58 0.1 | 47 440.65                | 3196.05     | 98          | 0.150  | 458.05                   | 3359.40         | 138 0.155        | 455.74        | 3222.45    |
| 19           | 0.146         | 463.68                   | 3508.34                 | 59 0.1 | 48 415.49                | 3103.42     | 99          | 0.151  | 423.17                   | 3229.33         | 139 0.153        | 436.39        | 31 35 . 51 |
| 20           | 0.138         | 453.42                   | 3540.18                 | 60 0.1 | 43 408.33                | 3155.89     | 100         | 0.151  | 445.48                   | 3236.18         | 140 0.145        | 432.52        | 3282.56    |
| 21           | 0.149         | 490.38                   | 3568.23                 | 61 0.1 | 47 473.55                | 3400.27     | 101         | 0.147  | 407.34                   | 3247.41         | 141 0.146        | 415.10        | 31 22 . 08 |
| 22           | 0.148         | 465.23                   | 3423.26                 | 62 0.1 | 43 467.44                | 3522.28     | 102         | 0.162  | 424.78                   | 3186.28         | 142 0.144        | 468.32        | 3412.51    |
| 23           | 0.144         | 462.32                   | 3532.53                 | 63 0.1 | 53 431.03                | 3089.14     | 103         | 0.142  | 456.71                   | 3393.92         | 143 0.145        | 478.97        | 3714,24    |
| 24           | 0.144         | 445.29                   | 3297.57                 | 64 0.1 | 51 450.85                | 3175.97     | 104         | 0.166  | 394.78                   | 2843.71         | 144 0.146        | 462.50        | 3344.71    |
| 25           | 0.163         | 491.93                   | 3232.02                 | 65 0.1 | 59 471.22                | 3226.03     | 105         | 0.151  | 477.03                   | 3483.74         | 145 0.161        | 466.39        | 3260.06    |
| 26           | 0.141         | 470.44                   | 3576.53                 | 66 0.1 | 53 469.10                | 3247.92     | 106         | 0.146  | 383.17                   | 3117.20         | 146 0.151        | 448.97        | 3113.30    |
| 27           | 0.140         | 472.39                   | 3755.21                 | 67 0.1 | 41 404.45                | 3110.45     | 107         | 0.150  | 430.34                   | 2976.19         | 147 0.153        | 441.94        | 3052.15    |
| 28           | 0.156         | 476.52                   | 3364.4                  | 68 0.1 | 46 411.43                | 3258.69     | 108         | 0.143  | 349.50                   | 2949.75         | 148 0.137        | 456.69        | 3546.30    |
| 29           | 0.152         | 456.90                   | 4256.27                 | 69 0.1 | 46 423.81                | 3190.62     | 109         | 0.153  | 418.96                   | 3121.02         | 149 0.143        | 403.49        | 2991.45    |
| 30           | 0.153         | 460.58                   | 3182.95                 | 70 0.1 | 40 425.73                | 3150.19     | 110         | 0.145  | 440.26                   | 3507.10         | 150 0.150        | 433.49        | 3214.43    |
| 31           | 0.152         | 450.50                   | 3186.36                 | 71 0.1 | 41 441.42                | 3358.16     | 111         | .0.151 | 392.84                   | 2972.82         |                  |               |            |
| 32           | 0.148         | 399.43                   | 2944.23                 | 72 0.1 | 51 463.29                | 3157.83     | 112         | 0.155  | 477.99                   | 3507.06         |                  |               |            |
| 33           | 0.143         | 441.81                   | 3207.90                 | 73 0.1 | 48 460.77                | 3159.80     | 113         | 0.141  | 403.49                   | 3161.03         |                  |               |            |
| <b>34</b>    | 0.148         | 439.68                   | 3253.27                 | 74 0.1 | 51 420.90                | 2952.00     | 114         | 0.143  | 412.20                   | 3024.48         |                  |               |            |
| 35           | 0.150         | 424.58                   | 3807.18                 | 75 0.1 | 54 446.84                | 3122.30     | 115         | 0.147  | 382.58                   | 3400.55         |                  |               |            |
| 36           | 0.148         | 430.97                   | 3265.71                 | 76 0.1 | 61 439.87                | 2981.56     | 116         | 0.150  | 411.23                   | 3117.32         |                  |               |            |
| 37           | 0.143         | 520.77                   | 4043.57                 | 77 0.1 | <b>434.4</b> 6           | 3087.90     | 117         | 0.155  | 433.48                   | 3354.16         | 1                |               |            |
| 38           | 0.151         | 427.68                   | 3205.80                 | 78 0.1 | 38 392.65                | 2945.95     | 118         | 0.147  | 407.75                   | 3144.50         |                  |               |            |
| 39           | 0.150         | 456.32                   | 3274.89                 | 79 0.1 | 465.40                   | 3366.23     | 119         | 0.145  | 392.65                   | 2950.50         |                  |               |            |
| 40           | 0.143         | 440.13                   | 3362.80                 | 80 0.1 | 484.77                   | 3699.15     | 120         | 0.147  | 431.92                   | 3321.96         |                  |               |            |

TABLE A.2.--Mean Values of the Three Different Parameters ( $\epsilon_{f}$ ,  $\sigma$ , and E)--Group II

| Date :            | Nov 15, 1978 (Group II) | Sample Height             | : 1.27 Cm              |
|-------------------|-------------------------|---------------------------|------------------------|
| Variety :         | Jona than               | Sample Diameter           | : 1.27 Cm              |
| liead Speed of Ir | nstron : 1.2° Cm/min    | Sample Cross-section area | $: 1.266 \text{ Cm}^2$ |
| Chart Speed       | : 25.4 Cm/min           |                           |                        |

| Apple<br>No. | 573   | <sup>C</sup> FA<br>(Kpa) | E<br>(Rpa) | Apple<br>No. | <sup>6</sup> 97A | <sup>0</sup> FA<br>(Rps) | EA<br>(Rps) | Apple<br>No. | <b>رام</b> | <sup>0</sup> FA<br>(Rpa) | EA<br>(Rps) | Apple CTA<br>No. | <sup>O</sup> FA<br>(Kpa) | Е <sub>А</sub><br>(Кра) |
|--------------|-------|--------------------------|------------|--------------|------------------|--------------------------|-------------|--------------|------------|--------------------------|-------------|------------------|--------------------------|-------------------------|
| 1            | 0.115 | 258.35                   | 2867.15    | 41           | 0.121            | 248.68                   | 2506.14     | 81           | 0.125      | 243.84                   | 2486.79     | 121 0.116        | 250.61                   | 2457.76                 |
| 2            | 0.116 | 271.90                   | 3252.83    | 42           | 0.121            | 239.97                   | 2296.49     | 82           | 0.120      | 226.42                   | 2167.48     | 122 0.130        | 292.22                   | 2464.21                 |
| 3            | 0.096 | 241.90                   | 2914.60    | 43           | 0.123            | 253.51                   | 2670.64     | 83           | 0.108      | 227.39                   | 2409.38     | 123 0.123        | 202.23                   | 2238.43                 |
| 4            | 0.112 | 297.06                   | 3206.06    | 44           | 0.121            | 269.96                   | 2534.25     | 84           | 0.113      | 236.10                   | 2338.42     | 124 0.134        | 270.93                   | 2644.38                 |
| 5            | 0.115 | 249.64                   | 2502.92    | 45           | 0.119            | 248.68                   | 2429.20     | 85           | 0.111      | 222.55                   | 2032.01     | 125 0.133        | 249.64                   | 2206.18                 |
| 6            | 0.113 | 254.16                   | 2644.84    | 46           | 0.118            | 239.97                   | 2406.16     | 86           | 0.123      | 240.93                   | 2270.69     | 126 0.121        | 261.25                   | 2451.31                 |
| 7            | 0.147 | 323.18                   | 2640.54    | 47           | 0.110            | 271.90                   | 2673.87     | 87           | 0.110      | 216.74                   | 2183.60     | 127 0.128        | 267.06                   | 2515.82                 |
| 8            | 0.117 | 308.67                   | 3119.59    | 48           | 0.120            | 316.41                   | 2808.92     | 88           | 0.116      | 245.77                   | 2290.45     | 128 0.121        | 228.36                   | 2219.08                 |
| 9            | 0.118 | 276.74                   | 3031.88    | 49           | 0.125            | 238.03                   | 2125.55     | 89           | 0.116      | 229.32                   | 2306.17     | 129 0.104        | 212.87                   | 2353.63                 |
| 10           | 0.117 | 275.77                   | 2919.00    | 50           | 0.119            | 278.67                   | 2757.73     | İ            | 0.105      | 233.19                   | 2509.37     | 130 0.123        | 239.97                   | 2115.87                 |
| 11           | 0.112 | 274.80                   | 2773.85    | 51           | 0.113            | 257.38                   | 2831.91     | 91           | 0.115      | 244.80                   | 2535.17     | 131 0.126        | 224.48                   | 2156.42                 |
| 12           | 0.116 | 302.86                   | 3193.16    | 52           | 0.118            | 223.52                   | 2164.25     | 92           | 0.114      | 223.52                   | 2244 . 89   | 132 0.116        | 265.12                   | 2725.47                 |
| 13           | 0.115 | 303.83                   | 3048.01    | 53           | 0.108            | 234.16                   | 2459.76     | 93           | 0.126      | 233.19                   | 2090.07     | 133 0.129        | 230.29                   | 2154.57                 |
| 14           | 0.107 | 287.38                   | 3096.40    | 54           | 0.113            | 306.73                   | 3067.37     | 94           | 0.108      | 241.90                   | 2451.31     | 134 0.123        | 262.22                   | 2599.91                 |
| 15           | 0.123 | 245.77                   | 2560.98    | 55           | 0.116            | 284.48                   | 2878.68     | 95           | 0.094      | 243.84                   | 3096.40     | 135 0.124        | 242.87                   | 2238.43                 |
| 16           | 0.125 | 291.25                   | 2924.07    | 56           | 0.109            | 290.28                   | 2862.55     | 96           | 0.121      | 235.13                   | 2273.91     | 136 0.125        | 256.42                   | 2361.00                 |
| 17           | 0.111 | 291.55                   | 3077.04    | 57           | 0.119            | 285.44                   | 2626.41     | 97           | 0.113      | 232.23                   | 2341.66     | 137 0.135        | 245.77                   | 2061.04                 |
| 18           | 0.114 | 249.61                   | 2574.34    | 58           | 0.121            | 262.22                   | 2486.79     | 98           | 0.118      | 249.64                   | 2309.71     | 138 0.111        | 204.16                   | 2118.63                 |
| 10           | 0.126 | 297.06                   | 2739.30    | 59           | 0.133            | 195.46                   | 1699.79     | iðu (        | 0.100      | 220.61                   | 2467.44     | 139 0.128        | 214.81                   | 2051.36                 |
| 20           | 0.123 | 322.21                   | 2955.01    | 60           | 0.121            | 227.30                   | 2144.90     | 100          | 0.120      | 265.12                   | 2464.21     | 140 0.120        | 252.55                   | 2361.00                 |
| 21           | 0.100 | 214.80                   | 2372.06    | 61           | 0.113            | 248.6                    | 2454.54     | 101          | 0.111      | 247.71                   | 2409.38     | 141 0.124        | 226.42                   | 2181.76                 |
| 22           | 0.118 | 246.74                   | 2528.72    | 62           | 0.113            | 229.32                   | 2215.40     | 102          | 0.113      | 231.26                   | 2298.10     | 142 0.129        | 250.61                   | 2425.50                 |
| 23           | 0.113 | 257.38                   | 2738.37    | 63           | 0.108            | 252.55                   | 2496.47     | 103          | 0.118      | 205.13                   | 2051.36     | 143 0.125        | 235.13                   | 2148.12                 |
| 24           | 0.125 | 259.32                   | 2305.71    | 64           | 0.106            | 248.68                   | 3141.55     | 104          | 0.125      | 290.28                   | 2902.87     | 144 0.145        | 260.29                   | 2315.84                 |
| 25           | 0.120 | 266.09                   | 2443.25    | 65           | 0.121            | 269.96                   | 2406.16     | 105          | 0.108      | 227.39                   | 2386.80     | 145 0.135        | 263.19                   | 2352.71                 |
| 26           | 0.133 | 281.57                   | 2354.55    | 66           | 0.118            | 243.84                   | 2317.46     | 106          | 0.106      | 243.84                   | 2580.33     | 146 0.138        | 254.48                   | 2354.55                 |
| 27           | 0.114 | 239.00                   | 2696.44    | 67           | 0.127            | 247.71                   | 2225.53     | 107          | 0.118      | 254.48                   | 2419.06     | 147 0.128        | 248.68                   | 2379.18                 |
| 28           | 0.129 | 311.57                   | 2823.62    | 68           | 0.113            | 244.80                   | 2361.00     | 108          | 0.123      | 223.52                   | 2061.04     | 148 0.126        | 221.58                   | 2161.02                 |
| 29           | 0126  | 237.06                   | 2113.56    | 69           | 0.128            | 270.93                   | 2419.06     | 109          | 0.125      | 241.90                   | 2277.14     | 149 0.133        | 231.45                   | 2302.94                 |
| 30           | 0.125 | 275.77                   | 2509.37    | 70           | 0.111            | 219.65                   | 2273.91     | 110          | 0.109      | 231.26                   | 2457.76     | 150 0.139        | 251.58                   | 2115.74                 |
| 21           | 0.114 | 282.54                   | 2846.43    | 71           | 0.113            | 249.64                   | 2346.49     | 111          | 0.124      | 219.65                   | 2080.39     |                  |                          |                         |
| 32           | 0.119 | 285.44                   | 2889.97    | 72           | 0.118            | 278.67                   | 2609.36     | 112          | 0.125      | 238.03                   | 2151.35     |                  |                          |                         |
| 33           | 0.119 | 251.58                   | 2788.37    | 73           | 0.104            | 276.74                   | 3028.66     | 113          | 0.115      | 262.22                   | 2657.74     |                  |                          |                         |
| 34           | 0.118 | 222.55                   | 2161.02    | 74           | 0.116            | 231.26                   | 2219.08     | 114          | 0.120      | 248.68                   | 2509.37     | 1                |                          |                         |
| 35           | 0.121 | 325.12                   | 2994.79    | 75           | 0.120            | 263.19                   | 2541.62     | 115          | 0.124      | 250.61                   | 2247.65     |                  |                          |                         |
| 36           | 0.121 | 267.06                   | 2515.82    | 76           | 0.113            | 256.42                   | 2660.96     | 116          | 0.131      | 228.34                   | 2115.87     |                  |                          |                         |
| 37           | 0.105 | 270.93                   | 3212.51    | 77           | 0.126            | 233.19                   | 2157.80     | 117          | 0.123      | 226.42                   | 2212.63     |                  |                          |                         |
| 38           | 0.125 | 248.68                   | 2222.08    | 78           | 0.121            | 230.29                   | 2132.92     | 118          | 0.121      | 224.48                   | 2112.64     |                  |                          |                         |
| 39           | 0.113 | 210.94                   | 2075.99    | 79           | 0.113            | 239.48                   | 2396.48     | 119          | 0.125      | 266.09                   | 2477.12     |                  |                          |                         |
| 40           | 0.119 | 310.6                    | 2967.38    | 80           | 0.111            | 256.42                   | 2612.58     | 120          | 0.114      | 217.71                   | 2264.24     |                  |                          |                         |

| TABLE | A.3Mean           | Values         | of tl | ie Thre | e Different | Parameters |
|-------|-------------------|----------------|-------|---------|-------------|------------|
|       | (ε <sub>f</sub> , | $\sigma$ , and | E)(   | Group I | II          |            |

•

|     |       |         | Date<br>Variety              | :Du<br>:J | ec 30,<br>Ionatha | 1978 (Group<br>n         | ; III)                  |           |                | Samap1<br>Samap1         | e Height<br>e Diameter  |              | : 1.27 Or<br>: 1.27 Or   | г.<br>Л                   |
|-----|-------|---------|------------------------------|-----------|-------------------|--------------------------|-------------------------|-----------|----------------|--------------------------|-------------------------|--------------|--------------------------|---------------------------|
|     |       |         | Head Speed of<br>Chart Speed | Instro    | 1 : 1<br>: 2      | .27 Om/min<br>5.4 Om/min |                         |           |                | Sampl                    | e Cross-Jec             | tion area    | : 1.266 0                | - سر                      |
|     |       |         |                              |           | -                 |                          |                         |           |                |                          |                         |              |                          |                           |
| [   |       | G       |                              | r         |                   |                          |                         | T         |                |                          |                         | <del>.</del> |                          |                           |
| No. | -FX   | (Rpa)   | <sup>—</sup> "А<br>(Кра)     | No.       | 593               | "77A<br>(16pa)           | <sup>Ε</sup> λ<br>(Rpa) | No.       | . <b>, 1</b> 7 | <sup>o</sup> PA<br>(Rps) | <sup>Е</sup> А<br>(Жра) | Apple TA     | <sup>0</sup> FA<br>(Kpa) | - Σ <sub>λ</sub><br>(Rpa) |
| 1   | 0.124 | 270.45  | 2443.24                      | 41        | 0.138             | 274.80                   | 2603.62                 | 81        | 0.141          | 208.52                   | 1915.27                 | 121 0.121    | 234.06                   | 2412.15                   |
| 2   | 0.130 | 220.03  | 2242.24                      | 42        | 0.124             | 228.74                   | 2139.83                 | 82        | 0.121          | 271.41                   | 2706.27                 | 122 0.128    | 229.81                   | 2262.39                   |
| 3   | 0.121 | 199.81  | 2056.20                      | 43        | 0.109             | 253.90                   | 2552.68                 | 83        | 0.130          | 188.97                   | 2381.04                 | 123 0.129    | 216.74                   | 2438.26                   |
| 4   | 0.105 | 219.65  | 2556.14                      | 44        | 0.129             | 202.23                   | 2073.22                 | 84        | 0.121          | 223.90                   | 2184.95                 | 124 0.134    | 229.81                   | 2550.22                   |
| 5   | 0.135 | 240.84  | 2180.38                      | 45        | 0.121             | 230.68                   | 2290.68                 | 85        | 0.128          | 262.71                   | 2641.25                 | 125 0.121    | 224.00                   | 2583.55                   |
| 4   | 0.124 | 240.45  | 2162.64                      | 46        | 0.133             | 221.10                   | 2026. 36                | 86        | 0.123          | 252.55                   | 2577.10                 | 126 0.120    | 234.64                   | 2484.72                   |
| 7   | 0.123 | 200.87  | 2161.02                      | 47        | 0.116             | 258.84                   | 3019.60                 | 87        | 0.124          | 241.42                   | 2434.95                 | 127 0.135    | 240,93                   | 2631.01                   |
| P   | 0.125 | 207.07  | 2019.11                      | 48        | 0.121             | 200.07                   | 2052.74                 | 88        | 0.110          | 283.51                   | 3096.40                 | 128 0.115    | 241.90                   | 2741.60                   |
| 0   | 0.120 | 201.76  | 1949.76                      | 49        | 0.130             | 233.00                   | 2102.63                 | 89        | 0.131          | 205.13                   | 2077.32                 | 129 0.121    | 216.74                   | 2333.77                   |
| 10  | 0.143 | 234.46  | 2038.48                      | 50        | 0.126             | 223.03                   | 2297.82                 | 90        | 0.121          | 225.84                   | 2409.23                 | 130 0.135    | 244.03                   | 2532.41                   |
| 11  | 0.113 | 209.39  | 1911.05                      | 51        | 0.120             | 222.55                   | 2380.35                 | 91<br>[9] | 0.119          | 201.65                   | 2061.77                 | 131 0.131    | 228.84                   | 2218.62                   |
| 12  | 0.125 | 233.68  | 2006.20                      | 52        | 0.118             | 264.16                   | 2589.08                 | 92        | o.120          | 231.64                   | 2519.28                 | 132 0.120    | <b>19</b> 0.62           | 2411.25                   |
| 13  | 0.116 | 233.98  | 2136.83                      | 53        | 0.118             | 285.44                   | 2733.54                 | 93        | 0.125          | 216.74                   | 2273.15                 | 133 0.111    | 235.67                   | 2609.36                   |
| 14  | 0.125 | 194.49  | 1902.99                      | 54        | 0.120             | 239.97                   | 2429.04                 | 94        | 0.124          | 222.55                   | 2303.86                 | 134 0.111    | 195.46                   | 2068.29                   |
| 15  | 0.136 | 229. "1 | 1957.44                      | 55        | 0.120             | 244.32                   | 2407.54                 | 95        | 0.125          | 218.68                   | 2322.00                 | 135 0.125    | 250.13                   | 2525.04                   |
| 16  | 0.138 | 225.94  | 2068.29                      | 56        | 0.119             | 220.42                   | 2280.95                 | 96        | 0.134          | 230.19                   | 2182.91                 | 136 0.124    | 248.19                   | 2773.85                   |
| 1-  | 0.120 | 229.23  | 2128.                        | 57        | 0.124             | 251.87                   | 2546.04                 | 97        | 0.125          | 241.32                   | 2451.31                 | 137 0.131    | 268.51                   | 2482.03                   |
| 18  | 0.125 | 268.41  | 2335.20                      | 58        | 0.116             | 245.77                   | 2999.63                 | 98        | 0.130          | 237.06                   | 2410.99                 | 138 0.139    | 283.32                   | 2921.48                   |
| 19  | 0.118 | 246.74  | 2185.22                      | 59        | 0.125             | 243.35                   | 2113.80                 | 99        | 0.125          | 262.71                   | 2584.88                 | 139 0.136    | 206.58                   | 2135.22                   |
| 20  | 0.118 | 220.61  | 2215.86                      | 60        | 0.124             | 260.29                   | 2563.68                 | 100       | 0.125          | 217.23                   | 2414.58                 | 140 0.129    | 195.99                   | 1998.99                   |
| 21  | 0.125 | 213.89  | 1858.64                      | 61        | 0.121             | 214.81                   | 2096.52                 | 101       | 0.130          | 232.71                   | 2318.84                 | 141 0.121    | 214.81                   | 2130.56                   |
| 22  | 0.121 | 195.46  | 20~5.55                      | 62        | 0.125             | 237.06                   | 2483.57                 | 102       | 0.128          | 219.16                   | 2292.47                 | 142 0.114    | 218.19                   | 2528.72                   |
| 23  | 0.126 | 164.98  | 1678.82                      | 63        | 0.120             | 266.09                   | 2882.14                 | 103       | 0.138          | 255.93                   | 2455.54                 | 143 0.126    | 234.16                   | 2489.33                   |
| 24  | 0.124 | 224.19  | 2197.31                      | 64        | 0.128             | 239.00                   | 2267.46                 | 104       | 0.124          | 239.00                   | 2603.37                 | 144 0.125    | 241.90                   | 2516.74                   |
| 25  | 0.124 | 261.16  | 2477.92                      | 65        | 0.121             | 235.61                   | 2459.38                 | 105       | 0.131          | 251.00                   | 2403.70                 | 145 0.129    | 244.32                   | 2591.62                   |
| 26  | 0.130 | 237.55  | 2386.80                      | 66        | 0.131             | 252.55                   | 2559.59                 | 106       | 0.134          | 264.16                   | 2538.86                 | 146 0.136    | 239.97                   | 2607.05                   |
| 27  | 0.130 | 217.13  | 2179.26                      | 67        | 0.136             | 291.25                   | 2716.90                 | 107       | 0.125          | 236.39                   | 2361.46                 | 147 0.125    | 239.97                   | <b>2689.9</b> 9           |
| 28  | 0.124 | 209.49  | 1969.80                      | 68        | 0.120             | 222.07                   | 2303,86                 | 108       | 0.131          | 211.91                   | 2175.42                 | 148 0.115    | 231.26                   | 2791.98                   |
| 29  | 0.140 | 221.58  | 2028.78                      | 69        | 0.131             | 243.84                   | 2393.56                 | 109       | 0.128          | 209.97                   | 2235.24                 | 149 0.125    | 275.77                   | 2912.55                   |
| 30  | 0.141 | 253.51  | 2685.15                      | 70        | 0.120             | 238.03                   | 2491.63                 | 110       | 0.128          | 234.16                   | 2295.36                 | 150 0.120    | <b>199</b> .33           | 2004.36                   |
| 31  | 0.120 | 228.74  | 2289.12                      | 71        | 0.121             | 229.81                   | 2305.26                 | 111       | 0.118          | 236.10                   | 2690.91                 |              |                          |                           |
| 32  | 0.145 | 245.96  | 2478.96                      | 72        | 0.116             | 224.48                   | 2332.78                 | 112       | 0.119          | 221.58                   | 2509.37                 |              |                          |                           |
| 33  | 0.118 | 240.93  | 2483.57                      | 73        | 0.126             | 287.86                   | 2890.43                 | 113       | 0.111          | 247.71                   | 2659.68                 |              |                          |                           |
| 34  | 0.124 | 231.46  | 2092.48                      | 74        | 0.128             | 259.80                   | 2502.51                 | 114       | 0.129          | 232.23                   | 2394.58                 |              |                          |                           |
| 35  | 0.149 | 230.77  | 2061.57                      | 75        | 0.129             | 254.00                   | 2554.53                 | 115       | 0.126          | 250.03                   | 2589.54                 |              |                          |                           |
| 36  | 0.133 | 229.81  | 2239.36                      | 76        | 0.120             | 252.55                   | 2451.31                 | 116       | 0.109          | 215.58                   | 2316.78                 |              |                          |                           |
| 37  | 0.136 | 243.84  | 2148.74                      | 77        | 0.128             | 255.93                   | 2422.67                 | 117       | 0.123          | 203.68                   | 2083.54                 |              |                          |                           |
| 38  | 0.129 | 245.58  | 2179.56                      | 78        | 0.129             | 242.87                   | 2522.73                 | 118       | 0.135          | 214.32                   | 2153.92                 |              |                          |                           |
| 39  | 0.129 | 211.32  | 2013.98                      | 79        | 0.123             | 230.29                   | 2488.17                 | 119       | 0.131          | 249.64                   | 2518.40                 |              |                          |                           |
| 40  | 0.129 | 241.90  | 1981.51                      | 80        | 0.115             | 223.03                   | 2540.01                 | 120       | 0.118          | 209,97                   | 2105.16                 |              |                          |                           |

#### APPENDIX B

# COMPUTER PROGRAM FOR CALCULATION OF HEIGHT AND PERCENT BRUISE RELATION-SHIP

Computer Program for Calculation of Height and Percent Bruise Relation-ship, Case I, Group I :

Computer Program for Calculation of Height and percent Bruise •• Relation-ship, Case II, Group I

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PRC6PAM CPDEF(INFLT.(LTFLT.TAFE=IAFLT.TAFES=CUTPUT)

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                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                            0
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#### APPENDIX C

# A COMPUTER PROGRAM FOR CALCULATION OF CONTACT FORCES AND NODAL DEFLECTIONS IN A TWO DIMENSIONAL RHOMBOHEDRAL ASSEMBLAGE OF SPHERES

#### APPENDIX C

# A COMPUTER PROGRAM FOR CALCULATION OF CONTACT FORCES AND NODAL DEFLECTIONS IN A TWO DIMENSIONAL RHOMBOHEDRAL ASSEMBLAGE OF SPHERES

The following computer program was written in the basic language for the CDC 6500 to solve Equation (6.6). The given data are the transformation matrix  $[N] = (32 \times 50)$  and the force matrix  $\{P\} = (32 \times 1)$ . The program compares the contact forces as well as nodal deflections after each iteration with those calculated previously to determine when to half the solution. The covergence to the solution was obtained when both contact forces and nodal deflections after each iteration with those calculated previously to determine to the solution when to halt the solution. The convergence to the solutions after each iteration with those calculated previously to determine when to halt the solution. The convergence to the solution was obtained when both contact forces and nodal deflections had the same last three decimal points and it was achieved in the 18th iteration.

100HASE1 110/IM E(32,50),A1(32,50),B(50,50),C(50,32),F(32,32),G(50),F1(32,32) 1201/IM F(32)+1/(32)+F2(50)+M(50) 13001M0(50+50) 140MARGIN 132 150SETUIGITS 5 160MAT 0=10N(50,50) 170MAT E= (-39.44)\*Q 180MAT READ A1 190FRINT \*A1\* 200REMMAT PRINT A1; 210MAT READ P 220FRINT\*F\* 230FEMMAT PRINT PF 240J=1 250NAT C=TRN(A1) 2601FINT \*C(\*;J;\*)\* 27URENMAT PRINT CF 280hAT E=A1\*B 290PRINT \*E(\*#J#\*)\* BOOREMMAT PRINT E: 310MAT F=E\*C 3POPRINT "F(";J;")" 330FEMMAT FRINT FF 340HAT F1=1NV(F) 350FRINT \*F1(\*;J;\*)\* 360FERMAT PRINT F1; 370MAT 1=F1\*F 3801F INT(J/3) >> J/3 THEN 410 390FRINT\*D(\*;J;\*)\* 400HAT PRINT DE 410MAT G=C\*U 420FOR 1=11050 430F2(I)=78.26\*SQR(ABS(G(I)\*\*3)) 440NEXT I 450IF INT(J/3) > J/3 THEN490 460FHINT \*F2(\*;J;\*)\* 470MAT FRINT F2; 480FR1N1.G(.+11+.). APOFEMMAT PRINT GF 500F0R I=1T050 5101:(I,I)=78,26\*SQR(ABS(G(I))) SZONEXT I 530FRINT \*B(\*;J;\*)\* **SAOREMMAT FRINT BF** 550J=J+1 56060 TO 250 570REM NOTHING

## APPENDIX D

# CALCULATION OF LOCATION OF MAXIMUM STRAIN(Z) IN A SINGLE APPLE UNDER CONTACT LOAD

#### APPENDIX D

## CALCULATION OF LOCATION OF MEXIMUM STRAIN

(Z) IN A SINGLE APPLE UNDER CONTACT LOAD

Case I: Two Apples in Contact

In this case two apples are in contact with two different modulii of elasticity,  $E_1$  and  $E_2$ . Measured values of strain at failure ( $\varepsilon_f$ ) and the modulus of elasticity of 150 apples were used to determine the depth at which the maximum strain occurs (Z).

The following computer program calculates the value of Z for different loads (F). The program is set up in such a way that each time two modulus of elasticity values and their coupled corresponding strains at failure is randomly chosen (out-off 150 available data for each testing date. This is done by RANF (-1) in the program). Substituting randomly chosen  $E_1$  and  $E_2$  in the place of E in Equation 5.2 will give  $\varepsilon_{max1}$  and  $\varepsilon_{max2}$ , respectively. The program compares these two values (calculated maximum strain at failure) with the actual (measured) values of strain at failure and counts the value of calculated strain if it is larger than the measured one. This process is repeated for different depth (Z) from 0.1 mm up

to 5 mm. Each time maximum strain and its corresponding Z value under a given load (F) will be printed.

Figure 5.1 shows the relation between depth (Z) and maximum strain  $\varepsilon_{max}$  under different loads (F). It was found that the maximum strain for apples group I occurs around Z = 3.56 mm and for group II and III it is around Z = 3.0 mm.

```
FECCE FM GENERAL PROTOUTE UT TAPES INFORT TAPES OUTFUT Y

FINTHER STOCK (15), FOR (17)

C = I C THE FAILURE STATION

RF40FF (1 TAPES (1), J=1, 15), (SFS(I), J=1, 15),

TO TO TAPES (1), J=1, 15), (SFS(I), J=1, 15),

TO TO TAPES (1), J=1, 15), (SFS(I), J=1, 15),

TO TO TAPES (1), J=1, 15), (SFS(I), J=1, 15),

TO TO TAPES (1), J=1, 15), (SFS(I), J=1, 15),

TO TO TAPES (1), J=1, 15), (SFS(I), J=1, 15),

TO TO TAPES (1), J=1, 15), (SFS(I), J=1, 15),

TO TO TAPES (1), J=1, 15), (SFS(I), J=1, 15),

TO TO TAPES (1), J=1, 15), (SFS(I), J=1), (SFS(I
```

### Case II: Apple in Contact with Flat Surface

In this case the computer program uses Equation 5.2 and considers  $R_1$  of the flat surface to be inifity and the modulus of elasticity of enamel steel as 206850 Mpa (30 x  $10^6$  Psi). Z for group I was determined to be 3.88 mm and the other groups were around 3.0 mm.

Computer Program for Calculation of Z in Case II, Group I :

```
FPOGRAM URIFRINDUT GUTPLT TAFE 5= 14FUT TAPE 4= 0UTPUT)

FFOGRAM URIFRINDUT GUTPLT TAFE 5= 14FUT TAPE 4= 0UTPUT)

FFORMAT 6= 10

FFORMAT
                                                                                                                                                                                                                                                                                                                                                                                                                                 S
                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                                          1000
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## APPENDIX E

# LOADING OF CYLINDRICAL SPECIMENS AT DIFFERENT DEFORMATION RATES

#### APPENDIX E

## LOADING OF CYLINDRICAL SPECIMENS AT DIFFERENT DEFORMATION RATES

To determine the effect of deformation rate on strain and stress at failure, a separate test was conducted. This time 20 apples were chosen and rates of deformation, 2.5 cm/min (1.0 in/min), 1.25 cm/min (0.5 in/min) and 0.5 cm/min (0.2 in/min) were used. The test was done on four different groups of apples. Group I were tested on October 1 (immediately after harvest), Group II were tested on November 15 (1.5 months of 2.2°C storage), Group III were tested on December 30 (3 months of 2.2°C storage), and Group IV were tested on Febaruary 15 (4.5 months of 2.2°C storage).

All apples which were stored at 2.2°C were removed from storage 24 hours prior to testing in order to get to the equilibrium with the laboratory temperature.

Table El gives the mean and standard deviation of stress and strain at failure for this test.

Compression test of apple tissue (Jonathan) at three different deformation rates Table El---

|                              | - <b>IS</b>     | TAIN AT FAI     | LURE            |                 | 5           | ESS AT FAIL | URE (Kpa)           |             | Shcant         | Modulus of     | ' Elasticit    | y (Kpa)       |  |
|------------------------------|-----------------|-----------------|-----------------|-----------------|-------------|-------------|---------------------|-------------|----------------|----------------|----------------|---------------|--|
| DEI CAMATICIN INTE MIN MILLI | GROUP 1         | 11              | 111             | 2               | I           | 11          | 111                 | Ŋ           | 1              | 11             | 111            | 21            |  |
| 25                           | 0.15<br>(9.010) | 0.12<br>(0.006) | 0.13<br>(0.014) | 0.12<br>(0.006) | 450<br>(42) | 250<br>(25) | 240<br>(25)         | 215<br>(28) | 3000           | 2083.33<br>( ) | 1846.15<br>( ) | 1791.6<br>( ) |  |
| 12.5                         | 0.15<br>(0.008) | 0.12<br>(0.005) | 0.13<br>(0.008) | 0.12<br>(0.008) | 440<br>(28) | 240<br>(24) | 2 <b>30</b><br>(29) | 210<br>(31) | 2933.33<br>( ) | 2000           | 1769.23<br>( ) | 1750 )<br>( ) |  |
| S                            | n.15<br>(9.010) | 0.12<br>(0.006) | 0.13<br>(0.004) | 0.12<br>(0.007) | 430<br>(43) | 230<br>(23) | 220<br>(25)         | 200<br>(24) | 2866.66<br>( ) | 1916.66<br>( ) | 1692.30<br>( ) | 1666.6<br>( ) |  |
|                              |                 |                 |                 |                 |             | 00. 0.      |                     |             |                |                |                |               |  |

Group I: Tested immediately after harvest. Group II: Tested with 1.5 months of storage (2.2°C) after harvest.

Group III: Tested with 3 months of storage (2.2<sup>o</sup>C) after harvest. Group IV: Tested with 4.5 months of storage (2.2<sup>o</sup>C) after harvest.

( ) Indicates Standard Deviation.

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